

CONFIDENTIAL471
Copy
RM L57J11

RESEARCH MEMORANDUM

AERODYNAMIC LOAD DISTRIBUTION OVER A 45° SWEPT WING
HAVING A SPOILER-SLOT-DEFLECTOR AILERON AND
OTHER SPOILER AILERONS FOR MACH NUMBERS
FROM 0.60 TO 1.03

By F. E. West, Jr., Charles F. Whitcomb,
and James W. Schmeer

Langley Aeronautical Laboratory
Langley Field, Va.

CLASSIFICATION AUTHORITY: NASA CHANGED TO UNCLASSIFIED
EFFECTIVE DATE: MARCH 18, 1960
DISTRIBUTION: ANNOUNCEMENTS # 17
KJL

CLASSIFIED DOCUMENT

This material contains information affecting the National Defense of the United States within the meaning of the espionage laws, Title 18, U.S.C., Secs. 793 and 794, the transmission or revelation of which in any manner to an unauthorized person is prohibited by law.

NATIONAL ADVISORY COMMITTEE
FOR AERONAUTICS

WASHINGTON

December 5, 1957

CONFIDENTIAL

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

RESEARCH MEMORANDUM

AERODYNAMIC LOAD DISTRIBUTION OVER A 45° SWEPT WING

HAVING A SPOILER-SLOT-DEFLECTOR AILERON AND
OTHER SPOILER AILERONS FOR MACH NUMBERS

FROM 0.60 TO 1.03

By F. E. West, Jr., Charles F. Whitcomb,
and James W. Schmeer

SUMMARY

A loads investigation was made with flap spoiler ailerons, deflector ailerons, and a spoiler-slot-deflector aileron on a 45° swept-wing—body combination. These controls were located in the vicinity of the 70-percent wing chord line and extended outboard to 87 percent of the wing semispan. Control projection was varied only for the deflector ailerons. The wing had an aspect ratio of 4, a taper ratio of 0.60, and NACA 65A006 airfoil sections parallel to the plane of symmetry. Static pressures were measured on the wing and some of the controls at several spanwise stations for Mach numbers from 0.60 to 1.03, Reynolds numbers from 5.05×10^6 to 6×10^6 , and an angle-of-attack range of about 0° to 20°.

This investigation is a closely related extension of a pressure study (see NACA RM L54C17a) of retractable spoiler ailerons. The results of these two investigations indicate that the effects of flap spoiler ailerons and retractable spoiler ailerons on the wing pressures are generally similar. Adding a slot deflector to a spoiler-aileron configuration creates low pressures on the deflector rear surface and the subsequent wing lower surface which have a favorable effect on rolling moment, particularly at high angles of attack. At low angles of attack the low pressures associated with a slot deflector have their largest effects on the spanwise and chordwise loadings for a swept wing at the inboard stations. These low pressures tend to be of the same magnitude as the upper-surface pressures behind a spoiler which are lowest at the inboard stations. At high angles of attack where a spoiler has a rather small effect on wing loading, the low pressures associated with a

CONFIDENTIAL

slot deflector affect the loading over most of the wing span. The control spanwise load distributions were similar in shape for the spoiler aileron and the spoiler and deflector components of the spoiler-slot-deflector aileron at low angles of attack.

INTRODUCTION

Very few pressure data for spoiler-aileron configurations have been available at transonic speeds for use in load calculations or in studying the effects of spoiler ailerons on the flow about wings. A test program was, therefore, initiated in the Langley 16-foot transonic tunnel to provide static-pressure data for various spoiler-aileron configurations in the transonic speed range for a large angle-of-attack range.

The initial results of this test program (see ref. 1) were for a 45° swept-wing-body combination having retractable and plug spoiler ailerons. The most recent phase of this test program, reported herein, was primarily the investigation at 0° yaw of spoiler-slot-deflector ailerons and flap spoiler ailerons on essentially the same model used in the initial phase. These controls were located in the vicinity of the 70-percent wing chord line and extended from the vicinity of the body to 87 percent of the wing semispan (same as the spoiler ailerons used in the initial phase of the investigation).

In this paper the wing normal-force characteristics, centers of load, and span-load and chordwise pressure distributions for one of the spoiler-slot-deflector aileron configurations and one of the flap spoiler-aileron configurations are compared with those for the basic model. Tabulated wing pressure coefficients and spoiler and deflector loading characteristics are also given for these two control configurations. The effects of wing leading-edge chord-extensions on the spoiler-slot-deflector loads are also included. There are shown, in addition, the effects of varying projection of a deflector aileron and of removing inboard segments of the flap spoiler aileron on the wing chordwise pressure distributions. Except for the configurations having only a deflector aileron, control projection was not varied.

A few of these pressure data results have been presented in reference 2 and six-component force balance data obtained simultaneously with the present pressure data have been presented in reference 3.

SYMBOLS

All chords are parallel to the vertical plane of symmetry. Wing coefficients include contribution of controls.

b	wing span
b_i	inboard-end location of various controls
c	local basic wing chord
\bar{c}	average basic wing chord
c'	basic wing mean aerodynamic chord
c_d	local chord of deflector rear face
\bar{c}_d	average chord of deflector rear face
c_s	local chord of spoiler front face
\bar{c}_s	average chord of spoiler front face
c_n	wing section normal-force coefficient, $\frac{\text{Wing section normal force}}{qc}$
c_{n_d}	deflector section normal-force coefficient, $\frac{\text{Deflector section normal force}}{qc_d}$
c_{n_s}	spoiler section normal-force coefficient, $\frac{\text{Spoiler section normal force}}{qc_s}$
C_N	wing-panel normal-force coefficient, $\int_{0.135}^{1.0} c_n \frac{c}{c} d\left(\frac{y}{b/2}\right)$
C_p	pressure coefficient, $\frac{p_1 - p}{q}$

d	distance along c_d measured from deflector leading edge
d_{cp}	distance along c_d measured from deflector leading edge to deflector section center of load
M	free-stream Mach number
p	free-stream static pressure
p_1	local static pressure
q	free-stream dynamic pressure
s	distance along c_s measured from spoiler hinge line
s_{cp}	distance along c_s measured from spoiler hinge line to spoiler section center of load
S	total wing area
x	distance along c measured from wing leading edge, positive downstream
x_{cp}	longitudinal location of wing section center of load measured from wing leading edge, positive downstream
X_{cp}	longitudinal location of wing panel center of load measured from leading edge along c' , positive downstream
y	lateral distance from plane of symmetry
Y_{cp}	lateral location of wing panel center of load measured from plane of symmetry
α	angle of attack
δ_d	projection of deflector into airstream, fraction of c , measured from wing surface perpendicular to wing chord line
δ_s	projection of spoiler into airstream, fraction of c , measured from wing surface perpendicular to wing chord line

APPARATUS

Tunnel and Model

The investigation was conducted in the Langley 16-foot transonic tunnel, the airflow and power characteristics of which are presented in reference 4. Figure 1 presents the geometric details of the sting-supported model. The steel wing had a quarter-chord line sweep of 45° , a taper ratio of 0.60, an aspect ratio of 4.0, and NACA 65A006 airfoil sections parallel to the plane of symmetry. The wing had no geometric incidence, dihedral, or twist, and was mounted in a midwing position on the fuselage. The steel fuselage was a body of revolution with a fineness ratio of 10. (See ref. 1 for fuselage ordinates.) The quarter-chord point of the wing mean aerodynamic chord was located at the longitudinal position of the maximum fuselage diameter.

Lateral-Control and Chord-Extension Configurations

Figure 2 shows some of the geometry of the lateral-control and chord-extension configurations used in the test program. More extensive geometric sectional details for the spoiler-slot-deflector aileron and one of the deflector ailerons are shown in figure 1. All the control devices were made of steel and were mounted on the left wing.

Except for modifications involving the removal of small inboard segments, the same spoiler was used for all configurations that included a spoiler. It was projected 7.8 percent of the wing local chord above the wing upper surface and was hinged at the 68.1-percent wing chord line. (See fig. 1(b).) In the undeflected position, this spoiler would extend over 12 percent of the wing chord. When the inboard end of the spoiler extended to the fuselage ($b_i \approx 0.14 \frac{b}{2}$), the fuselage-spoiler juncture was sealed. The wing slot (3.8 percent of the local wing chord) extended from 15 to 87 percent of the semispan. Ribs, which had a height of 2.4 percent of the local wing chord and a width of 0.25 inch, were located in the wing slot parallel to the plane of symmetry at the 20-, 30-, 39-, 48-, 57-, 66-, 75-, and 83-percent-semispan stations. Braces for the flap-type spoiler were mounted on top of these ribs. (See fig. 1(b).)

The deflector of the spoiler-slot-deflector aileron projected 5.5 percent of the local wing chord below the wing lower surface with the inboard end located at the fuselage. In its undeflected position, this deflector would extend over 8.6 percent of the wing chord. The deflector was hinged at the 73.8-percent wing chord line and was fastened to the wing by seven braces. (See fig. 1(b).)

The configurations having only a deflector utilized a deflector that had a chord length of about 7.8 percent of the local wing chord when in the undeflected position. Although this deflector extended inboard to the fuselage, it was not contoured to fit the fuselage closely. For the deflector configurations having a gap between the deflector and wing surface, spacers with a height of 2 percent of the local wing chord and a width of 5/16 inch were located at seven semispan stations. (See fig. 1(c).)

The leading-edge chord-extensions, which are similar to those discussed in references 5 and 6, extended forward 15 percent of the local wing chord from the 65-percent-semispan station to the wing tip. The chord extensions had the same section ordinates back to their maximum thickness as did the basic airfoil sections at corresponding spanwise stations. The airfoil contour paralleled the wing chord line between the maximum thickness of the chord extensions and the maximum thickness of the wing.

Pressure Instrumentation

Static-pressure orifices were distributed parallel to the vertical plane of symmetry at the 13.5- (average), 25-, 40-, 55-, 70-, 85-, and 95-percent-semispan stations on the left wing panel. A total of 45 orifices were located on the upper and lower wing surfaces at each spanwise station. The orifices at the wing-fuselage juncture (average 0.135 semispan station) were actually located on the fuselage 0.1 inch from the wing surface. Orifices located on the front and rear faces of the spoiler (with and without the slot deflector) and the deflector used in the spoiler-slot-deflector combination were only utilized for pressure measurements when the controls extended inboard to the fuselage. These orifices were distributed parallel to the vertical plane of symmetry at the 14-, 25-, 40-, 55-, 70-, and 85-percent wing semispan stations. Eight orifices and six orifices were located on the spoiler and deflector, respectively, at each spanwise station. (See fig. 1(b).) The pressures were transmitted by means of small tubing through the model and its support system to mercury manometer boards.

TESTS

The configurations shown in figure 2 were generally tested through an angle-of-attack range from 0° to approximately 21.5° for Mach numbers from 0.60 to 0.94 and up to angles of 19.4° , 15.4° , and 13.2° for Mach numbers of 0.98, 1.00, and 1.03, respectively. These maximum angles were not attained for all of the configurations because of model stress

or tunnel power limitations. The variation of Reynolds number (based on wing mean aerodynamic chord) with Mach number is presented in figure 3.

DATA REDUCTION

The pressure data were generally reduced by the methods described in reference 1. For some conditions, however, the normally used rectangular-step integration did not provide sufficiently accurate section pitching-moment coefficients for the spoiler or deflector. Hence, for these conditions the moments were obtained by manually integrating the faired pressure distributions for the controls.

No corrections have been applied to the data. Sting interference effects on the flow over the relatively remote wing and lateral controls are believed to be negligible. Also tunnel wall effects are small for the present test Mach number range. (See ref. 7.) Angle-of-attack accuracy is estimated to be within $\pm 0.1^\circ$.

RESULTS

Wing static-pressure coefficients are given for the configuration with the spoiler aileron extending to the fuselage in table I and for the spoiler-slot-deflector aileron configuration in table II. Tabulated basic wing pressure coefficients are available in reference 1.

Wing chordwise pressure distributions showing the effects of lateral control projection or the effects of changes in lateral control configuration are shown in figures 4 to 10. In figure 4 the upper-surface pressure coefficients for the spoiler-aileron configuration are based on both wing pressures and the pressures obtained on the front surface of the spoiler aileron. The pressures on the spoiler aileron rear surface and on the portion of the wing surface over which the spoiler aileron extends were not utilized in obtaining the pressure distributions. These latter spoiler aileron and wing pressures tend to be equal. They have, therefore, been omitted to simplify the pressure distributions. Their values were approximately equal to the upper-surface wing pressures obtained immediately downstream of the spoiler.

For the spoiler-slot-deflector aileron configuration in figure 7, the pressures on the front surface of the spoiler and on the rear surface of the deflector have been utilized in the chordwise pressure distributions. In this case the pressures between the wing and controls have been omitted.

Since control pressures were obtained only for the two configurations where the lateral controls extended into the body, control pressures are not available for the chordwise pressure distributions shown for the remaining control configurations in figures 5, 6, 8, 9, and 10. For these configurations the pressure distributions have not been extended over the control portion of the wing.

Wing spanwise load distributions and wing section center-of-pressure locations are presented in figures 11 and 12, respectively, for the basic model, for the spoiler-slot-deflector configuration, and for the configuration with the spoiler aileron extending to the fuselage. Wing-panel normal-force characteristics, lateral center-of-pressure locations, and longitudinal center-of-pressure locations for these three configurations are presented in figures 13 to 15, respectively.

Control load characteristics are shown in figures 16 to 21 for the spoiler-slot-deflector aileron. These characteristics are with and without the leading-edge chord-extensions added to the model. Figures 16, 18, and 19 also include the control load characteristics for the spoiler aileron that extended to the fuselage.

In figures which show comparisons at one angle of attack, the angle given is an average for the compared configurations. The maximum deviation from this angle is $\pm 0.1^\circ$.

DISCUSSION

Wing Chordwise Pressure Distributions

Spoiler ailerons.- At $M = 0.60$, figure 4(a) shows that at low angles of attack the addition of the spoiler aileron to the model caused an increase in the upper-surface pressures ahead of the spoiler, a decrease in the upper-surface pressures behind the spoiler, and a decrease in the lower-surface pressures. The effects were about the same as those shown in reference 1 with retractable spoilers deflected $0.04c$. In both cases the flow was separated ahead of and behind the spoilers. The primary difference was that, behind the retractable spoiler aileron, complete recovery of the flow generally occurred before the flow reached the wing trailing edge at $\frac{y}{b/2} = 0.135$ and 0.25 . In the present case similar flow recovery was not obtained probably because of the higher deflection of the spoiler aileron above the wing surface.

The region of fairly large pressure rise ahead of the spoiler aileron extended forward of the spoiler in approximately the shape of a

spanwise wedge with the apex and the highest pressure occurring at $\frac{y}{b/2} = 0.135$. The peculiar shape of the pressure distribution upstream of the spoiler-aileron top edge at $\frac{y}{b/2} = 0.25$ was a result of the pressures near the leading-edge front surface of the spoiler aileron being less than either the immediately preceding wing pressures or the following spoiler-aileron pressures. This pressure pattern is believed to be associated with a vortex-type flow lying in the separated region ahead of the spoiler-aileron trailing edge. (See ref. 1.)

As the angle of attack was increased for $M = 0.60$, the effect of the spoiler aileron on the wing pressures did not change appreciably until angles of attack were reached where flow separation on the basic wing began to progress inboard from the wing tip. (See fig. 4(a).) At these angles of attack the spoiler-aileron influence was reduced, as would be expected, since the spoiler aileron was in a separated flow region where the local flow was predominantly spanwise. (A discussion pertaining to flow separation and other flow phenomena over the basic wing may be found in ref. 6.)

Figure 4 shows that, as the Mach number was increased from 0.60 to 1.03, the influence of the spoiler aileron on the lower-surface pressures became confined to the trailing-edge region. This confinement probably occurred because shocks on the lower surface opposed the upstream transmission of pressure changes. On the upper surface, the forward extent of the spoiler-aileron influence became more confined at the inboard stations, and at $\frac{y}{b/2} = 0.135$ some pressure recovery occurred

behind the spoiler aileron at the higher Mach numbers. The confinement on the upper surface was probably caused by the presence of a shock wave associated with the separation point ahead of the spoiler aileron. The presence of this shock would have opposed the upstream transmission of the spoiler pressure field except outboard where the boundary layer was thickened.

Figure 4 also shows some unexplained effects of increasing Mach number from 0.60 at angles of attack of about 8.5° and 13° . At an angle of attack of about 8.5° and Mach numbers of 0.80 and 0.90, the spoiler aileron has much less effect on the upper-surface pressures for a short distance upstream of its trailing edge at $\frac{y}{b/2} = 0.55$ than at $\frac{y}{b/2} = 0.40$ or 0.70. A similar effect occurs at $\frac{y}{b/2} = 0.70$ for the higher Mach numbers. At an angle of attack of about 13° and some of the higher Mach numbers, the addition of the spoiler aileron to the model caused an increase in normal force at the two outboard stations where flow

separation existed on the basic model. With increasing Mach number this normal-force increase reached a maximum at a Mach number of about 0.94 and then decreased to a very low value at a Mach number of 1.00. Although the reasons for the normal-force increase are not known, the relative projection of the spoiler to the boundary-layer thickness must certainly be a factor.

Since inboard lateral controls do not usually extend into the fuselage on actual aircraft, inboard portions of the spoiler aileron were removed to determine the fuselage end-plate effect on the wing pressures. The results for the inboard stations are presented without discussion in figures 5 and 6 for two representative Mach numbers. Since moving the inboard end location to $0.16b/2$ or $0.22b/2$ had negligible effects on the wing pressures at the outboard stations, these pressures are not shown. The effects of inboard end location on the rolling moments were small, as shown in reference 3.

The qualitative effects on rolling-moment effectiveness of the usual pressure changes due to the spoiler ailerons were as follows: The pressure rise on the upper surface ahead of the spoiler-aileron trailing edge and the pressure reduction on the lower surface decreased normal force and thus contributed to a favorable rolling moment. The pressure reduction on the upper surface behind the spoiler aileron, however, was adverse. At higher angles of attack rolling-moment effectiveness decreased because of wing flow separation. The actual rolling-moment coefficients resulting from the pressure changes are available in reference 3.

Spoiler-slot-deflector aileron.- In figure 7 the compression region on the lower surface, which is due to deflector projection, is shown as terminating at about the 70-percent chord line. This chordwise position corresponds to the deflector leading edge. The compression region on the upper surface, which is due to spoiler projection, is shown as terminating at about the 74-percent chord line. This position corresponds to the spoiler trailing edge.

Comparison of figures 4 and 7 shows that adding the slot deflector to the spoiler generally had negligible effects on the upper-surface pressures ahead of the spoiler. Behind the spoiler at the $0.135b/2$ station, the addition generally resulted in more upper-surface pressure recovery for the lower Mach numbers. In many cases, for stations outboard of $0.25b/2$, the effect of the addition was also to reduce slightly the upper-surface pressures behind the spoiler.

The unexplained effects on the pressures which were discussed for a spoiler-aileron configuration at angles of attack of about 8.5° and 13° also occurred for the spoiler-slot-deflector aileron configuration.

As shown in figure 7(a), the lower-surface pressures ahead of the deflector became more positive. Figure 7(a) also shows for a Mach number of 0.60 that on the lower surface behind the deflector there was considerable pressure recovery at the $0.135b/2$ station for all angles of attack and at the $0.25b/2$ station for the higher angles of attack. At the other spanwise stations flow separation apparently existed over the deflector rear surface and the following wing surface for all angles of attack. In the outboard portion of this separated region, the pressures approached those existing on the upper surface behind the spoiler.

The unusual shape of the pressure pattern existing over the deflector rear surface and the downstream wing surface at the $0.25b/2$ station may be associated with a separated vortex-type flow such as that believed to occur ahead of the spoiler and also discussed in reference 1. Increasing angle of attack or Mach number generally had very little effect on the type of flow existing on the lower surface of the spoiler-slot-deflector configuration. (See fig. 7.)

With respect to rolling-moment effectiveness, the most important pressure changes due to the addition of the slot deflector to the spoiler-aileron configuration were on the wing lower surface. The pressure rise on the wing surface ahead of the deflector was detrimental; whereas, the pressure reduction on the deflector rear surface and the following wing surface was beneficial. Unpublished results show that the magnitude of the pressure rise, which corresponded to an increase in normal force, was reduced appreciably by decreasing the deflector chord so that its deflection was $0.04c$ instead of $0.055c$. If the deflector chord is decreased so that its leading edge is at the slot leading edge for the unprojected case, then the pressure rise can probably be eliminated. (See ref. 8.)

At low angles of attack the pressure reduction on the deflector rear surface and the subsequent wing lower surface resulted in increased rolling moment as it caused the normal-force load over the wing trailing-edge region to decrease from a positive to an almost negligible value. The magnitude of the pressure reduction was largest at the inboard stations. At high angles of attack the pressure reduction resulted in the wing trailing-edge loading decreasing considerably over the entire span of the aileron.

Deflector ailerons. - One of the problems often associated with spoiler-slot-deflector ailerons is that the yawing moments due to control projection, although favorable, are larger than desirable. A possible method of reducing these yawing moments is to project the deflector part of the control on one wing panel simultaneously with projection of the complete control on the opposite wing panel. A shortcoming in this solution, however, is that the roll contribution of the deflector part of the control would probably be reversed at moderately high angles

of attack. The effects on this reversal problem of deflector projection and of a gap between the deflector trailing edge and the wing surface were, therefore, studied in a brief investigation.

Results presented in figure 8 at a Mach number of 0.90 for a low deflector projection ($\delta_d = 0.016c$) show an expected favorable pressure rise on the lower wing surface ahead of the deflector. On the lower wing surface behind the deflector the pressures were decreased by adding the deflector. The flow in this region, however, was not generally separated. As the angle of attack was increased, the region of pressure reduction became larger at the outboard stations and the region of pressure rise tended to become smaller. These pressure changes resulted in rolling-moment reversal at moderate angles of attack as shown by the force data of reference 3.

As would be expected from the chordwise pressure distributions of figure 9, the force data presented in reference 3 also show reversals in rolling-moment effectiveness at moderately high angles of attack when the deflector projection was increased to $0.045c$. The effects on the wing pressures of adding a $0.02c$ gap between the deflector trailing edge and the wing (see fig. 10) were also too small to prevent reversals in rolling-moment effectiveness at the moderately high angles of attack. (See ref. 3.)

Wing-Section Loading

At $M = 0.60$ for low angles of attack, figure 11(a) shows that the addition of the spoiler aileron to the model caused the largest reductions in the semispan load distributions between 40 and 80 percent of the wing semispan. Apparently, the large reduction in pressure which occurred on the wing surface behind the spoiler caused considerable reduction in control effectiveness inboard of $\frac{y}{b/2} = 0.40$. (See fig. 4.) Increasing Mach number caused the region of maximum load change at low angles to shift outboard to about $\frac{y}{b/2} = 0.80$. (See fig. 11.) As indicated in the discussion of the chordwise pressure distributions, the reduction in loading due to the spoiler aileron was much smaller at high angles of attack than at the low angles.

For low angles of attack, figure 11 shows that adding the spoiler-slot-deflector aileron to the model caused larger decreases in loading at the inboard stations and smaller decreases in loading at the outboard stations than did the spoiler aileron. At high angles of attack, the spoiler-slot-deflector aileron caused a reduction in the outboard loading which was as large or larger than that caused by the spoiler aileron.

Reasons for these reductions were indicated in the pressure-distribution discussion of figure 7.

At several Mach numbers and angles near 13° , figure 11 also shows load increases over about the outboard 20 percent of the wing semispan that were caused by adding either lateral control to the model. These increases were also considered in the discussion of figures 4 and 7.

Figure 12 shows that, below an angle of attack of about 8° , addition of the spoiler aileron to the model usually caused the section centers of pressure to shift forward at stations outboard of $\frac{y}{b/2} = 0.25$.

This shift occurred because the wing loading behind the spoiler aileron became progressively less across the wing semispan. (See fig. 4.) At high angles of attack where its effectiveness was considerably reduced the addition of the spoiler aileron had only small effects on the section center-of-pressure locations.

Addition of the spoiler-slot-deflector aileron to the model generally caused a larger forward shift of the section centers of pressure than did the addition of the spoiler aileron. This shift generally occurred at all spanwise stations for all angles of attack except those near 0° . The larger forward shift resulted primarily because the addition of the slot deflector greatly reduced the trailing-edge loading.

Wing-Panel Loading

The effects of the spoiler aileron and spoiler-slot-deflector aileron on the wing-panel loading characteristics are analogous to their effects on the section pressures and loadings. These characteristics (see figs. 13 to 15) are, therefore, presented without discussion.

Spoiler and Deflector Pressure Distributions

The pressure distributions of figure 16 show that the pressures on the front of the spoiler decreased from positive values at low angles of attack to negative values at high angles. This trend seemed to be influenced mainly by wing flow separation spreading inboard from the wing tip as angle of attack was increased. The chord-extension effects on wing flow separation (see ref. 6) delayed the trend in the spoiler pressures at the outboard station for angles of attack of about 12° to 17° . With this exception, figure 16 shows very little effect of configuration on the pressures over the front of the spoiler.

At low angles of attack the spoiler front-face pressures at the inboard stations were apparently influenced considerably by circulation in the separated flow upstream of the spoiler trailing edge. As indicated in the discussion of figure 4, this circulatory flow probably resulted in the pressures in the region of the spoiler base being less than the immediately preceding wing pressures or the pressures at about the midchord of the spoiler. A somewhat similar pressure pattern is shown in reference 1 for a spoiler which essentially extended perpendicular to the wing surface. This reference shows, however, that the two high-pressure regions occurred at the top and bottom of the spoiler front face. This effect of spoiler deflection angle on the pressures also agrees with results obtained at supersonic speeds. (See ref. 9.) It seems probable that the position of the high-pressure region nearest the upper edge of the spoilers gives an indication of the height of the separated flow.

Figure 16 also shows that the negative pressures on the rear face of the spoiler were affected by the addition of the slot deflector; that is, flow through the slot caused the pressures nearest to the wing to increase.

The pressure distributions for the deflector, as shown in figure 17, were only slightly affected by angle-of-attack changes. Also, adding the chord-extensions had very little effect on these distributions. The positive pressures on the front of the deflector usually decreased as the distance to the wing became less probably because of flow through the slot. On the rear face of the deflector the pressure variations shown in figure 17 at the two inboard stations were probably caused by the separated flow phenomena indicated in the discussion of figure 7.

Spoiler and Deflector Spanwise Loading

Some of the results for the spoiler and deflector span-load distributions shown in figures 18 and 19, respectively, have been summarized in reference 2. This reference shows that at low angles of attack the shape of the load distributions was similar for the deflector and the spoiler (with and without the slot deflector added to the model). It also indicates that Mach number had very little effect on the shape of the span-load distributions.

The large loads at the inboard stations of the spoiler and deflector (see figs. 18 and 19) occurred because the typical effects of sweep did not exist at these stations. Similar results have also been found at supersonic speeds. (See ref. 9.) The spoiler loads decreased with increasing angle of attack primarily because wing-flow separation caused the spoiler front-face pressures to decrease. (See fig. 16.) The effect of the flow through the slot was to reduce the spoiler loads at all angles

of attack. The chord-extensions caused a slight increase in the spoiler loading at the outboard stations for angles of attack from about 12° to 17° . Chord-extension effects on deflector loading, however, were negligible. The unexplained flow phenomena noted at an angle of attack of 8.5° in the discussion of the chordwise pressure distributions of figures 4 and 7 resulted in decreased spoiler loads as shown in figure 18.

With regard to total control normal force, reference 2 indicates that, for large control deflections, tests need only be made at low angles of attack to establish the maximum control loads; that is, the spoiler loads decreased and the deflector loads varied only slightly with increasing angle of attack. Increasing Mach number caused small increases in the total control normal-force coefficients (ref. 2).

Although no methods are known for estimating the control loads, references 10 and 1 show that, for retractable spoiler ailerons, spoiler load at any spanwise position can be determined by measurement at the wing surface of the pressure drop across the spoiler. Studies utilizing the wing orifices at $0.65c$ and $0.75c$ indicate that this statement also generally applies to the spoiler aileron of this paper. It does not apply, however, in the case of the spoiler and deflector of a spoiler-slot-deflector aileron. In these cases, flow through the wing slot, as previously indicated, has a large effect on the control pressure distributions. This effect will probably vary with control geometry.

In figure 20, spoiler center-of-pressure locations for the spoiler configuration have been omitted in many cases at the higher angles of attack because the spoiler loads became very small. For the spoiler (without the slot deflector on the model) or the deflector (see figs. 20 and 21, respectively), the section center-of-pressure locations generally varied only slightly across the control span. In both cases these center-of-pressure locations were usually slightly ahead of the control 50-percent chord line.

The addition of the slot deflector to the spoiler configuration, however, caused large rearward shifts in the spoiler center-of-pressure locations below an angle of attack of about 8° (see fig. 20); these shifts were caused by the flow through the wing slot. (See fig. 16.) At higher angles where the spoiler loads were small, inconsistent shifts were caused by both wing flow separation and the flow through the wing slot.

Adding the chord-extensions to the spoiler-slot-deflector configurations had a small effect on the spoiler center-of-pressure locations except for angles of attack where the spoiler loads became small. In the case of the deflector, however, the effect was negligible through the angle-of-attack and Mach number range.

CONCLUSIONS

An investigation was made with spoiler ailerons, deflector ailerons, and a spoiler-slot-deflector aileron mounted on a 45° swept-wing-fuselage combination. These controls were located in the vicinity of the 70-percent wing chord line and extended outboard to 87 percent of the wing semispan. Pressures were measured on the wing and some of the control components at several spanwise stations for Mach numbers from 0.60 to 1.03 and for an angle-of-attack range of about 0° to 20°. This investigation is a closely related extension of a previously reported study of retractable spoiler ailerons (NACA RM L54C17a).

1. Effects of flap spoiler ailerons and retractable spoiler ailerons on the wing pressures are generally similar.
2. Adding a slot deflector to a spoiler-aileron configuration creates low pressures on the rear surface of the deflector and the subsequent wing lower surface, which have a favorable effect on rolling moment, particularly at high angles of attack.
3. At low angles of attack the low pressures associated with a slot deflector have their largest effects on the spanwise and chordwise loadings for a swept wing at the inboard stations. These low pressures tend to be of the same magnitude as the upper-surface pressures behind a spoiler which are lowest at the inboard stations.
4. At high angles of attack where a spoiler has a rather small effect on wing loading, the low pressures associated with a slot deflector affect the loadings over most of the wing span.
5. Flow through the wing slot had a large effect on the chordwise loadings of the spoiler and deflector, but at low angles of attack the spanwise load distributions were similar in shape for the spoiler aileron and the spoiler and deflector components of the spoiler-slot-deflector aileron.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., September 12, 1957.

REFERENCES

1. Hallissy, Joseph M., Jr., West, F. E., Jr., and Liner, George: Effects of Spoiler Ailerons on the Aerodynamic Load Distribution Over a 45° Sweptback Wing at Mach Numbers From 0.60 to 1.03. NACA RM L54C17a, 1954.
2. West, F. E., Jr., and Czarnecki, K. R.: Loads Due to Controls at Transonic and Low Supersonic Speeds. NACA RM L57D26a, 1957.
3. West, F. E., Jr., Whitcomb, Charles F., and Schmeer, James W.: Investigation of Spoiler-Slot-Deflector Ailerons and Other Spoiler Ailerons on a 45° Sweptback-Wing-Fuselage Combination at Mach Numbers From 0.60 to 1.03. NACA RM L56F15, 1956.
4. Ward, Vernon G., Whitcomb, Charles F., and Pearson, Merwin D.: Air-Flow and Power Characteristics of the Langley 16-Foot Transonic Tunnel With Slotted Test Section. NACA RM L52E01, 1952.
5. West, F. E., Jr., Liner, George, and Martz, Gladys S.: Effect of Leading-Edge Chord-Extensions on the Aerodynamic Characteristics of a 45° Sweptback Wing-Fuselage Combination at Mach Numbers of 0.40 to 1.03. NACA RM L53B02, 1953.
6. West, F. E., Jr., and Henderson, James H.: Relationship of Flow Over a 45° Sweptback Wing With and Without Leading-Edge Chord-Extensions to Longitudinal Stability Characteristics at Mach Numbers From 0.60 to 1.03. NACA RM L53H18b, 1953.
7. Whitcomb, Charles F., and Osborne, Robert S.: An Experimental Investigation of Boundary Interference on Force and Moment Characteristics of Lifting Models in the Langley 16- and 8-Foot Transonic Tunnels. NACA RM L52L29, 1953.
8. Hammond, Alexander D.: Low-Speed Pressure Distribution Investigation of a Spoiler and a Spoiler-Slot-Deflector on a 30° Sweptback Wing-Fuselage Model Having an Aspect Ratio of 3, a Taper Ratio of 0.5, and NACA 65A004 Airfoil Section. NACA RM L55I29, 1956.
9. Lord, Douglas R., and Czarnecki, K. R.: Aerodynamic Loadings Associated With Swept and Unswept Spoilers on a Flat Plate at Mach Numbers of 1.61 and 2.01. NACA RM L55L12, 1956.
10. Hammond, Alexander D., and West, F. E., Jr.: Loads Due to Flaps and Spoilers on Sweptback Wings at Subsonic and Transonic Speeds. NACA RM L53D29a, 1953.

TABLE I. - WING WITH SPOILER

Percent chord		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.60 \quad \alpha = 0.26^\circ$															
Upper surface															
0.0	+194	+459	+447	+415	+252	+344	+110	+187	+042	+121	+101	+248	+222	+391	
1.2	+116	+007	+081	+191	+232	+283	+254	+189	+793	+818	+700	+630	+522	+443	
2.4	+079	+046	+043	+108	+173	+198	+157	+206	+654	+509	+444	+350	+317	+308	
5.0	+031	+035	+029	+084	+126	+140	+094	+224	+373	+336	+269	+230	+207	+208	
7.5	+007	+041	+029	+065	+097	+114	+048	+224	+305	+257	+222	+197	+163	+171	
10.0	+016	+048	+013	+051	+074	+086	+021	+229	+286	+229	+171	+137	+128	+113	
15.0	+029	+048	+013	+046	+067	+077	+004	+216	+214	+147	+147	+074	+043	+072	
19.6	+036	+048	+006	+056	+105	+011	+011	+197	+214	+166	+166	+059	+046	+064	
24.5	+055	+025	+049	+063	+084	+076	+023	+193	+224	+180	+185	+192	+163	+032	
29.5	+020	+006	+066	+078	+093	+083	+023	+147	+158	+085	+009	+020	+027	+028	
34.5	+040	+021	+089	+109	+117	+092	+023	+126	+020	+064	+072	+081	+070	+030	
39.5	+026	+062	+122	+136	+148	+110	+026	+128	+128	+129	+132	+135	+116	+024	
44.5	+018	+130	+164	+177	+184	+135	+013	+111	+069	+129	+132	+135	+116	+024	
49.5	+017	+211	+229	+226	+232	+171	+056	+175	+180	+185	+192	+163	+032		
54.5	+082	+309	+293	+295	+296	+223	+074	+005	+264	+205	+213	+224	+184	+040	
59.5	+200	+374	+321	+345	+326	+249	+084	+129	+277	+196	+221	+234	+175	+047	
64.5	+348	+353	+317	+360	+325	+228	+093	+313	+067	+183	+225	+230	+150	+067	
69.6	+741						+096	+716							
74.6	+760	+742	+426	+366	+311	+223	+115	+734	+726	+443	+372	+303	+214	+036	
79.5	+760	+734	+426	+356	+309	+231	+120	+738	+719	+439	+358	+305	+223	+086	
84.6	+697	+735	+455	+371	+299	+222	+126	+683	+719	+459	+372	+297	+215	+091	
89.6	+734	+722	+476	+372	+295	+205	+130	+725	+719	+483	+379	+291	+205	+091	
94.6	+741	+619	+464	+369	+296	+195	+118	+748	+656	+465	+374	+293	+200	+082	
Lower surface															
1.3	+138	+073	+151	+271	+420	+517	+431	+313	+370	+365	+336	+292	+266	+205	
5.0	+092	+093	+151	+252	+288	+325	+327	+269	+302	+276	+243	+199	+199	+128	
7.6	+003	+079	+122	+172	+224	+263	+286	+234	+198	+193	+163	+140	+110	+070	
10.1	+020	+098	+125	+175	+212	+249	+250	+196	+149	+125	+104	+082	+061	+019	
15.1	+056	+120	+165	+199	+228	+244	+235	+166	+109	+094	+060	+041	+043	+025	
19.6	+088	+124	+179	+188	+252	+245	+203	+214	+066	+052	+021	+004	+016	+075	
24.5	+088	+159	+201	+219	+242	+250	+214	+194	+031	+038	+006	+001	+044	+085	
29.5	+114	+172	+212	+230	+251	+250	+194	+190	+001	+064	+100	+110	+118	+110	
34.5	+129	+190	+238	+246	+256	+243	+190	+190	+001	+064	+100	+110	+118	+117	
39.5	+146	+204	+252	+247	+260	+247	+194	+202	+086	+128	+122	+126	+140	+130	
44.5	+146	+220	+251	+259	+265	+247	+194	+202	+052	+112	+140	+145	+146	+141	
49.5	+176	+233	+262	+272	+226	+250	+194	+207	+067	+136	+152	+170	+195	+172	
54.5	+194	+217	+270	+267	+269	+244	+185	+205	+085	+126	+181	+177	+176	+135	
59.5	+194	+232	+263	+252	+257	+228	+164	+197	+149	+178	+183	+173	+129		
64.5	+194	+233	+256	+263	+249	+214	+154	+160	+183	+190	+179	+171	+123		
69.5	+186														
74.6	+185	+236	+250	+228	+213	+194	+148	+127	+185	+194	+176	+165	+160	+119	
79.5	+237	+230	+252	+232	+213	+173	+126	+165	+193	+214	+189	+174	+150	+096	
84.6	+110	+253	+243	+226	+210	+180	+119	+072	+223	+213	+191	+187	+163	+082	
89.7	+231	+285	+265	+226	+214	+156	+111	+178	+263	+249	+198	+185	+137	+074	
94.6	+284	+336	+298	+244	+230	+150	+049	+265	+319	+291	+229	+209	+125	+053	
Upper surface															
1.3	+164	+380	+203	+529	+351	+467	+056	+125	+705	+565	+1031	+713	+535	+136	
5.0	+373	+1071	+733	+1011	+1290	+1076	+054	+596	+1045	+820	+1089	+993	+634	+636	
7.6	+386	+4975	+673	+759	+666	+642	+650	+426	+615	+601	+007	+788	+1021	+922	
10.0	+387	+723	+601	+532	+479	+440	+442	+591	+913	+748	+1002	+787	+547	+551	
15.0	+364	+4588	+523	+426	+378	+345	+341	+551	+852	+686	+791	+736	+494	+519	
19.6	+359	+506	+472	+351	+319	+285	+276	+524	+832	+641	+558	+631	+462	+485	
24.5	+333	+404	+361	+257	+226	+195	+183	+466	+738	+562	+345	+395	+408	+466	
29.5	+285	+332	+294	+175	+175	+122	+121	+291	+606	+511	+275	+277	+371	+399	
34.5	+285	+266	+211	+143	+110	+049	+065	+315	+333	+335	+221	+176	+289	+207	
39.5	+228	+206	+146	+103	+066	+087	+003	+045	+239	+261	+175	+143	+238	+144	
44.5	+150	+085	+060	+036	+038	+039	+034	+271	+156	+193	+140	+099	+182	+104	
49.5	+172	+017	+022	+042	+077	+070	+027	+243	+072	+125	+115	+059	+122	+078	
54.5	+032	+115	+068	+090	+118	+102	+036	+182	+011	+071	+092	+025	+063	+078	
59.5	+081	+207	+120	+128	+149	+124	+046	+092	+112	+025	+068	+005	+011	+084	
64.5	+289	+226	+143	+148	+148	+121	+055	+050	+149	+003	+049	+014	+027	+093	
69.6	+703	+000	+036	+043	+042	+053	+076	+240	+035	+030	+021	+028	+041	+109	
74.6	+718	+699	+447	+362	+288	+217	+100	+719	+676	+436	+344	+276	+195	+130	
79.5	+723	+690	+446	+349	+285	+210	+107	+719	+672	+433	+336	+275	+207	+131	
84.6	+668	+703	+458	+368	+273	+203	+119	+671	+700	+451	+356	+274	+197	+131	
89.6	+716	+712	+483	+376	+276	+219	+119	+718	+708	+477	+370	+278	+188	+128	
94.6	+657	+459	+371	+283	+190	+107	+742	+641	+446	+366	+286	+183	+122		
Lower surface															
1.3	+369	+450	+442	+443	+432	+425	+359	+400	+491	+474	+476	+470	+456	+404	
5.0	+320	+407	+374	+359	+351	+348	+280	+343	+445	+445	+435	+426	+414	+342	
7.6	+286	+298	+291	+280	+274	+263	+200	+362	+386	+370	+365	+364	+329	+276	
10.1	+261	+249	+224	+211	+197	+131	+363	+334	+302	+299	+302	+273	+228		
15.1	+261	+205	+193	+170	+162	+156	+077	+350	+289	+275	+263	+263	+227	+146	
19.6	+162	+109	+083	+084	+045	+007	+009	+247	+180	+158	+159	+186	+164	+057	
24.5	+137	+061	+040	+027	+026	+005	+066	+217	+133	+112	+103	+103	+080	+028	
29.5	+101	+028	+007	+010	+009	+030	+077	+175	+097	+055	+053	+03	+036	+038	
34.5	+071	+000	+036	+043	+042	+053	+090	+140	+066	+030	+025	+025	+029	+007	
39.5	+039	+029	+062	+065	+061	+084	+102	+102	+034	+007	+005	+005	+025	+066	
44.5	+009	+057	+083	+086	+087	+103	+114	+070	+005	+023	+025	+025	+053	+080	
49.5	+015	+083	+101	+111	+152	+120	+119	+036	+025	+047	+056	+103	+080	+090	
54.5	+027	+077	+126	+127	+127	+140	+117	+036	+025	+076	+070	+074	+097	+085	
59.5	+049	+107	+134	+148	+131	+137	+113	+004	+055	+086	+072	+080	+096	+082	
64.5	+059	+123	+146	+152	+140	+142	+105	+010	+071	+099	+103	+093	+107	+079	
69.5	+074														
74.6	+094	+150	+166	+149	+134	+139	+105	+066	+107	+126	+106	+096	+113	+084	
79.6	+131	+173	+184	+163	+146	+128	+083	+083	+087	+145	+122	+108	+106	+068	
84.6	+058	+202	+193	+169	+146	+147	+076	+040	+161	+156	+134	+117	+129	+067	
89.7	+148	+243	+232	+178	+171	+123	+072	+118	+209	+191	+148	+142	+110	+067	
94.6	+244	+304	+273	+212	+189	+063	+216	+216	+270	+230	+				

TABLE I. - WING WITH SPOILER - Continued

Percent chord		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	
		M = 0.60	a = 12.32°	M = 0.60											
Upper surface	0.0	-0.018	-1.717	-1.193	-0.951	-0.656	-0.552	-0.211	-0.230	-2.098	-1.292	-0.725	-0.651	-0.516	-0.404
	1.2	-1.110	-1.631	-1.010	-0.736	-0.560	-0.473	-0.312	-1.863	-1.917	-1.088	-0.673	-0.508	-0.427	-0.378
	2.4	-1.156	-1.661	-1.014	-0.747	-0.589	-0.525	-0.308	-1.946	-1.908	-1.091	-0.676	-0.524	-0.432	-0.381
	5.0	-1.531	-1.756	-1.024	-0.787	-0.649	-0.430	-0.308	-2.099	-1.890	-1.100	-0.668	-0.535	-0.439	-0.381
	7.5	-1.120	-1.756	-1.015	-0.778	-0.664	-0.402	-0.312	-1.710	-1.916	-1.084	-0.670	-0.543	-0.445	-0.381
	10.0	-0.965	-1.844	-1.031	-0.794	-0.573	-0.395	-0.310	-1.338	-1.974	-1.088	-0.668	-0.546	-0.449	-0.381
	15.0	-0.99	-1.791	-1.043	-0.749	-0.438	-0.406	-0.310	-1.165	-1.955	-1.068	-0.673	-0.553	-0.452	-0.381
	19.6	-0.676	-1.583	-1.150	-0.737	-0.405	-0.411	-0.325	-0.985	-1.908	-1.092	-0.680	-0.556	-0.456	-0.384
	24.0	-0.617	-1.014	-1.160	-0.743	-0.405	-0.423	-0.328	-0.825	-1.772	-1.080	-0.694	-0.552	-0.454	-0.386
	29.5	-0.525	-0.329	-1.175	-0.768	-0.411	-0.430	-0.325	-0.718	-1.568	-1.094	-0.700	-0.552	-0.455	-0.390
	34.5	-0.489	-0.143	-1.148	-0.802	-0.419	-0.429	-0.321	-0.675	-1.278	-1.095	-0.709	-0.544	-0.454	-0.394
	39.5	-0.418	-0.109	-1.078	-0.836	-0.440	-0.421	-0.312	-0.587	-0.946	-1.088	-0.713	-0.533	-0.448	-0.396
	44.5	-0.362	-0.060	-0.942	-0.831	-0.472	-0.413	-0.302	-0.531	-0.647	-1.045	-0.597	-0.522	-0.443	-0.393
	49.5	-0.277	0.019	-0.746	-0.810	-0.486	-0.397	-0.290	-0.425	-0.436	-0.985	-0.675	-0.502	-0.436	-0.392
	54.5	-0.148	0.129	-0.545	-0.758	-0.489	-0.380	-0.271	-0.305	-0.271	-0.898	-0.655	-0.489	-0.433	-0.386
	59.5	-0.030	0.184	-0.414	-0.693	-0.479	-0.359	-0.256	-0.212	-0.210	-0.820	-0.639	-0.478	-0.429	-0.39
	64.5	0.068	0.164	-0.326	-0.608	-0.465	-0.334	-0.248	-0.143	-0.267	-0.763	-0.634	-0.466	-0.426	-0.368
	69.6	-0.713													-0.341
	74.6	-0.708	-0.671	-0.377	-0.430	-0.348	-0.282	-0.229	-0.927	-0.840	-0.602	-0.470	-0.433	-0.371	-0.342
	79.5	-0.682	-0.665	-0.384	-0.445	-0.338	-0.285	-0.225	-0.907	-0.825	-0.597	-0.469	-0.436	-0.378	-0.331
	84.6	-0.656	-0.656	-0.424	-0.527	-0.355	-0.278	-0.219	-0.745	-0.836	-0.652	-0.501	-0.417	-0.379	-0.314
	89.6	-0.708	-0.677	-0.447	-0.505	-0.360	-0.262	-0.211	-0.845	-0.828	-0.617	-0.512	-0.428	-0.369	-0.298
	94.6	-0.723	-0.641	-0.426	-0.418	-0.355	-0.253	-0.205	-0.827	-0.655	-0.542	-0.509	-0.426	-0.354	-0.278
Lower surface	1.3	0.414	0.463	0.475	0.478	0.490	0.483	0.422	0.356	0.374	0.423	0.442	0.454	0.445	0.411
	2.6	0.398	0.530	0.525	0.514	0.503	0.467	0.398	0.396	0.524	0.539	0.531	0.505	0.475	0.418
	5.0	0.429	0.518	0.497	0.493	0.455	0.419	0.347	0.439	0.587	0.559	0.535	0.499	0.443	0.382
	7.6	0.496	0.479	0.439	0.430	0.406	0.373	0.279	0.594	0.546	0.323	0.409	0.455	0.410	0.325
	10.1	0.514	0.434	0.418	0.391	0.362	0.328	0.238	0.638	0.539	0.311	0.462	0.425	0.374	0.279
	15.1	0.468	0.367	0.335	0.317	0.296	0.251	0.128	0.576	0.474	0.32	0.387	0.356	0.298	0.172
	19.6	0.402	0.315	0.284	0.281	0.232	0.194	0.073	0.504	0.418	0.378	0.344	0.308	0.237	0.116
	24.5	0.364	0.267	0.236	0.224	0.205	0.160	0.032	0.464	0.372	0.331	0.298	0.266	0.200	0.054
	29.5	0.310	0.226	0.194	0.180	0.157	0.116	0.005	0.408	0.333	0.289	0.254	0.216	0.156	0.025
	34.5	0.270	0.189	0.143	0.142	0.121	0.075	0.000	0.365	0.289	0.243	0.211	0.174	0.110	-0.014
	39.5	0.231	0.155	0.111	0.108	0.091	0.037	0.045	0.322	0.252	0.199	0.173	0.140	0.085	-0.040
	44.5	0.190	0.116	0.084	0.076	0.059	0.000	0.068	0.287	0.214	0.177	0.133	0.099	0.023	-0.069
	49.5	0.155	0.053	0.052	0.046	0.004	0.029	0.086	0.246	0.177	0.140	0.100	0.040	-0.017	-0.088
	54.5	0.152	0.080	0.021	0.026	0.000	0.052	0.094	0.234	0.169	0.104	0.066	0.023	-0.049	-0.099
	59.5	0.113	0.043	-0.001	-0.010	-0.020	0.073	0.098	0.191	0.129	0.080	0.028	0.000	-0.073	-0.110
	64.5	0.084	0.020	-0.188	-0.1010	-0.046	-0.086	-0.105	0.166	0.101	0.057	0.003	-0.029	-0.100	-0.131
	69.5	0.053							-0.111	-0.123					
Upper surface	74.6	0.004	-0.027	-0.053	-0.022	-0.068	-0.110	-0.128	-0.067	-0.050	-0.010	-0.040	-0.083	-0.134	-0.152
	79.5	0.040	-0.047	-0.079	-0.041	-0.091	-0.114	-0.114	-0.08	-0.021	-0.023	-0.079	-0.112	-0.144	-0.141
	84.6	-0.004	-0.100	-0.094	-0.058	-0.036	-0.147	-0.127	-0.036	-0.017	-0.047	-0.117	-0.139	-0.189	-0.156
	89.7	-0.029	-0.152	-0.122	-0.078	-0.148	-0.133	-0.128	-0.014	-0.068	-0.104	-0.165	-0.183	-0.180	-0.161
	94.6	-0.144	-0.230	-0.166	-0.123	-0.195	-0.150	-0.140	-0.048	-0.136	-0.176	-0.232	-0.239	-0.200	-0.153
Lower surface	1.3	0.241	0.498	0.509	0.481	0.383	0.411	0.048	0.223	0.176	0.197	0.155	0.245	0.236	0.398
	2.6	0.290	-0.008	0.057	0.182	0.227	0.284	0.277	0.013	-0.103	-1.036	-0.871	-0.788	-0.667	-0.542
	5.0	0.062	-0.028	0.027	0.090	0.133	0.149	0.114	0.208	-0.160	-0.576	-0.586	-0.493	-0.401	-0.376
	7.5	0.033	-0.034	0.046	0.073	0.104	0.119	0.071	0.221	-0.333	-0.303	-0.245	-0.230	-0.195	-0.123
	10.0	0.003	0.045	0.025	0.062	0.086	0.099	0.034	0.234	-0.318	-0.264	-0.206	-0.197	-0.174	-0.139
	15.0	-0.013	-0.039	0.030	0.062	0.081	0.094	0.000	0.234	-0.281	-0.183	-0.148	-0.139	-0.127	-0.126
	19.6	-0.021	0.041	0.033	0.107	0.107	0.121	0.061	0.215	-0.246	-0.154	-0.085	-0.080	-0.060	-0.081
	24.5	-0.042	0.012	0.086	0.094	0.107	0.094	-0.005	0.225	-0.287	-0.107	-0.060	-0.045	-0.048	-0.067
	29.5	-0.006	0.010	0.103	0.103	0.118	0.099	-0.005	0.172	-0.143	-0.021	-0.023	-0.013	-0.016	-0.023
	34.5	-0.031	0.048	0.128	0.126	0.142	0.112	-0.011	0.182	-0.101	-0.033	-0.027	-0.027	-0.031	-0.0107
	39.5	-0.013	0.104	0.164	0.158	0.174	0.133	-0.015	0.155	-0.004	-0.080	-0.092	-0.082	-0.021	-0.021
	44.5	-0.003	0.180	0.213	0.201	0.214	0.161	-0.029	0.134	-0.029	-0.050	-0.035	-0.035	-0.008	-0.021
	49.5	0.038	0.276	0.275	0.263	0.201	0.155	-0.076	0.234	-0.181	-0.175	0.169	0.167	-0.035	-0.021
	54.5	0.120	0.387	0.326	0.324	0.247	0.147	-0.077	0.16	-0.168	-0.199	0.210	0.171	-0.054	-0.024
	59.5	0.255	0.423	0.332	0.360	0.252	0.159	-0.086	0.157	-0.171	-0.208	0.218	0.162	-0.073	-0.023
	64.5	0.444	0.367	0.320	0.371	0.340	0.231	-0.109	0.339	-0.054	-0.161	0.215	0.207	-0.097	-0.0207
	69.6	-0.736													-0.107
	74.6	-0.728	-0.705	-0.448	-0.371	-0.304	-0.252	-0.125	-0.717	-0.491	-0.384	-0.330	-0.247	-0.116	
	79.5	-0.710	-0.712	-0.443	-0.357	-0.300	-0.235	-0.133	-0.705	-0.491	-0.374	-0.326	-0.249	-0.118	
	84.6	-0.668	-0.658	-0.457	-0.364	-0.290	-0.226	-0.149	-0.673	-0.498	-0.385	-0.319	-0.243	-0.125	
	89.6	-0.677	-0.689	-0.485	-0.371	-0.291	-0.210	-0.147	-0.693	-0.504	-0.394	-0.319	-0.236	-0.125	
	94.6	-0.667	-0.640	-0.475	-0.366	-0.290	-0.199	-0.128	-0.710	-0.505	-0.394	-0.318	-0.229	-0.114	
Lower surface	1.3	0.184	-0.012	-0.078	-0.203	-0.374	-0.524	-0.468	-0.328	-0.380	-0.377	-0.369	-0.318	-0.287	-0.211
	2.6	0.147	-0.038	-0.091	-0.19										

TABLE I. - WING WITH SPOILER -Continued

		Pressure coefficient														
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2		
Percent chord		M = 0.80						M = 0.80								
M	α	-0.053	-0.015	-0.198	-0.044	-0.154	-0.172	-0.210	-0.318	-0.371	-0.489	-0.370	-0.443	-0.049		
0.0		*228	-0.053	-0.015	-0.198	-0.044	-0.154	-0.172	-0.210	-0.318	-0.371	-0.489	-0.370	-0.443	-0.049	
1.2		*075	-0.983	-0.744	-0.974	-0.954	-0.949	-0.880	-0.165	-1.091	-1.044	-0.891	-0.930	-0.826	-0.803	
2.4		*307	-0.903	-0.681	-0.769	-0.738	-0.701	-0.715	-0.494	-1.047	-1.004	-0.859	-0.883	-0.822	-0.478	
5.0		*355	-0.730	-0.611	-0.583	-0.567	-0.550	-0.557	-0.694	-0.946	-0.988	-0.971	-0.813	-0.761	-0.652	
7.5		*357	-0.561	-0.489	-0.410	-0.408	-0.362	-0.351	-0.454	-0.542	-0.884	-0.747	-0.676	-0.856	-0.651	
10.0		*357	-0.313	-0.257	-0.257	-0.262	-0.256	-0.188	-0.493	-0.876	-0.973	-0.436	-0.806	-0.593	-0.394	
15.0		*344	-0.448	-0.403	-0.257	-0.262	-0.256	-0.188	-0.493	-0.810	-0.980	-0.297	-0.675	-0.513	-0.353	
19.6		*313	-0.367	-0.335	-0.160	-0.173	-0.161	-0.112	-0.428	-0.810	-0.980	-0.297	-0.675	-0.513	-0.323	
24.5		*318	-0.295	-0.231	-0.133	-0.114	-0.115	-0.087	-0.416	-0.517	-0.810	-0.251	-0.238	-0.449	-0.287	
29.5		*251	-0.221	-0.164	-0.104	-0.077	-0.074	-0.065	-0.446	-0.446	-0.427	-0.241	-0.209	-0.392	-0.259	
34.5		*260	-0.141	-0.098	-0.068	-0.039	-0.029	-0.022	-0.342	-0.129	-0.072	-0.239	-0.138	-0.336	-0.244	
39.5		*221	-0.051	-0.040	-0.025	-0.003	-0.011	-0.004	-0.294	-0.034	-0.085	-0.261	-0.114	-0.192	-0.223	
44.5		*193	-0.049	-0.016	-0.012	-0.011	-0.004	-0.004	-0.209	-0.064	-0.103	-0.287	-0.092	-0.119	-0.222	
49.5		*127	-0.158	-0.067	-0.035	-0.080	-0.002	-0.002	-0.079	-0.082	-0.162	-0.302	-0.068	-0.048	-0.221	
54.5		*022	-0.255	-0.112	-0.063	-0.109	-0.09	-0.079	-0.079	-0.162	-0.302	-0.068	-0.048	-0.221		
59.5		*125	-0.260	-0.130	-0.083	-0.132	-0.115	-0.091	-0.089	-0.157	-0.315	-0.063	-0.043	-0.211		
64.5		*328	-0.105	-0.145	-0.107	-0.145	-0.107	-0.113	-0.235	-0.048	-0.157	-0.306	-0.019	-0.036	-0.215	
69.5		*697	-0.688	-0.483	-0.389	-0.308	-0.241	-0.131	-0.705	-0.682	-0.500	-0.356	-0.347	-0.223	-0.193	
74.6		*697	-0.688	-0.481	-0.362	-0.306	-0.230	-0.131	-0.691	-0.682	-0.505	-0.356	-0.335	-0.224	-0.181	
79.5		*688	-0.688	-0.485	-0.371	-0.300	-0.224	-0.134	-0.658	-0.684	-0.505	-0.391	-0.322	-0.217	-0.175	
84.6		*655	-0.681	-0.384	-0.304	-0.216	-0.128	-0.701	-0.696	-0.520	-0.416	-0.328	-0.213	-0.172		
89.7		*686	-0.688	-0.500	-0.382	-0.308	-0.208	-0.117	-0.715	-0.667	-0.497	-0.405	-0.315	-0.209	-0.165	
94.6		*700	-0.668	-0.487	-0.382	-0.308	-0.208	-0.117	-0.715	-0.667	-0.497	-0.405	-0.315	-0.209		
Upper surface		M = 0.80						M = 0.80								
Lower surface		$\alpha = 6.29^\circ$						$\alpha = 8.33^\circ$								
0.0		*390	*669	*457	*454	*428	*414	*355	*425	*521	*501	*484	*477	*478	*409	
1.2		*346	*416	*388	*369	*341	*341	*270	*365	*490	*463	*433	*426	*427	*343	
2.4		*341	*309	*304	*294	*266	*259	*195	*406	*396	*385	*367	*357	*347	*277	
5.0		*308	*260	*240	*226	*205	*187	*123	*393	*345	*320	*303	*300	*289	*206	
7.5		*227	*167	*139	*130	*097	*081	*027	*316	*312	*304	*267	*251	*240	*151	
10.0		*280	*227	*222	*189	*168	*153	*063	*374	*312	*217	*199	*186	*162	*045	
15.0		*227	*167	*139	*130	*097	*081	*027	*316	*247	*217	*199	*125	*115	*004	
19.6		*178	*121	*094	*091	*046	*013	*058	*261	*196	*163	*155	*125	*115		
24.5		*154	*080	*048	*039	*013	*011	*107	*232	*150	*119	*106	*090	*067	*052	
29.5		*119	*048	*014	*003	*-0.022	*-0.048	*-0.115	*190	*115	*080	*065	*053	*027	*067	
34.5		*084	*017	*-0.031	*-0.038	*-0.055	*-0.081	*-0.125	*152	*083	*035	*023	*022	*010	*082	
39.5		*053	*013	*-0.063	*-0.084	*-0.114	*-0.138	*-0.120	*051	*003	*-0.004	*-0.010	*-0.047	*-0.097		
44.5		*023	*042	*085	*091	*113	*144	*148	*082	*018	*-0.023	*-0.034	*-0.043	*-0.083	*-0.109	
49.5		*006	*066	*105	*128	*127	*170	*153	*052	*-0.008	*-0.048	*-0.066	*-0.097	*-0.109	*-0.118	
54.5		*015	*055	*134	*142	*156	*182	*143	*052	*-0.008	*-0.081	*-0.087	*-0.093	*-0.131	*-0.113	
59.5		*042	*092	*147	*160	*163	*180	*138	*015	*-0.038	*-0.098	*-0.109	*-0.102	*-0.136	*-0.107	
64.5		*054	*110	*161	*168	*175	*179	*123	*001	*-0.058	*-0.111	*-0.122	*-0.119	*-0.142	*-0.098	
69.5		*062							*119	*021					*095	
74.6		*091	*139	*192	*165	*164	*173	*125	*057	*-0.096	*-0.150	*-0.125	*-0.118	*-0.144	*-0.105	
79.5		*121	*158	*211	*180	*176	*161	*101	*077	*-0.108	*-0.173	*-0.142	*-0.133	*-0.134	*-0.091	
84.6		*046	*194	*219	*186	*189	*173	*094	*025	*-0.151	*-0.186	*-0.149	*-0.132	*-0.154	*-0.095	
89.7		*135	*255	*257	*193	*196	*150	*087	*099	*-0.220	*-0.227	*-0.160	*-0.160	*-0.137	*-0.110	
94.6		*212	*331	*294	*225	*222	*180	*075	*-0.184	*-0.301	*-0.271	*-0.193	*-0.190	*-0.150		
Upper surface		M = 0.80						$\alpha = 12.34^\circ$								
Lower surface		M = 0.80						$\alpha = 16.83^\circ$								
0.0		*165	*817	*879	*730	*409	*777	*347	*043	*-1.284	*-1.284	*-1.759	*-1.611	*-1.570	*-1.429	
1.2		*314	*1666	*978	*630	*424	*787	*602	*556	*1.494	*1.066	*709	*509	*501	*446	
2.4		*799	*1657	*968	*617	*465	*882	*596	*506	*1.113	*1.047	*702	*521	*507	*431	
5.0		*862	*1648	*968	*602	*459	*574	*590	*506	*1.148	*1.047	*692	*532	*510	*432	
7.5		*970	*1659	*959	*590	*564	*709	*587	*578	*1.147	*1.046	*685	*542	*524	*436	
10.0		*922	*1675	*971	*581	*548	*746	*578	*578	*1.147	*1.045	*685	*542	*524		
15.0		*894	*1609	*972	*559	*532	*738	*559	*559	*1.159	*1.041	*678	*552	*531		
19.6		*765	*1526	*988	*565	*534	*746	*556	*556	*1.160	*1.040	*668	*564	*541		
24.5		*735	*1326	*960	*561	*536	*704	*524	*524	*1.160	*1.039	*664	*571	*551		
29.5		*635	*240	*1053	*699	*417	*621	*506	*506	*1.161	*1.023	*654	*571	*551		
34.5		*589	*476	*1135	*705	*441	*583	*493	*493	*1.202	*1.021	*659	*573	*550		
44.5		*476	*166	*1005	*678	*466	*544	*481	*481	*1.202	*1.092	*649	*564	*540		
49.5		*055	*18	*912	*636	*476	*505	*462	*462	*1.202	*1.092	*647	*564	*540		
54.5		*171	*097	*774	*600	*479	*464	*447	*447	*1.202	*1.070	*646	*564	*541		
59.5		*059	*110	*648	*572	*472	*428	*404	*404	*1.202	*1.070	*645	*564	*541		
64.5		*045	*172	*207	*150	*101	*074	*024	*024	*1.202	*1.070	*644	*563	*541		
69.5		*068	*008	*-0.079	*-0.054	*-0.101	*-0.150	*-0.20	*-0.20	*1.202	*1.070	*643	*562	*541		
74.6		*052	*035	*-0.109	*-0.082	*-0.127	*-0.150	*-0.136	*-0.269	*1.202	*1.070	*642	*561	*540		
79.5		*008	*089	*-0.125	*-0.101	*-0.143	*-0.179	*-0.146	*-0.405	*-0.030	*-0.075	*-0.122	*-0.150	*-0.225	*-0.199	
84.6		*-0.007	*-0.156	*-0.167	*-0.131	*-0.184	*-0.167	*-0.148	*-0.031	*-0.089	*-0.141	*-0.170	*-0.211	*-0.218	*-0.204	
89.7		*-0.080	*-0.250	*-0.208	*-0.194	*-0.225	*-0.180	*-0.141	*-0.028	*-0.060	*-0.130	*-0.240	*-0.267	*-0.240	*-0.198	
94.6																
Upper surface		M = 0.80						$\alpha = 6.29^\circ$								
Lower surface		$\alpha = 8.33^\circ$						$\alpha = 16.83^\circ$								
1.2		*466	*555	*522	*528	*507										

TABLE I. - WING WITH SPOILER α Continued

Percent chord	Pressure coefficient														
	0.135b/2		0.25b/2		0.40b/2		0.55b/2		0.70b/2		0.85b/2		0.95b/2		
	M = 0.90	$\alpha = 0.45^\circ$	M = 0.90	$\alpha = 4.46^\circ$	M = 0.90	$\alpha = 6.47^\circ$	M = 0.90	$\alpha = 8.54^\circ$	M = 0.90	$\alpha = 10.55^\circ$	M = 0.90	$\alpha = 12.56^\circ$	M = 0.90	$\alpha = 14.57^\circ$	
Upper surface	0.0	+241	+499	+525	+498	+363	+432	+032	+244	+268	+262	+201	+287	+233	+382
	1.2	+201	-046	-003	+131	+155	+220	+230	+091	-092	+098	+085	+066	+093	+043
	2.4	+127	-082	+021	+047	+131	+140	+144	+071	+072	+086	+070	+081	+065	+027
	5.0	+069	-056	+111	+059	+077	+102	+076	+155	+170	+150	+050	+037	+029	+027
	7.5	+033	-060	+007	+047	+051	+074	+032	+128	+127	+123	+025	+025	+024	+026
	10.0	+004	-072	+011	+037	+039	+051	+006	+207	+344	+289	+027	+027	+023	+027
	15.0	-023	-072	+017	+046	+038	+049	+046	+225	+313	+187	+158	+173	+153	+147
	19.6	-034	-075	+027	+083	+064	+089	+051	+213	+266	+141	+090	+112	+117	+102
	29.5	-065	-040	+080	+076	+071	+056	+051	+243	+229	+049	+072	+076	+067	+083
	39.5	-032	-009	+107	+085	+084	+064	+034	+201	+150	+005	+038	+032	+026	+056
	44.5	-054	+040	+133	+107	+108	+073	+036	+225	+065	+058	+008	+020	+021	+040
	39.5	-037	+108	+171	+140	+145	+098	+040	+195	+042	+116	+056	+081	+066	+030
	44.5	-024	+202	+223	+184	+188	+127	+057	+157	+167	+159	+109	+130	+114	+037
	49.5	-027	+309	+281	+241	+245	+174	+097	+075	+286	+182	+151	+182	+148	+064
	54.5	+120	+421	+316	+310	+298	+218	+133	+043	+336	+177	+188	+203	+158	+095
	59.5	+274	+441	+311	+339	+314	+233	+146	+191	+297	+162	+207	+209	+155	+118
	64.5	+390	+293	+293	+353	+320	+213	+163	+365	+079	+154	+220	+201	+135	+147
	69.6	-765	-780	-543	-418	-363	-283	-182	-760	-753	-561	-433	-357	-268	-163
	79.5	-773	-776	-534	-411	-360	-283	-189	-740	-763	-561	-425	-355	-271	-164
	84.6	-708	-767	-538	-412	-354	-276	-203	-704	-751	-561	-428	-349	-268	-170
	89.6	-693	-750	-549	-417	-350	-265	-191	-704	-737	-576	-433	-347	-262	-168
	94.6	-672	-722	-543	-415	-353	-259	-173	-711	-726	-570	-430	-349	-257	-156
Lower surface	1.3	+204	+014	+025	+138	+306	+459	+468	+342	+384	+377	+372	+315	+289	+223
	2.6	+167	+011	+062	+155	+226	+308	+386	+308	+316	+298	+271	+225	+214	+139
	5.0	+098	+026	+053	+099	+214	+284	+355	+278	+223	+213	+196	+157	+123	+072
	7.6	+056	+035	+085	+135	+209	+281	+337	+233	+177	+154	+132	+097	+077	+004
	10.1	+036	+051	+081	+164	+230	+282	+353	+205	+144	+135	+099	+054	+044	+055
	15.1	-002	-077	+134	+192	+260	+321	+436	+155	+097	+071	+051	+004	+031	+168
	19.6	-042	-089	+155	+187	+296	+340	+385	+112	+061	+027	+019	+048	+053	+188
	24.5	-054	+133	+190	+231	+292	+363	+338	+093	+019	+014	+036	+078	+117	+223
	29.5	-079	+154	+214	+257	+308	+380	+312	+058	+011	+046	+071	+117	+153	+220
	34.5	-101	+178	+246	+283	+330	+411	+283	+027	+041	+087	+110	+151	+188	+226
	39.5	+26	+200	+272	+299	+355	+434	+266	+002	+067	+119	+136	+184	+229	+233
	44.5	+32	+227	+282	+318	+393	+468	+258	+031	+096	+137	+170	+217	+268	+233
	49.5	+170	+250	+306	+339	+249	+372	+263	+056	+120	+158	+205	+250	+301	+239
	54.5	+194	+238	+323	+343	+451	+382	+251	+070	+115	+184	+230	+271	+311	+226
	59.5	+210	+256	+315	+365	+454	+296	+223	+093	+136	+195	+244	+277	+303	+205
	64.5	+218	+271	+308	+379	+455	+261	+197	+101	+140	+203	+274	+285	+287	+181
	69.5	+212	-	-	-	-	-	-	-	-	-	-	-	-	-
Upper surface	74.6	+199	+273	+315	+376	+338	+424	+196	+130	+161	+284	+278	+246	+243	+175
	79.5	+283	+261	+336	+385	+312	+229	+162	+169	+166	+273	+295	+273	+217	+150
	84.6	+123	+259	+348	+373	+273	+233	+156	+083	+190	+285	+299	+279	+227	+134
	89.7	+256	+294	+384	+345	+264	+206	+152	+152	+250	+334	+280	+292	+200	+126
	94.6	+267	+363	+421	+326	+272	+190	+132	+192	+324	+381	+288	+288	+200	+123
	0.0	+260	+097	+050	+062	+044	+051	+211	+281	+081	+175	+336	+213	+304	+070
	1.2	+122	+157	+1191	+1009	+117	+793	+815	+054	+1205	+1362	+941	+1049	+756	+562
	2.4	+204	+192	+1139	+1759	+848	+675	+688	+327	+1490	+1317	+929	+973	+742	+534
	5.0	+336	+716	+1041	+576	+615	+549	+547	+464	+1065	+1272	+953	+926	+732	+507
	7.5	+292	+588	+902	+510	+500	+462	+504	+435	+958	+1168	+910	+957	+684	+491
	10.0	+315	+648	+848	+435	+490	+405	+452	+449	+869	+1031	+660	+915	+656	+469
	15.0	+326	+490	+363	+273	+351	+338	+318	+442	+756	+884	+357	+483	+623	+425
	19.6	+308	+452	+244	+185	+222	+290	+234	+410	+676	+817	+299	+273	+602	+398
	24.5	+332	+410	+108	+168	+149	+246	+177	+426	+606	+566	+310	+243	+578	+363
	29.5	+283	+268	+104	+140	+111	+201	+137	+378	+524	+234	+296	+250	+538	+342
	34.5	+332	+106	+027	+108	+081	+156	+100	+405	+324	+081	+275	+242	+489	+322
	39.5	+313	+001	+070	+074	+042	+113	+078	+402	+128	+001	+261	+220	+431	+315
	44.5	+275	+122	+108	+045	+009	+061	+068	+441	+013	+049	+254	+198	+364	+312
	49.5	+127	+235	+134	+026	+023	+004	+078	+338	+123	+068	+257	+173	+267	+310
	54.5	+009	+307	+146	+002	+047	+046	+096	+057	+206	+071	+260	+152	+165	+312
	59.5	+171	+297	+145	+007	+074	+077	+109	+189	+070	+267	+123	+074	+294	+294
	64.5	+342	+143	+145	+022	+088	+092	+129	+243	+107	+073	+263	+098	+038	+291
	69.6	-746	-735	-548	-407	-351	-256	-154	-751	-748	-563	-400	-365	-261	-252
	74.6	-735	-726	-548	-407	-351	-256	-154	-751	-748	-563	-400	-365	-261	-252
	79.5	-708	-740	-548	-403	-351	-263	-156	-729	-765	-567	-405	-358	-267	-242
	84.6	-681	-727	-548	-406	-345	-260	-161	-744	-744	-567	-421	-356	-264	-233
	89.6	-698	-715	-556	-414	-348	-254	-154	-717	-739	-576	-440	-365	-261	-224
	94.6	-707	-703	-552	-411	-348	-246	-146	-730	-724	-566	-434	-356	-254	-215
Lower surface	1.3	+407	+479	+461	+459	+420	+405	+351	+452	+543	+509	+503	+477	+469	+405
	2.6	+368	+427	+394	+376	+341	+394	+274	+401	+505	+463	+447	+430	+415	+347
	5.0	+368	+328	+308	+301	+262	+253	+203	+449	+413	+391	+395	+359	+333	+281
	7.6	+334	+279	+249	+236	+157	+143	+067	+407	+324	+307	+281	+248	+234	+154
	10.1	+306	+253	+142	+197	+143	+145	+096	+065	+351	+263	+226	+213	+173	+149
	15.1	+248	+188	+111	+105	+042	+018	+018	+290	+215	+175	+175	+124	+112	+037
	19.6	+205	+144	+111	+105	+042	+018	+018	+210	+170	+134	+121	+082	+056	+102
	24.5	+181	+099	+068	+053	+037	+027	+027	+167	+216	+136	+097	+049	+010	+117
	29.5	+139	+068	+032	+012	+037	+027	+027	+167	+216	+136	+097	+049	+010	+117
	34.5	+104	+037	+008	+026	+071	+107	+108	+177	+104	+056	+037	+004	+031	+135
	39.5	+075	+009	+042	+057	+105	+147	+140	+145	+073	+021	+004	+031	+146	+145
	44.5	+041	+021	+066	+093	+142	+194	+189	+111	+042	+002	+034	+070	+142	+154
	49.5	+008	+047	+085	+131	+228	+238	+190	+073	+013	+027</				

TABLE I. - WING WITH SPOILER - Continued

Per- cent chord		Pressure coefficient															
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2		
		M = 0.90	a = 12.53°								M = 0.90	a = 16.98°					
Upper surface	0.0	+234	-+497	-+632	-+555	+404	-+723	-+333	+150	-+884	-1.019	+829	-+568	-+559	-+407		
	1.2	+126	-+438	-+114	-+496	+518	-+833	-+850	+308	-+1.19	+974	-+767	-+519	-+504	-+434		
	2.4	+452	-+415	-+106	-+489	+579	-+808	-+849	+805	-+1.419	+956	-+761	-+522	-+508	-+430		
	5.0	+599	-+343	-+1.052	-+485	+723	-+869	-+854	+851	-+1.397	+956	-+748	-+529	-+518	-+427		
	7.5	+754	-+392	-+1.060	-+486	+681	-+776	-+863	+1.048	-+1.379	+945	-+739	-+537	-+513	-+429		
	10.0	+738	-+343	-+1.052	-+485	+465	-+773	-+861	+1.023	-+1.367	+948	-+729	-+546	-+515	-+435		
	15.0	+732	-+259	-+1.022	-+475	+355	-+797	-+841	-+983	-+1.309	+956	-+712	-+558	-+511	-+444		
	19.6	+657	-+1223	-+0.999	-+495	+359	-+821	-+820	-+897	-+1.260	+947	-+696	-+564	-+514	-+448		
	24.5	+641	-+1192	-+0.971	-+559	+388	-+859	-+765	-+805	-+1.213	+921	-+714	-+566	-+511	-+454		
	29.5	+592	-+1095	-+0.942	-+611	+410	-+883	-+734	-+740	-+1.167	+913	-+722	-+570	-+509	-+461		
	34.5	+608	-+890	-+0.918	-+640	+432	-+893	-+715	-+700	-+1.122	+897	-+719	-+568	-+506	-+466		
	39.5	+575	-+436	-+0.900	-+640	+460	-+867	-+699	-+633	-+1.079	+904	-+708	-+563	-+503	-+475		
	44.5	+599	-+082	-+0.884	-+626	+487	-+796	-+687	-+620	-+994	+884	-+690	-+559	-+504	-+475		
	49.5	+489	-+061	-+0.835	-+614	+511	-+680	-+676	-+591	-+878	+857	-+674	-+555	-+504	-+469		
	54.5	+300	-+127	-+0.730	-+598	+527	-+534	-+648	-+501	-+745	+833	-+652	-+553	-+503	-+463		
	59.5	+220	-+099	-+0.618	-+582	+526	-+525	-+583	-+388	-+724	+809	-+651	-+545	-+503	-+439		
	64.5	+112	-+041	-+0.436	-+578	+519	-+363	-+535	-+237	-+517	+760	-+645	-+548	-+503	-+451		
	69.6	+803							-+488	-+857							
Lower surface	7.4	+797	-+800	-+532	-+546	+455	-+361	-+460	-+857	-+836	+663	-+574	-+513	-+447	-+423		
	79.5	+765	-+820	-+545	-+529	+465	-+399	-+440	-+817	-+844	+670	-+563	-+514	-+454	-+414		
	84.6	+723	-+791	-+590	-+537	+458	-+440	-+413	-+729	-+800	+670	-+564	-+505	-+459	-+398		
	89.6	+742	-+772	-+619	-+537	+455	-+458	-+376	-+778	-+780	+664	-+566	-+510	-+451	-+386		
	94.6	+756	-+755	-+579	-+530	+440	-+419	-+325	-+792	-+672	+651	-+563	-+498	-+436	-+363		
	1.3	+487	+587	+540	+534	+501	+475	+424	+504	+595	+536	+505	+502	+478	+440		
	2.6	+389	+592	+542	+525	+482	+453	+403	+351	+653	+597	+564	+548	+502	+452		
	5.0	+549	+534	+490	+491	+430	+400	+351	+653	+646	+587	+569	+523	+487	+415		
	7.6	+579	+492	+436	+428	+376	+350	+282	+745	+618	+544	+523	+487	+448	+367		
	10.1	+572	+447	+417	+385	+332	+308	+235	+742	+580	+536	+487	+452	+412	+322		
	15.1	+502	+382	+334	+316	+264	+229	+100	+654	+515	+459	+429	+390	+339	+201		
	19.6	+426	+326	+285	+271	+208	+169	+001	+574	+465	+408	+387	+343	+293	+129		
	24.5	+384	+281	+238	+215	+170	+126	+060	+532	+420	+363	+336	+294	+241	+044		
	29.5	+333	+241	+202	+172	+128	+080	+099	+478	+379	+324	+293	+246	+186	+005		
	34.5	+292	+200	+155	+126	+085	+033	+143	+431	+337	+278	+245	+208	+159	+000		
	39.5	+252	+166	+114	+090	+048	+015	+174	+390	+303	+236	+213	+170	+100	+010		
	44.5	+214	+137	+087	+050	+013	+064	+202	+350	+268	+206	+171	+137	+108	+015		
	49.5	+173	+101	+057	+010	+069	+114	+225	+307	+227	+170	+127	+185	+145	+103		
	54.5	+177	+094	+016	+020	+059	+158	+224	+303	+219	+129	+097	+046	+060	+198		
	59.5	+128	+060	+005	+045	+081	+188	+221	+257	+177	+104	+048	+013	+104	+213		
	64.5	+104	+037	+033	+079	+104	+220	+156	+225	+150	+072	+025	+025	+139	+232		
	69.5	+658							-+214	-+181					-+245		
Lower surface	74.6	+009	+002	-+0.93	-+0.97	-+141	-+248	-+217	-+138	-+098	-+005	-+022	-+073	-+201	-+281		
	79.6	+054	-+024	-+133	-+121	-+180	-+248	-+209	-+163	-+061	-+035	-+063	-+120	-+226	-+256		
	84.6	+087	-+073	-+159	-+142	-+202	-+272	-+212	-+074	-+014	-+041	-+103	-+140	-+281	-+271		
	89.7	-+001	-+142	-+216	-+170	-+253	-+237	-+198	-+070	-+059	-+121	-+143	-+207	-+283	-+269		
	94.6	-+037	-+241	-+269	-+235	-+305	-+240	-+198	-+027	-+162	-+199	-+216	-+265	-+300	-+257		
	1.3	+232	+050	+026	+080	+230	+367	+384	+360	+393	+390	+397	+330	+301	+257		
	2.6	+200	+037	-+016	-+124	+140	+197	+200	+098	-+838	+969	-+727	-+833	-+831	-+687		
	5.0	+157	-+072	-+030	-+065	+096	+131	+115	-+044	-+797	+914	-+541	-+720	-+723	-+573		
	7.5	+054	-+048	-+003	-+053	+049	+068	+017	-+143	-+295	+374	-+242	-+250	-+237	-+286		
	10.0	+024	-+061	-+007	-+045	+041	+047	-+024	-+173	-+300	+291	-+188	-+204	-+238	-+250		
	15.0	+006	-+065	-+031	-+050	+040	+046	-+071	-+189	-+315	+099	-+133	-+179	-+170	-+255		
	19.6	+018	-+070	-+046	-+087	+064	+090	-+078	-+184	-+292	+086	-+062	-+117	-+089	-+102		
	24.5	+052	-+032	-+107	-+086	+070	+058	-+070	-+214	-+285	+002	-+042	-+080	-+081	-+078		
	29.5	+021	-+002	-+128	-+093	+085	+069	-+047	-+180	-+196	+053	-+013	-+042	-+047	-+058		
	34.5	+048	+058	+157	+114	+104	+083	-+044	-+221	-+211	+015	-+029	-+012	-+005	-+042		
	39.5	+029	+133	+196	+145	+144	+100	-+048	-+225	-+209	+153	-+071	-+071	-+048	-+032		
	44.5	+016	+231	+243	+191	+188	+133	-+066	-+200	-+210	+195	-+116	-+123	-+094	-+041		
	49.5	+041	+343	+298	+246	+247	+175	-+114	-+052	-+320	+205	-+155	-+165	-+131	-+074		
	54.5	+144	+448	+327	+315	+304	+224	-+168	-+080	-+360	+193	-+192	-+193	-+147	-+118		
	59.5	+295	+468	+320	+338	+315	+240	-+192	-+221	-+308	+176	-+206	-+204	-+145	-+148		
	64.5	+407	+301	+299	+351	+324	+227	-+214	-+383	+120	+171	-+219	-+201	-+173	-+165		
	69.6	+793						-+212	-+800								
Lower surface	74.6	+809	-+807	-+588	-+460	-+369	-+279	-+216	-+800	-+808	-+602	-+481	-+376	-+286	-+172		
	79.5	-+809	-+801	-+583	-+448	-+369	-+278	-+216	-+797	-+808	-+602	-+463	-+379	-+284	-+173		
	84.6	+746	-+807	-+587	-+444	-+363	-+273	-+223	-+745	-+800	-+600	-+455	-+373	-+282	-+179		
	89.6	+694	-+792	-+582	-+443	-+359	-+266	-+204	-+728	-+788	-+608	-+458	-+368	-+274	-+174		
	94.6	+633	-+751	-+580	-+440	-+356	-+258	-+185	-+705	-+763	-+605	-+454	-+366	-+266	-+166		
	1.3	+232	+050	+026	+080	+230	+367	+384	+360	+393	+390	+397	+330	+301	+257		
	2.6	+193	+024	-+010	-+104	+166	+266	+341	+328	+331	+310	+299	+242	+225	+171		
	5.0	+126	+008	-+013	-+056	+165	+229	+310	+298	+244	+230	+227	+173	+133	+104		
	7.6	+086	-+008	-+046	-+131	+168	+228	+291	+259	+196	+175	+165	+154	+131	+067		
	10.1	+068	-+023	-+043	-+131	+223	+286	+385	+328	+245	+162	+154	+131	+103	+015		
	15.1	+025	-+046	-+098	-+147	+274	+334	+437	+372	+246	+135	+108	+094	+081	-+016		
	19.6	+015	-+058	-+121	-+150	+269	+305	+437	+374	+246	+116	+038	+011	-+005	-+016		
	24.5	+027	-+099	-+156	-+196	+290	+317	+431	+380	+211	+116	+038	-+005	-+060	-+014		
	29.5	+051	-+121	-+113	-+222	+310	+360	+437	+404	+202	+117	+022	-+039	-+097	-+143		
	34.5	+074	-+147	-+218	-+254	+310	+360	+437	+404	+202	+117	+021	-+039				

TABLE I. - WING WITH SPOILER - Continued

Percent chord	Pressure coefficient														
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2		
	M = 0.94	$\alpha = 6.47^\circ$						M = 0.94				$\alpha = 8.54^\circ$			
Upper surface	0.0	+282	+156	+108	-0.036	+0.058	-0.033	+203	+291	-0.001	+0.084	+265	+140	+242	+068
	1.2	+072	-1.125	-1.219	-1.704	-1.097	-1.806	-1.960	+0.095	-1.230	-1.327	-0.991	+0.946	+739	+773
	2.4	-1.158	-1.103	-1.189	-1.655	-1.969	-1.700	-1.860	+0.265	-1.212	-1.305	-0.987	+0.883	+707	+711
	5.0	-2.299	+0.827	-1.098	-1.570	+1.611	+1.567	+1.653	+0.390	+1.046	+1.253	-0.921	+0.838	+699	+662
	7.5	+2.250	+0.533	-1.013	-1.518	+1.497	+1.471	+1.571	+0.376	+1.016	+1.195	-0.863	+0.871	+660	+644
	10.0	+2.279	+0.476	-1.942	-1.479	+1.464	+1.418	+1.514	+0.401	+1.006	+1.152	-0.818	+0.828	+635	+620
	15.0	+2.298	+0.442	+0.483	+1.411	+1.369	+1.361	+1.399	+0.399	+0.622	+1.035	+0.728	+0.368	+610	+543
	19.6	+2.281	+0.417	+0.09	-1.348	+1.221	+1.323	+1.275	+0.367	+0.583	+1.23	+0.666	+0.227	+592	+486
	24.5	+2.312	+0.399	+0.09	-1.296	+1.28	+1.287	+1.194	+0.387	+0.550	+1.21	+0.646	+0.217	+576	+423
	29.5	+2.309	+0.369	+0.401	+0.402	+1.222	+1.096	+1.153	+0.399	+0.526	+1.206	+0.676	+0.206	+542	+400
	34.5	+2.288	+0.349	+0.4	+0.409	+1.049	+1.240	+1.184	+0.372	+0.479	+0.007	+0.443	+1.188	+0.504	+348
	44.5	+3.109	+0.111	+0.060	+0.403	+1.204	+1.119	+1.124	+0.408	+0.001	+0.077	+0.301	+1.167	+0.454	+325
	49.5	+3.1	+0.152	+0.143	+0.008	+0.203	+1.139	+1.112	+0.420	+1.145	+0.080	+0.006	+1.166	+0.294	+305
	54.5	+3.282	+0.263	+0.062	+0.007	+0.063	+1.113	+1.158	+0.213	+0.076	+0.079	+0.134	+0.191	+0.067	+283
	59.5	+1.66	+0.295	+1.42	+1.77	+0.105	+0.042	+1.37	+0.466	+1.174	+0.071	+0.180	+0.229	+0.024	+288
	64.5	+1.34	+0.154	+1.40	+0.204	+0.18	+0.073	+1.153	+0.239	+1.116	+0.063	+0.180	+0.212	+0.024	+268
	69.5	+0.829						+1.71	+0.823						
	74.6	+0.804	+0.806	+0.614	+0.508	+0.371	+0.292	+1.182	+0.822	+0.814	+0.622	+0.520	+0.387	+0.306	+0.264
	79.5	+0.779	+0.815	+0.615	+0.484	+0.371	+0.291	+1.189	+0.785	+0.829	+0.619	+0.501	+0.389	+0.319	+0.259
	84.6	+0.754	+0.803	+0.612	+0.477	+0.369	+0.292	+1.196	+0.758	+0.812	+0.621	+0.494	+0.408	+0.317	+0.255
	89.6	+0.742	+0.793	+0.619	+0.476	+0.372	+0.288	+1.191	+0.757	+0.803	+0.630	+0.494	+0.413	+0.313	+0.249
	94.6	+0.729	+0.771	+0.615	+0.468	+0.374	+0.275	+1.179	+0.749	+0.780	+0.625	+0.485	+0.415	+0.293	+0.233
Lower surface	1.3	+419	+486	+463	+471	+433	+409	+355	+460	+546	+509	+499	+482	+461	+399
	2.6	+384	+432	+398	+391	+357	+339	+285	+414	+508	+459	+441	+431	+405	+342
	5.0	+383	+334	+314	+321	+278	+263	+211	+417	+380	+375	+365	+326	+276	
	7.6	+350	+285	+255	+219	+190	+142	+440	+366	+324	+313	+305	+266	+202	
	10.1	+323	+261	+246	+218	+184	+151	+083	+414	+327	+297	+288	+228	+145	
	15.1	+268	+196	+165	+162	+115	+080	+049	+357	+272	+227	+222	+182	+148	+016
	19.6	+216	+151	+117	+124	+057	+022	+150	+298	+220	+180	+181	+125	+095	+071
	24.5	+190	+106	+076	+072	+025	+021	+236	+265	+173	+136	+130	+090	+049	+156
	29.5	+151	+075	+044	+033	+022	+063	+163	+225	+139	+103	+090	+047	+006	+193
	34.5	+116	+043	+003	+006	+057	+103	+290	+183	+106	+058	+049	+006	+038	+025
	39.5	+087	+015	+027	+036	+091	+145	+311	+147	+074	+027	+017	+032	+085	+027
	44.5	+053	+013	+046	+071	+130	+302	+197	+115	+043	+002	+020	+065	+137	+028
	49.5	+017	+041	+070	+108	+204	+260	+310	+073	+015	+017	+058	+171	+198	+026
	54.5	+016	+037	+096	+135	+196	+317	+307	+084	+015	+049	+087	+137	+249	+026
	59.5	+0.15	+063	+108	+161	+211	+346	+280	+041	+010	+065	+106	+159	+273	+022
	64.5	+0.208	+0.074	+1.121	+1.188	+1.232	+1.376	+1.210	+0.924	+0.981	+1.142	+1.186	+1.502	+1.172	+0.139
	69.5	+0.041						+1.64	+0.003						
	74.6	+0.072	+0.089	+0.165	+0.200	+0.248	+0.384	+0.130	+0.041	+0.046	+0.132	+0.163	+0.206	+0.311	+0.139
	79.6	+0.090	+0.095	+0.196	+0.226	+0.275	+0.349	+0.117	+0.034	+0.044	+0.164	+0.195	+0.237	+0.284	+0.129
	84.6	+0.023	+0.120	+0.208	+0.243	+0.303	+0.298	+0.124	+0.003	+0.092	+0.183	+0.218	+0.254	+0.264	+0.148
	89.7	+0.083	+0.173	+0.261	+0.265	+0.331	+0.199	+0.130	+0.043	+0.152	+0.237	+0.243	+0.291	+0.206	+0.154
	94.6	+0.132	+0.252	+0.308	+0.305	+0.355	+0.130	+0.096	+0.232	+0.289	+0.292	+0.319	+0.190	+0.143	
Upper surface	0.0	+262	+413	+546	+659	+392	+567	+315	+196	+741	+906	+746	+594	+565	+420
	1.2	+071	+1.337	+1.136	+1.619	+1.497	+1.792	+1.859	+211	+1.369	+1.033	+0.892	+0.559	+0.511	+441
	2.4	+498	+1.337	+1.106	+1.605	+1.551	+1.846	+1.864	+1.368	+1.020	+0.888	+0.589	+0.510	+441	
	5.0	+508	+1.327	+1.089	+1.584	+1.658	+1.855	+1.858	+1.358	+1.017	+0.888	+0.580	+0.523	+441	
	7.5	+671	+1.294	+1.060	+1.564	+1.666	+1.874	+1.873	+1.351	+1.017	+0.888	+0.595	+0.527	+444	
	10.0	+674	+1.242	+1.052	+1.562	+1.516	+1.756	+1.801	+1.316	+1.039	+0.887	+0.603	+0.526	+448	
	15.0	+664	+1.168	+1.043	+1.542	+1.376	+1.774	+1.886	+1.296	+1.022	+0.886	+0.612	+0.527	+453	
	19.6	+600	+1.133	+1.055	+1.573	+1.362	+1.795	+1.876	+1.288	+1.016	+0.886	+0.613	+0.529	+457	
	24.5	+1.102	+1.043	+1.043	+1.542	+1.390	+1.828	+1.874	+1.279	+1.011	+0.887	+0.613	+0.532	+463	
	29.5	+1.541	+1.044	+1.043	+1.712	+1.423	+1.849	+1.757	+1.676	+1.023	+0.920	+0.619	+0.525	+471	
	34.5	+1.557	+0.894	+1.025	+1.738	+1.429	+1.862	+1.727	+1.662	+1.144	+0.898	+0.730	+0.617	+480	
	39.5	+1.521	+0.604	+1.034	+1.722	+1.483	+1.856	+1.708	+1.613	+1.091	+0.898	+0.723	+0.613	+489	
	44.5	+1.551	+0.168	+1.015	+1.690	+1.521	+1.822	+1.693	+1.638	+0.898	+0.884	+0.713	+0.605	+492	
	49.5	+1.520	+0.059	+0.939	+1.661	+1.555	+1.766	+1.679	+1.580	+0.576	+0.853	+0.701	+0.589	+495	
	54.5	+1.317	+0.124	+0.758	+1.638	+1.580	+1.648	+1.643	+1.576	+0.576	+0.821	+0.681	+0.576	+520	+0.495
	59.5	+1.225	+0.087	+0.573	+1.618	+1.585	+1.595	+1.578	+1.572	+0.545	+0.789	+0.678	+0.566	+522	+0.478
	64.5	+1.148	+0.066	+0.482	+1.581	+1.527	+1.537	+1.541	+1.521	+0.527	+0.767	+0.676	+0.558	+521	+0.492
	69.5	+0.850						+1.506	+0.951						
	74.6	+0.829	+0.826	+0.578	+0.495	+0.429	+0.491	+0.925	+0.879	+0.676	+0.609	+0.549	+0.491	+0.472	
	79.5	+0.790	+0.848	+0.582	+0.525	+0.473	+0.483	+0.859	+0.894	+0.688	+0.598	+0.554	+0.499	+0.468	
	84.6	+0.761	+0.817	+0.518	+0.518	+0.461	+0.474	+0.845	+0.845	+0.692	+0.598	+0.549	+0.504	+0.453	
	89.6	+0.749	+0.814	+0.649	+0.590	+0.514	+0.540	+0.436	+0.779	+0.851	+0.683	+0.594	+0.552	+0.500	+0.442
	94.6	+0.758	+0.792	+0.620	+0.586	+0.502	+0.497	+0.775	+0.761	+0.666	+0.587	+0.544	+0.486	+0.422	
Lower surface	1.3	+512	+614	+563	+549	+521	+495	+443	+532	+632	+568	+536	+515	+499	+455
	2.6	+409	+620	+565	+545	+506	+478	+424	+369	+625	+589	+561	+523	+471	
	5.0	+588	+561	+518	+515	+456	+426	+370	+611	+675	+612	+592	+541	+507	+438
	7.6	+620	+520	+461	+451	+406	+378	+310	+779	+646	+571	+546	+505	+473	+389
	10.1	+611	+477	+443	+411	+359	+336	+262	+774	+608	+563	+511	+471	+439	+348
	15.1	+538	+416	+364	+345	+295	+259	+131	+679	+547	+487	+445	+410	+366	+231
	19.6	+461	+362	+312	+303	+239	+209	+008	+604	+494	+441	+412	+3		

TABLE I. - WING WITH SPOILER - Continued

Percent chord		Pressure coefficient												
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2
M = 0.98 $\alpha = 0.45^\circ$														
Upper surface														
0.0	*280	*533	*563	*541	*426	*477	*161	*295	*377	*356	*295	*357	*303	*418
1.2	*231	*048	-0.002	*153	*168	*210	*217	*150	-0.708	-0.882	-0.708	-0.730	-0.708	-0.591
2.4	*186	-0.047	-0.012	*094	*128	*140	*132	*015	-0.662	-0.827	-0.618	-0.617	-0.604	-0.484
5.0	*087	-0.030	-0.018	*083	*073	*076	*039	-0.086	-0.250	-0.338	-0.148	-0.229	-0.225	-0.256
7.5	*053	-0.046	-0.020	*076	*066	*055	*005	-0.116	-0.247	-0.330	-0.121	-0.165	-0.180	-0.216
10.0	*019	-0.054	-0.057	*080	*063	*057	-0.062	-0.137	-0.265	-0.143	-0.077	-0.127	-0.146	-0.212
15.0	*006	-0.060	-0.070	*117	*084	*100	-0.107	-0.137	-0.246	*018	-0.015	-0.057	-0.044	-0.146
19.6	-0.034	-0.023	-0.138	*113	*086	*064	-0.089	-0.168	-0.247	*072	-0.008	-0.041	-0.043	-0.175
24.5	-0.001	-0.020	-0.155	*120	*098	*072	-0.056	-0.137	-0.282	*109	*012	-0.001	-0.012	-0.049
29.5	-0.031	*084	*187	*140	*118	*085	-0.052	-0.180	-0.356	*156	*044	*025	*025	-0.36
34.5	-0.017	*160	*224	*166	*152	*099	-0.049	-0.189	-0.319	*200	*086	*097	*069	-0.30
39.5	*002	*256	*273	*209	*195	*127	-0.075	-0.221	-0.367	*231	*194	*150	*117	-0.46
44.5	*071	*366	*326	*263	*249	*167	-0.126	-0.117	-0.359	*229	*178	*192	*149	-0.095
49.5	*179	*475	*354	*329	*307	*218	-0.204	*093	-0.387	*207	*218	*219	*162	-0.157
54.5	*332	*495	*344	*346	*323	*235	-0.276	-0.245	-0.373	*190	*232	*231	*158	-0.218
59.5	*434	*410	*319	*358	*327	*223	-0.332	-0.396	-0.475	*179	*243	*229	*138	-0.283
64.5	-0.893						-0.333	-0.485						-0.294
69.5	-0.907	-0.887	-0.875	-0.576	-0.482	-0.396	-0.482	-0.885	-0.893	-0.657	-0.567	-0.488	-0.389	-0.309
74.6	-0.915	-0.875	-0.664	-0.564	-0.483	-0.395	-0.327	-0.880	-0.877	-0.652	-0.554	-0.490	-0.391	-0.298
79.5	-0.855	-0.862	-0.658	-0.568	-0.473	-0.392	-0.333	-0.841	-0.865	-0.658	-0.555	-0.481	-0.391	-0.298
84.6	-0.705	-0.870	-0.664	-0.565	-0.470	-0.383	-0.329	-0.742	-0.869	-0.671	-0.558	-0.476	-0.383	-0.303
89.6	-0.609	-0.846	-0.657	-0.560	-0.470	-0.306	-0.444	-0.823	-0.669	-0.558	-0.474	-0.372	-0.298	
Lower surface														
0.0	*254	*070	0.054	-0.046	-0.211	-0.335	-0.376	-0.400	-0.384	-0.402	-0.335	-0.311	-0.260	
2.6	*217	*042	*016	-0.072	*151	*243	*308	*348	*339	*304	*254	*251	*239	*184
5.0	*154	*027	*012	-0.028	*193	*213	*283	*317	*251	*228	*232	*186	*150	*125
7.6	*109	*012	-0.021	-0.062	*144	*205	*255	*277	*209	*176	*172	*127	*111	*064
10.1	*089	-0.001	-0.018	-0.099	*170	*210	*250	*250	*174	*150	*135	*086	*078	*009
15.1	*050	-0.025	-0.072	-0.128	*162	*256	*339	*199	*133	*098	*087	*040	*009	-0.102
19.6	*011	-0.033	-0.094	-0.133	*243	*268	*345	*155	*099	*058	*058	-0.008	-0.029	-0.181
24.5	-0.001	-0.076	-0.128	-0.172	*245	*303	*372	*135	*054	*019	*004	-0.033	-0.074	-0.252
29.5	-0.024	-0.096	-0.152	-0.195	*263	*322	*356	*100	*023	*013	-0.030	-0.063	-0.111	-0.254
34.5	-0.048	-0.121	-0.190	-0.229	*263	*338	*360	*070	-0.002	*048	-0.068	-0.101	-0.140	-0.271
39.5	-0.070	-0.145	-0.247	-0.288	*208	*367	*378	*044	-0.030	*081	-0.094	-0.132	-0.176	-0.293
44.5	-0.093	-0.171	-0.236	-0.275	*343	*406	*395	*012	-0.060	*101	-0.118	-0.168	-0.219	-0.303
49.5	-0.115	-0.197	-0.257	-0.309	*155	*446	*424	*026	-0.082	*125	-0.144	-0.180	-0.270	-0.322
54.5	-0.135	-0.186	-0.274	-0.328	*405	*479	*441	*031	-0.078	*148	-0.162	-0.233	-0.326	-0.335
59.5	-0.164	-0.204	-0.278	-0.327	*397	*490	*439	*054	-0.098	*156	-0.179	-0.244	-0.355	-0.338
64.5	-0.163	-0.216	-0.295	-0.340	*407	*482	*428	*063	-0.114	*156	-0.206	-0.258	-0.376	-0.323
69.5	-0.164													-0.325
74.6	-0.156	-0.241	-0.283	-0.318	*348	*424	*396	*090	-0.128	*168	-0.217	-0.249	-0.367	-0.325
79.5	-0.234	-0.232	-0.285	-0.317	*369	*393	*365	*136	-0.121	*189	-0.232	-0.271	-0.352	-0.314
84.6	-0.085	-0.230	-0.274	-0.311	*365	*391	*350	*028	-0.125	*200	-0.239	-0.286	-0.363	-0.317
89.7	-0.222	-0.243	-0.303	-0.315	*371	*360	*334	*116	-0.160	*242	-0.256	-0.301	-0.338	-0.316
94.6	-0.232	-0.286	-0.329	-0.343	*368	*350	*312	*144	-0.225	*283	-0.293	-0.340	-0.340	
Upper surface														
0.0	*311	*223	*184	*033	*124	*025	*260	*320	*079	*002	-0.157	-0.116	-0.185	*118
1.2	*095	-1.019	-1.098	-1.051	*929	*618	*922	*508	-1.181	*1221	-1.202	-0.952	-0.570	-0.839
2.4	-0.993	-0.992	-1.070	-1.031	*715	*561	*847	*201	-1.160	*1200	-1.187	-0.906	-0.585	-0.772
5.0	-0.236	*810	*991	-0.944	*553	*493	*631	*314	-1.079	*1160	-1.143	-0.868	-0.568	-0.724
7.5	-0.193	-0.456	-0.908	-0.895	*487	*438	*519	*313	-0.866	*1097	-1.103	-0.884	-0.532	-0.73
10.0	-0.220	*403	*857	-0.647	*462	*395	*456	*346	-0.587	*1073	-0.921	-0.902	-0.526	-0.714
13.0	-0.243	*380	*514	-0.345	*342	*356	*382	*339	-0.507	*1019	-0.657	-0.594	-0.535	-0.574
19.6	-0.229	*359	*403	-0.167	*213	*213	*263	*138	-0.473	*808	-0.73	-0.213	-0.544	-0.453
24.5	-0.261	*344	*010	-0.060	*119	*292	*170	*337	-0.446	*447	-0.518	-0.179	-0.559	-0.360
29.5	-0.226	*353	*083	-0.013	*074	*259	*135	*305	-0.437	*044	-0.446	-0.178	-0.565	-0.320
34.5	-0.261	*351	*019	-0.045	*214	*116	*338	*440	-0.440	*058	-0.356	-0.165	-0.560	-0.294
39.5	-0.261	*078	*173	*048	*012	*153	*110	*329	-0.318	*092	-0.248	-0.135	-0.532	-0.282
44.5	-0.295	*152	*199	*083	*014	*080	*120	*360	*008	*122	-0.094	-0.134	-0.475	-0.287
49.5	-0.312	*278	*199	*121	*034	*012	*152	*383	*174	*123	*066	-0.151	-0.387	-0.313
54.5	-0.090	*334	*185	*166	*046	*043	*196	*14	*244	*116	*162	-0.197	-0.266	-0.337
59.5	-0.143	*238	*173	*204	*042	*075	*240	*023	*184	*110	*227	-0.244	-0.159	-0.345
64.5	-0.324	*186	*167	*234	*024	*091	*290	*173	*158	*102	*277	-0.296	-0.124	-0.374
69.6	-0.924													-0.381
74.6	-0.906	-0.904	-0.669	-0.584	*510	-0.417	-0.311	*940	-0.926	-0.695	-0.604	-0.558	-0.490	-0.388
79.5	-0.888	-0.901	-0.665	-0.571	*507	-0.416	-0.316	*894	-0.932	-0.687	-0.596	-0.558	-0.507	-0.380
84.6	-0.850	-0.890	-0.670	-0.571	*501	-0.416	-0.319	*878	-0.923	-0.692	-0.592	-0.578	-0.494	-0.377
89.6	-0.784	-0.890	-0.687	-0.577	*499	-0.412	-0.311	*804	-0.919	-0.713	-0.596	-0.601	-0.476	-0.373
94.6	-0.706	-0.884	-0.684	-0.577	*498	-0.405	-0.299	*755	-0.845	-0.708	-0.596	-0.604	-0.453	-0.363
Lower surface														
1.3	*438	*505	*474	*470	*456	*429	*384	*476	*568	*523	*515	*491	*485	*428
2.6	*407	*452	*409	*389	*383	*362	*313	*437	*529	*470	*450	*442	*429	*370
5.0	*409	*354	*326	*319	*309	*288	*251	*482	*435	*393	*385	*378	*352	*304
7.6	*374	*306	*268	*258	*245	*220	*185	*461	*386	*336	*327	*316	*297	*238
10.1	*349	*282	*254	*222	*211	*177	*132	*441	*348	*318	*300	*269	*256	*189
15.1	*289	*219	*177	*111	*142	*109	*002	*377	*291	*245	*234	*201	*180	*065
19.6	*241	*177	*111	*136	*090	*043	*110	*321	*242	*195	*198	*147	*122	*039
24.5	*219	*126	*090	*090	*053	*014	*187	*290	*195	*154	*148	*112	*080	*-0.121
29.5	*178	*098	*040	*056	*014	*022	*063	*262	*129	*079	*073	*035	*006	*-0.204
34.5	*144	*065	*019	*020	*006	*058	*100	*298	*176	*097	*048	*004	*-0.001	*-0.043
39.5	*102	*010	-0.006	-0.058	-0.096	*146	*303	*139	*064	*025	*010	*-0.037	*-0.098	-0.268
44.5</td														

TABLE I. - WING WITH SPOILER - Continued

Percent chord		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	
		M = 0.98	a = 13.21°	M = 0.98 a = 17.47°											
Upper surface	0.0	*303	-*311	-*434	-*585	-*418	-*598	-*255	*217	-*657	-*816	-*882	-*577	-*788	-*546
	1.2	*001	-1.233	-1.089	-*621	-*654	-*749	-*921	-*138	-1.318	-1.165	-*783	-*497	-*715	-*592
	2.4	-*401	-1.228	-1.070	-*615	-*683	-*791	-*930	-*612	-1.317	-1.147	-*778	-*517	-*724	-*603
	5.0	-*407	-1.202	-1.070	-*601	-*745	-*834	-*940	-*691	-1.309	-1.152	-*762	-*532	-*738	-*614
	7.5	-*572	-1.163	-1.045	-*595	-*833	-*756	-*961	-*856	-1.312	-1.126	-*759	-*553	-*730	-*616
	10.0	-*591	-1.114	-1.050	-*600	-*757	-*737	-*980	-*847	-1.289	-1.126	-*756	-*573	-*725	-*622
	15.0	-*580	-1.045	-1.045	-*607	-*499	-*63	-*985	-*823	-1.250	-1.130	-*735	-*616	-*701	-*623
	19.6	-*512	-1.006	-1.042	-*588	-*411	-*793	-*983	-*742	-1.223	-1.122	-*725	-*662	-*690	-*626
	24.5	-*495	-0.977	-*987	-*786	-*379	-*838	-*907	-*670	-1.200	-1.088	-*743	-*698	-*686	-*626
	29.5	-*483	-0.949	-*917	-*910	-*439	-*911	-*758	-*610	-1.168	-1.084	-*709	-*548	-*687	-*636
	34.5	-*447	-*593	-*906	-*974	-*470	-*921	-*746	-*562	-1.156	-1.082	-*829	-*741	-*686	-*640
	44.5	-*482	-*254	-*915	-*1.010	-*534	-*885	-*745	-*595	-*983	-*1.078	-*806	-*740	-*683	-*637
	49.5	-*470	-*074	-*915	-*1.003	-*598	-*803	-*746	-*604	-*451	-*1.151	-*796	-*736	-*679	-*631
	54.5	-*281	-*149	-*760	-*910	-*633	-*705	-*738	-*480	-*138	-*1.21	-*791	-*729	-*673	-*623
	59.5	-*186	-*115	-*540	-*813	-*647	-*636	-*681	-*389	-*089	-*1.072	-*795	-*723	-*667	-*597
	64.5	-*135	-*105	-*337	-*764	-*635	-*578	-*657	-*281	-*067	-*1.013	-*795	-*724	-*656	-*610
	69.5	-*948	-*917	-*687	-*636	-*573	-*594	-*598	-*961	-*942	-*696	-*703	-*686	-*615	-*591
	74.6	-*897	-*914	-*680	-*549	-*579	-*607	-*589	-*898	-*935	-*743	-*696	-*681	-*617	-*586
	79.5	-*853	-*914	-*680	-*657	-*589	-*614	-*585	-*786	-*919	-*863	-*727	-*665	-*616	-*576
	84.6	-*786	-*910	-*685	-*688	-*598	-*625	-*584	-*786	-*930	-*849	-*730	-*660	-*608	-*567
	89.6	-*742	-*852	-*665	-*653	-*593	-*620	-*571	-*774	-*858	-*715	-*724	-*652	-*594	-*557
Lower surface	1.3	*538	*650	*591	*568	*538	*519	*473	*546	*659	*589	*554	*533	*505	*469
	2.6	*440	*653	*589	*562	*526	*502	*453	*373	*706	*642	*603	*573	*532	*494
	5.0	*628	*594	*541	*533	*479*	*450	*406	*727	*699	*626	*606	*553	*517	*464
	7.6	*653	*552	*483	*468	*433	*404	*349	*812	*668	*587	*560	*516	*487	*418
	10.1	*645	*507	*467	*430	*390	*362	*298	*799	*630	*577	*529	*482	*455	*379
	15.1	*568	*446	*385	*368	*324	*287	*169	*703	*569	*502	*474	*422	*388	*265
	19.6	*496	*392	*338	*328	*271	*255	*083	*629	*514	*455	*432	*379	*333	*173
	24.5	*459	*345	*294	*275	*231	*192	*010	*586	*468	*411	*385	*340	*295	*098
	29.5	*407	*307	*256	*233	*188	*144	*067	*533	*430	*376	*341	*294	*252	*043
	34.5	*362	*271	*215	*193	*149	*106	*114	*486	*393	*330	*297	*255	*204	*011
	39.5	*324	*235	*180	*160	*115	*059	*165	*447	*357	*292	*264	*212	*158	*061
	44.5	*284	*201	*154	*120	*074	*010	*202	*405	*319	*263	*225	*178	*109	*110
	49.5	*243	*170	*126	*080	*076	*045	*240	*363	*283	*229	*184	*182	*054	*152
	54.5	*248	*163	*090	*052	*004	*096	*255	*364	*274	*192	*153	*106	*008	*183
	59.5	*201	*131	*073	*021	*-021	*136	*271	*314	*237	*167	*116	*079	*029	*209
	64.5	*174	*111	*047	*-013	*-013	*172	*272	*285	*211	*137	*082	*044	*074	*235
	69.5	*134	*907	*010	*-042	*-086	*227	*289	*171	*169	*071	*039	*009	*144	*309
	74.6	*074	*083	*-010	*-042	*-086	*227	*289	*171	*169	*071	*039	*009	*144	*309
	79.6	*127	*046	*-047	*-078	*-119	*236	*283	*225	*139	*030	*003	*038	*172	*308
	84.6	*055	*024	*-071	*-102	*-151	*269	*320	*077	*097	*-003	*-003	*-028	*-071	*-341
	89.7	*058	*-037	*-131	*-131	*-189	*262	*354	*113	*030	*-066	*-063	*-130	*-241	*-374
	94.6	*034	*-123	*-168	*-188	*-247	*280	*378	*076	*064	*-131	*-119	*-196	*-280	*-422
Upper surface		M = 1.00 a = 0.26°													
		M = 1.00	a = 0.26°	M = 1.00 a = 4.22°											
Upper surface	0.0	*298	*557	*584	*558	*426	*487	*163	*310	*383	*365	*269	*334	*294	*417
	1.2	*257	*023	*022	*182	*198	*235	*244	*159	*734	*898	*559	*780	*777	*652
	2.4	*217	*-014	*011	*124	*159	*165	*158	*028	*276	*850	*410	*673	*690	*556
	5.0	*154	*008	*032	*122	*128	*131	*107	*041	*276	*885	*303	*227	*223	*257
	7.5	*119	*001	*045	*114	*104	*104	*069	*072	*241	*328	*239	*208	*193	*223
	10.0	*085	*-016	*045	*104	*095	*082	*039	*105	*238	*324	*175	*187	*179	*207
	15.0	*039	*-030	*085	*112	*091	*085	*025	*127	*252	*321	*089	*114	*137	*194
	19.6	*039	*-037	*102	*150	*111	*125	*048	*127	*231	*308	*022	*065	*046	*134
	24.5	*002	*000	*166	*145	*115	*091	*074	*158	*237	*317	*003	*028	*030	*054
	29.5	*023	*047	*17	*14	*10	*099	*027	*147	*242	*314	*005	*008	*032	*032
	34.5	*006	*112	*216	*145	*109	*097	*057	*189	*219	*319	*061	*053	*042	*018
	39.5	*006	*192	*252	*195	*179	*125	*026	*176	*213	*266	*100	*043	*043	*013
	44.5	*023	*289	*300	*238	*221	*149	*052	*212	*274	*249	*149	*156	*126	*028
	49.5	*095	*398	*353	*291	*270	*186	*106	*173	*359	*240	*200	*160	*077	*077
	54.5	*203	*502	*380	*348	*325	*235	*190	*071	*390	*212	*198	*220	*178	*148
	59.5	*355	*525	*372	*369	*340	*251	*271	*239	*220	*195	*252	*244	*174	*219
	64.5	*453	*439	*345	*380	*351	*238	*355	*385	*204	*183	*240	*247	*152	*290
	69.6	*907	*886	*562	*513	*441	*361	*906	*892	*656	*583	*502	*422	*335	*309
	74.6	*914	*872	*665	*565	*444	*361	*891	*883	*649	*572	*501	*425	*326	
	79.5	*833	*852	*644	*565	*432	*360	*846	*864	*656	*570	*488	*420	*324	
	84.6	*681	*857	*648	*568	*497	*418	*732	*872	*670	*575	*493	*409	*319	
	89.6	*587	*847	*648	*568	*499	*404	*634	*829	*670	*578	*493	*402	*312	
Lower surface	1.3	*275	*094	*082	*-018	*-183	*-318	*-310	*-393	*-421	*-407	*-429	*-366	*-338	*-293
	2.6	*240	*066	*045	*040	*-001	*-116	*-188	*-253	*-277	*-248	*-261	*-204	*-178	*-153
	5.0	*180	*054	*040	*001	*-116	*-184	*-229	*-299	*-231	*-196	*-201	*-157	*-144	*-095
	7.6	*137	*043	*008	*-032	*-115	*-184	*-229	*-274	*-196	*-174	*-169	*-110	*-106	*-047
	10.1	*121	*029	*011	*-070	*-140	*-184	*-225	*-274	*-196	*-174	*-169	*-110	*-106	*-047
	15.1	*081	*005	*-042	*-102	*-175	*-228	*-304	*-222	*-154	*-120	*-116	*-067	*-033	*-061
	19.6	*039	*-004	*-064	*-102	*-216	*-243	*-315	*-176	*-119	*-076	*-086	*-015	*-003	*-153
	24.5	*028	*-044	*-097	*-141	*-221	*-276	*-340	*-157	*-075	*-039	*-032	*-009	*-047	*-217
	29.5	*004	*-065	*-121	*-167	*-236	*-291	*-320	*-121	*-047	*-008	*-003	*-039	*-085	*-229
	34.5	*-018	*-090	*-156	*-194	*-255	*-314	*-329	*-090	*-018	*-029	*-041	*-074	*-11	

TABLE I. - WING WITH SPOILER - Continued

		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	
Percent chord		M = 1.00						M = 1.00							
		M = 1.00	$\alpha = 6.47^\circ$						M = 1.00	$\alpha = 8.52^\circ$					
Upper surface	0.0	*327	*255	*220	*076	*113	*047	*283	*345	*127	*053	*104	*067	*139	*154
	1.2	*222	*991	-1.056	-1.019	-0.919	-0.847	-0.647	-0.173	-1.113	-1.154	-1.140	-0.937	-0.525	-0.713
	2.4	-0.63	-0.960	-1.040	-1.012	-0.705	-0.767	-0.505	-0.157	-1.03	-1.135	-1.139	-0.898	-0.512	-0.632
	5.0	-203	-786	-0.971	-0.955	-0.502	-0.530	-0.516	-0.266	-1.020	-1.096	-1.086	-0.861	-0.507	-0.601
	7.5	-166	-423	-0.889	-0.901	-0.430	-0.407	-0.478	-0.264	-0.99	-0.526	-1.008	-0.972	-0.843	-0.470
	10.0	-192	-373	-0.846	-0.785	-0.460	-0.358	-0.446	-0.299	-0.944	-0.444	-0.948	-0.593	-0.631	-0.479
	15.0	-213	-352	-0.482	-0.377	-0.346	-0.300	-0.404	-0.296	-0.904	-0.417	-0.795	-0.505	-0.225	-0.496
	19.6	-201	-333	-0.401	-0.124	-0.203	-0.266	-0.373	-0.277	-0.817	-0.430	-0.434	-0.149	-0.521	-0.335
	24.5	-231	-314	-0.020	-0.019	-0.106	-0.228	-0.277	-0.294	-0.892	-0.389	-0.028	-0.363	-0.146	-0.532
	29.5	-200	-324	-0.98	-0.31	-0.07	-0.153	-0.197	-0.263	-0.892	-0.389	-0.028	-0.363	-0.146	-0.532
	34.5	-235	-334	-0.155	-0.061	-0.37	-0.086	-0.131	-0.297	-0.395	-0.095	-0.286	-0.129	-0.526	-0.297
	39.5	-239	-101	-0.193	-0.022	-0.001	-0.043	-0.091	-0.291	-0.316	-0.130	-0.203	-0.104	-0.502	-0.300
	44.5	-270	-159	-0.216	-0.109	-0.032	-0.001	-0.090	-0.323	-0.32	-0.158	-0.095	-0.096	-0.452	-0.312
	49.5	-291	-297	-0.214	-0.144	-0.057	-0.033	-0.128	-0.343	-0.200	-0.160	-0.037	-0.109	-0.367	-0.339
	54.5	-0.099	-342	-0.199	-0.186	-0.074	-0.055	-0.185	-0.147	-0.272	-0.147	-0.151	-0.150	-0.252	-0.375
	59.5	-150	-223	-0.185	-0.217	-0.079	-0.066	-0.234	-0.044	-0.210	-0.136	-0.233	-0.205	-0.150	-0.380
	64.5	-321	-201	-0.178	-0.244	-0.073	-0.063	-0.289	-0.191	-0.191	-0.125	-0.292	-0.264	-0.113	-0.408
	69.6	-97	-0.908	-0.661	-0.593	-0.517	-0.431	-0.345	-0.919	-0.908	-0.676	-0.605	-0.564	-0.519	-0.424
	74.6	-908	-0.907	-0.656	-0.582	-0.514	-0.442	-0.343	-0.891	-0.908	-0.670	-0.597	-0.566	-0.531	-0.418
	79.5	-879	-0.901	-0.656	-0.577	-0.504	-0.431	-0.349	-0.858	-0.901	-0.673	-0.593	-0.596	-0.521	-0.413
	84.6	-853	-0.890	-0.682	-0.582	-0.509	-0.423	-0.343	-0.770	-0.896	-0.697	-0.594	-0.615	-0.502	-0.406
	89.6	-752	-0.890	-0.682	-0.509	-0.423	-0.343	-0.770	-0.896	-0.697	-0.594	-0.621	-0.481	-0.397	-0.397
	94.6	-685	-0.826	-0.679	-0.583	-0.509	-0.414	-0.328	-0.713	-0.817	-0.693	-0.593	-0.621	-0.481	-0.397
Lower surface	1.3	*449	*515	*488	*481	*471	*443	*400	*497	*587	*543	*528	*505	*491	*446
	2.6	-426	-460	-419	-400	-397	-379	-336	-459	-546	-488	-465	-453	-440	-391
	5.0	-423	-367	-339	-330	-322	-304	-270	-503	-452	-412	-400	-387	-364	-325
	7.6	-389	-319	-282	-270	-267	-238	-210	-483	-406	-354	-340	-336	-306	-262
	10.1	-362	-295	-265	-231	-234	-199	-161	-460	-367	-339	-312	-288	-267	-210
	15.1	-307	-231	-194	-181	-166	-132	-027	-397	-312	-264	-251	-218	-200	-091
	19.6	-258	-188	-148	-149	-119	-077	-068	-342	-262	-213	-214	-167	-142	-000
	24.5	-233	-144	-107	-101	-078	-042	-015	-313	-218	-175	-165	-111	-107	-092
	29.5	-196	-111	-074	-069	-040	-001	-019	-271	-182	-136	-131	-093	-063	-136
	34.5	-158	-077	-035	-036	-003	-034	-023	-232	-150	-100	-091	-054	-024	-176
	39.5	-128	-050	-005	-013	-030	-073	-0263	-198	-119	-069	-064	-019	-017	-219
	44.5	-092	-020	-016	-018	-069	-120	-0272	-162	-086	-049	-032	-016	-066	-238
	49.5	-048	-009	-034	-049	-1.07	-172	-0287	-113	-058	-029	-006	-077	-122	-258
	54.5	-059	-007	-055	-073	-1.33	-218	-0288	-131	-057	-000	-030	-076	-167	-270
	59.5	-023	-032	-058	-088	-1.47	-257	-0290	-085	-031	-008	-053	-093	-209	-278
	64.5	-006	-048	-064	-1.24	-1.65	-290	-0278	-065	-017	-023	-085	-118	-240	-267
	69.5	-012	-017	-027	-027	-0.92	-295	-0275	-007	-0104	-0046	-0595	-017	-022	-073
Upper surface	1.3	-0.058	-0.056	-0.056	-0.139	-0.170	-0.317	-0.291	-0.021	-0.007	-0.067	-0.098	-0.138	-0.278	-0.286
	2.6	-0.054	-0.050	-0.052	-0.126	-0.160	-0.320	-0.279	-0.038	-0.006	-0.097	-0.123	-0.164	-0.289	-0.270
	5.0	-0.012	-0.062	-0.136	-0.170	-0.206	-0.337	-0.291	-0.002	-0.020	-0.110	-0.139	-0.178	-0.319	-0.279
	84.6	-0.049	-0.105	-0.180	-0.187	-0.238	-0.313	-0.307	-0.005	-0.073	-0.156	-0.162	-0.215	-0.302	-0.300
	89.7	-0.078	-0.178	-0.226	-0.231	-0.276	-0.319	-0.313	-0.013	-0.149	-0.205	-0.206	-0.259	-0.320	-0.302
	94.6	-0.078	-0.178	-0.226	-0.231	-0.276	-0.319	-0.313	-0.013	-0.149	-0.205	-0.206	-0.259	-0.320	-0.302
Lower surface	1.3	*541	*654	*599	*569	*539	*518	*475	*563	*678	*606	*566	*551	*523	*485
	2.6	-440	-658	-593	-564	-529	-500	-456	-388	-725	-655	-614	-592	-549	-507
	5.0	-635	-598	-542	-536	-479	-450	-407	-750	-715	-643	-617	-571	-534	-482
	7.6	-664	-557	-488	-476	-433	-403	-347	-827	-685	-605	-572	-536	-503	-437
	10.1	-652	-515	-470	-430	-389	-361	-303	-817	-647	-592	-539	-501	-471	-399
	15.1	-576	-451	-393	-369	-323	-293	-173	-724	-583	-519	-405	-442	-406	-287
	19.6	-502	-397	-344	-328	-274	-247	-071	-646	-531	-447	-391	-357	-314	-126
	24.5	-464	-351	-301	-281	-235	-200	-002	-604	-486	-429	-399	-353	-314	-072
	29.5	-412	-312	-265	-240	-193	-153	-062	-553	-447	-391	-357	-314	-268	-072
	34.5	-369	-276	-222	-195	-155	-110	-113	-508	-408	-346	-314	-275	-225	-015
	39.5	-331	-242	-187	-162	-114	-063	-161	-447	-349	-311	-279	-239	-183	-036
	44.5	-289	-207	-161	-125	-081	-019	-200	-426	-336	-281	-243	-201	-136	-084
	49.5	-248	-175	-134	-088	-076	-036	-236	-361	-303	-251	-204	-186	-082	-123
	54.5	-253	-167	-101	-056	-011	-005	-258	-365	-292	-214	-172	-126	-043	-151
	59.5	-204	-137	-080	-008	-015	-0127	-273	-335	-256	-190	-129	-096	-001	-180
	64.5	-176	-117	-056	-006	-045	-0165	-289	-303	-231	-162	-104	-062	-044	-203
	69.5	-134	-033	-024	-017	-017	-0177	-247	-305	-247	-162	-104	-062	-044	-269
	74.6	-1070	-090	-002	-037	-087	-0224	-1327	-185	-191	-095	-058	-006	-0108	-278
	79.6	-130	-071	-034	-068	-110	-0241	-315	-246	-169	-056	-023	-011	-136	-278
	84.6	-024	-036	-058	-092	-133	-0290	-333	-075	-120	-024	-009	-048	-197	-315
	89.7	-044	-024	-0117	-0122	-181	-0289	-357	-055	-037	-004	-008	-0205	-0346	-380
	94.6	-026	-0110	-0173	-0177	-238	-0392	-093	-038	-103	-110	-150	-264	-0264	-0302

TABLE I. - WING WITH SPOILER - Continued

		Pressure coefficient														
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
Percent chord	Upper surface	M = 1.03							M = 1.03							
		M = 1.03	a = 0.45°						M = 1.03	a = 4.48°						
0.0		*.229	*.553	*.590	*.591	*.510	*.529	*.213	*.246	*.401	*.385	*.286	*.376	*.312	*.439	
1.2		*.198	*.009	*.003	*.214	*.243	*.281	*.293	*.124	*.691	*.824	*.715	*.721	*.722	*.600	
2.4		*.173	*.026	*.003	*.160	*.203	*.214	*.214	*.010	*.655	*.789	*.609	*.623	*.645	*.515	
5.0		*.116	*.005	*.039	*.156	*.172	*.174	*.157	*.072	*.220	*.695	*.348	*.225	*.201	*.238	
7.5		*.075		*.077	*.144	*.142	*.129	*.098	*.123	*.243	*.312	*.211	*.158	*.159	*.189	
10.0		*.046	*.037	*.087	*.144	*.142	*.129	*.098	*.147	*.236	*.341	*.211	*.158	*.154	*.165	
15.0		*.020	*.051	*.124	*.151	*.136	*.129	*.020	*.159	*.237	*.317	*.208	*.158	*.159	*.158	
19.6		*.015	*.057	*.130	*.193	*.154	*.174	*.033	*.159	*.237	*.317	*.208	*.158	*.159	*.159	
24.5		*.025	*.033	*.196	*.181	*.153	*.138	*.030	*.159	*.231	*.313	*.207	*.158	*.159	*.159	
29.5		*.002	*.044	*.222	*.187	*.165	*.144	*.012	*.159	*.249	*.312	*.207	*.158	*.159	*.159	
34.5		*.032	*.149	*.251	*.208	*.184	*.153	*.016	*.167	*.254	*.327	*.204	*.158	*.159	*.159	
39.5		*.028	*.227	*.288	*.233	*.218	*.168	*.016	*.178	*.296	*.320	*.204	*.158	*.159	*.159	
44.5		*.002	*.321	*.335	*.276	*.260	*.191	*.009	*.209	*.273	*.278	*.187	*.182	*.153	*.006	
49.5		*.108	*.429	*.385	*.327	*.310	*.228	*.061	*.218	*.347	*.257	*.234	*.226	*.189	*.045	
54.5		*.236	*.533	*.411	*.384	*.364	*.276	*.148	*.201	*.367	*.230	*.270	*.257	*.210	*.121	
59.5		*.388	*.557	*.407	*.401	*.379	*.289	*.234	*.219	*.152	*.201	*.287	*.274	*.209	*.189	
64.5		*.510	*.477	*.387	*.409	*.385	*.276	*.316	*.369	*.247	*.178	*.295	*.273	*.188	*.264	
69.6		*.852													*.290	
74.6		*.861	*.837	*.592	*.522	*.474	*.404	*.346	*.852	*.849	*.613	*.548	*.483	*.409	*.313	
79.5		*.868	*.814	*.585	*.523	*.470	*.405	*.333	*.836	*.838	*.607	*.538	*.482	*.415	*.302	
84.6		*.759	*.776	*.581	*.523	*.461	*.400	*.329	*.790	*.819	*.613	*.536	*.472	*.403	*.304	
89.6		*.597	*.787	*.585	*.525	*.458	*.382	*.328	*.690	*.831	*.630	*.542	*.473	*.388	*.296	
94.6		*.565	*.782	*.591	*.525	*.459	*.371	*.323	*.690	*.790	*.627	*.546	*.473	*.380	*.290	
1.3	Lower surface	*.224	*.096	*.108	*.013	*.159	*.277	*.282	*.328	*.407	*.403	*.431	*.400	*.369	*.320	
2.6		*.195	*.061	*.015	*.097	*.180	*.244	*.317	*.349	*.327	*.338	*.309	*.298	*.248		
5.0		*.143	*.044	*.063	*.023	*.071	*.145	*.214	*.311	*.262	*.252	*.272	*.239	*.211	*.186	
7.6		*.107	*.034	*.028	*.007	*.079	*.143	*.190	*.276	*.221	*.202	*.218	*.187	*.175	*.130	
10.1		*.084	*.026	*.032	*.039	*.101	*.138	*.196	*.254	*.193	*.180	*.191	*.145	*.141	*.079	
15.1		*.049	*.002	*.020	*.072	*.137	*.182	*.264	*.209	*.149	*.127	*.144	*.097	*.066	*.022	
19.6		*.022	*.010	*.040	*.075	*.176	*.201	*.274	*.166	*.115	*.087	*.112	*.048	*.028	*.070	
24.5		*.013	*.042	*.072	*.112	*.173	*.230	*.296	*.152	*.074	*.048	*.064	*.023	*.012	*.185	
29.5		*.006	*.061	*.093	*.136	*.194	*.243	*.281	*.118	*.047	*.022	*.033	*.004	*.049	*.196	
34.5		*.028	*.081	*.128	*.166	*.211	*.261	*.287	*.087	*.019	*.014	*.006	*.037	*.083	*.211	
39.5		*.045	*.098	*.154	*.186	*.233	*.289	*.305	*.062	*.005	*.038	*.030	*.071	*.114	*.237	
44.5		*.069	*.120	*.174	*.214	*.267	*.323	*.318	*.032	*.032	*.057	*.057	*.101	*.154	*.246	
49.5		*.090	*.144	*.190	*.244	*.366	*.355	*.301	*.059	*.072	*.082	*.082	*.205	*.266		
54.5		*.135	*.195	*.208	*.257	*.330	*.410	*.377	*.008	*.053	*.088	*.093	*.168	*.256	*.274	
59.5		*.114	*.147	*.208	*.268	*.325	*.416	*.377	*.027	*.073	*.094	*.116	*.179	*.286	*.279	
64.5		*.125	*.151	*.221	*.274	*.335	*.408	*.363	*.045	*.085	*.095	*.129	*.196	*.312	*.267	
69.5		*.132													*.262	
74.6		*.150	*.169	*.210	*.251	*.280	*.357	*.344	*.125	*.089	*.101	*.142	*.193	*.306	*.270	
79.5		*.162	*.153	*.211	*.246	*.288	*.318	*.299	*.052	*.072	*.120	*.161	*.209	*.297	*.259	
84.6		*.169	*.155	*.195	*.235	*.281	*.306	*.291	*.069	*.073	*.129	*.167	*.219	*.309	*.268	
89.7		*.157	*.168	*.222	*.242	*.285	*.288	*.274	*.079	*.100	*.171	*.184	*.235	*.274	*.274	
94.6		*.167	*.203	*.248	*.267	*.303	*.280	*.251	*.087	*.164	*.206	*.220	*.269	*.280	*.282	
		M = 1.03														
		a = 6.47°														
0.0		*.287	*.298	*.246	*.107	*.141	*.079	*.307								
1.2		*.183	*.941	*.106	*.006	*.988	*.858	*.942	*.624							
2.4		*.046	*.926	*.988	*.982	*.685	*.873	*.561								
5.0		*.176	*.798	*.928	*.921	*.483	*.527	*.490								
7.5		*.155	*.411	*.850	*.867	*.392	*.352	*.449								
10.0		*.183	*.349	*.811	*.780	*.403	*.295	*.415								
15.0		*.199	*.325	*.552	*.388	*.311	*.239	*.389								
19.6		*.189	*.308	*.373	*.125	*.176	*.199	*.360								
24.5		*.217	*.293	*.010	*.011	*.086	*.177	*.254								
29.5		*.185	*.291	*.134	*.061	*.042	*.111	*.154								
34.5		*.214	*.308	*.195	*.099	*.012	*.056	*.086								
39.5		*.219	*.103	*.232	*.116	*.023	*.019	*.050								
44.5		*.249	*.178	*.249	*.139	*.055	*.013	*.053								
49.5		*.270	*.308	*.244	*.172	*.082	*.045	*.095								
54.5		*.095	*.363	*.225	*.213	*.093	*.068	*.153								
59.5		*.162	*.218	*.204	*.246	*.102	*.078	*.203								
64.5		*.325	*.230	*.187	*.276	*.095	.077	*.259								
69.6		*.844														
74.6		*.855	*.858	*.614	*.560	*.495	*.424	*.313								
79.5		*.838	*.855	*.611	*.552	*.493	*.431	*.318								
84.6		*.798	*.839	*.619	*.550	*.489	*.417	*.327								
89.6		*.715	*.843	*.638	*.554	*.486	*.402	*.322								
94.6		*.638	*.773	*.638	*.557	*.487	*.392	*.310								
1.3	Lower surface	*.399	*.527	*.506	*.498	*.487	*.467	*.424								
2.6		*.389	*.473	*.441	*.418	*.418	*.403	*.364								
5.0		*.420	*.379	*.362	*.349	*.344	*.330	*.298								
7.6		*.396	*.338	*.305	*.291	*.291	*.264	*.235								
10.1		*.375	*.312	*.292	*.252	*.259	*.225	*.190								
15.1		*.324	*.246	*.219	*.203	*.192	*.157	*.058								
19.6		*.307	*.207	*.176	*.149	*.141	*.111	*.071								
24.5		*.253	*.166	*.104	*.091	*.070	*.026	*.065								
29.5		*.213	*.136	*.104	*.091	*.070	*.026	*.165								
34.5		*.179	*.104	*.066	*.059	*.030	*.009	*.198								
39.5		*.149	*.076	*.038	*.037	*.000	*.045	*.232								
44.5		*.118	*.049	*.017	*.008	*.037	*.090	*.240								
49.5		*.082	*.024	*.005	*.023	*.013	*.137	*.255								
54.5		*.091	*.024													

TABLE I. - WING WITH SPOILER - Concluded

Percent chord	Pressure coefficient													
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
	M = 1.03	a = 8.54°	M = 1.03 a = 13.19°											
Upper surface														
0.0	+313	+162	+091	-+061	-+015	-+086	+188	+330	-+227	-+340	-+479	+339	-+522	-+181
1.2	+149	-+1047	-+1089	-+1078	-+871	-+469	-+653	+052	-+136	-+993	-+527	-+632	-+782	-+894
2.4	-+107	-+1047	-+1068	-+1075	-+838	-+453	-+572	-+320	-+113	-+975	-+515	-+652	-+816	-+904
5.0	-+236	-+900	-+969	-+982	-+769	-+415	-+543	-+330	-+102	-+975	-+499	-+723	-+869	-+913
7.5	-+218	-+977	-+1029	-+1025	-+798	-+443	-+543	-+319	-+101	-+963	-+499	-+833	-+818	-+929
10.0	-+270	-+889	-+950	-+924	-+774	-+411	-+542	-+319	-+101	-+963	-+499	-+760	-+765	-+935
15.0	-+281	-+390	-+905	-+549	-+573	-+418	-+459	-+50	-+95	-+949	-+520	-+483	-+754	-+924
19.6	-+254	-+370	-+823	-+458	-+593	-+432	-+398	-+455	-+913	-+939	-+612	-+420	-+778	-+909
24.5	-+266	-+349	-+407	-+389	-+511	-+457	-+325	-+432	-+884	-+881	-+704	-+405	-+815	-+813
29.5	-+242	-+350	-+076	-+330	-+107	-+472	-+294	-+404	-+856	-+848	-+776	-+426	-+853	-+735
34.5	-+266	-+356	-+192	-+25	-+094	-+474	-+272	-+406	-+756	-+815	-+829	-+457	-+883	-+710
39.5	-+264	-+310	-+117	-+193	-+069	-+451	-+268	-+370	-+549	-+4796	-+893	-+499	-+891	-+699
44.5	-+293	-+056	-+200	-+098	-+066	-+400	-+278	-+405	-+197	-+810	-+932	-+544	-+838	-+703
49.5	-+314	-+220	-+113	-+295	-+184	-+168	-+131	-+301	-+403	-+139	-+828	-+932	-+599	-+749
54.5	-+113	-+295	-+184	-+168	-+131	-+221	-+329	-+248	-+208	-+704	-+917	-+629	-+679	-+722
59.5	-+055	-+233	-+171	-+263	-+193	-+127	-+342	-+167	-+150	-+509	-+884	-+624	-+592	-+694
64.5	-+203	-+222	-+155	-+329	-+246	-+094	-+365	-+055	-+137	-+292	-+812	-+544	-+508	-+666
69.6	-+872	-+857	-+623	-+565	-+525	-+488	-+391	-+909	-+900	-+660	-+665	-+557	-+625	-+636
74.6	-+861	-+857	-+618	-+555	-+529	-+501	-+385	-+886	-+890	-+660	-+677	-+559	-+619	-+620
79.5	-+835	-+857	-+618	-+555	-+529	-+501	-+381	-+840	-+884	-+658	-+700	-+586	-+613	-+616
84.6	-+795	-+848	-+625	-+554	-+550	-+493	-+381	-+840	-+884	-+658	-+752	-+593	-+609	-+621
89.6	-+719	-+846	-+647	-+557	-+585	-+477	-+375	-+756	-+881	-+668	-+685	-+588	-+605	-+625
94.6	-+663	-+761	-+647	-+553	-+593	-+451	-+368	-+682	-+813	-+656	-+685	-+588	-+605	-+625
Lower surface														
1.3	+447	+603	+565	+552	+537	+517	+475	+560	+685	+632	+602	+566	+545	+507
2.6	+425	+561	+516	+489	+490	+466	+417	+465	+687	+624	+593	+556	+527	+487
5.0	+513	+472	+437	+424	+420	+386	+355	+668	+630	+574	+561	+512	+477	+437
7.6	+504	+424	+381	+367	+367	+335	+292	+694	+589	+521	+502	+461	+429	+383
10.1	+485	+387	+363	+339	+320	+297	+245	+682	+548	+502	+460	+419	+393	+339
15.1	+425	+333	+293	+276	+252	+228	+132	+606	+483	+424	+400	+352	+319	+216
19.6	+365	+287	+247	+240	+201	+183	+016	+535	+431	+379	+360	+303	+275	+114
24.5	+337	+242	+206	+194	+166	+138	-+053	+496	+384	+334	+311	+269	+231	+043
29.5	+294	+211	+172	+157	+127	+099	-+099	+447	+344	+300	+271	+225	+185	-+018
34.5	+254	+178	+134	+123	+091	+062	-+140	+399	+311	+256	+228	+188	+144	-+067
39.5	+225	+150	+100	+094	+054	+022	-+182	+366	+277	+223	+198	+151	+099	-+114
44.5	+190	+118	+083	+062	+024	-+029	-+204	+326	+240	+197	+160	+118	+058	-+151
49.5	+148	+088	+061	+026	+040	-+075	-+230	+283	+209	+170	+123	+101	+004	-+185
54.5	+160	+088	+038	-+001	-+035	-+127	-+237	+288	+203	+138	+093	+051	+004	-+212
59.5	+116	+062	+029	-+020	-+054	-+162	-+244	+241	+170	+120	+065	+024	-+081	-+232
64.5	+093	+048	+014	-+052	-+079	-+199	-+238	+212	+153	+094	+034	-+008	-+121	-+245
69.5	+061	-+016	+039	-+024	-+066	-+102	-+234	-+254	-+100	-+128	-+042	-+006	-+047	-+175
74.6	+078	-+001	+033	-+057	-+092	-+119	-+246	-+242	-+166	-+105	-+006	-+028	-+066	-+197
79.6	-+020	-+001	+020	-+071	-+107	-+139	-+278	-+250	-+041	-+076	-+015	-+049	-+102	-+243
84.6	-+020	-+029	+020	-+114	-+130	-+168	-+265	-+268	-+078	-+017	-+072	-+076	-+132	-+242
89.7	-+012	-+012	+013	-+164	-+176	-+213	-+260	-+278	-+060	-+064	-+124	-+133	-+187	-+300

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR

Pressure coefficient														
Percent chord	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
	M = 0.60	a = -0.06°						M = 0.60	a = 4.02°					
Upper surface	0.0	+204	+444	+460	+439	+361	+440	+176	+200	+097	+165	+102	+206	+082
	1.2	+262	+033	+112	+198	+212	+222	+194	+090	+752	+775	+681	+671	+639
	2.4	+116	-010	+072	+123	+164	+157	+113	+159	+620	+460	+412	+353	+348
	5.0	+055	-005	+057	+100	+127	+114	+061	+203	+346	+302	+244	+229	+361
	7.5	+031	-013	+057	+080	+096	+086	+032	+203	+282	+223	+197	+184	+195
	10.0	+003	-027	+042	+068	+081	+068	+009	+211	+263	+193	+162	+155	+166
	15.0	+008	-027	+042	+062	+073	+068	+001	+180	+221	+137	+116	+108	+107
	19.6	+017	-024	+029	+073	+086	+091	+002	+180	+194	+153	+057	+059	+046
	24.5	+032	+001	+076	+079	+091	+091	+012	+184	+141	+049	+037	+031	+026
	29.5	+012	+017	+096	+090	+103	+078	+012	+146	+109	+009	+006	+002	+001
	34.5	+027	+046	+116	+122	+125	+091	+015	+150	+061	+038	+040	+045	+030
	39.5	+006	+091	+147	+140	+156	+100	+010	+154	+086	+086	+086	+081	+081
	44.5	+046	+155	+194	+182	+182	+136	+021	+088	+091	+117	+146	+147	+107
	49.5	+046	+238	+250	+231	+244	+172	+049	+043	+203	+205	+191	+197	+170
	54.5	+107	+338	+314	+302	+303	+223	+070	+031	+287	+223	+219	+220	+045
	59.5	+233	+401	+327	+342	+322	+254	+085	+151	+302	+209	+224	+228	+175
	64.5	+381	+375	+324	+352	+322	+237	+108	+323	+079	+190	+229	+223	+094
	p5.4	+011	+011	+011	+011	+011	+011	+011	+106	+001	+001	+006	+006	+006
	74.6	+944	+784	+570	+455	+359	+288	+123	+887	+769	+539	+436	+369	+273
	79.5	+883	+756	+595	+449	+377	+288	+119	+845	+742	+554	+426	+382	+269
	84.6	+780	+747	+625	+494	+364	+281	+119	+772	+728	+588	+466	+377	+263
	89.6	+698	+731	+641	+506	+372	+271	+117	+710	+714	+609	+467	+377	+252
	94.6	+637	+684	+606	+512	+363	+259	+110	+672	+653	+583	+472	+376	+243
Lower surface	1.3	+131	-095	+137	+203	+249	+242	+196	+308	+367	+374	+342	+347	+278
	2.6	+887	+099	+130	+182	+156	+154	+155	+269	+301	+292	+289	+262	+276
	5.0	+029	+074	+087	+100	+099	+118	+129	+231	+199	+213	+201	+199	+144
	7.6	+011	-070	+101	+106	+093	+101	+111	+193	+157	+155	+147	+153	+149
	10.1	+032	+086	+081	+115	+098	+090	+098	+159	+127	+129	+116	+122	+053
	15.1	+052	+108	+106	+114	+097	+090	+101	+119	+080	+088	+076	+094	+002
	19.6	+076	+103	+106	+106	+096	+073	+109	+081	+052	+063	+077	+064	+005
	24.5	+082	+124	+111	+099	+082	+074	+088	+062	+020	+035	+046	+059	+033
	29.5	+099	+124	+102	+087	+068	+058	+060	+036	+001	+023	+036	+049	+025
	34.5	+109	+124	+088	+075	+045	+035	+054	+018	+008	+018	+031	+049	+032
	39.5	+120	+110	+067	+045	+013	+017	+055	+000	+007	+020	+040	+059	+045
	44.5	+120	+088	+034	+016	+022	+019	+062	+020	+001	+033	+052	+077	+056
	49.5	+117	+050	+012	+030	+082	+071	+071	+029	+022	+063	+081	+103	+097
	54.5	+093	+034	+085	+100	+144	+133	+079	+005	+088	+122	+136	+159	+147
	59.5	+013	+127	+173	+157	+204	+194	+089	+045	+157	+194	+184	+221	+102
	64.5	+151	+204	+213	+245	+218	+183	+095	+206	+234	+228	+275	+260	+116
	69.5	+311	+011	+011	+011	+011	+011	+011	+089	+404	+340	+340	+340	+324
	74.6	+1383	+846	+679	+502	+357	+294	+098	+1400	+821	+656	+517	+384	+273
	79.6	+894	+1015	+683	+524	+370	+302	+097	+874	+999	+656	+537	+388	+292
	84.6	+415	+1487	+675	+531	+355	+293	+092	+333	+1216	+653	+546	+381	+282
	89.7	+332	+1047	+662	+556	+371	+276	+094	+248	+996	+630	+561	+386	+269
	94.6	+156	+738	+654	+584	+369	+266	+083	+114	+579	+631	+581	+386	+258
	M = 0.60	a = 6.04°						M = 0.60	a = 8.05°					
Upper surface	0.0	+168	+340	+183	+552	+366	+547	+032	+137	+649	+504	+1077	+737	+133
	1.2	+059	+1067	+741	+1016	+1188	+979	+675	+257	+1059	+755	+1619	+952	+524
	2.4	+326	+951	+666	+748	+670	+676	+584	+525	+1020	+724	+1067	+892	+498
	5.0	+363	+664	+588	+507	+478	+470	+488	+578	+911	+682	+833	+771	+470
	7.5	+345	+528	+500	+403	+382	+360	+421	+521	+837	+623	+661	+735	+443
	10.0	+341	+457	+436	+324	+321	+307	+357	+498	+801	+579	+521	+631	+539
	15.0	+298	+362	+325	+234	+229	+225	+254	+436	+683	+483	+339	+383	+370
	19.6	+274	+301	+265	+160	+155	+159	+177	+379	+527	+439	+253	+253	+254
	24.5	+271	+336	+231	+129	+129	+129	+123	+371	+527	+439	+253	+253	+254
	29.5	+220	+181	+111	+063	+063	+078	+089	+295	+527	+439	+253	+253	+254
	34.5	+200	+051	+039	+020	+031	+003	+052	+296	+527	+439	+253	+253	+254
	39.5	+181	+046	+000	+020	+031	+029	+044	+251	+527	+439	+253	+253	+254
	44.5	+143	+043	+052	+056	+075	+039	+044	+222	+527	+439	+253	+253	+254
	49.5	+095	+149	+105	+097	+114	+077	+059	+152	+494	+035	+064	+018	+017
	54.5	+015	+234	+145	+129	+140	+108	+079	+065	+153	+003	+037	+002	+020
	59.5	+113	+247	+154	+153	+159	+115	+096	+088	+187	+032	+007	+014	+027
	64.5	+306	+083	+164	+168	+170	+108	+123	+266	+076	+052	+020	+025	+021
	69.5	+007	+011	+011	+011	+011	+011	+011	+143	+008	+008	+008	+008	+219
	74.6	+855	+747	+532	+430	+371	+272	+155	+809	+694	+501	+402	+353	+253
	79.5	+819	+725	+542	+419	+387	+269	+163	+769	+672	+511	+392	+369	+209
	84.6	+762	+717	+574	+457	+364	+263	+169	+723	+661	+538	+430	+369	+202
	89.6	+705	+697	+593	+457	+384	+253	+163	+675	+649	+555	+430	+377	+194
	94.6	+677	+641	+574	+461	+384	+248	+151	+656	+602	+536	+433	+373	+183
Lower surface	1.3	+363	+456	+446	+452	+455	+444	+355	+401	+495	+476	+476	+480	+394
	2.6	+317	+407	+380	+375	+380	+377	+295	+348	+479	+448	+446	+445	+351
	5.0	+306	+300	+302	+304	+312	+298	+235	+356	+395	+373	+384	+394	+288
	7.6	+280	+266	+241	+242	+253	+241	+165	+356	+350	+318	+326	+344	+225
	10.1	+244	+224	+210	+208	+222	+207	+114	+334	+305	+305	+285	+302	+181
	15.1	+201	+158	+157	+164	+157	+158	+045	+291	+232	+227	+234	+236	+091
	19.6	+156	+126	+120	+146	+128	+129	+004	+238	+195	+184	+207	+198	+054
	24.5	+136	+082	+087	+100	+116	+110	+012	+209	+152	+150	+159	+175	+013
	29.5	+104	+051	+070	+082	+093	+084	+013	+170	+125	+126	+126	+151	+006
	34.5	+079	+043	+056	+072	+090	+075	+029	+141	+104	+107	+121	+136	+013
	39.5	+053	+035	+056	+072	+092	+067	+045	+114	+093	+100	+118	+136	+026
	44.5	+027	+039	+063	+080	+102	+079	+061	+085	+093	+100	+121	+137	+043
	49.5	+014	+054	+086	+106	+113	+102	+075	+065	+100	+112	+136	+145	+063
	54.5	+043	+111	+130	+145	+171	+141	+091	+092	+151	+153	+172	+190	+080
	59.5	+078	+180	+199	+221	+231	+216	+109	+116	+207	+219	+232	+246	+099
	64.5	+232	+250	+236	+285	+265	+214	+121	+259	+280	+253	+302	+280	+228
	69.5	+455	+1431	+872	+542	+391	+294	+130	+152	+790	+633	+527	+367	+212
	74.6	+79.6	+1009	+658	+550	+405	+301	+132	+806	+992	+639	+538	+392	+287
	84.6	+287	+219	+550	+552	+385	+291	+132	+243	+180	+634	+538	+371	+275
	89.7	+203	+960	+562	+563	+401	+275	+138	+142	+876	+609	+546	+387	+259
	94.6	+087	+501	+624	+582	+399	+259	+151	+052	+396	+602	+556	+388	+238

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

Per-cent chord		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2
		M = 0.60				M = 0.60				M = 0.60				M = 0.60	
Upper surface	0.0	-0.002	-1.649	-1.129	+0.976	+0.692	+0.547	+0.253	-1.187	-2.053	-1.235	-0.685	+0.587	-0.498	+0.393
	1.2	-0.778	-1.589	-0.973	+0.778	+0.529	+0.460	+0.341	-1.207	-1.853	-1.029	-0.613	+0.456	-0.434	+0.349
	2.4	-1.029	-1.592	+0.973	+0.784	+0.569	+0.474	+0.336	-1.732	-1.860	-1.037	-0.614	+0.465	-0.429	+0.360
	5.0	-1.239	-1.645	-0.982	+0.802	+0.643	+0.422	+0.336	-2.023	-1.849	-1.046	-0.608	+0.488	-0.427	+0.360
	7.5	-1.057	-1.693	+0.975	+0.796	+0.621	+0.409	+0.336	-1.646	-1.873	-1.036	-0.608	+0.496	-0.423	+0.360
	10.0	-0.977	-1.787	-0.985	+0.771	+0.510	+0.415	+0.329	-1.335	-1.939	-1.036	-0.608	+0.502	-0.433	+0.359
	15.0	-0.783	-1.707	-0.992	+0.698	+0.372	+0.431	+0.328	-1.129	-1.908	-1.021	-0.614	+0.510	-0.441	+0.359
	19.6	-0.648	-1.468	-1.094	+0.650	+0.346	+0.437	+0.338	-0.941	-1.845	-1.049	-0.608	+0.518	-0.444	+0.359
	24.5	-0.585	-0.879	-1.081	+0.665	+0.351	+0.437	+0.339	-0.792	-1.691	-1.037	-0.640	+0.522	-0.452	+0.359
	29.5	-0.492	-0.246	-1.102	+0.694	+0.359	+0.461	+0.344	-0.667	-1.476	-1.058	-0.655	+0.532	-0.452	+0.359
	34.5	-0.411	+0.128	+1.076	+0.739	+0.366	+0.458	+0.344	-0.613	-1.138	-1.069	-0.682	+0.537	-0.445	+0.359
	39.5	-0.389	-0.094	-1.003	+0.776	+0.386	+0.446	+0.341	-0.530	-0.811	-1.064	-0.695	+0.529	-0.440	+0.357
	44.5	-0.326	-0.029	-0.862	+0.787	+0.422	+0.430	+0.330	-0.463	-0.525	-1.034	-0.682	+0.521	-0.430	+0.348
	49.5	-0.238	-0.054	-0.653	+0.766	+0.449	+0.413	+0.317	-0.361	-0.316	-0.973	-0.667	+0.507	-0.420	+0.331
	54.5	-0.108	+0.173	-0.460	+0.726	+0.466	+0.397	+0.303	-0.240	-0.163	-0.870	-0.640	+0.494	-0.410	+0.331
	59.5	-0.000	+0.219	-0.330	+0.663	+0.456	+0.377	+0.284	-0.152	-0.111	-0.780	-0.615	+0.479	-0.400	+0.308
	64.5	-0.092	+0.188	+0.247	+0.560	+0.451	+0.343	+0.279	-0.078	-0.146	-0.704	-0.584	+0.469	-0.383	+0.297
	69.5	-0.021	-0.224	-0.106	+0.268	+0.222	+0.222	+0.228	-0.022	-0.081	-0.669	-0.582	+0.477	-0.382	+0.282
	74.6	-0.744	-0.658	-0.481	+0.457	+0.333	+0.271	+0.259	-0.864	-0.727	-0.490	-0.384	+0.339	-0.293	+0.264
	79.5	-0.714	-0.638	-0.512	+0.477	+0.356	+0.311	+0.251	-0.784	-0.693	-0.538	-0.395	+0.367	-0.339	+0.257
	84.6	-0.668	-0.617	-0.575	+0.558	+0.362	+0.303	+0.243	-0.727	-0.654	-0.670	-0.458	+0.382	-0.330	+0.244
	89.6	-0.623	-0.601	-0.616	+0.571	+0.377	+0.288	+0.238	-0.639	-0.649	-0.643	-0.458	+0.379	-0.305	+0.236
	94.6	-0.614	-0.566	-0.523	+0.530	+0.365	+0.273	+0.224	-0.631	-0.604	-0.497	-0.458	+0.368	-0.287	+0.231
Lower surface	1.3	+0.410	+0.473	+0.482	+0.480	+0.494	+0.492	+0.418	+0.360	+0.391	+0.437	+0.457	+0.471	+0.465	+0.417
	2.6	+0.396	+0.538	+0.530	+0.512	+0.485	+0.441	+0.397	+0.400	+0.532	+0.548	+0.521	+0.490	+0.431	+0.396
	5.0	+0.426	+0.516	+0.499	+0.491	+0.447	+0.411	+0.355	+0.441	+0.586	+0.557	+0.509	+0.474	+0.439	+0.342
	7.6	+0.486	+0.483	+0.446	+0.434	+0.421	+0.397	+0.387	+0.428	+0.620	+0.526	+0.466	+0.439	+0.342	+0.311
	10.1	+0.494	+0.434	+0.424	+0.398	+0.387	+0.356	+0.356	+0.446	+0.569	+0.433	+0.406	+0.375	+0.346	+0.207
	15.1	+0.458	+0.374	+0.344	+0.334	+0.319	+0.292	+0.277	+0.376	+0.563	+0.433	+0.375	+0.303	+0.263	+0.162
	19.6	+0.393	+0.325	+0.299	+0.304	+0.277	+0.276	+0.276	+0.317	+0.499	+0.410	+0.382	+0.370	+0.337	+0.203
	24.5	+0.355	+0.278	+0.260	+0.255	+0.251	+0.249	+0.249	+0.317	+0.455	+0.373	+0.339	+0.320	+0.263	+0.092
	29.5	+0.309	+0.246	+0.227	+0.223	+0.219	+0.190	+0.190	+0.248	+0.408	+0.335	+0.306	+0.285	+0.263	+0.093
	34.5	+0.238	+0.197	+0.183	+0.182	+0.191	+0.146	+0.146	+0.205	+0.326	+0.272	+0.237	+0.255	+0.236	+0.032
	39.5	+0.202	+0.188	+0.173	+0.173	+0.184	+0.146	+0.137	+0.287	+0.258	+0.228	+0.229	+0.213	+0.198	+0.024
	44.5	+0.175	+0.188	+0.176	+0.179	+0.185	+0.149	+0.167	+0.258	+0.246	+0.219	+0.208	+0.205	+0.153	+0.064
	49.5	+0.191	+0.224	+0.202	+0.204	+0.210	+0.175	+0.194	+0.268	+0.275	+0.234	+0.221	+0.220	+0.177	+0.088
	54.5	+0.204	+0.263	+0.251	+0.256	+0.261	+0.222	+0.115	+0.272	+0.299	+0.273	+0.277	+0.261	+0.212	+0.109
	59.5	+0.327	+0.322	+0.284	+0.317	+0.300	+0.229	+0.138	+0.379	+0.343	+0.300	+0.332	+0.310	+0.223	+0.132
	64.5	+0.586	+0.704	+0.595	+0.508	+0.421	+0.335	+0.150	+0.098	+0.628	+0.648	+0.618	+0.495	+0.356	+0.152
	74.6	-0.1373	-0.704	-0.595	-0.508	+0.421	+0.335	+0.150	+0.098	+0.628	+0.648	+0.618	+0.495	+0.356	+0.152
	79.6	-0.632	-0.920	-0.600	-0.527	+0.413	+0.324	+0.150	+0.098	+0.628	+0.684	+0.650	+0.502	+0.356	+0.157
	84.6	-0.155	-0.702	-0.590	-0.554	+0.424	+0.312	+0.150	+0.081	+0.988	+0.690	+0.691	+0.495	+0.347	+0.157
	89.7	-0.026	-0.715	-0.545	-0.604	+0.441	+0.291	+0.166	+0.093	+0.547	+0.609	+0.729	+0.483	+0.325	+0.167
	94.6	+0.017	-0.247	-0.531	-0.595	+0.438	+0.253	+0.174	+0.081	+0.098	+0.562	+0.665	+0.631	+0.317	+0.183
Upper surface	0.0	+0.260	-1.030	+0.987	+0.764	+0.720	+0.613	+0.872	+0.245	+0.498	+0.516	+0.497	+0.474	+0.487	+0.234
	1.2	+0.842	-1.029	+0.909	+0.742	+0.687	+0.606	+0.547	+0.291	+0.103	+0.082	+0.184	+0.195	+0.219	+0.192
	2.4	+0.117	-1.026	+0.897	+0.747	+0.682	+0.609	+0.507	+0.148	+0.123	+0.051	+0.154	+0.155	+0.116	+0.116
	5.0	-1.103	-1.022	+0.893	+0.741	+0.684	+0.614	+0.498	+0.086	+0.013	+0.049	+0.102	+0.121	+0.118	+0.122
	7.5	-1.090	-1.022	+0.883	+0.741	+0.686	+0.622	+0.498	+0.042	+0.022	+0.061	+0.085	+0.098	+0.093	+0.040
	10.0	-1.033	-1.027	+0.874	+0.740	+0.686	+0.622	+0.498	+0.013	+0.033	+0.044	+0.074	+0.084	+0.076	+0.014
	15.0	-1.025	-1.027	+0.874	+0.740	+0.690	+0.615	+0.498	+0.008	+0.027	+0.049	+0.074	+0.081	+0.079	+0.009
	19.6	-1.025	-1.027	+0.865	+0.733	+0.686	+0.613	+0.503	+0.013	+0.030	+0.042	+0.095	+0.110	+0.117	+0.003
	24.5	-1.015	-1.016	+0.868	+0.739	+0.684	+0.613	+0.511	+0.036	+0.001	+0.105	+0.099	+0.106	+0.111	+0.009
	29.5	-0.997	-1.008	+0.864	+0.735	+0.678	+0.607	+0.512	+0.015	+0.025	+0.121	+0.116	+0.116	+0.093	+0.007
	34.5	-0.977	-1.007	+0.859	+0.733	+0.672	+0.603	+0.511	+0.029	+0.062	+0.148	+0.141	+0.140	+0.109	+0.002
	39.5	-0.943	-0.996	+0.849	+0.729	+0.665	+0.591	+0.506	+0.010	+0.119	+0.184	+0.124	+0.176	+0.127	+0.004
	44.5	-0.930	-0.973	+0.832	+0.723	+0.657	+0.574	+0.504	+0.025	+0.199	+0.242	+0.208	+0.216	+0.155	+0.020
	49.5	-0.902	-0.953	+0.821	+0.719	+0.645	+0.561	+0.497	+0.053	+0.298	+0.246	+0.268	+0.268	+0.197	+0.051
	54.5	-0.873	-0.925	+0.800	+0.710	+0.645	+0.556	+0.471	+0.134	+0.142	+0.334	+0.325	+0.325	+0.246	+0.088
	59.5	-0.832	-0.893	+0.783	+0.710	+0.652	+0.559	+0.426	+0.271	+0.428	+0.329	+0.355	+0.339	+0.269	+0.107
	64.5	-0.676	-0.851	+0.761	+0.705	+0.659	+0.563	+0.403	+0.231	+0.321	+0.315	+0.360	+0.336	+0.256	+0.141
	69.5	-0.993	-0.822	+0.615	+0.589	+0.443	+0.368	+0.355	+0.927	+0.806	+0.598	+0.462	+0.370	+0.298	+0.144
	74.6	-0.928	-0.808	+0.641	+0.567	+0.447	+0.403	+0.341	+0.857	+0.780	+0.653	+0.635	+0.475	+0.369	+0.292
	79.5	-0.844	-0.802	+0.666	+0.595	+0.434	+0.399	+0.326	+0.771	+0.653	+0.670	+0.651	+0.484	+0.370	+0.279
	84.6	-0.765	-0.784	+0.666	+0.578	+0.434	+0.376	+0.315	+0.707	+0.644	+0.621	+0.670	+0.477	+0.356	+0.252
	89.6	-0.710	-0.725	+0.666	+0.559	+0.422	+0.353	+0.301	+0.664	+0.715	+0.631	+0.662	+0.490	+0.359	+0.268
Lower surface	1.3	+0.304	+0.385	+0											

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

		Pressure coefficient														
Percent chord		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
	M = 0.80	$\alpha = 4.17^\circ$							M = 0.80	$\alpha = 6.17^\circ$						
Upper surface	0.0	+233	+221	+231	+157	+238	+116	+319	+237	+021	+010	+208	+087	+229	+106	
	1.2	+160	-899	-938	-816	-835	+799	-684	+064	-979	-699	-1031	-900	+836	+785	
	2.4	-102	-622	-525	-441	-380	-391	-416	-242	-897	-643	-750	-725	+697	+658	
	5.0	-151	-346	-350	-259	-250	-266	-274	-291	-701	-570	-550	-559	+573	+538	
	7.5	-184	-283	-261	-209	-209	-205	-222	-315	-588	-500	-449	-494	+438	+473	
	10.0	-200	-272	-220	-172	-178	-178	-186	-329	-512	-446	-354	-407	+356	+380	
	15.0	-180	-235	-148	-321	-123	-122	-117	-287	-402	-358	-233	-238	+248	+231	
	19.6	-181	-204	-143	-057	-065	-058	-063	-282	-327	-287	-134	-143	+167	+146	
	24.5	-198	-147	-036	-034	-036	-020	-055	-292	-254	-178	-110	-090	+107	+104	
	29.5	-156	-102	015	002	001	-001	-024	-235	-185	-108	-081	-052	+084	+077	
	34.5	-132	-046	068	049	048	044	-011	-237	-105	-046	-042	-018	+048	+058	
	39.5	-128	-043	027	100	107	090	-002	-196	-015	012	001	022	+013	+048	
	44.5	-085	154	184	157	155	135	-004	-147	092	062	038	060	+025	+051	
	49.5	-271	213	188	194	163	030	-009	204	113	066	096	058	+073	+073	
	54.5	-051	340	213	209	213	170	-065	012	289	149	090	119	+082	+100	
	59.5	-185	325	191	215	218	160	-093	156	284	160	115	139	+098	+125	
	64.5	-360	082	172	210	134	-127	-349	-125	169	139	151	107	+152	+170	
	69.5	-014	-879	-772	-597	-471	-394	-295	-163	-829	-741	-568	-453	+377	+281	
	74.6	-822	-755	-604	-463	-401	-296	-169	-783	-724	-573	-442	-382	+184	+184	
	79.5	-762	-734	-620	-487	-395	-290	-179	-737	-708	-588	-466	-372	+277	+186	
	84.6	-692	-715	-630	-487	-398	-282	-177	-668	-685	-592	-466	-383	+267	+177	
	89.6	-686	-670	-616	-489	-396	-275	-169	-671	-653	-578	-466	-385	+262	+168	
Lower surface	1.3	+331	+390	+396	+398	+371	+368	+305	+395	+476	+475	+478	+467	+462	+395	
	2.6	+294	+327	+314	+297	+287	+299	+229	+352	+427	+410	+401	+397	+400	+330	
	5.0	+263	+230	+233	+233	+219	+214	+169	+327	+327	+328	+329	+327	+318	+260	
	7.6	+219	+186	+180	+178	+171	+165	+110	+315	+299	+270	+274	+275	+264	+198	
	10.1	+181	+165	+153	+146	+136	+151	+070	+275	+244	+247	+236	+243	+232	+144	
	15.1	+146	+110	+112	+110	+107	+102	+007	+237	+183	+190	+193	+190	+185	+169	
	19.6	+105	+083	+081	+101	+076	+090	+014	+191	+149	+149	+137	+136	+160	+013	
	24.5	+089	+052	+059	+064	+066	+055	-035	+171	+115	+126	+131	+137	+121	-002	
	29.5	+052	+036	+044	+051	+055	+046	-026	+138	+095	+107	+114	+119	+114	-001	
	34.5	+039	+029	+040	+046	+059	+043	-034	+110	+083	+095	+102	+112	+092	+014	
	39.5	+023	+036	+046	+054	+067	+047	-040	+086	+084	+093	+104	+116	+087	+025	
	44.5	+004	+053	+062	+070	+091	+065	-054	+063	+095	+100	+111	+130	+098	+040	
	49.5	-001	+083	+091	+104	+130	+100	-083	+053	+121	+126	+135	+151	+123	+071	
	54.5	+034	+153	+146	+163	+179	+156	-105	+088	+180	+173	+180	+202	+172	+095	
	59.5	+101	+215	+222	+221	+242	+231	-127	+138	+235	+240	+254	+257	+241	+118	
	64.5	+278	+426	+428	+429	+459	+421	-141	+311	+286	+280	+311	+385	+296	+132	
	69.5	+467	-879	-772	-597	-471	-394	-295	-163	-829	-741	-568	-453	+377	+139	
	74.6	-1099	-817	-669	-523	-415	-318	-190	-129	-751	-635	-514	-401	+296	+146	
	79.6	-767	-922	-582	-528	-416	-314	-195	-801	-902	-623	-516	-408	+301	+180	
	84.6	-458	-1029	-650	-528	-416	-307	-195	-406	-1052	-616	-516	-404	+294	+150	
	89.7	-322	-994	-634	-542	-418	-295	-169	-252	-1013	-609	-530	-409	+283	+166	
	94.6	-233	-785	-633	-580	-414	-282	-169	-178	-744	-604	-563	-404	+261	+166	
	M = 0.80	$\alpha = 8.20^\circ$							M = 0.80	$\alpha = 12.19^\circ$						
Upper surface	0.0	+226	-274	-337	-493	-389	-514	-039	+163	-818	-915	-735	-465	+669	+333	
	1.2	-052	-1084	-978	-969	-879	-852	-511	+306	-1625	-1054	-638	-452	+618	+508	
	2.4	-401	-1050	-944	-983	-846	-896	-499	+685	-1612	-1042	-632	-485	+681	+498	
	5.0	-500	-946	-930	-805	-799	-783	-484	+817	-1544	-1037	-622	-538	+600	+494	
	7.5	-497	-885	-888	-916	-589	-825	-543	+929	-1540	-1018	-613	-559	+578	+487	
	10.0	-434	-864	-914	-368	-751	-484	-451	+917	-1554	-1016	-604	-496	+577	+479	
	15.0	-402	-745	-745	-747	-226	-438	-447	+835	-1496	-997	-589	-370	+599	+463	
	19.6	-389	-634	-728	-492	-192	-283	-420	+750	-1388	-1012	-589	-334	+599	+463	
	24.5	-332	-297	-319	-182	-182	-370	-353	+714	-1222	-981	-517	-351	+599	+564	
	29.5	-344	-181	-067	-182	-110	-335	-316	+629	-787	-981	-672	-378	+607	+446	
	34.5	-293	-084	-121	-190	-110	-155	-275	+574	-337	-979	-731	-409	+594	+436	
	39.5	-231	-016	134	-208	-015	-195	-183	+504	-177	-979	-765	-439	+569	+423	
	44.5	-159	-117	151	-225	-073	-120	-259	+450	-076	-950	-755	-477	+542	+411	
	49.5	-032	-207	156	-228	-051	-085	-257	+291	-042	-848	-732	-506	+515	+395	
	54.5	-141	-183	154	-228	-024	-014	-251	+153	-114	-703	-698	-519	+490	+381	
	59.5	-253	-136	166	-213	-003	-012	-255	+059	-079	-552	-646	-517	+463	+356	
	64.5	-006	-797	-700	-546	-436	-346	-287	+287	-234	-492	-391	-339	+333	+333	
	69.5	-749	-684	-554	-442	-358	-285	-226	+714	-660	-599	-506	-404	+379	+309	
	74.6	-709	-666	-559	-446	-357	-280	-222	+679	-641	-570	-453	-385	+385	+296	
	79.5	-645	-645	-555	-450	-363	-272	-217	+634	-622	-690	-549	-416	+377	+285	
	84.6	-650	-617	-540	-457	-360	-265	-210	+630	-596	-470	-531	-408	+362	+270	
Lower surface	1.3	+434	+534	+524	+505	+501	+506	+440	+458	+549	+524	+521	+516	+489	+443	
	2.6	+379	+503	+485	+458	+465	+467	+389	+385	+57:	+545	+529	+511	+479	+429	
	5.0	+413	+419	+410	+397	+405	+402	+328	+504	+507	+507	+502	+467	+436	+389	
	7.6	+399	+376	+352	+339	+353	+346	+267	+539	+486	+486	+455	+444	+422	+329	
	10.1	+367	+330	+328	+307	+312	+314	+212	+530	+442	+437	+404	+380	+358	+233	
	15.1	+325	+265	+262	+253	+255	+247	+114	+476	+378	+360	+341	+323	+295	+177	
	19.6	+271	+224	+220	+227	+216	+218	+075	+407	+332	+316	+317	+283	+255	+127	
	24.5	+245	+186	+183	+182	+191	+178	+027	+369	+291	+275	+266	+250	+217	+067	
	29.5	+208	+162	+163	+161	+170	+152	+018	+323	+242	+247	+234	+219	+188	+042	
	34.5	+174	+149	+146	+146	+157	+135	-003	+284	+237	+224	+205	+198	+164	+009	
	39.5	+150	+143	+139	+142	+157	+124	-015	+251	+224	+205	+198	+188	+140	+019	
	44.5	+126	+148	+139	+147	+164	+127	-039	+220	+220	+196	+192	+186	+137	+055	
	49.5	+106	+163	+155	+164	+177	+145	-069	+195	+225	+201	+197	+193	+142	+094	
	54.5	+142	+216	+196	+204	+224	+185	-093	+222	+263	+224	+222	+224	+171	+130	
	59.5	+179	+263	+260	+264	+278	+247	-117	+246	+300	+276	+273	+279	+223	+162	
	64.5	+344	+313	+294	+329	+307	+246	+132	+391	+347	+312	+331	+307	+224	+180	
	69.5	+598	-1235	-707	-607	-510	-384	-292	-143	-1148	-641	-579	-519	+429	+370	
	74.6	-784	-853	-599	-512	-386	-297	-144	-689	-829	-576	-528	-449	+370	+193	
	79.6	-383	-1064	-593	-507	-374	-291	-145	-337	-1019	-562	-558	-460	+359	+193	

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

Percent chord		Pressure coefficient													
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	
		M = 0.80	α = 16.73°	M = 0.80 α = 20.95°											
Upper surface	0.0	*065	-1.256	-1.225	-0.558	-0.529	-0.529	-0.421	-0.052	-0.875	-0.732	-0.692	-0.642	-0.928	
	1.2	-0.536	-1.371	-0.977	-0.465	-0.492	-0.389	-0.591	-0.853	-0.821	-0.714	-0.660	-0.629	-0.579	
	2.4	-1.024	-1.378	-0.969	-0.610	-0.473	-0.497	-0.389	-0.910	-0.855	-0.811	-0.714	-0.660	-0.637	-0.526
	5.0	-1.177	-1.347	-0.963	-0.601	-0.485	-0.497	-0.386	-0.880	-0.848	-0.810	-0.710	-0.660	-0.637	-0.511
	7.5	-1.267	-1.336	-0.947	-0.598	-0.495	-0.493	-0.389	-0.868	-0.848	-0.803	-0.707	-0.660	-0.633	-0.511
	10.0	-1.228	-1.324	-0.935	-0.592	-0.507	-0.497	-0.394	-0.874	-0.848	-0.803	-0.706	-0.661	-0.642	-0.509
	15.0	-1.062	-1.303	-0.928	-0.584	-0.524	-0.499	-0.402	-0.854	-0.850	-0.797	-0.698	-0.663	-0.640	-0.500
	19.6	-0.891	-1.289	-0.909	-0.578	-0.531	-0.503	-0.407	-0.859	-0.850	-0.794	-0.698	-0.663	-0.626	-0.503
	24.5	-0.834	-1.266	-0.879	-0.611	-0.537	-0.492	-0.415	-0.852	-0.851	-0.785	-0.693	-0.663	-0.612	-0.506
	29.5	-0.799	-1.222	-0.871	-0.630	-0.543	-0.499	-0.421	-0.845	-0.851	-0.782	-0.687	-0.662	-0.614	-0.512
	34.5	-0.780	-1.165	-0.864	-0.645	-0.546	-0.488	-0.424	-0.830	-0.851	-0.775	-0.685	-0.660	-0.600	-0.514
	39.5	-0.731	-1.107	-0.864	-0.546	-0.428	-0.413	-0.353	-0.849	-0.775	-0.660	-0.600	-0.514	-0.423	
	44.5	-0.688	-1.011	-0.845	-0.698	-0.543	-0.467	-0.421	-0.826	-0.763	-0.678	-0.655	-0.580	-0.515	
	49.5	-0.577	-0.901	-0.813	-0.626	-0.536	-0.461	-0.410	-0.819	-0.844	-0.751	-0.672	-0.648	-0.560	-0.507
	54.5	-0.447	-0.791	-0.769	-0.611	-0.534	-0.458	-0.394	-0.806	-0.834	-0.736	-0.669	-0.648	-0.546	-0.488
	59.5	-0.317	-0.688	-0.719	-0.599	-0.523	-0.453	-0.355	-0.785	-0.821	-0.727	-0.667	-0.653	-0.553	-0.453
	64.5	-0.175	-0.489	-0.673	-0.579	-0.525	-0.430	-0.371	-0.725	-0.803	-0.708	-0.665	-0.658	-0.565	-0.446
	69.5	-0.006						-0.36							
Lower surface	74.6	-0.895	-0.830	-0.581	-0.517	-0.411	-0.355	-0.340	-0.845	-0.792	-0.615	-0.577	-0.470	-0.401	-0.408
	79.5	-0.853	-0.800	-0.590	-0.519	-0.415	-0.374	-0.332	-0.815	-0.787	-0.625	-0.573	-0.472	-0.397	
	84.6	-0.778	-0.705	-0.619	-0.541	-0.409	-0.371	-0.317	-0.777	-0.762	-0.632	-0.573	-0.463	-0.427	-0.381
	89.6	-0.710	-0.769	-0.597	-0.519	-0.416	-0.357	-0.307	-0.756	-0.752	-0.632	-0.564	-0.461	-0.415	-0.366
	94.6	-0.663	-0.705	-0.601	-0.495	-0.401	-0.341	-0.299	-0.746	-0.728	-0.621	-0.551	-0.451	-0.399	-0.357
	1.3	*466	*536	*501	*492	*494	*478	*426	*383	*514	*460	*420	*426	*411	*381
	2.6	*367	*618	*584	*562	*515	*458	*204	*628	*578	*539	*493	*449		
	5.0	*596	*630	*585	*567	*531	*501	*428	*648	*682	*609	*577	*550	*515	*445
	7.6	*496	*602	*549	*526	*498	*472	*380	*789	*668	*590	*552	*531	*505	*412
	10.1	*696	*565	*539	*491	*466	*440	*344	*781	*638	*585	*528	*501	*479	*379
	15.1	*628	*506	*470	*434	*411	*377	*242	*709	*587	*521	*479	*452	*418	*276
	19.6	*590	*457	*421	*399	*369	*344	*174	*637	*537	*477	*443	*418	*385	*199
	24.5	*510	*417	*382	*350	*332	*295	*115	*596	*493	*439	*398	*381	*337	*143
	29.5	*461	*380	*349	*314	*298	*256	*082	*541	*458	*402	*361	*342	*296	*099
	34.5	*416	*349	*319	*279	*267	*224	*039	*499	*424	*369	*326	*307	*258	*049
	39.5	*379	*328	*291	*256	*251	*198	*005	*460	*396	*340	*299	*285	*226	*006
	44.5	*346	*311	*271	*239	*239	*184	*046	*423	*370	*315	*272	*264	*206	*049
	49.5	*316	*305	*257	*229	*281	*179	*092	*389	*358	*291	*256	*263	*193	*157
	54.5	*329	*332	*266	*245	*257	*198	*132	*397	*376	*294	*260	*265	*200	*158
	59.5	*341	*352	*302	*287	*304	*237	*163	*403	*390	*315	*297	*299	*233	*207
	64.5	*467	*384	*332	*345	*341	*250	*192	*514	*414	*341	*347	*335	*247	
	69.5	*681													
Lower surface	74.6	-0.931	-0.588	-0.703	-0.678	-0.450	-0.385	-0.222	-0.868	-0.530	-0.790	-0.704	-0.546	-0.458	-0.278
	79.5	-0.520	-0.749	-0.831	-0.665	-0.466	-0.384	-0.213	-0.840	-0.600	-0.687	-0.621	-0.579	-0.561	-0.278
	84.6	-0.171	-0.910	-0.825	-0.648	-0.469	-0.381	-0.210	-0.838	-0.652	-0.852	-0.800	-0.785	-0.555	-0.457
	89.7	-0.015	-0.651	-0.731	-0.632	-0.462	-0.373	-0.225	-0.819	-0.522	-0.702	-0.775	-0.562	-0.452	-0.283
	94.6	-0.140	-0.221	-0.655	-0.592	-0.450	-0.352	-0.219	-0.814	-0.585	-0.695	-0.537	-0.426	-0.283	
	1.3	*4224	*037	*023	*034	*119	*157	*155	*356	*399	*417	*422	*388	*383	*324
	2.6	*4185	*013	*014	*051	*061	*097	*130	*319	*339	*338	*330	*316	*316	*250
	5.0	*4121	*009	*003	*005	*047	*063	*121	*290	*248	*261	*259	*243	*234	*183
	7.6	*078	*001	*017	*024	*052	*075	*109	*252	*202	*209	*208	*193	*192	*119
	10.1	*047	*015	*006	*042	*065	*070	*114	*213	*167	*184	*175	*153	*163	*074
	15.1	*026	*039	*055	*070	*076	*111	*175	*129	*145	*136	*124	*116	*116	*004
	19.6	-0.014	-0.045	*035	*035	*070	*074	*088	*131	*101	*116	*120	*098	*086	
	24.5	-0.021	*071	*041	*052	*063	*072	*078	*117	*067	*093	*086	*080	*065	-0.047
	29.5	-0.043	*073	*031	*044	*051	*057	*050	*085	*058	*081	*072	*067	*053	-0.037
	34.5	-0.062	*064	*018	*034	*030	*034	*040	*062	*055	*076	*065	*067	*046	-0.037
	39.5	-0.074	*035	*001	*010	*003	*010	*036	*044	*065	*079	*070	*074	*048	-0.043
	44.5	-0.083	*009	*030	*026	*037	*031	*047	*025	*091	*091	*086	*095	*065	-0.057
	49.5	-0.059	*059	*075	*077	*114	*088	*079	*025	*129	*120	*120	*126	*100	-0.097
	54.5	-0.018	*159	*146	*147	*172	*163	*117	*068	*199	*173	*170	*189	*161	-0.140
	59.5	-0.097	*240	*238	*203	*227	*230	*147	*142	*257	*246	*216	*245	*237	-0.173
	64.5	-0.267	*306	*267	*264	*235	*207	*160	*327	*292	*266	*289	*261	*215	-0.189
	69.5	-0.413													
Lower surface	74.6	L1961	-0.494	-0.691	-0.521	-0.421	-0.346	-0.160	-0.964	-0.880	-0.673	-0.573	-0.450	-0.223	-0.196
	79.6	-0.727	-0.936	-0.684	-0.518	-0.419	-0.341	-0.163	-0.762	-0.917	-0.662	-0.567	-0.456	-0.338	-0.204
	84.6	-0.525	-0.925	-0.673	-0.510	-0.422	-0.337	-0.159	-0.500	-0.950	-0.654	-0.555	-0.451	-0.335	-0.208
	89.7	-0.464	-0.940	-0.652	-0.513	-0.421	-0.322	-0.170	-0.570	-0.975	-0.637	-0.554	-0.448	-0.322	-0.226
	94.6	-0.365	-0.836	-0.652	-0.532	-0.416	-0.305	-0.168	-0.571	-0.832	-0.631	-0.580	-0.443	-0.306	-0.218

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

Percent chord		Pressure coefficient												
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2
M = 0.90														
Upper surface														
0.0	+260	+091	+026	-015	+060	+160	+114	+266	+094	+197	+405	+268	+402	+008
1.2	+120	-1421	-1+21	-1+004	-1+085	+639	+763	+042	-1+307	-1+407	+981	+953	+752	+527
2.4	+116	-1421	-1+21	-1+004	-1+085	+584	+692	+279	-1+278	-1+362	+972	+885	+774	+514
5.0	+286	-1+723	-1+047	+581	+648	+552	+627	+423	-1+040	-1+322	+1+035	+879	+749	+497
7.5	+293	-1+565	+851	+555	+536	+488	+594	+432	+865	+1+193	+955	+934	+629	+488
10.0	+302	-1+520	+607	+552	+536	+446	+536	+453	+795	+1+193	+463	+900	+583	+475
15.0	+317	-1+475	+304	+241	+358	+400	+364	+422	+699	+1+803	+314	+493	+553	+454
19.6	+304	-1+431	+196	+149	+213	+369	+251	+410	+640	+1+734	+277	+305	+537	+448
24.5	+337	-1+369	+059	+141	+136	+314	+181	+430	+580	+1+382	+287	+259	+523	+426
29.5	+298	-1+206	+004	+118	+100	+286	+135	+396	+466	+011	+262	+245	+516	+411
34.5	+334	-094	+053	+091	+074	+229	+116	+428	+236	+046	+241	+219	+487	+400
39.5	+294	+007	+094	+062	+036	+160	+100	+412	+077	+064	+229	+188	+436	+394
44.5	+210	+122	+127	+035	+006	+083	+103	+406	+046	+076	+221	+171	+374	+385
49.5	+102	+235	+139	+014	+020	+018	+120	+218	+147	+074	+217	+149	+289	+385
54.5	+025	+303	+134	+007	+042	+039	+156	+009	+205	+064	+211	+133	+206	+380
59.5	+183	+276	+127	+025	+068	+078	+180	+125	+179	+049	+200	+111	+140	+360
64.5	+343	+148	+115	+045	+082	+095	+215	+246	+144	+036	+183	+092	+102	+353
69.6	+000						+222	+006						+332
74.6	-877	-1+799	+632	+508	+403	+338	+238	+842	+779	+623	+487	+394	+343	+317
79.5	+831	-1+787	+631	+508	+409	+340	+240	+811	+766	+623	+490	+404	+347	+308
84.6	+783	-1+777	+635	+510	+409	+336	+244	+769	+755	+632	+492	+404	+343	+297
89.6	+709	-1+750	+635	+509	+415	+331	+239	+713	+729	+632	+491	+413	+337	+287
94.6	+709	-1+720	+625	+507	+419	+324	+234	+697	+701	+610	+490	+415	+330	+278
Lower surface														
1.3	+402	+482	+483	+461	+451	+387	+442	+538	+516	+504	+498	+496	+437	
2.6	+364	+432	+421	+398	+390	+327	+392	+460	+476	+460	+457	+455	+389	
5.0	+366	+333	+343	+343	+327	+300	+262	+440	+411	+405	+398	+400	+387	+332
7.6	+332	+283	+287	+286	+269	+256	+205	+421	+363	+349	+341	+345	+337	+268
10.1	+296	+251	+265	+251	+232	+216	+149	+388	+324	+322	+309	+304	+301	+221
15.1	+253	+202	+212	+206	+189	+174	+051	+338	+271	+266	+256	+244	+238	+177
19.6	+202	+160	+176	+181	+156	+142	+003	+284	+227	+226	+225	+215	+208	+043
24.5	+183	+127	+146	+144	+134	+109	+032	+256	+190	+194	+184	+187	+165	+008
29.5	+148	+111	+129	+125	+113	+089	+028	+215	+170	+172	+164	+162	+151	+001
34.5	+117	+104	+118	+112	+104	+075	+036	+183	+157	+155	+146	+152	+122	+021
39.5	+098	+108	+115	+109	+108	+067	+046	+155	+134	+134	+129	+149	+106	+035
44.5	+071	+127	+120	+119	+120	+078	+067	+129	+166	+149	+142	+152	+111	+057
49.5	+066	+156	+144	+145	+144	+100	+104	+115	+186	+161	+162	+168	+124	+101
54.5	+109	+216	+187	+188	+197	+157	+154	+159	+239	+197	+203	+218	+170	+140
59.5	+166	+267	+255	+234	+253	+226	+175	+198	+284	+280	+256	+270	+237	+177
64.5	+344	+303	+283	+308	+1274	+211	+188	+369	+321	+295	+291	+290	+287	
69.6	+564													+189
74.6	-1+002	-1+818	+659	+570	+448	+343	+198	+990	+769	+665	+571	+432	+348	+192
79.5	+785	+878	+652	+567	+453	+351	+208	+772	+685	+658	+569	+443	+348	+204
84.6	+491	+972	+652	+553	+446	+346	+214	+505	+1+033	+648	+561	+439	+344	+210
89.7	+367	+1+020	+637	+551	+446	+334	+231	+436	+1+021	+629	+569	+439	+329	+228
94.6	+246	+795	+626	+571	+445	+315	+227	+183	+674	+611	+588	+436	+316	+223
M = 0.90														
Upper surface														
0.0	+250	-1+473	+607	+660	+435	+664	+343	+158	+889	+1+013	+767	+528	+539	+429
1.2	+108	+1+379	+1+216	+593	+504	+654	+649	+302	+1+378	+947	+711	+477	+502	+410
2.4	+456	+1+373	+1+110	+585	+545	+696	+645	+743	+1+382	+931	+705	+483	+507	+412
5.0	+546	+1+319	+1+103	+570	+601	+682	+641	+886	+1+356	+931	+692	+491	+512	+411
7.5	+701	+1+285	+1+072	+566	+626	+648	+633	+1+008	+1+345	+921	+684	+502	+513	+411
10.0	+705	+1+236	+1+061	+565	+545	+658	+618	+993	+1+338	+915	+673	+513	+515	+414
15.0	+659	+1+151	+1+030	+560	+389	+676	+585	+937	+1+287	+926	+653	+528	+516	+412
19.6	+628	+1+093	+1+017	+562	+335	+683	+572	+864	+1+228	+914	+638	+538	+518	+410
24.5	+609	+1+058	+969	+567	+343	+686	+545	+791	+1+181	+889	+662	+547	+509	+411
29.5	+572	+1+064	+939	+666	+381	+702	+532	+737	+1+133	+883	+673	+558	+515	+415
34.5	+581	+813	+923	+723	+406	+692	+522	+712	+1+077	+864	+673	+562	+510	+421
39.5	+546	+534	+911	+740	+433	+663	+510	+658	+1+042	+866	+670	+562	+502	+427
44.5	+559	+207	+887	+731	+472	+624	+500	+627	+965	+841	+658	+558	+494	+428
49.5	+453	+000	+801	+713	+500	+572	+493	+574	+847	+813	+650	+554	+486	+428
54.5	+306	+059	+649	+684	+521	+529	+480	+472	+725	+783	+632	+553	+482	+421
59.5	+233	+055	+845	+635	+526	+489	+440	+384	+694	+748	+625	+544	+479	+404
64.5	+058	+071	+308	+574	+532	+432	+420	+244	+699	+684	+606	+551	+464	+415
69.6	+011						+402							+400
74.6	+763	+734	+596	+524	+421	+387	+387	+884	+868	+810	+526	+456	+403	+393
79.5	+757	+723	+635	+542	+432	+414	+376	+901	+845	+828	+526	+458	+413	+392
84.6	+728	+713	+727	+549	+432	+423	+363	+875	+860	+848	+536	+450	+413	+383
89.6	+669	+685	+676	+555	+439	+419	+350	+760	+798	+842	+529	+457	+406	+370
94.6	+650	+663	+545	+553	+433	+404	+360	+717	+765	+807	+517	+445	+394	+366
Lower surface														
1.3	+490	+596	+551	+537	+522	+500	+440	+511	+598	+545	+508	+503	+485	+442
2.6	+403	+600	+557	+537	+514	+485	+430	+368	+658	+611	+571	+556	+520	+470
5.0	+555	+542	+514	+505	+474	+377	+383	+658	+655	+604	+580	+539	+507	+442
7.6	+581	+497	+462	+448	+427	+400	+323	+749	+626	+566	+539	+508	+480	+392
10.1	+562	+458	+444	+414	+387	+365	+288	+737	+588	+557	+508	+449	+436	
15.1	+502	+395	+374	+330	+311	+301	+170	+662	+528	+486	+451	+416	+385	+250
19.6	+346	+346	+330	+297	+275	+224	+053	+538	+439	+403	+369	+341	+307	+118
24.5	+393	+306	+297	+275	+260	+224	+053	+538	+439	+403	+369	+341	+307	+118
29.5	+346	+282	+266	+246	+229	+193	+025	+487	+407	+370	+335	+304	+266	+079
34.5	+307	+260	+247	+221	+209	+170	+001	+444	+379	+341	+302	+277	+236	+034
39.5	+274	+248	+228	+206	+198	+146	+031	+410	+360	+316	+276	+258	+211	+002
44.5	+240	+248	+221	+198	+193	+139	+073	+377	+344	+298	+254	+243	+191	+053
49.5	+221	+253	+221	+207	+203	+142	+131	+345	+340	+285	+249	+293	+185	+121
54.5	+255	+296	+244	+232	+234	+175	+185	+363	+368	+293	+262	+258	+205	+189
59.5	+281	+327	+293	+274	+280	+228	+224	+376	+386	+329	+305	+300	+247	+244
64.5	+429	+362	+323	+338	+312	+232	+241	+511	+414	+356	+359	+336		

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued.

Percent chord	Pressure coefficient														
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
	M = 0.90				M = 0.94				M = 0.26°						
Upper surface	0, 0	*040	*895	-872	-737	-603	-507	-972	*275	*535	*567	*563	*489	*515	*297
	1, 2	*444	*867	-838	-714	-635	-404	-591	*324	*005	*040	*164	*158	*183	*170
	2, 4	*814	*867	-826	-716	-630	-403	-550	*191	-032	*025	*109	*191	*181	*101
	5, 0	*856	*867	-826	-709	-628	-406	-528	*123	-008	*041	*106	*108	*103	*060
	7, 5	*838	*866	-816	-708	-627	-409	-528	*081	-019	*049	*099	*087	*084	*027
	10, 0	*849	*868	-809	-703	-629	-413	-528	*049	-028	*049	*090	*084	*064	*001
	15, 0	*856	*863	-825	-695	-631	-517	-528	*025	-028	*072	*093	*085	*072	*035
	19, 6	*859	*863	-816	-685	-630	-530	-528	*010	-031	*083	*110	*105	*108	*024
	24, 5	*854	*867	-804	-695	-628	-534	-528	*025	*008	*141	*115	*102	*094	*018
	29, 5	*849	*867	-810	-693	-628	-534	-534	*004	*040	*170	*131	*117	*094	*003
	34, 5	*844	*857	-803	-693	-627	-532	-537	*018	*099	*198	*167	*140	*109	*013
	39, 5	*809	*859	-800	-692	-627	-540	-542	*005	*172	*235	*187	*174	*129	*013
	44, 5	*809	*849	-786	-688	-627	-519	-537	*033	*265	*281	*231	*216	*160	*008
	49, 5	*797	*854	-771	-688	-623	-502	-530	*087	*372	*340	*285	*270	*205	*061
	54, 5	*771	*841	-749	-674	-627	-587	-516	*186	*479	*376	*348	*326	*258	*133
	59, 5	*715	*836	-734	-670	-623	-585	-484	*334	*505	*371	*370	*342	*281	*190
	64, 5	*598	*807	-718	-650	-634	-585	-494	*471	*435	*353	*376	*345	*270	*242
	69, 6	*066						-479	*029						*248
	74, 6	*868	*821	-641	-616	-526	-445	-457	*963	-837	*680	*560	*458	*358	*295
	79, 5	*846	*811	-658	-609	-526	-446	-456	*899	*816	*688	*544	*454	*359	*247
	84, 6	*815	*737	-666	-602	-518	-446	-435	*780	-805	*694	*549	*451	*359	*245
	89, 6	*768	*778	-672	-595	-513	-459	-421	*696	*800	*702	*548	*446	*352	*234
	94, 6	*749	*754	-655	-586	-501	-443	-413	*695	*755	*683	*547	*444	*346	*221
Lower surface	1, 3	*401	*574	*508	*456	*449	*434	*380	*244	*063	*048	*001	*088	*139	*158
	2, 6	*202	*677	*615	*568	*554	*508	*465	*205	*036	*012	*020	*043	*081	*132
	5, 0	*714	*722	*644	*607	*571	*529	*493	*143	*037	*024	*035	*019	*061	*122
	7, 6	*838	*708	*624	*581	*552	*515	*422	*099	*015	*024	*002	*033	*070	*113
	10, 1	*824	*679	*620	*558	*524	*490	*393	*166	*011	*015	*014	*051	*067	*127
	15, 1	*751	*625	*559	*511	*478	*432	*294	*039	*020	*013	*028	*058	*1077	*193
	19, 6	*683	*580	*519	*476	*441	*405	*227	*003	*029	*011	*009	*058	*146	*082
	24, 5	*639	*538	*479	*432	*402	*351	*152	*005	*061	*015	*029	*055	*1074	*050
	29, 5	*591	*502	*445	*395	*362	*310	*103	*027	*061	*007	*023	*043	*066	*038
	34, 5	*548	*470	*415	*358	*330	*275	*053	*047	*007	*014	*028	*047	*047	*038
	39, 5	*511	*446	*386	*333	*295	*243	*009	*060	*012	*012	*009	*003	*023	*030
	44, 5	*473	*422	*362	*306	*281	*219	*048	*074	*039	*056	*043	*033	*021	*037
	49, 5	*440	*412	*342	*290	*319	*204	*132	*058	*099	*099	*094	*101	*083	*069
	54, 5	*420	*433	*342	*292	*281	*212	*204	*002	*189	*163	*158	*172	*166	*126
	59, 5	*456	*440	*366	*312	*311	*246	*265	*120	*272	*253	*215	*230	*244	*184
	64, 5	*718	*462	*384	*362	*340	*260	*318	*290	*337	*293	*269	*231	*230	*212
	69, 6	*727	*447	*453	*472	*586	*495	*360	*907	-929	*710	*560	*467	*355	*210
	74, 6	*321	*570	*723	*873	*614	*497	*347	*935	*933	*707	*565	*467	*361	*205
	79, 6	*070	*698	*689	*911	*588	*490	*351	*948	-948	*707	*558	*464	*358	*197
	84, 6	*253	*452	*609	*853	*583	*482	*353	*407	*972	*680	*555	*464	*344	*209
	89, 7	*254	*98	*547	*678	*545	*472	*358	*290	*884	*655	*577	*458	*321	*197
Upper surface	1, 3	*270	*320	*303	*213	*277	*142	*328	*293	*176	*116	*048	*051	*084	*165
	2, 6	*218	*814	*953	*727	*834	*890	*750	*750	*102	*1162	*615	*1005	*643	*725
	2, 4	*417	*767	*887	*515	*667	*773	*610	*106	*1029	*1134	*568	*772	*584	*647
	5, 0	*089	*338	*519	*304	*291	*294	*380	*220	*696	*1022	*509	*583	*521	*575
	7, 6	*137	*287	*277	*236	*217	*233	*321	*226	*485	*920	*478	*487	*443	*551
	10, 1	*163	*299	*176	*187	*209	*233	*280	*257	*441	*831	*439	*463	*393	*525
	15, 0	*187	*305	*115	*109	*150	*153	*173	*253	*412	*266	*366	*304	*359	*465
	19, 6	*174	*280	*084	*037	*093	*087	*064	*257	*386	*056	*290	*174	*347	*358
	24, 5	*210	*247	*017	*016	*057	*051	*052	*288	*373	*031	*255	*107	*338	*253
	29, 5	*188	*090	*073	*018	*015	*034	*029	*260	*364	*065	*187	*071	*327	*179
	34, 5	*218	*005	*057	*037	*010	*016	*295	*127	*104	*107	*045	*268	*116	
	39, 5	*195	*102	*172	*098	*092	*057	*010	*287	*056	*144	*032	*175	*078	
	44, 5	*117	*219	*216	*141	*106	*106	*025	*283	*184	*162	*031	*1078	*073	
	49, 5	*021	*334	*227	*172	*181	*144	*066	*169	*283	*156	*083	*20	*100	
	54, 5	*098	*380	*215	*201	*207	*162	*126	*045	*398	*148	*130	*033	*031	
	59, 5	*238	*336	*200	*218	*214	*163	*180	*207	*271	*144	*166	*044	*081	
	64, 5	*392	*127	*187	*226	*212	*149	*236	*363	*174	*134	*197	*044	*103	
	69, 6	*009							*258	*012					
	74, 6	*966	*847	*678	*572	*445	*362	*279	*903	-821	*654	*541	*437	*352	*255
	79, 5	*880	*834	*571	*445	*364	*278	*858	*806	*651	*541	*438	*356	*258	
	84, 6	*803	*824	*584	*570	*443	*364	*279	*793	*797	*660	*541	*436	*355	*262
	89, 6	*720	*809	*592	*566	*443	*361	*269	*706	*777	*664	*537	*442	*352	*257
	94, 6	*678	*676	*679	*559	*442	*358	*259	*682	*741	*650	*534	*443	*347	*252
Lower surface	1, 3	*359	*393	*413	*427	*390	*372	*315	*421	*495	*487	*499	*468	*455	*392
	2, 6	*328	*333	*335	*338	*307	*302	*243	*385	*441	*425	*427	*401	*394	*328
	5, 0	*299	*245	*261	*249	*243	*200	*184	*390	*244	*348	*359	*333	*309	*265
	7, 6	*256	*202	*212	*213	*190	*180	*122	*354	*296	*295	*303	*279	*259	*201
	10, 1	*215	*189	*183	*153	*155	*071	*071	*316	*267	*277	*266	*237	*222	*148
	15, 0	*176	*128	*150	*146	*121	*103	*031	*270	*213	*223	*221	*197	*180	*042
	19, 6	*139	*090	*125	*129	*097	*088	*066	*222	*179	*191	*196	*167	*156	*007
	24, 5	*122	*067	*101	*092	*076	*051	*066	*202	*142	*164	*156	*142	*117	*063
	29, 5	*091	*057	*090	*078	*064	*036	*049	*169	*127	*148	*135	*122	*094	*042
	34, 5	*064	*056	*085	*070	*060	*029	*048	*138	*121	*138	*123	*114	*077	*037
	39, 5	*046	*070	*088	*075	*070	*029	*048	*118	*129	*134	*120	*114	*069	*037
	44, 5	*025	*101	*100	*091	*087	*044	*061	*092	*153	*141	*129	*125	*074	*053
	49, 5	*027	*143	*127	*123	*115	*078	*101	*086	*182	*158	*155	*146	*100	*096
	54, 5	*079	*214	*177	*172	*184	*149	*161	*137	*244	*202	*197	*207	*161	*158
	59,														

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

Percent chord	Pressure coefficient														
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
	M = 0.94	a = 8.25°								M = 0.94	a = 12.73°				
Upper surface	0.0	*302	*008	-*089	-*315	-*160	-*307	*016	*279	-*414	-*563	-*699	-*467	-*569	-*268
	1.2	*102	-*1208	-*1280	-*765	-*768	-*681	-*613	-*061	-*1304	-*133	-*657	-*556	-*508	-*462
	2.4	*200	-*1182	-*1259	-*4760	-*726	-*697	-*592	-*403	-*1299	-*108	-*652	-*564	-*521	-*455
	5.0	*330	-*1046	-*1202	-*726	-*765	-*686	-*579	-*494	-*1244	-*102	-*641	-*582	-*534	-*451
	7.5	*357	-*775	-*131	-*720	-*874	-*571	-*579	-*662	-*1191	-*1075	-*645	-*600	-*533	-*446
	10.0	*385	-*663	-*1091	-*721	-*767	-*542	-*575	-*662	-*1128	-*1069	-*649	-*601	-*533	-*437
	15.0	*368	-*578	-*929	-*709	-*319	-*534	-*554	-*618	-*1059	-*1046	-*652	-*524	-*539	-*423
	19.6	*352	-*536	-*458	-*676	-*202	-*544	-*523	-*587	-*1021	-*1028	-*661	-*461	-*537	-*428
	24.5	*370	-*505	-*336	-*684	-*169	-*537	-*471	-*570	-*996	-*969	-*722	-*440	-*531	-*425
	29.5	*343	-*482	-*131	-*631	-*157	-*555	-*436	-*534	-*949	-*936	-*779	-*461	-*540	-*426
	34.5	*374	-*424	*053	-*542	-*139	-*529	-*415	-*543	-*841	-*908	-*822	-*480	-*537	-*423
	39.5	*360	-*145	*115	-*385	-*126	-*472	-*405	-*502	-*676	-*902	-*830	-*505	-*529	-*417
	44.5	*372	*050	*132	-*159	-*126	-*373	-*402	-*526	-*384	-*896	-*816	-*540	-*522	-*407
	49.5	*371	*175	*127	*018	-*129	-*239	-*412	-*489	-*133	-*852	-*799	-*560	-*510	-*399
	54.5	*086	*224	*116	*108	-*148	-*094	-*427	-*328	-*041	-*739	-*760	-*567	-*497	-*388
	59.5	*058	*167	*100	*163	-*178	*020	-*411	-*253	-*032	-*571	-*702	-*559	-*482	-*367
	64.5	*227	*175	*084	*198	-*221	-*071	-*413	-*125	-*018	-*402	-*617	-*554	-*456	-*373
	69.6	*009													
	74.6	*867	-*805	-*635	-*505	-*441	-*344	-*356	-*831	-*799	-*641	-*553	-*448	-*387	-*355
	79.5	*843	-*792	-*638	-*507	-*440	-*353	-*333	-*831	-*782	-*663	-*572	-*455	-*402	-*347
	84.6	*785	-*778	-*645	-*510	-*438	-*350	-*318	-*797	-*738	-*588	-*545	-*399	-*340	
	89.6	*717	-*756	-*648	-*507	-*448	-*347	-*309	-*718	-*746	-*715	-*599	-*466	-*390	-*334
	94.6	*674	-*725	-*628	-*505	-*457	-*343	-*293	-*681	-*721	-*629	-*588	-*458	-*378	-*326
Lower surface	1.3	*463	*561	*532	*527	*514	*501	*437	*514	*628	*578	*547	*534	*516	*466
	2.6	*418	*523	*487	*468	*458	*408	*426	*536	*585	*554	*538	*506	*455	
	5.0	*467	*432	*418	*418	*418	*386	*337	*598	*583	*542	*530	*496	*498	*414
	7.6	*447	*377	*355	*364	*364	*356	*356	*576	*549	*549	*494	*428	*379	
	10.1	*411	*347	*342	*333	*317	*303	*225	*612	*497	*477	*436	*412	*398	*317
	15.1	*363	*293	*285	*278	*259	*240	*106	*547	*436	*405	*378	*353	*325	*202
	19.6	*308	*252	*247	*251	*228	*214	*006	*476	*390	*364	*347	*317	*289	*145
	24.5	*281	*214	*217	*208	*197	*170	*004	*436	*350	*329	*300	*283	*251	*075
	29.5	*194	*197	*186	*171	*144	*011	*390	*320	*300	*272	*249	*215	*045	
	34.5	*207	*181	*181	*167	*158	*122	*024	*350	*302	*277	*244	*226	*188	*016
	39.5	*183	*184	*172	*162	*159	*109	*032	*320	*290	*260	*226	*214	*166	*-013
	44.5	*154	*197	*174	*162	*159	*110	*055	*284	*288	*251	*216	*208	*155	*-053
	49.5	*143	*218	*187	*179	*174	*125	*101	*265	*294	*249	*219	*232	*157	*-113
	54.5	*190	*273	*221	*214	*226	*175	*169	*295	*333	*268	*242	*248	*187	*-189
	59.5	*233	*316	*284	*257	*276	*242	*219	*323	*360	*315	*284	*291	*246	*-251
	64.5	*406	*346	*307	*331	*296	*238	*235	*471	*390	*339	*348	*319	*252	*-281
	69.5	*663													
	74.6	*920	-*775	-*715	-*595	-*451	-*343	-*220	-*772	-*577	-*765	-*642	-*521	-*446	-*273
	79.6	*778	-*870	-*715	-*599	-*460	-*352	-*232	-*674	-*746	-*772	-*651	-*527	-*421	-*268
	84.6	*474	-*1008	-*695	-*590	-*456	-*351	-*238	-*354	-*921	-*737	-*641	-*521	-*409	-*274
	89.7	*395	-*972	-*664	-*582	-*455	-*342	-*252	-*192	-*788	-*649	-*648	-*514	-*393	-*290
	94.6	*144	-*636	-*641	-*586	-*586	-*455	-*315	-*441	-*561	-*639	-*503	-*365	-*293	
Upper surface	M = 0.94	a = 17.16°													
	0.0	*200	-*750	-*912	-*707	-*549	-*530	-*450	*070	-*1008	-*1052	-*791	-*609	-*612	-*893
	1.2	-*207	-*1329	-*1421	-*658	-*505	-*496	-*432	-*368	-*102	-*940	-*761	-*623	-*606	-*616
	2.4	-*625	-*1329	-*1402	-*655	-*515	-*502	-*438	-*833	-*106	-*925	-*757	-*623	-*605	-*583
	5.0	-*771	-*1317	-*1406	-*654	-*527	-*509	-*437	-*966	-*1077	-*930	-*748	-*623	-*605	-*565
	7.5	-*903	-*1320	-*986	-*651	-*540	-*509	-*437	-*1068	-*1072	-*920	-*743	-*624	-*608	-*561
	10.0	-*886	-*1304	-*971	-*642	-*551	-*513	-*438	-*1073	-*1072	-*913	-*735	-*627	-*612	-*561
	15.0	-*823	-*1254	-*977	-*634	-*560	-*518	-*438	-*1001	-*1053	-*923	-*721	-*633	-*609	-*557
	19.6	-*771	-*1214	-*961	-*642	-*565	-*523	-*436	-*934	-*1040	-*911	-*705	-*635	-*613	-*561
	24.5	-*713	-*1188	-*918	-*679	-*579	-*518	-*434	-*876	-*1044	-*890	-*709	-*635	-*613	-*563
	29.5	-*667	-*1157	-*904	-*594	-*583	-*528	-*434	-*819	-*1034	-*891	-*710	-*635	-*629	-*566
	34.5	-*655	-*1110	-*877	-*596	-*589	-*525	-*435	-*800	-*1010	-*883	-*715	-*634	-*632	-*571
	39.5	-*619	-*1053	-*871	-*592	-*586	-*519	-*441	-*745	-*1002	-*872	-*719	-*634	-*626	-*577
	44.5	-*624	-*892	-*857	-*584	-*582	-*513	-*441	-*759	-*977	-*851	-*722	-*632	-*619	-*573
	49.5	-*555	-*545	-*827	-*573	-*570	-*506	-*444	-*746	-*956	-*833	-*725	-*629	-*612	-*569
	54.5	-*438	-*432	-*786	-*554	-*557	-*499	-*419	-*661	-*970	-*812	-*721	-*628	-*607	-*541
	59.5	-*379	-*290	-*739	-*542	-*547	-*494	-*423	-*589	-*953	-*795	-*713	-*622	-*609	-*537
	64.5	-*253	-*307	-*698	-*543	-*547	-*485	-*436	-*540	-*854	-*767	-*696	-*624	-*615	-*547
	69.6	-*029													
	74.6	-*883	-*850	-*650	-*544	-*475	-*422	-*420	-*1004	-*914	-*723	-*655	-*580	-*521	-*521
	79.6	-*896	-*828	-*671	-*544	-*480	-*420	-*420	-*971	-*899	-*745	-*655	-*582	-*535	-*516
	84.6	-*874	-*772	-*686	-*550	-*475	-*431	-*413	-*931	-*832	-*749	-*655	-*577	-*538	-*502
	89.6	-*749	-*791	-*688	-*543	-*479	-*423	-*410	-*777	-*871	-*749	-*644	-*577	-*532	-*490
	94.6	-*681	-*681	-*657	-*595	-*474	-*417	-*408	-*739	-*826	-*750	-*633	-*566	-*520	-*476
Lower surface	1.3	*537	*638	*577	*534	*518	*500	*450	*408	*606	*530	*478	*470	*446	*407
	2.6	*387	*691	*636	*594	*570	*531	*479	*214	*703	*638	*587	*572	*522	*480
	5.0	*696	*685	*628	*600	*556	*520	*452	*746	*752	*669	*628	*590	*559	*479
	7.6	*780	*657	*593	*559	*523	*495	*408	*866	*735	*650	*606	*572	*530	*449
	10.1	*767	*618	*583	*527	*492	*462	*375	*853	*706	*648	*584	*545	*506	*422
	15.1	*689	*561	*515	*473	*436	*400	*267	*783	*655	*589	*538	*494	*452	*318
	19.6	*612	*511	*471	*441	*400	*368	*203	*710	*611	*548	*505	*464	*420	*266
	24.5	*567	*472	*435	*395	*359	*320	*127	*669	*570	*509	*461	*425	*371	*176
	29.5	*520	*440	*402	*357	*325	*282	*088	*622	*536	*476	*423	*386	*330	*126
	34.5	*477	*412	*374	*327	*297	*220	*007	*541	*480	*415	*361	*329	*258	*026
	39.5	*444	*393	*351	*303	*277	*220</								

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

Percent chord	Pressure coefficient														
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
	M = 0.98	$\alpha = 0.26^\circ$							M = 0.98	$\alpha = 4.22^\circ$					
Upper surface	0.0	*293	*548	*581	*580	*478	*523	*331	*295	*362	*342	*224	*280	*165	*342
	1.2	*337	*014	*043	*174	*165	*183	*173	*250	*757	*930	*812	*802	*857	*716
	2.4	*211	-*021	*031	*123	*141	*132	*106	*024	*709	*873	*678	*640	*771	*600
	5.0	*142	*000	*049	*123	*119	*109	*070	*051	*297	*711	*254	*300	*296	*390
	7.5	*101	-*008	*060	*114	*101	*086	*040	*094	*258	*345	*181	*243	*227	*326
	10.0	*068	-*020	*059	*109	*099	*071	*010	*131	*258	*329	*146	*186	*185	*276
	15.0	*036	-*023	*091	*113	*100	*079	*049	*112	*272	*036	*078	*133	*152	*220
	19.6	-*027	-*027	*105	*130	*122	*118	*045	*144	*251	*004	*014	*072	*080	*085
	24.5	-*015	*016	*168	*135	*119	*118	*023	*178	*247	*067	*002	*030	*049	*032
	29.5	*007	*054	*190	*154	*131	*100	*006	*159	*258	*112	*025	*013	*022	*009
	34.5	-*018	*194	*222	*149	*155	*114	*033	*195	*007	*159	*057	*017	*003	
	39.5	*015	*194	*260	*105	*188	*131	*024	*191	*152	*211	*096	*107	*059	*005
	44.5	*041	*231	*306	*248	*230	*161	*003	*196	*274	*246	*139	*147	*106	*013
	49.5	*105	*396	*361	*300	*281	*200	*053	*369	*252	*177	*186	*149	*060	
Lower surface	54.5	*210	*504	*394	*361	*334	*256	*139	*120	*400	*235	*205	*210	*173	*133
	59.5	*360	*527	*390	*380	*353	*280	*227	*261	*317	*219	*221	*219	*175	*204
	64.5	*480	*452	*372	*385	*358	*270	*302	*396	*173	*207	*232	*221	*161	*278
	69.6	*336	-*051	-*844	-*684	-*603	-*500	-*450	-*353	-*969	-*857	-*703	-*620	-*529	-*429
	74.6	-*947	-*828	-*708	-*614	-*512	-*453	-*339	-*921	-*838	-*715	-*626	-*529	-*428	-*355
	79.5	-*800	-*815	-*733	-*619	-*503	-*452	-*331	-*815	-*832	-*734	-*627	-*524	-*425	-*360
	84.6	-*645	-*809	-*760	-*623	-*505	-*448	-*317	-*703	-*821	-*761	-*624	-*523	-*422	-*350
	89.7	-*542	-*777	-*723	-*619	-*502	-*441	-*305	-*613	-*781	-*758	-*622	-*521	-*420	-*340
	94.6	-*777	-*723	-*619	-*502	-*441	-*305	-*327	-*020	-*969	-*857	-*703	-*620	-*529	-*429
	1.3	*265	*085	*075	*042	-*041	-*099	*112	*379	*417	*428	*451	*414	*397	*345
	2.6	*226	*057	*035	*014	-*001	-*051	-*093	*350	*356	*353	*365	*336	*329	*271
	5.0	*166	*047	*048	*068	*018	-*037	-*087	*327	*272	*279	*297	*271	*238	*211
	7.6	*123	*031	*025	*036	*001	-*041	-*081	*285	*226	*232	*244	*221	*208	*155
Upper surface	10.1	*092	*019	*037	*017	-*019	-*045	-*089	*248	*194	*210	*209	*180	*181	*103
	15.1	*062	-*003	*012	*003	-*030	-*055	-*158	*209	*158	*179	*173	*148	*127	*001
	19.6	*024	-*013	*016	*019	-*030	-*050	-*187	*163	*128	*155	*123	*118	*067	
	24.5	*016	-*047	*016	*000	-*030	-*058	-*122	*150	*094	*134	*118	*099	*074	*-125
	29.5	-*005	-*049	*027	*004	-*022	-*051	-*028	*120	*086	*124	*101	*086	*061	*073
	34.5	-*028	-*031	*041	*009	-*011	-*035	-*018	*091	*087	*121	*092	*081	*049	*034
	39.5	-*044	*011	*058	*031	*011	-*021	-*012	*074	*107	*123	*104	*098	*084	*033
	44.5	-*060	*071	*088	*063	*044	-*001	-*018	*054	*141	*154	*107	*099	*048	*040
	49.5	-*045	*131	*123	*115	*116	*063	-*049	*184	*158	*140	*131	*072	*083	
	54.5	-*024	*223	*190	*173	*182	*160	-*121	*107	*255	*207	*186	*195	*139	*160
	59.5	-*153	*304	*274	*222	*244	*246	-*216	*191	*309	*274	*243	*246	*221	*250
	64.5	-*318	*364	*306	*275	*242	*246	-*282	*377	*385	*207	*292	*300	*258	*289
	69.5	-*455	-*1016	-*778	-*673	-*523	-*447	-*295	-*922	-*934	-*798	-*667	-*542	-*453	-*314
	74.6	-*770	-*1013	-*755	-*681	-*542	-*450	-*302	-*707	-*951	-*769	-*674	-*550	-*444	-*322
	79.5	-*478	-*1007	-*738	-*669	-*530	-*450	-*291	-*460	-*984	-*748	-*670	-*539	-*438	-*327
	84.6	-*412	-*1072	-*722	-*672	-*531	-*440	-*291	-*353	-*1004	-*731	-*665	-*533	-*424	-*345
	89.7	-*227	-*953	-*701	-*691	-*528	-*408	-*286	-*180	-*827	-*711	-*680	-*532	-*403	-*343
Lower surface	0.0	*306	*219	*165	*040	*077	*058	*185	*330	*083	*-005	*235	*144	*260	*052
	1.2	*192	-*1002	-*1495	-*979	-*933	-*506	-*778	-*149	-*1129	-*187	-*1087	-*783	-*658	-*637
	2.4	-*066	-*1059	-*109	-*942	-*722	-*541	-*607	-*135	-*1106	-*167	-*1074	-*745	-*675	-*616
	5.0	-*203	-*784	-*988	-*765	-*559	-*444	-*479	-*612	-*472	-*119	-*971	-*778	-*670	-*605
	7.5	-*186	-*460	-*883	-*824	-*568	-*466	-*386	-*585	-*280	-*781	-*1052	-*860	-*859	-*609
	10.1	-*224	-*399	-*824	-*824	-*566	-*474	-*376	-*553	-*328	-*579	-*1022	-*777	-*810	-*534
	15.0	-*223	-*376	-*477	-*475	-*294	-*366	-*464	-*319	-*431	-*493	-*955	-*679	-*527	-*553
	19.6	-*227	-*357	-*245	-*339	-*154	-*361	-*347	-*301	-*467	-*645	-*622	-*153	-*526	-*503
	24.5	-*263	-*334	-*041	-*170	-*091	-*348	-*252	-*320	-*439	-*353	-*605	-*151	-*528	-*444
	29.5	-*238	-*351	-*107	-*003	-*057	-*342	-*193	-*246	-*426	-*109	-*548	-*149	-*540	-*410
	34.5	-*171	-*330	-*151	-*058	-*035	-*271	-*137	-*326	-*422	-*056	-*458	-*132	-*522	-*391
	39.5	-*264	-*001	-*188	-*083	-*010	-*169	-*088	-*314	-*237	-*123	-*269	-*118	-*482	-*389
	44.5	-*276	-*178	-*200	-*115	-*010	-*074	-*075	-*327	-*041	-*146	-*008	-*120	-*411	-*396
	49.5	-*294	-*286	-*184	-*149	-*028	-*006	-*104	-*360	-*176	-*141	-*145	-*137	-*306	-*415
	54.5	-*030	-*339	-*170	-*184	-*027	-*048	-*164	-*128	-*237	-*135	-*183	-*166	-*438	
	59.5	-*174	-*226	-*158	-*215	-*009	-*082	-*216	-*042	-*159	-*123	-*217	-*232	-*046	-*438
	64.5	-*339	-*194	-*141	-*242	-*024	-*098	-*274	-*213	-*183	-*104	-*252	-*015	-*455	-*429
	69.6	-*166	-*848	-*592	-*533	-*425	-*329	-*903	-*824	-*661	-*574	-*561	-*440	-*418	
	74.6	-*909	-*828	-*689	-*595	-*542	-*424	-*334	-*882	-*671	-*585	-*580	-*450	-*403	
Upper surface	79.5	-*819	-*819	-*703	-*597	-*566	-*420	-*344	-*801	-*793	-*582	-*592	-*440	-*396	
	84.6	-*718	-*807	-*730	-*594	-*573	-*415	-*337	-*723	-*778	-*688	-*620	-*429	-*390	
	89.7	-*766	-*730	-*589	-*587	-*541	-*410	-*647	-*744	-*694	-*580	-*632	-*423	-*379	
	94.6	-*152	-*729	-*697	-*664	-*540	-*392	-*612	-*857	-*754	-*825	-*737	-*592	-*502	-*323
	1.3	*444	*497	*501	*487	*470	*404	*482	*576	*549	*536	*522	*510	*452	
	2.6	*401	*456	*435	*434	*422	*407	*346	*441	*536	*502	*484	*465	*411	
	5.0	*407	*362	*360	*370	*352	*319	*282	*490	*445	*432	*424	*427	*395	
	7.6	*373	*314	*307	*314	*290	*278	*222	*470	*398	*380	*370	*370	*329	
	10.1	*335	*286	*297	*282	*261	*240	*168	*434	*360	*361	*342	*342	*311	
	15.1	*293	*231	*240	*236	*215	*198	*067	*387	*310	*304	*290	*270	*247	
	19.6	*241	*194	*207	*212	*186	*170	*014	*331	*268	*266	*222	*237	*202	
	24.5	*223	*161	*184	*177	*159	*131	*-080	*305	*230	*239	*222	*207	*176	
	29.5	*187	*146	*168	*156	*138	*109	*-078	*266	*212	*219	*198	*182	*147	
	34.5	*156	*142	*159	*139	*127	*089	*-049	*231	*202	*205	*179	*167	*126	
	39.5	*135	*												

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

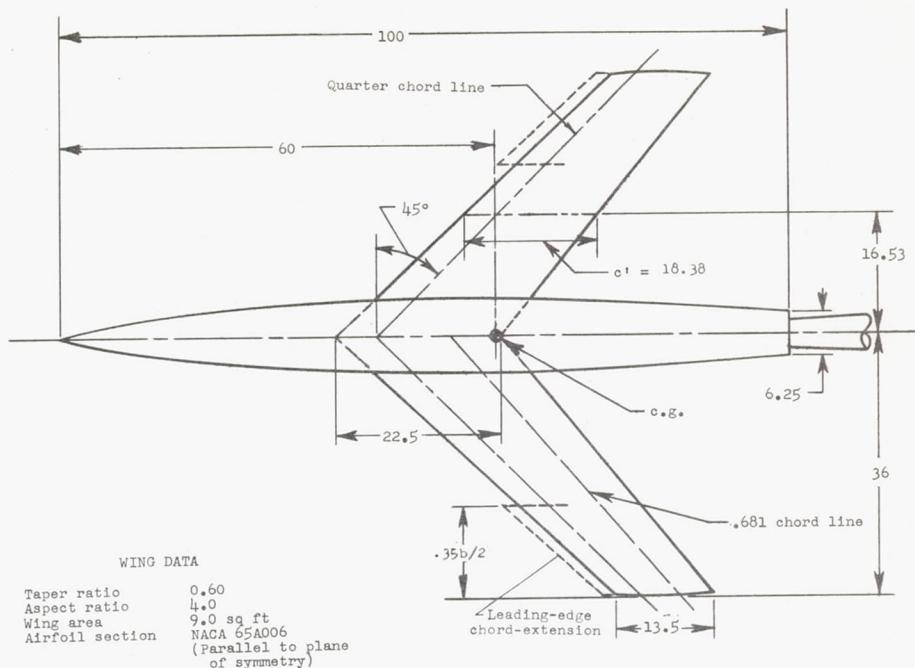
Percent chord	Pressure coefficient													
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
	M = 0.98	a = 12.82°	M = 0.98							a = 17.34°				
Upper surface														
0.0	+307	-+329	-+460	-+643	-+473	-+605	-+273	+226	-+660	-+834	-+839	-+526	-+701	-+532
1.2	-+002	-+1209	-+153	-+669	-+670	-+565	-+500	+129	-+277	-+117	-+748	-+437	-+663	-+530
2.4	-+333	-+204	-+121	-+669	-+673	-+570	-+500	+549	-+277	-+101	-+738	-+454	-+675	-+540
5.0	-+416	-+141	-+125	-+661	-+691	-+585	-+495	+694	-+266	-+096	-+718	-+476	-+679	-+549
7.5	-+586	-+079	-+096	-+667	-+715	-+576	-+495	+826	-+275	-+068	-+709	-+502	-+670	-+549
10.0	-+590	-+998	-+075	-+671	-+731	-+581	-+490	+818	-+248	-+062	-+696	-+532	-+672	-+549
15.0	-+558	-+902	-+027	-+689	-+661	-+585	-+486	+767	-+196	-+058	-+672	-+584	-+656	-+549
19.6	-+524	-+860	-+975	-+719	-+574	-+585	-+484	+711	-+173	-+058	-+655	-+616	-+647	-+549
24.5	-+507	-+839	-+884	-+778	-+525	-+591	-+480	+654	-+151	-+051	-+680	-+635	-+633	-+546
29.5	-+473	-+837	-+829	-+812	-+522	-+590	-+482	+601	-+130	-+080	-+727	-+655	-+639	-+551
34.5	-+481	-+766	-+789	-+828	-+525	-+589	-+482	+601	-+118	-+076	-+768	-+673	-+638	-+554
39.5	-+446	-+620	-+765	-+833	-+533	-+583	-+478	+561	-+092	-+072	-+784	-+684	-+635	-+563
44.5	-+469	-+412	-+749	-+838	-+559	-+576	-+466	+579	-+978	-+065	-+775	-+690	-+628	-+563
49.5	-+469	-+195	-+707	-+842	-+589	-+569	-+453	+582	-+949	-+097	-+767	-+686	-+622	-+556
54.5	-+111	-+612	-+834	-+606	-+559	-+443	-+468	+439	-+139	-+049	-+758	-+680	-+613	-+545
59.5	-+225	-+100	-+475	-+798	-+609	-+550	-+418	+399	-+102	-+955	-+749	-+669	-+610	-+513
64.5	-+098	-+012	-+339	-+705	-+604	-+521	-+427	+266	-+085	-+834	-+733	-+672	-+598	-+519
69.5	-+051							-+422	-+179					
74.6	-+887	-+827	-+660	-+593	-+501	-+456	-+411	-+918	-+843	-+712	-+604	-+554	-+522	-+495
79.5	-+884	-+805	-+676	-+619	-+516	-+489	-+410	-+912	-+828	-+741	-+636	-+580	-+565	-+492
84.6	-+856	-+761	-+693	-+646	-+525	-+480	-+401	-+893	-+799	-+771	-+660	-+583	-+552	-+486
89.6	-+729	-+770	-+715	-+670	-+526	-+464	-+400	-+731	-+795	-+780	-+652	-+588	-+532	-+485
94.6	-+654	-+747	-+668	-+649	-+512	-+451	-+396	-+652	-+763	-+721	-+639	-+576	-+514	-+484
Lower surface														
1.3	+537	+653	+600	+567	+545	+528	+479	+553	+667	+597	+547	+540	+506	+446
2.6	+445	+658	+603	+570	+551	+520	+467	+392	+717	+554	+607	+586	+543	+500
5.0	+627	+602	+559	+537	+511	+494	+424	+730	+646	+614	+575	+533	+477	
7.6	+657	+555	+510	+488	+468	+444	+368	+811	+682	+610	+576	+539	+495	+436
10.1	+638	+517	+492	+454	+428	+406	+333	+792	+648	+600	+544	+507	+477	+398
15.1	+575	+462	+431	+401	+370	+342	+220	+716	+586	+533	+491	+451	+418	+294
19.6	+502	+414	+387	+366	+333	+324	+156	+638	+540	+493	+460	+416	+381	+252
24.5	+467	+375	+355	+325	+298	+264	+080	+596	+500	+457	+413	+380	+338	+146
29.5	+419	+347	+328	+294	+268	+229	+046	+545	+488	+424	+379	+346	+298	+101
34.5	+377	+327	+306	+268	+244	+202	+012	+504	+442	+397	+346	+316	+267	+058
39.5	+347	+317	+289	+252	+228	+176	+019	+470	+424	+373	+324	+294	+236	+021
44.5	+316	+317	+280	+240	+220	+162	+051	+435	+412	+356	+303	+277	+212	-+027
49.5	+293	+324	+280	+242	+242	+158	+107	+404	+408	+344	+295	+283	+197	-+096
54.5	+326	+363	+300	+282	+256	+185	+202	+428	+434	+351	+301	+285	+214	-+195
59.5	+351	+389	+346	+312	+298	+251	+307	+444	+450	+385	+347	+316	+261	-+319
64.5	+503	+416	+366	+357	+318	+255	+364	+576	+476	+403	+383	+336	+273	-+383
69.5	+775							-+374	-+782					
74.6	-+720	-+537	-+746	-+765	-+527	-+456	-+366	-+633	-+403	-+706	-+848	-+710	-+604	-+437
79.6	-+629	-+677	-+759	-+804	-+567	-+511	-+337	-+452	-+543	-+716	-+876	-+721	-+602	-+426
84.6	-+329	-+879	-+719	-+840	-+646	-+499	-+326	-+165	-+715	-+667	-+921	-+726	-+582	-+413
89.7	-+119	-+732	-+647	-+840	-+644	-+481	-+343	-+167	-+570	-+609	-+884	-+743	-+565	-+420
94.6	-+127	-+384	-+583	-+793	-+459	-+341	-+291	-+291	-+226	-+572	-+760	-+701	-+550	-+420
Upper surface														
1.3	+279	+099	+092	+058	+026	-+084	-+094	+400	+440	+451	+486	+445	+421	+368
2.6	+348	+034	+063	+192	+185	+203	+195	+270	+737	+895	+645	+786	+853	+723
5.0	+229	-+006	+047	+139	+160	+151	+130	+048	-+689	+846	-+495	-+620	-+766	-+612
7.6	+164	+016	+066	+142	+139	+128	+094	+025	-+268	-+708	-+312	-+301	-+284	-+381
10.0	+087	-+008	+080	+128	+120	+109	+094	+100	-+232	-+308	-+147	-+184	-+185	-+259
15.0	+053	-+014	+111	+131	+118	+098	-+027	+108	-+244	-+175	-+068	-+108	-+120	-+196
19.6	+043	-+017	+125	+148	+138	+136	-+040	+114	-+225	+046	+005	+051	-+064	-+074
24.5	+002	+026	+192	+154	+138	+128	-+015	+150	-+224	+115	+023	-+005	-+014	-+016
29.5	-+002	+130	+240	+198	+170	+130	-+038	+131	-+241	+150	-+048	-+040	-+001	-+009
34.5	-+025	+229	+222	+205	+144	+205	-+042	+186	-+239	+239	-+083	-+082	-+039	-+024
39.5	-+056	+305	+324	+263	+245	+169	-+07	+182	-+290	+270	-+169	-+164	-+123	-+010
44.5	-+211	+414	+379	+316	+298	+209	-+039	+195	-+320	+272	-+209	-+201	-+163	-+039
49.5	-+228	+518	+412	+375	+325	+265	-+123	+118	-+416	+252	-+237	-+226	-+188	-+110
54.5	-+376	+540	+407	+392	+367	+290	-+219	+272	-+286	+238	-+254	-+239	-+190	-+178
59.5	-+503	+466	+390	+396	+368	+282	-+315	+401	-+205	+224	-+243	-+180	-+250	
64.5	-+002						-+348	-+018						
69.5	-+944	-+837	-+677	-+531	-+531	-+473	-+374	-+925	-+835	-+666	-+608	-+531	-+430	-+320
74.6	-+958	-+823	-+711	-+646	-+563	-+478	-+365	-+211	-+811	-+684	-+540	-+428	-+326	
79.6	-+829	-+812	-+739	-+650	-+560	-+474	-+376	-+796	-+801	-+700	-+619	-+531	-+411	-+336
84.6	-+600	-+806	-+771	-+661	-+561	-+465	-+357	-+661	-+792	-+734	-+619	-+542	-+399	-+327
89.6	-+526	-+782	-+725	-+663	-+555	-+455	-+344	-+580	-+753	-+736	-+615	-+542	-+396	-+320
Lower surface														
1.3	+279	+099	+092	+058	+026	-+084	-+094	+400	+440	+451	+486	+445	+421	+368
2.6	+243	+071	+052	+031	+013	-+034	-+075	+372	+382	+374	+401	+357	+301	+272
5.0	+185	+060	+062	+085	+037	-+024	-+065	+351	+296	+305	+334	+301	+272	+238
7.6	+140	+046	+044	+054	+018	-+028	-+062	+310	+252	+259	+281	+253	+238	+185
10.1	+111	+033	+053	+037	+003	-+032	-+068	+277	+218	+239	+246	+212	+208	+138
15.1	+080	+008	+030	+022	+013	-+044	-+137	+234	+184	+208	+208	+181	+155	+032
19.6	+042	-+003	+036	+041	+013	-+040	-+140	+188	+150	+185	+192	+155	+141	-+017
24.5	+033	-+037	+038	+019	+013	-+043	-+126	+176	+119	+165	+157	+130	+100	-+099
29.5	-+011	-+042	+049	+023	-+006	-+034	-+045	+144	+110	+157	+140	+115	+087	-+083
34.5	-+014	-+020	+064	+028	+004	-+018	-+004	+117	+112	+153	+127	+109	+075	-+017
39.5	-+031	-+027	+083	+047	+024	-+003	-+008	+097	+131	+156	+131	+112	+069	-+009
44.5	-+049	+090	+110	+079	+056	-+011	-+001	+078	+168	+167	+141	+126	+069	-+018
49.5	-+040	+150	+149	+128	+119	-+063	-+033	+078	+208	+191	+154	+089	-+057	
54.5	-+041	+243	+208	+187	+195	-+161	-+104	+136	+279	+236	+214	+217	+153	-+137
59.5	-+168	+320	+292	+236	+261	-+253	-+208	+213	+333	+303	+271	+267	+235	-+238

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Continued

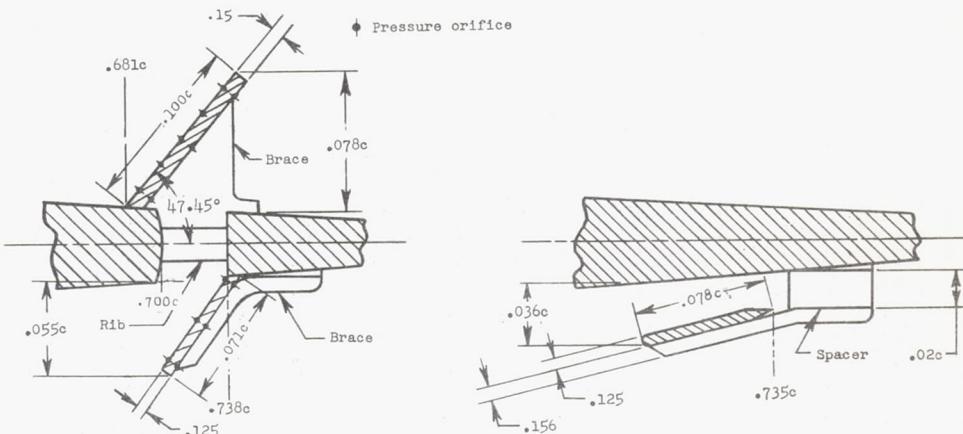
Percent chord		Pressure coefficient															
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2		
		M = 1.00	a = 6.22°							M = 1.00	a = 8.25°						
Upper surface	0.0	*333	*258	*1204	*002	*085	*032	*214	*349	*110	*030	*196	*141	*239	*065		
	1.2	*228	*957	-1*043	-*999	*812	*508	*755	*173	-1*098	-1*141	*115	*782	*560	*636		
	2.4	*030	-931	-1*023	-*950	*634	*460	*679	*113	-1*080	-1*123	*100	*752	*599	*625		
	5.0	-155	-766	-*950	-*902	*535	*416	*586	*237	-1*012	-1*080	*103	*777	-563	-623		
	7.5	-146	-424	-*855	-*723	*486	*378	*549	*256	-808	-1*019	*761	*840	-471	-622		
	10.0	-184	-358	-*805	-*557	*471	*363	*514	*303	-551	-995	*638	*345	-684	-543		
	15.0	-181	-334	-*467	-*408	*315	*348	*420	*286	-400	-45	*583	*155	-499	-476		
	19.6	-187	-318	-*353	-*265	*155	*354	*308	*276	-437	-409	*1560	*138	-518	-405		
	24.5	-221	-306	*037	-*100	*072	*345	*213	*293	-385	-450	*1520	*130	-553	-364		
	29.5	-200	-313	*127	*028	*035	*337	*158	*74	-398	-117	*1520	*130	-553	-364		
	34.5	-232	-315	*180	*074	*013	*251	*16	*30	-399	-069	*1459	*115	-556	-430		
	39.5	-226	-021	*217	*097	*014	*147	*072	*286	-265	-144	-326	-098	-524	-430		
	44.5	-242	*188	*229	*127	*035	*150	*059	*303	-041	*163	-080	*108	-443	-365		
	49.5	-272	*304	*218	*162	*054	*17	*085	*337	*188	*157	*153	*138	-316	-394		
	54.5	-039	*354	*200	*197	*053	*666	*144	*128	*248	*153	*214	*195	-176	-424		
	59.5	*183	*226	*183	*229	*038	*102	*196	*043	*165	*138	*249	*252	-063	-440		
	64.5	*346	*220	*161	*257	*010	*116	*254	*205	*113	*286	*305	-007	-466	-456		
	69.6	*031															
	74.6	*898	-812	-647	-*584	*538	*433	*433	*318	-870	-785	*644	*571	-622	-508		
	79.5	*885	-795	-659	-*596	*553	*433	*318	-767	-656	-656	-657	-649	-490	-434		
	84.6	-794	-781	-673	-601	*563	*419	*333	-804	-767	-679	-580	-671	-479	-429		
	89.6	-675	-770	-702	-598	*579	*407	*329	-689	-754	-679	-580	-671	-479	-429		
	94.6	-599	-732	-670	-593	*587	*402	-325	-616	-720	-669	-572	-682	-472	-422		
Lower surface	1.3	*457	*530	*520	*514	*494	*438	*498	*595	*569	*555	*538	*532	*476			
	2.6	*427	*478	*458	*454	*451	*434	*379	*460	*556	*523	*505	*498	*491	*431		
	5.0	*434	*388	*387	*392	*385	*355	*318	*511	*472	*453	*445	*444	*423	*377		
	7.6	*399	*378	*334	*339	*332	*307	*257	*489	*419	*403	*392	*390	*375	*312		
	10.1	*365	*311	*423	*307	*292	*268	*209	*459	*381	*382	*365	*349	*337	*271		
	15.1	*322	*257	*268	*261	*244	*229	*106	*409	*330	*329	*314	*290	*272	*150		
	19.6	*270	*220	*238	*239	*216	*201	*013	*354	*291	*292	*288	*257	*238	*066		
	24.5	*216	*173	*198	*184	*164	*138	*052	*288	*236	*244	*223	*200	*171	-011		
	29.5	*186	*168	*190	*169	*155	*118	*043	*252	*226	*230	*203	*186	*147	-021		
	34.5	*162	*178	*187	*165	*152	*110	*033	*227	*230	*223	*196	*179	*129	-030		
	39.5	*140	*206	*194	*169	*159	*104	*025	*201	*244	*226	*196	*179	*122	-041		
	44.5	*134	*242	*213	*191	*179	*112	*063	*189	*269	*236	*211	*198	*125	-084		
	49.5	*188	*301	*253	*228	*232	*166	*146	*237	*323	*269	*240	*169	*172			
	54.5	*248	*348	*314	*265	*282	*245	*251	*282	*365	*329	*284	*242	*295			
	59.5	*430	*373	*332	*337	*293	*236	*311	*457	*390	*346	*347	*296	*245	*375		
	64.5	*640															
	69.6	-858	-807	-798	-701	-592	-476	-309	-823	-719	-800	-777	-661	-542	-400		
	74.6	-627	-866	-793	-698	-589	-480	-297	-633	-797	-792	-783	-661	-542	-360		
	79.6	-431	-927	-782	-689	-579	-471	-297	-423	-916	-760	-787	-660	-538	-328		
	84.6	-303	-913	-724	-673	-586	-447	-315	-874	-874	-699	-806	-665	-518	-334		
	89.7	-106	-681	-655	-669	-604	-428	-309	-604	-578	-622	-683	-663	-508	-326		
Upper surface	1.3	M = 1.00	a = 12.80°							M = 1.00	a = 17.34°						
	0.0	*321	*287	*418	*603	*452	*614	*309	*242	*1616	*780	*809	*542	*801	*598		
	1.2	*025	-1*177	-1*158	-*638	*556	*614	*650	*092	-1*241	*1135	*749	*465	*744	*626		
	2.4	*302	-1*169	-1*126	-*638	*663	*625	*649	*717	-1*241	*1113	*743	*467	*763	*624		
	5.0	*388	-1*118	-1*132	-*632	*688	*649	*646	*650	-1*229	*1128	*733	*468	*780	*609		
	7.5	*551	-1*046	-1*104	-*635	*717	*640	*646	*649	-1*236	*1109	*749	*473	*785	*601		
	10.0	*561	-962	-1*077	-*641	*732	*644	*653	*646	-1*246	*1109	*761	*481	*800	*593		
	15.0	*537	-862	-1*012	-*661	*640	*653	*639	*673	-1*204	*1117	*761	*497	*791	*601		
	19.6	*496	-822	-945	-*710	*543	*543	*543	*626	-1*159	*1117	*761	*506	*774	*630		
	24.5	*481	-801	-859	-*769	*498	*480	*503	*680	-1*133	*1105	*786	*556	*774	*630		
	29.5	*451	-807	-794	-*803	*499	*492	*586	*581	-1*089	*1042	*916	*643	*728	*654		
	34.5	*456	-751	-754	-*822	*512	*497	*574	*567	-1*078	*1078	*940	*683	*703	*654		
	39.5	*421	-617	-733	-*836	*519	*486	*571	*533	-1*063	*1028	*933	-712	-691	-650		
	44.5	*444	-380	-719	-847	*544	*488	*560	*546	-1*031	*933	-727	-687	-641			
	49.5	*456	-123	-687	-851	*574	*677	*549	*557	-1*098	*885	-731	-684	-627			
	54.5	*309	-041	-603	-851	*604	*662	*532	*472	-1*081	*826	-732	-682	-612			
	59.5	*213	-041	-480	-822	*619	*649	*503	*378	-093	*983	-787	-732	-679	-578		
	64.5	*073	-032	-336	-732	*619	*621	*509	*242	-051	*871	-748	-725	-657	-577		
	69.6	*149															
	74.6	-859	-812	-658	-600	*524	*542	*488	*901	-822	*688	-886	-545	-577	-552		
	79.5	-887	-792	-677	-646	*549	-486	-896	-803	-717	-643	-589	-626	-541			
	84.6	-865	-764	-697	-678	*561	-587	-481	-881	-787	-691	-605	-591	-524			
	89.6	-727	-758	-724	-672	*566	-569	-485	-685	-773	-757	-701	-605	-591	-524		
	94.6	-622	-735	-674	-674	*557	-555	-483	-638	-747	-631	-687	-601	-571	-516		
Lower surface	1.3	*546	*672	*613	*576	*557	*538	*485	*563	*680	*613	*561	*553	*516	*475		
	2.6	*456	*674	*612	*577	*560	*533	*476	*401	*731	*670	*620	*600	*551	*511		
	5.0	*642	*615	*565	*543	*523	*498	*439	*747	*725	*660	*626	*592	*543	*490		
	7.6	*668	*568	*520	*498	*478	*453	*384	*826	*696	*623	*588	*555	*517	*450		
	10.1	*584	*479	*442	*411	*381	*357	*232	*727	*597	*546	*506	*469	*428	*310		
	15.1	*584	*479	*442	*411	*381	*357	*232	*727	*597	*471	*429	*398	*352	*163		
	19.6	*514	*430	*400	*380	*346	*325	*146	*650	*553	*506	*474	*433	*394	*230		
	24.5	*477	*390	*368	*335	*310	*279	*082	*608	*515	*471	*429	*398	*352	*163		
	29.5	*429	*364	*339	*304	*279	*240	*040	*559	*481	*441	*397	*358	*314	*16		
	34.5	*389	*344	*319	*278	*254	*215	*012	*518	*455	*414	*364	*332	*301	*070		
	39.5	*356	*334	*302	*263	*238	*188	*-016	*484	*436	*391	*340	*308	*250	*031		
	44.5	*326	*334	*295	*251	*231	*173	*-046	*451	*427	*374	*323	*268	*226	*017		

TABLE II. - WING WITH SPOILER-SLOT-DEFLECTOR - Concluded

Percent chord	Pressure coefficient															
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2		
	M = 1.03	$\alpha = 0.26^\circ$							M = 1.03	$\alpha = 4.22^\circ$						
Upper surface	0.0	*234	*550	*592	*624	*566	*579	*386	*254	*404	*392	*232	*292	*190	*381	
	1.2	*339	*010	*017	*233	*223	*249	*242	*236	*699	*837	*638	*763	*815	*705	
	2.4	*188	*030	*015	*180	*200	*198	*180	*040	*660	*797	*491	*594	*725	*589	
	5.0	*123	*012	*057	*185	*183	*177	*141	*050	*212	*712	*300	*294	*266	*351	
	7.5	*081	*024	*090	*177	*165	*156	*112	*072	*231	*413	*226	*199	*183	*277	
	10.0	*047	*043	*109	*171	*163	*142	*086	*111	*236	*291	*165	*162	*158	*242	
	15.0	*036	*055	*147	*176	*165	*146	*024	*115	*236	*295	*165	*162	*158	*156	
	19.6	*014	*061	*151	*197	*180	*185	*003	*124	*224	*402	*15	*154	*02	*156	
	24.5	*024	*040	*222	*197	*181	*181	*031	*154	*217	*159	*056	*036	*036	*046	
	29.5	*010	*047	*247	*218	*210	*185	*048	*125	*225	*253	*019	*080	*028	*038	
	34.5	*047	*166	*278	*244	*213	*183	*083	*146	*225	*253	*130	*121	*062	*053	
	39.5	*029	*243	*318	*267	*246	*192	*083	*146	*289	*168	*156	*097	*059		
	44.5	*032	*340	*444	*311	*286	*162	*027	*177	*301	*300	*217	*189	*137	*042	
Lower surface	49.5	*128	*451	*414	*362	*338	*286	*009	*182	*378	*287	*252	*223	*179	*004	
	54.5	*247	*540	*437	*416	*388	*308	*073	*078	*401	*266	*276	*249	*210	*074	
	59.5	*400	*563	*429	*450	*407	*334	*170	*254	*179	*246	*289	*259	*219	*138	
	64.5	*537	*408	*412	*453	*412	*324	*271	*380	*267	*224	*293	*263	*219	*208	
	69.5	*079							*312	*087					*231	
	74.6	*885	*766	*621	*576	*502	*494	*332	*860	*772	*606	*560	*496	*401	*281	
	79.5	*874	*757	*654	*593	*540	*440	*325	*854	*749	*627	*571	*507	*399	*286	
	84.6	*778	*723	*686	*599	*533	*433	*322	*745	*733	*647	*576	*507	*381	*297	
	89.6	*555	*737	*720	*610	*546	*423	*315	*621	*725	*686	*574	*513	*367	*291	
	94.6	*486	*715	*675	*612	*542	*413	*303	*521	*688	*682	*569	*512	*360	*283	
	1.3	*219	*084	*103	*098	*013	*033	*045	*335	*426	*457	*508	*488	*462	*410	
	2.6	*190	*052	*065	*073	*046	*014	*027	*324	*372	*387	*430	*414	*398	*344	
	5.0	*139	*038	*059	*124	*053	*038	*018	*324	*286	*319	*369	*347	*319	*283	
	7.6	*102	*026	*038	*094	*056	*020	*013	*291	*247	*274	*320	*299	*280	*230	
Lower surface	10.1	*071	*012	*044	*079	*038	*018	*024	*261	*213	*259	*286	*261	*251	*184	
	15.1	*052	*010	*027	*065	*029	*004	*086	*228	*180	*230	*252	*229	*202	*086	
	19.6	*016	*021	*063	*083	*031	*002	*079	*183	*153	*210	*234	*205	*180	*011	
	24.5	*012	*052	*083	*067	*029	*008	*082	*175	*124	*194	*199	*177	*147	*055	
	29.5	*008	*063	*094	*069	*038	*016	*019	*147	*120	*188	*182	*163	*130	*022	
	34.5	*030	*073	*105	*073	*049	*031	*050	*119	*125	*188	*169	*156	*119	*004	
	39.5	*047	*111	*126	*093	*065	*046	*052	*106	*153	*195	*173	*157	*113	*025	
	44.5	*071	*128	*154	*123	*097	*058	*046	*087	*194	*206	*184	*168	*112	*020	
	49.5	*081	*190	*193	*170	*163	*100	*015	*090	*238	*229	*211	*199	*129	*017	
	54.5	*002	*274	*253	*231	*233	*202	*057	*156	*308	*274	*253	*255	*184	*097	
	59.5	*172	*356	*331	*278	*296	*293	*166	*240	*360	*341	*302	*306	*268	*206	
	64.5	*348	*418	*349	*331	*296	*299	*256	*430	*385	*352	*359	*314	*254	*271	
	69.5	*481							*621						*272	
	74.6	*798	*912	*672	*625	*518	*466	*324	*812	*833	*737	*618	*547	*433	*372	
	79.6	*624	*907	*747	*637	*539	*471	*308	*608	*851	*726	*625	*556	*440	*342	
	84.6	*428	*901	*722	*638	*530	*467	*290	*409	*866	*724	*616	*549	*433	*360	
	89.7	*337	*958	*693	*657	*533	*445	*287	*260	*886	*717	*606	*537	*411	*274	
	94.6	*163	*846	*668	*681	*523	*437	*274	*098	*721	*637	*612	*543	*399	*271	
Upper surface	M = 1.03	$\alpha = 6.26^\circ$							M = 1.03	$\alpha = 8.23^\circ$						
	0.0	*287	*292	*238	*036	*080	*-015	*238	*318	*155	*072	*-149	*-113	*187	*103	
	1.2	*200	*905	*982	*-981	*-702	*-493	*-534	*166	*-103	*-085	*-1074	*-726	*-493	*-504	
	2.4	*013	*894	*968	*951	*547	*445	*445	*067	*-138	*-102	*-043	*-693	*-531	*-584	
	5.0	*136	*794	*910	*904	*478	*409	*579	*-076	*-066	*-032	*-1000	*-721	*-499	*-584	
	7.5	*144	*451	*829	*834	*375	*379	*498	*-219	*-868	*-969	*-861	*-799	*-418	*-596	
	10.0	*182	*344	*786	*580	*477	*344	*477	*-264	*-552	*-946	*-712	*-790	*-422	*-586	
	15.0	*179	*313	*576	*345	*294	*345	*346	*-410	*-248	*-903	*-584	*-298	*-439	*-493	
	19.6	*177	*296	*380	*199	*169	*168	*328	*-244	*-37	*-747	*-442	*-119	*-458	*-421	
	24.5	*208	*281	*361	*261	*169	*069	*059	*273	*-227	*-259	*-356	*-361	*-509	*-478	
	29.5	*212	*288	*343	*030	*023	*234	*164	*243	*-351	*-129	*-474	*-474	*-513	*-317	
	34.5	*217	*296	*217	*081	*-003	*159	*-109	*249	*-356	*-110	*-419	*-070	*-517	*-306	
	39.5	*209	*049	*255	*107	*031	*087	*-066	*253	*-223	*-192	*-316	*-051	*-485	*-311	
	44.5	*224	*220	*257	*141	*054	*022	*-052	*268	*-081	*-207	*-074	*-063	*-405	*-330	
	49.5	*257	*314	*240	*178	*071	*028	*-083	*299	*-224	*-198	*-148	*-097	*-275	*-336	
	54.5	*067	*354	*222	*213	*069	*070	*-138	*109	*-272	*-188	*-251	*-163	*-127	*-399	
	59.5	*178	*174	*194	*251	*056	*102	*-185	*071	*-174	*-172	*-293	*-225	*-015	*-420	
	64.5	*342	*256	*148	*279	*031	*120	*-239	*234	*-245	*-144	*-336	*-281	*-041	*-451	
	69.6	*100							*258	*-125					*446	
	74.6	*849	*762	*596	*544	*-518	*-420	*-303	*842	*-759	*-586	*-523	*-556	*-447		
	79.5	*838	*740	*613	*555	*-424	*-312	*-312	*831	*-739	*-600	*-538	*-602	*-493	*-435	
	84.6	*760	*719	*631	*563	*-544	*-407	*-329	*790	*-716	*-613	*-546	*-626	*-477	*-425	
	89.6	*629	*707	*670	*560	*-566	*-393	*-326	*634	*-700	*-639	*-546	*-665	*-465	*-417	
	94.6	*546	*671	*556	*556	*-576	*-390	*-321	*553	*-667	*-628	*-540	*-675	*-459	*-410	
Lower surface	1.3	*395	*539	*538	*549	*543	*520	*465	*456	*614	*591	*576	*561	*555	*496	
	2.6	*388	*488	*481	*484	*487	*466	*407	*431	*575	*545	*529	*520	*514	*452	
	5.0	*430	*397	*407	*422	*421	*389	*348	*521	*487	*476	*470	*470	*445	*396	
	7.6	*404	*350	*359	*370	*371	*338	*287	*511	*440	*428	*419	*413	*400	*339	
	10.1	*373	*328	*350	*339	*334	*300	*239	*480	*400	*407	*390	*375	*364	*292	
	15.1	*334	*275	*295	*294	*281	*264	*143	*335	*354	*355	*340	*317	*299	*178	
	19.6	*284	*238	*267	*271	*252	*238	*068	*377	*314	*321	*314	*286	*271	*116	
	24.5	*267	*211	*244	*237	*223	*197	*-013	*351	*279	*291	*275	*227	*033		
	29.5	*234	*197	*231	*219	*201	*172	*-029	*312	*261	*272	*251	*231	*201	*012	
	34.5	*203	*195	*223	*203	*189	*156	*-024	*275	*253	*253	*262	*232	*214	*003	
	39.5	*184	*209	*222	*201	*185	*142	*-017	*253	*255	*255	*225	*208	*181	*005</	



(a) Model.

(b) Spoiler-slot-deflector.
(Section parallel to plane of symmetry.)

(c) Deflector. (Section parallel to plane of symmetry.)

Figure 1.- Diagram and dimensional details of wing-fuselage model and cross-sectional detail of two different type spoiler control configurations. (All linear dimensions are in inches, except as noted.)

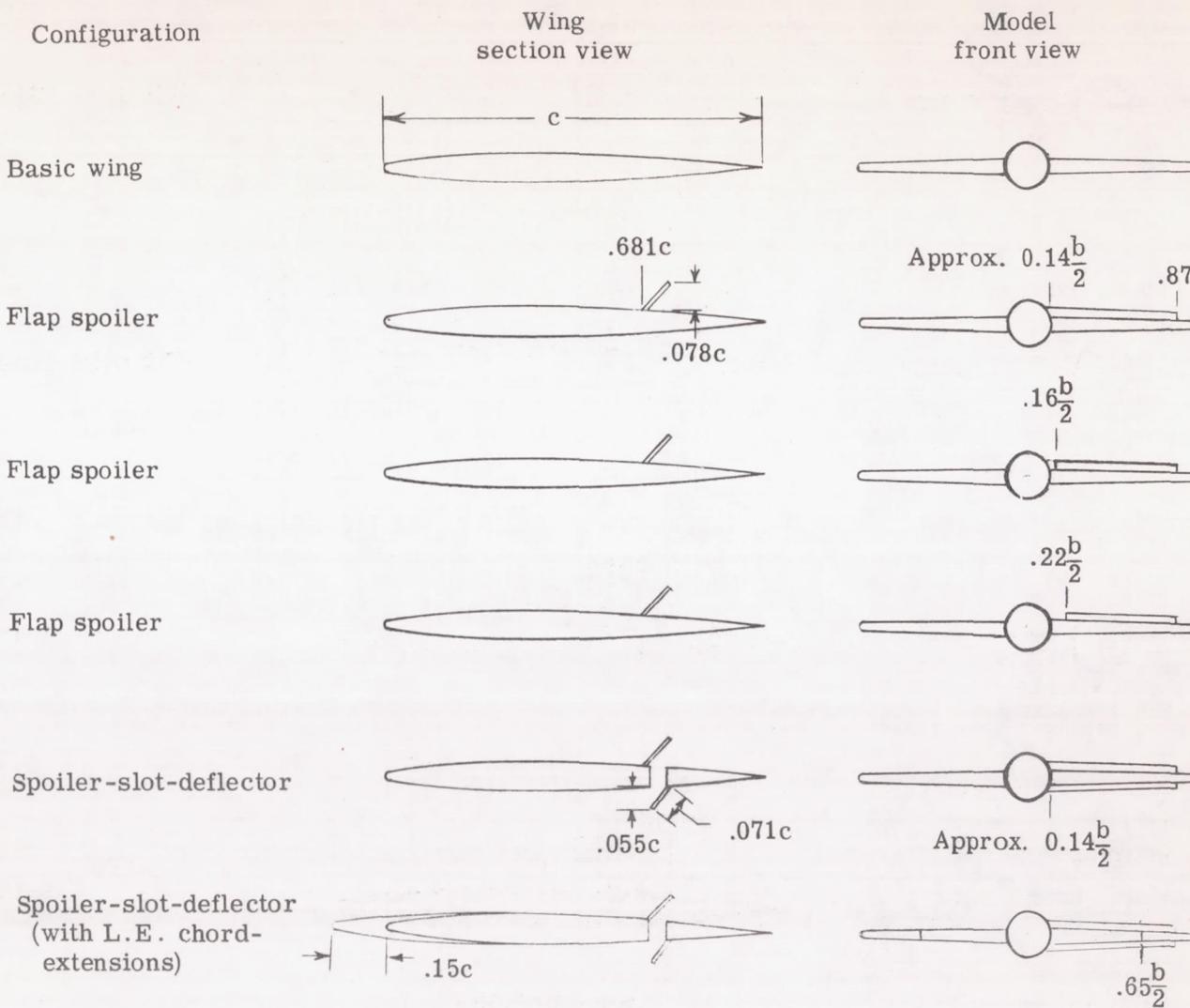


Figure 2.- Geometry of test configurations. (Not to scale.)

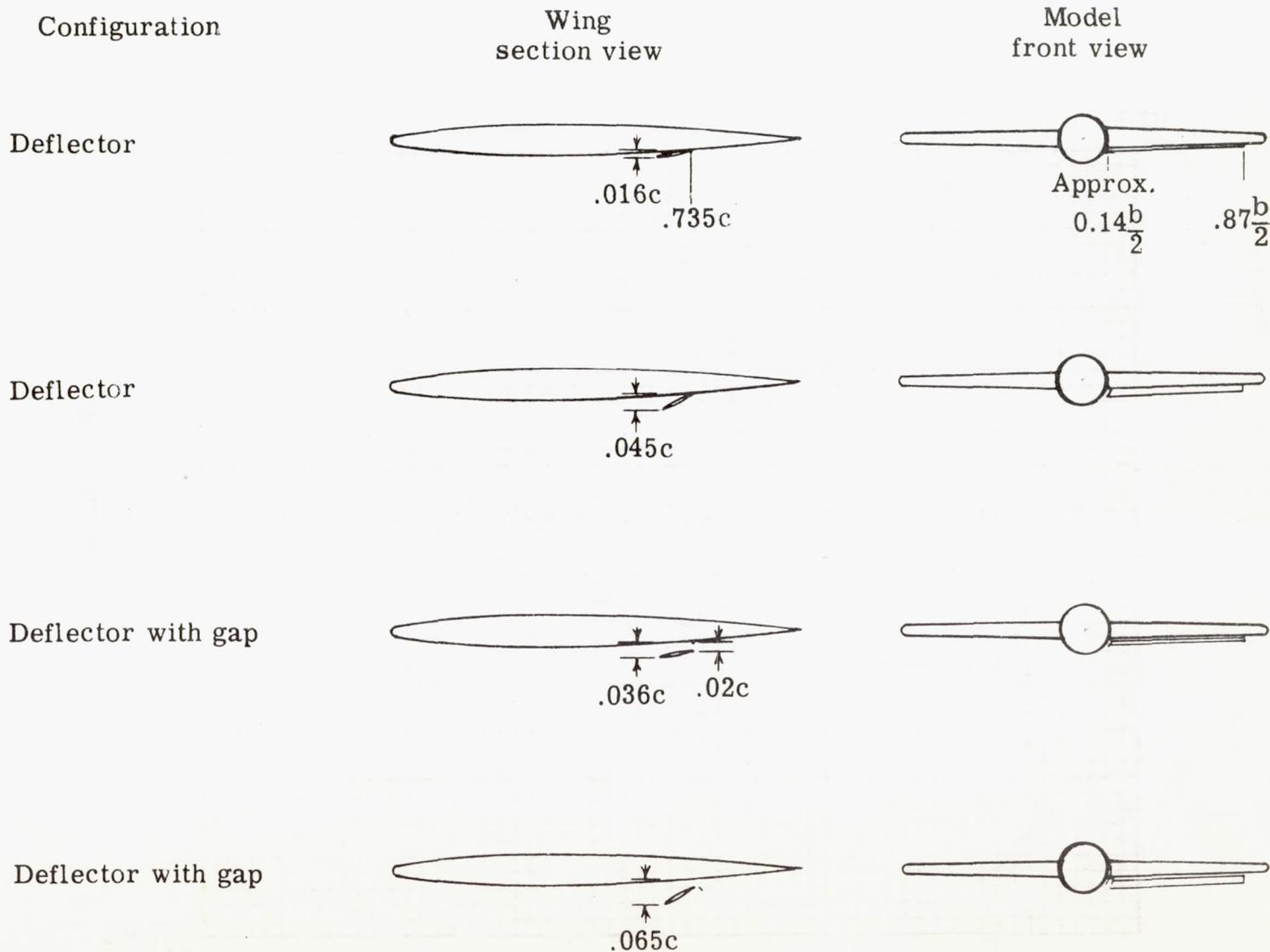


Figure 2.- Concluded.

CONFIDENTIAL

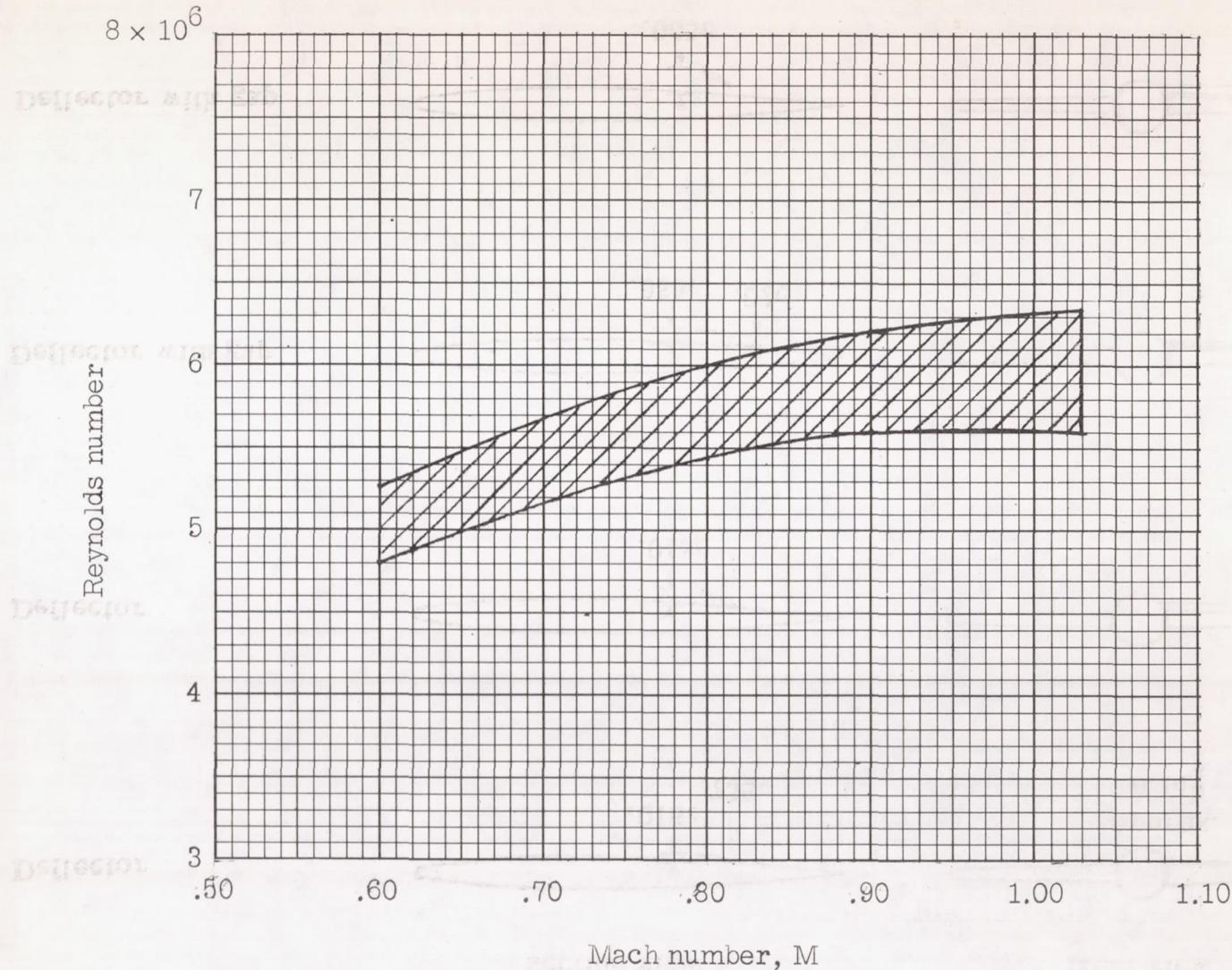


Figure 3.- Variation of Reynolds number (based on wing mean aerodynamic chord) with Mach number.

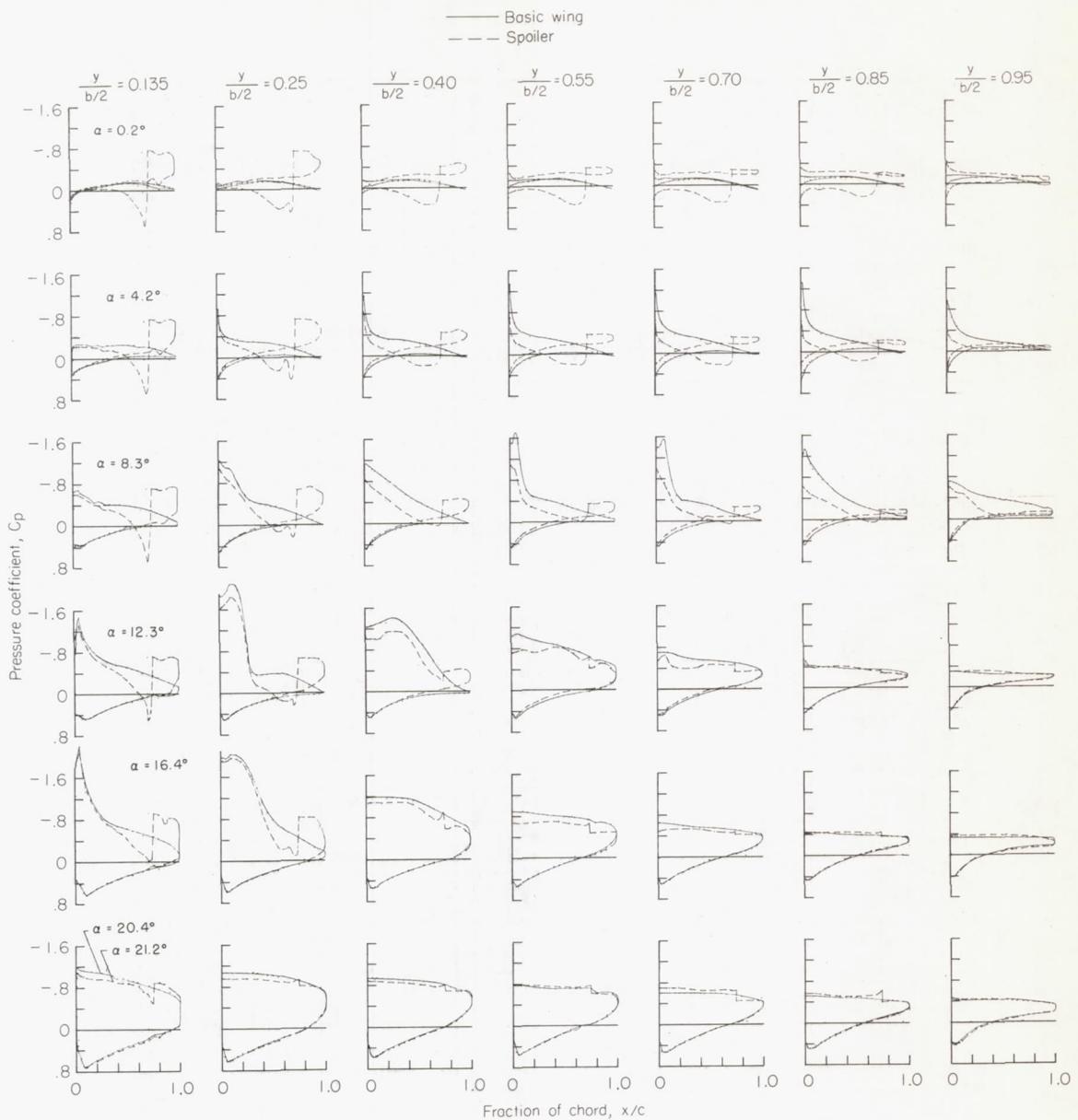
(a) $M = 0.60$.

Figure 4.- Wing chordwise pressure distributions for the basic model and a spoiler aileron configuration.

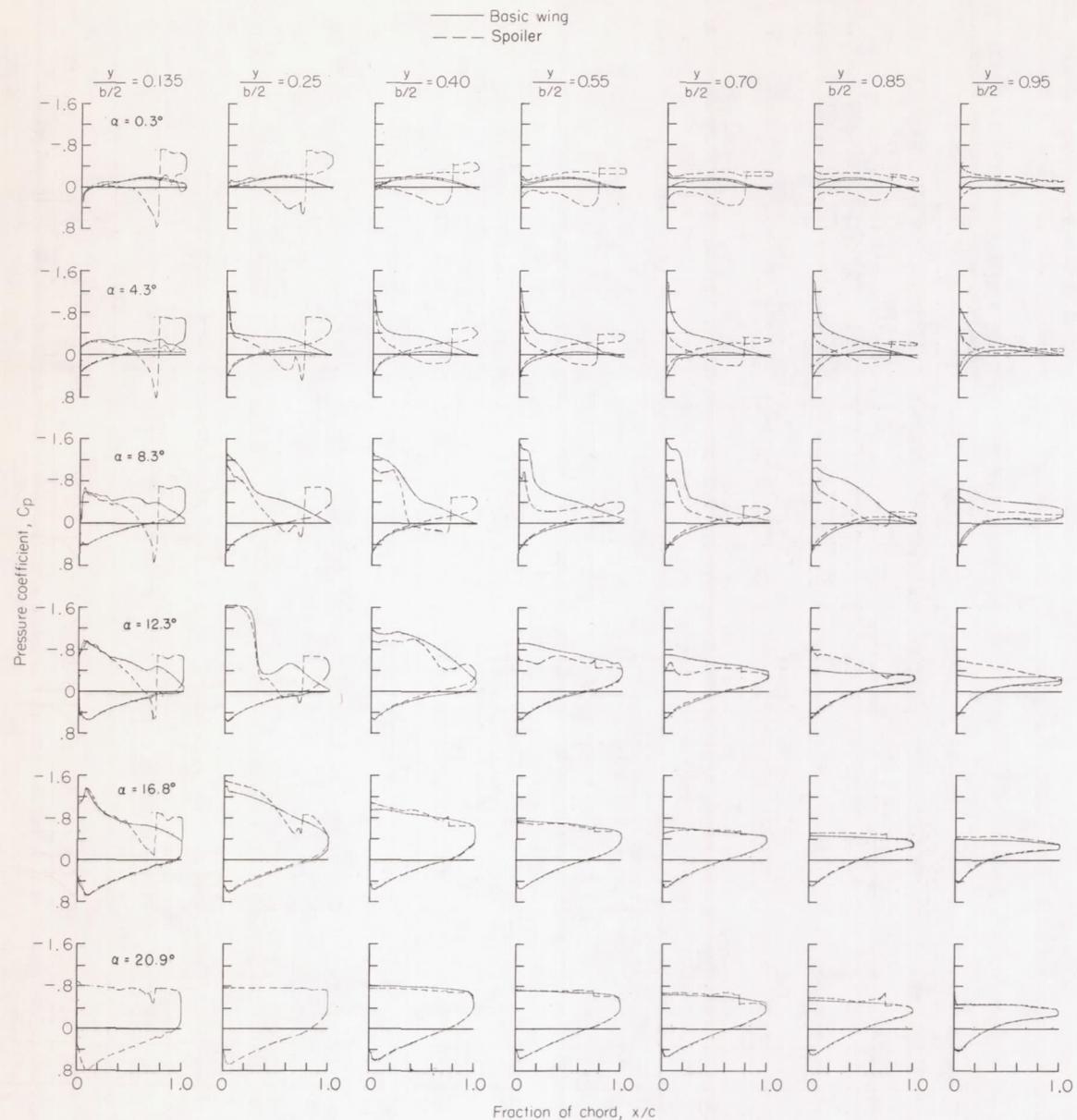
(b) $M = 0.80.$

Figure 4.- Continued.

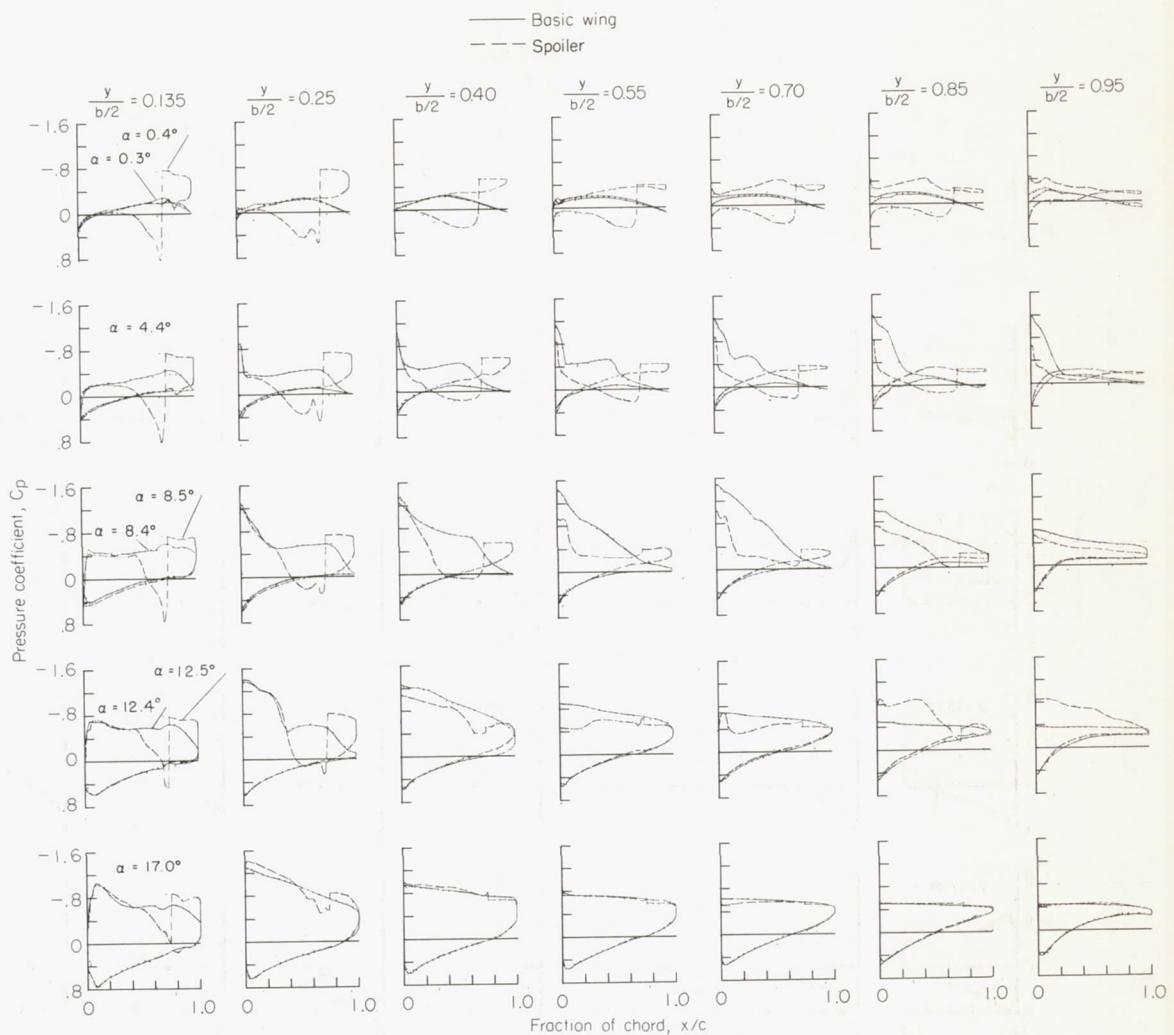
(c) $M = 0.90$.

Figure 4.- Continued.

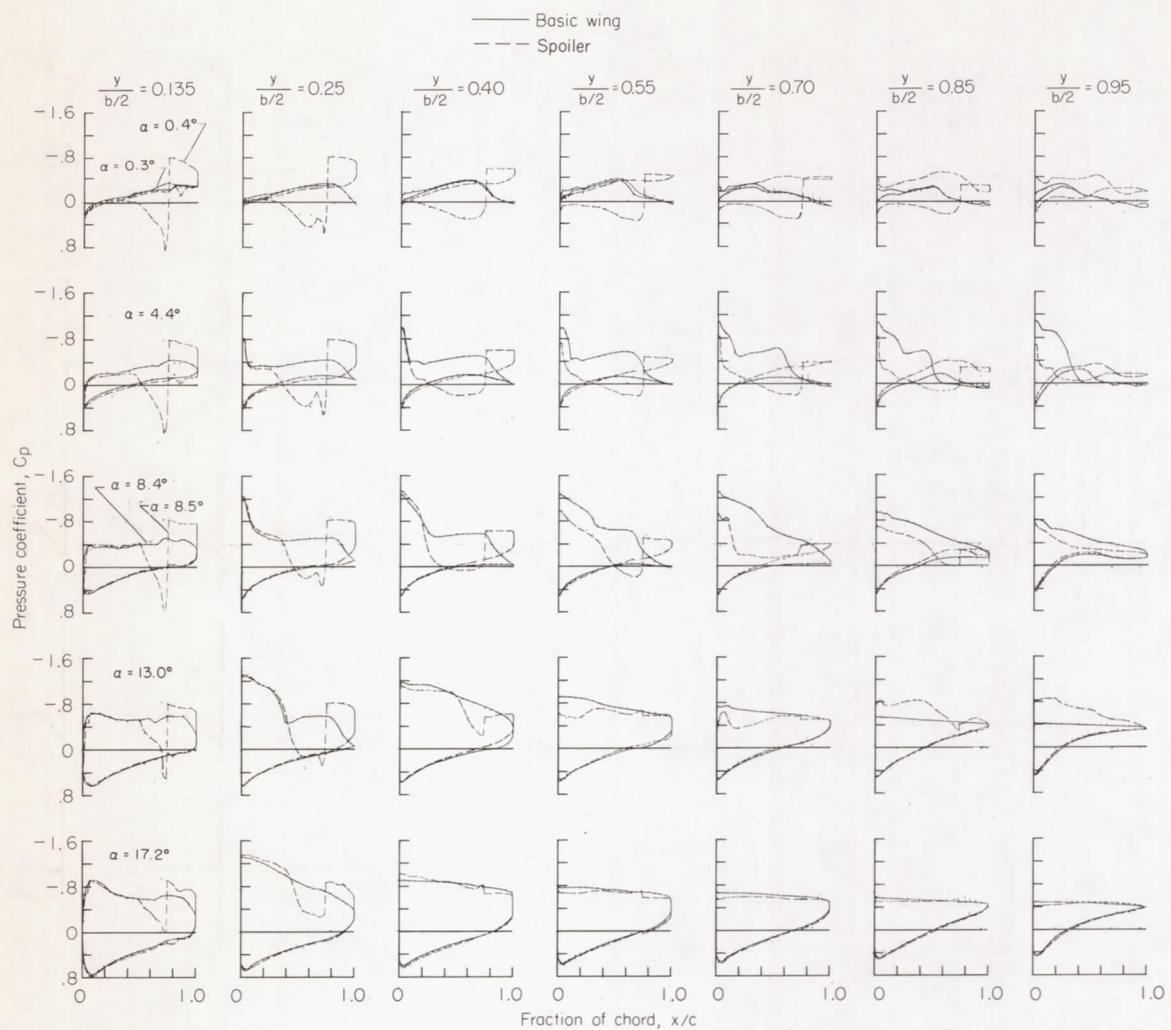
(d) $M = 0.94$.

Figure 4.- Continued.

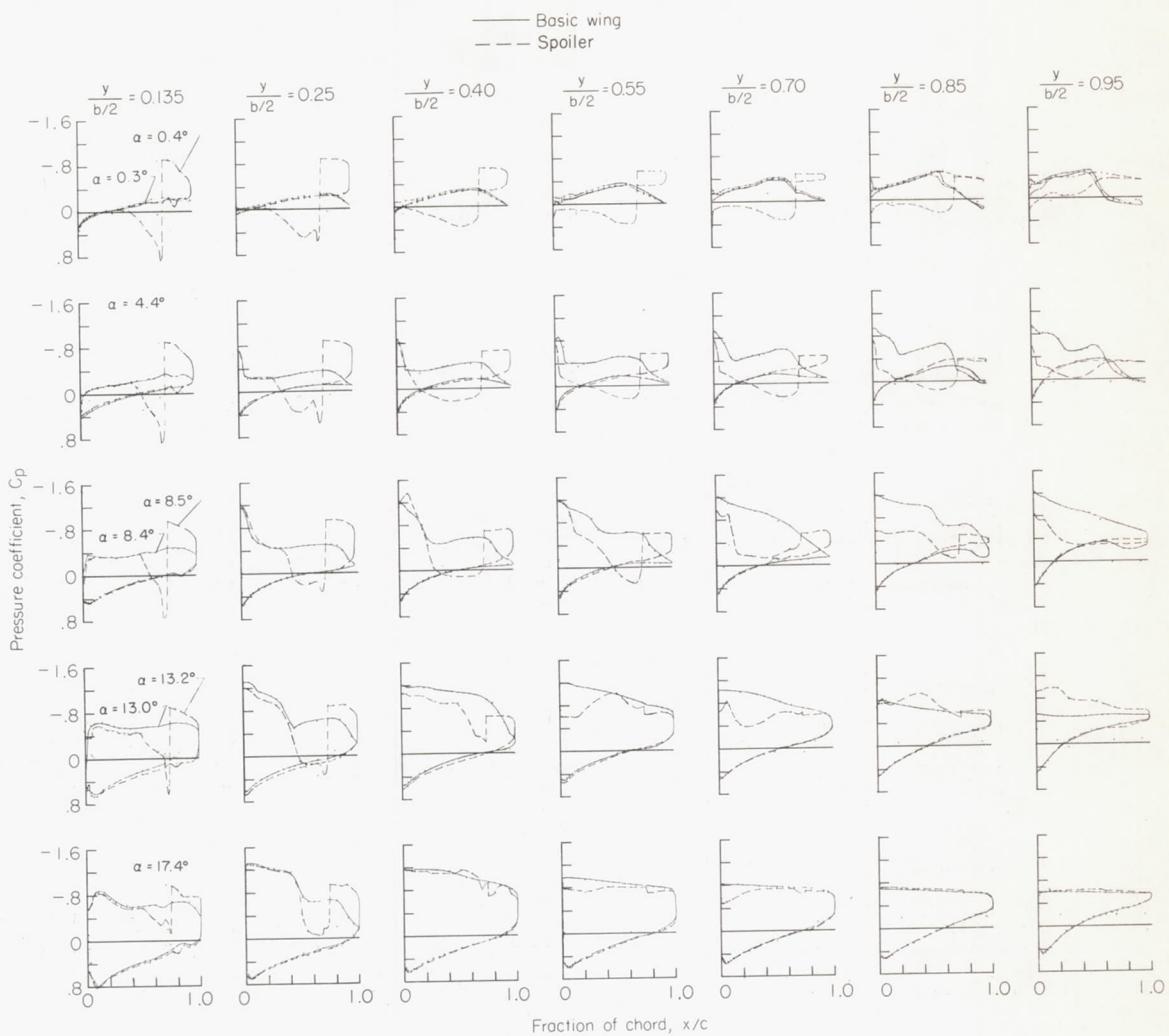
(e) $M = 0.98$.

Figure 4.- Continued.

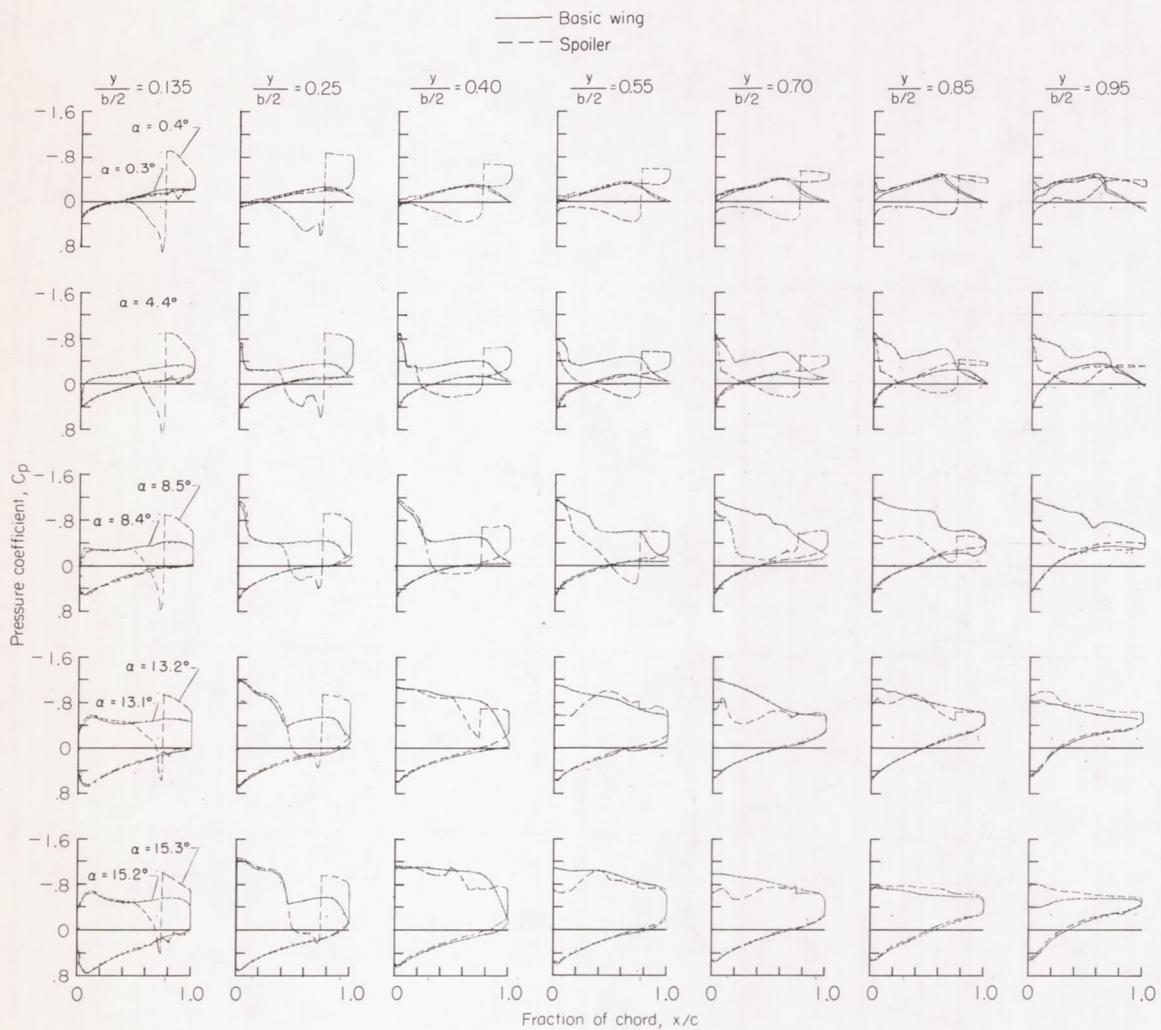
(f) $M = 1.00$.

Figure 4.- Continued.

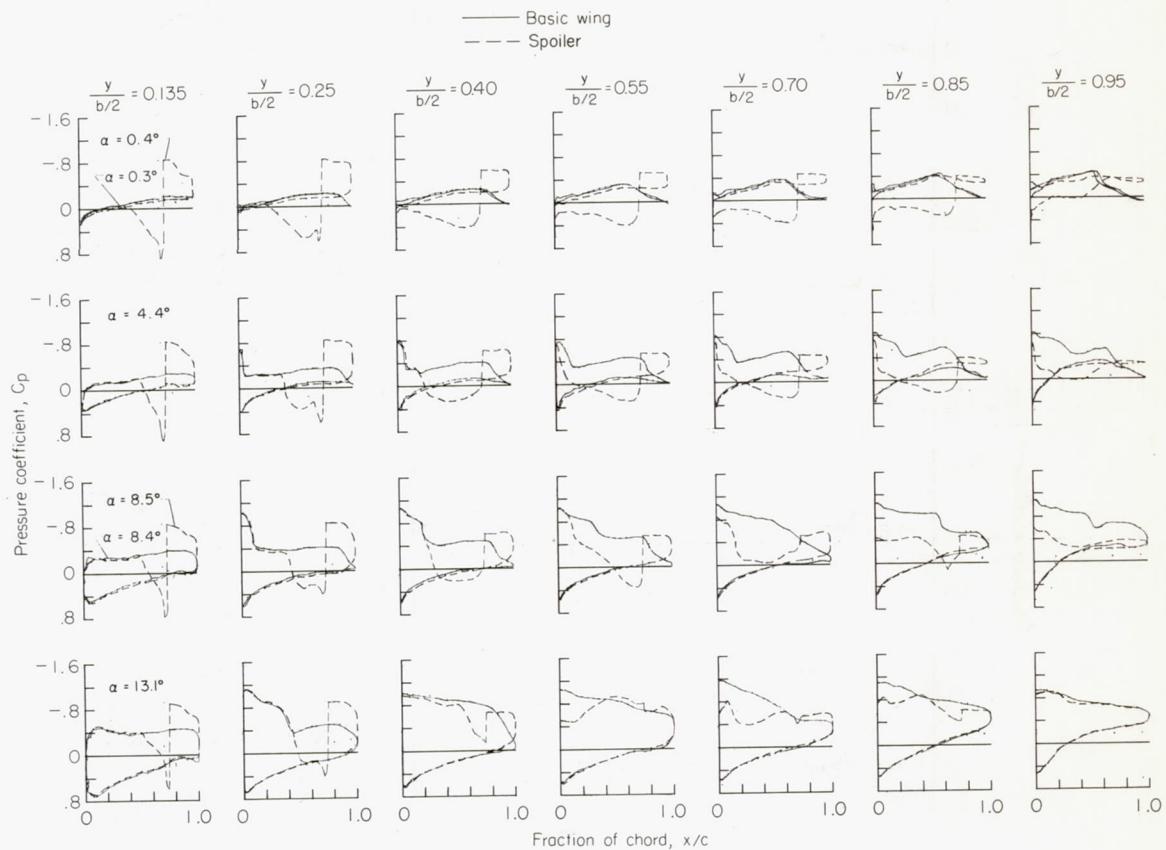
(g) $M = 1.03.$

Figure 4.- Concluded.

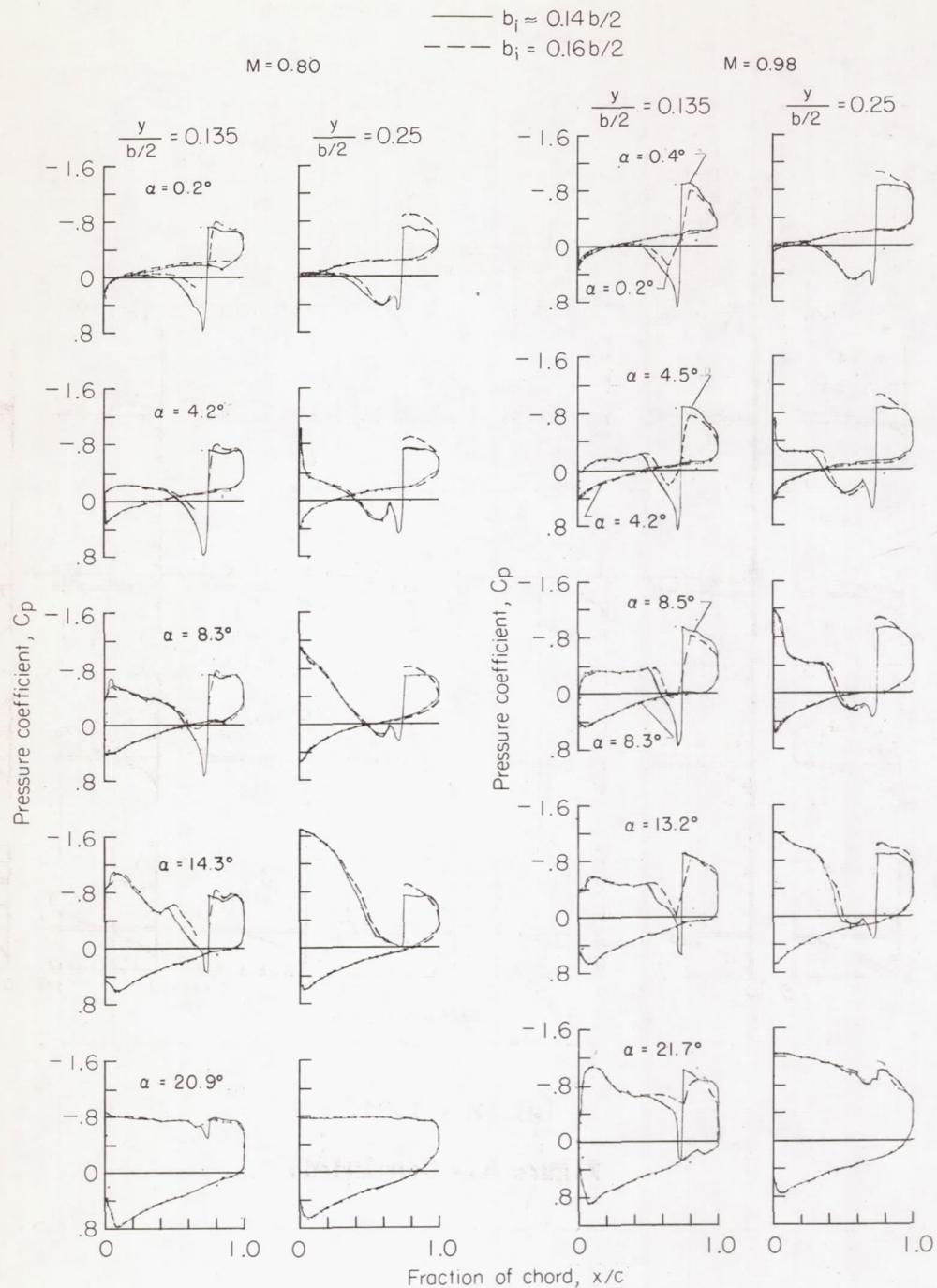


Figure 5.- Wing chordwise pressure distributions showing the effect of changing the inboard end position of a spoiler aileron from approximately 14 to 16 percent of the semispan.

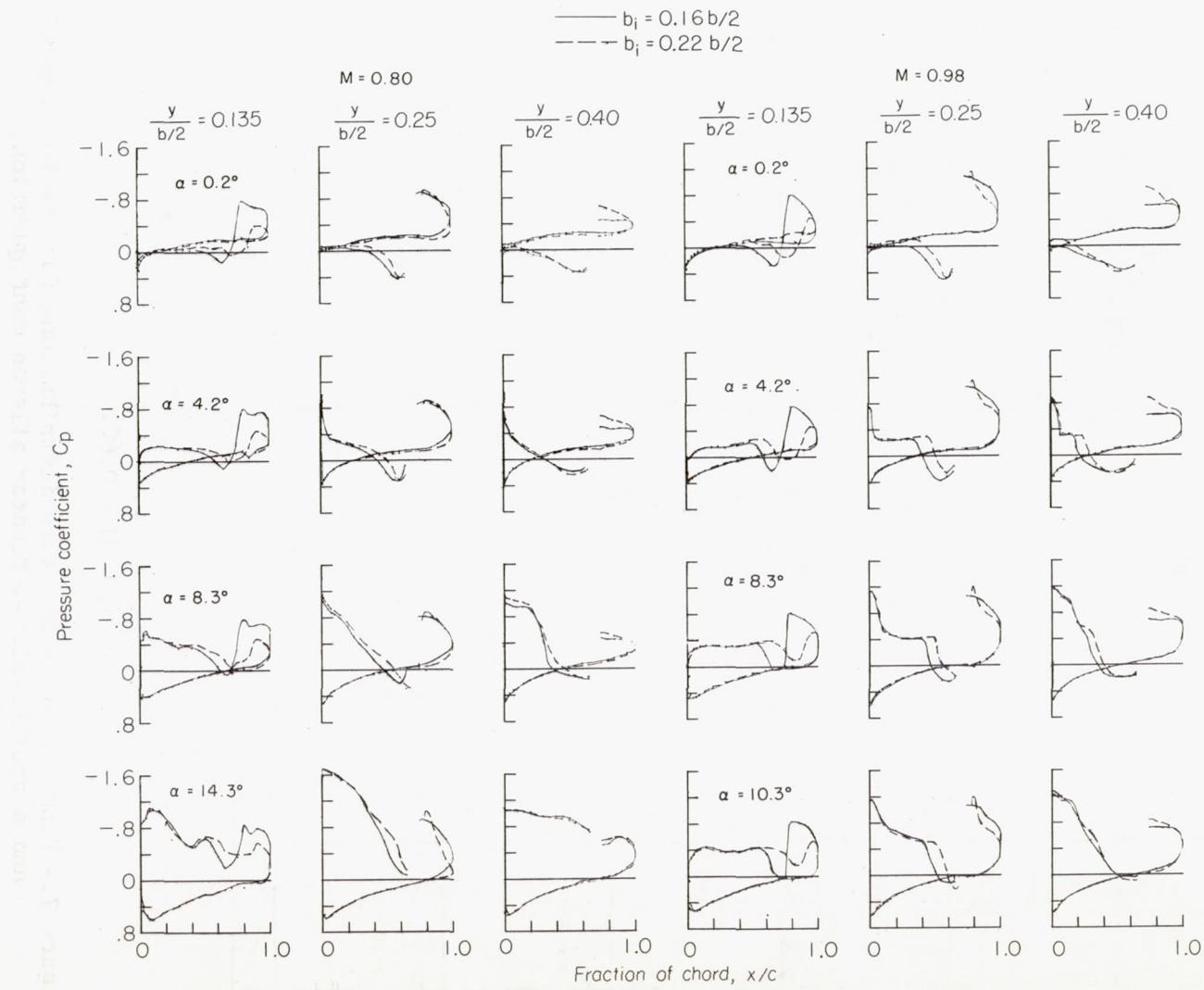


Figure 6.- Wing chordwise pressure distributions showing the effect of changing the inboard end position of a spoiler aileron from 16 to 22 percent of the semispan.

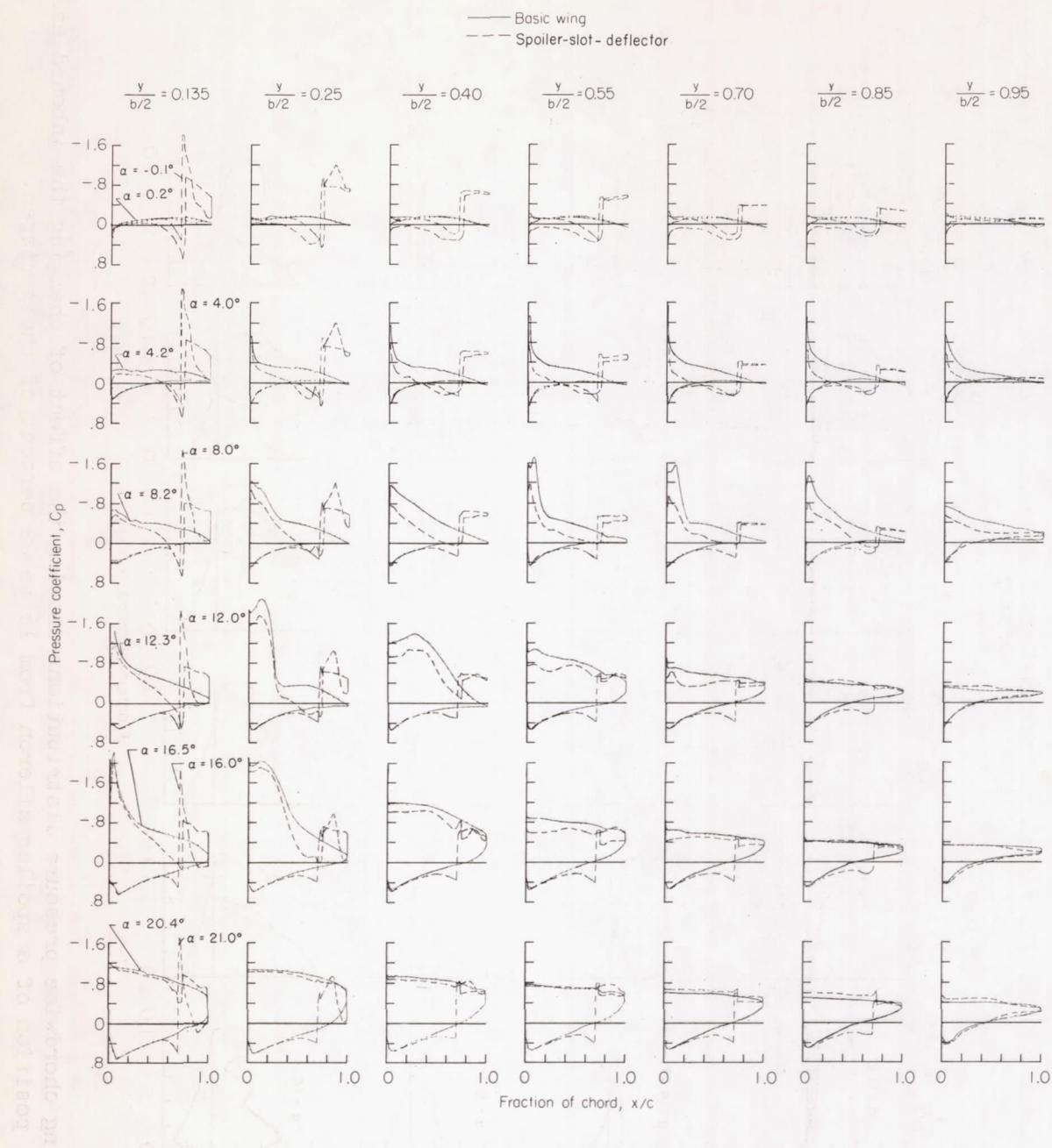
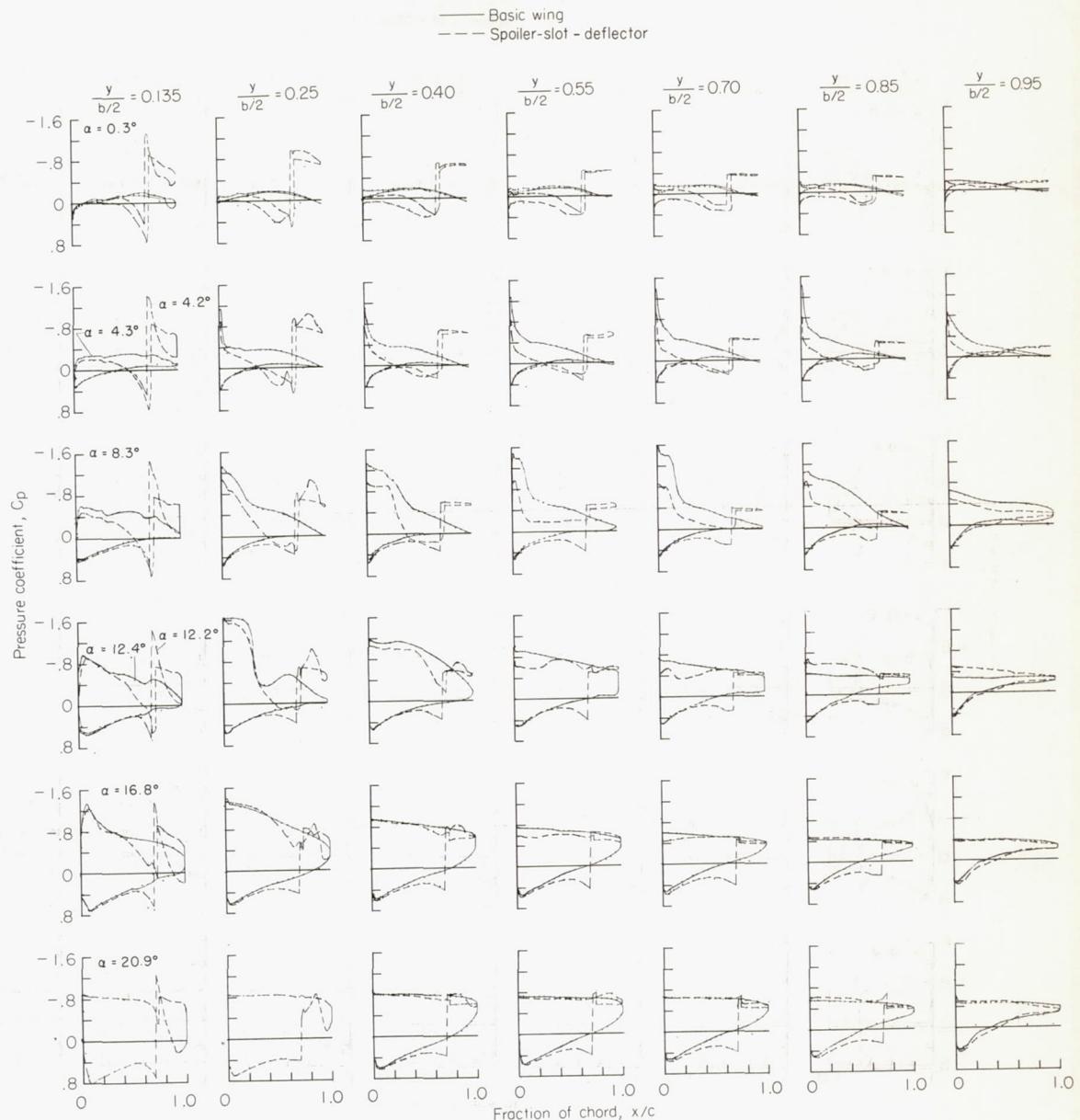
(a) $M = 0.60.$

Figure 7.- Wing chordwise pressure distributions for the basic model and a spoiler-slot-deflector aileron configuration.



(b) M = 0.80.

Figure 7.- Continued.

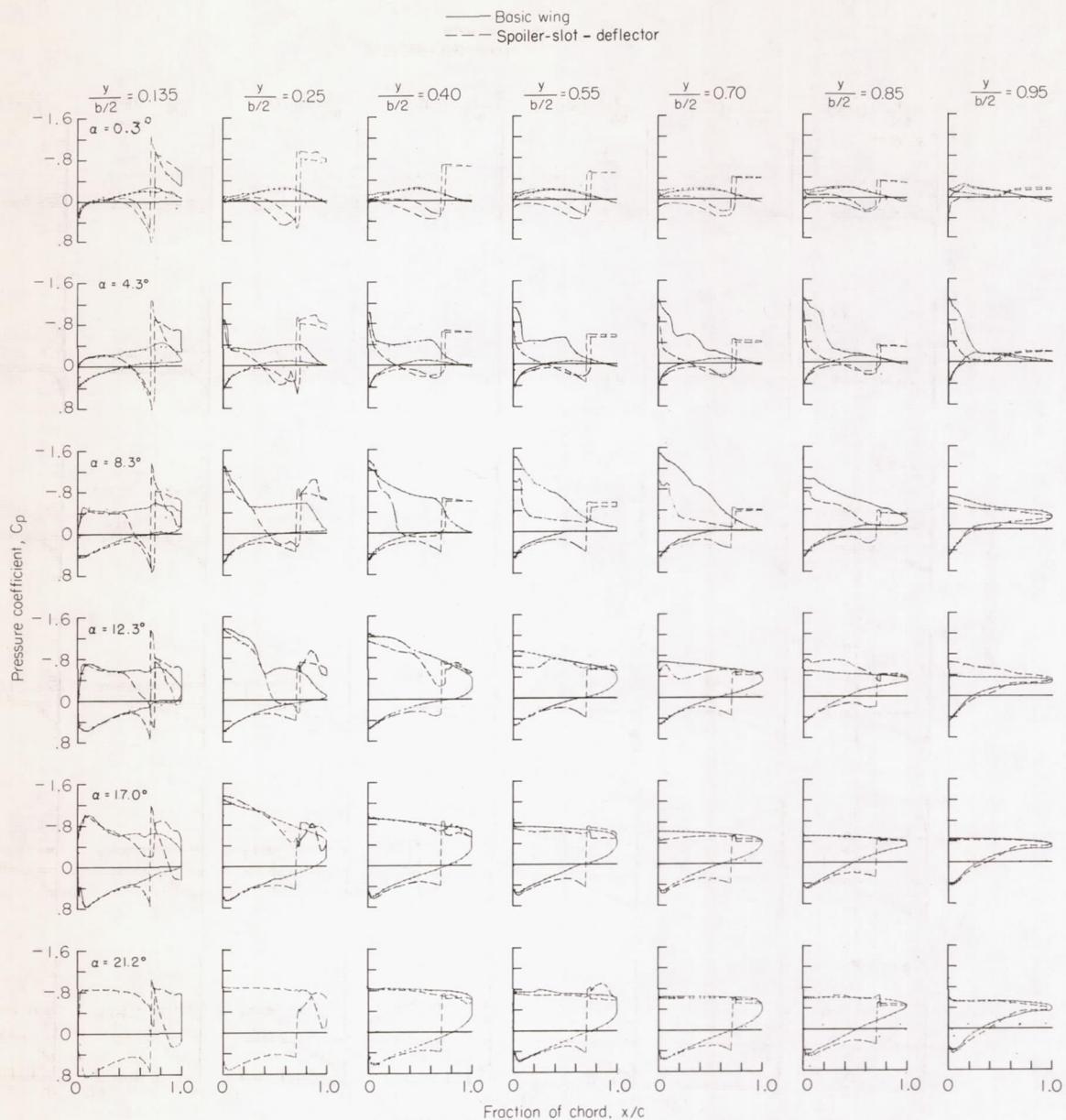
(c) $M = 0.90.$

Figure 7.- Continued.

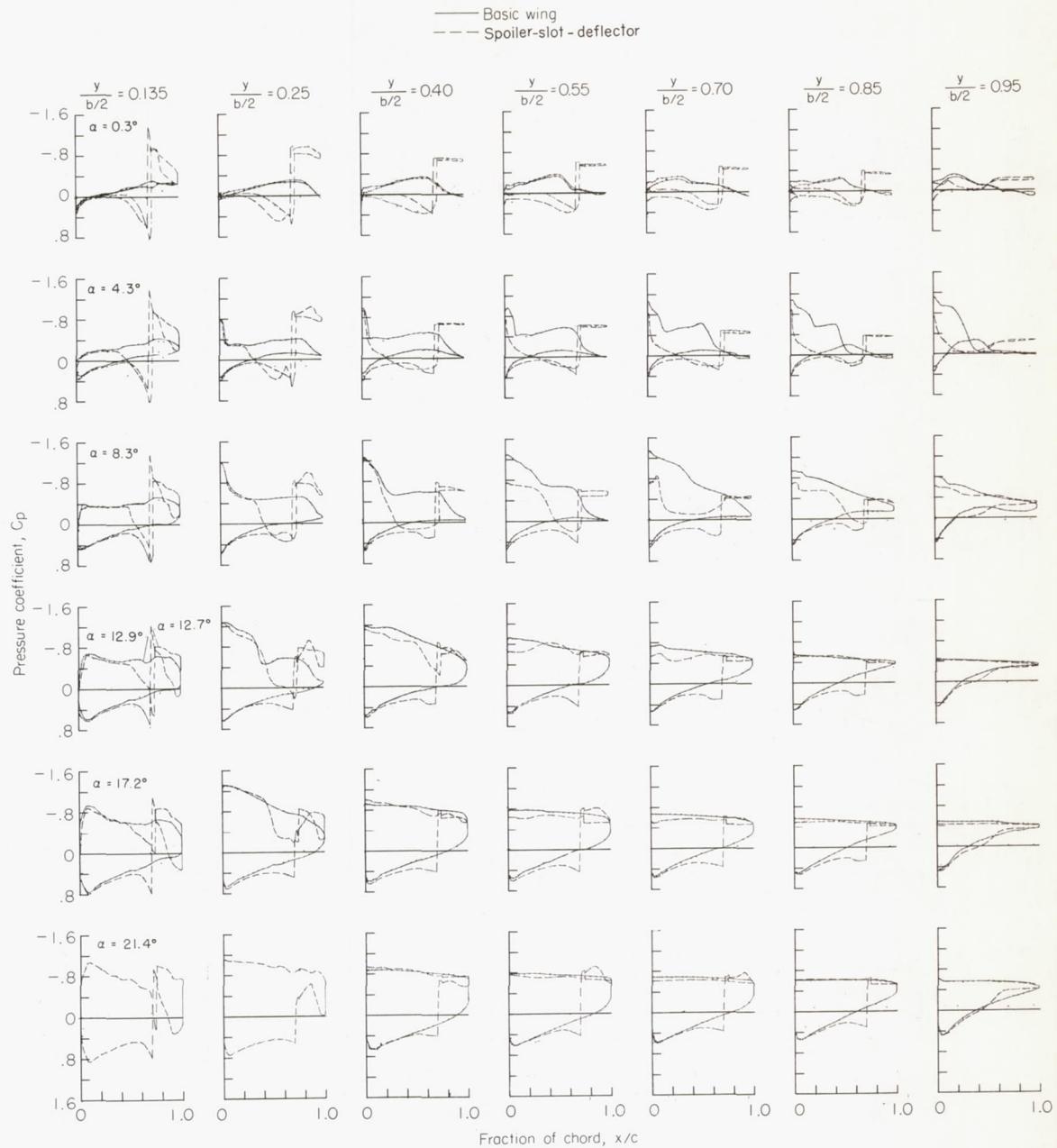
(d) $M = 0.94.$

Figure 7.- Continued.

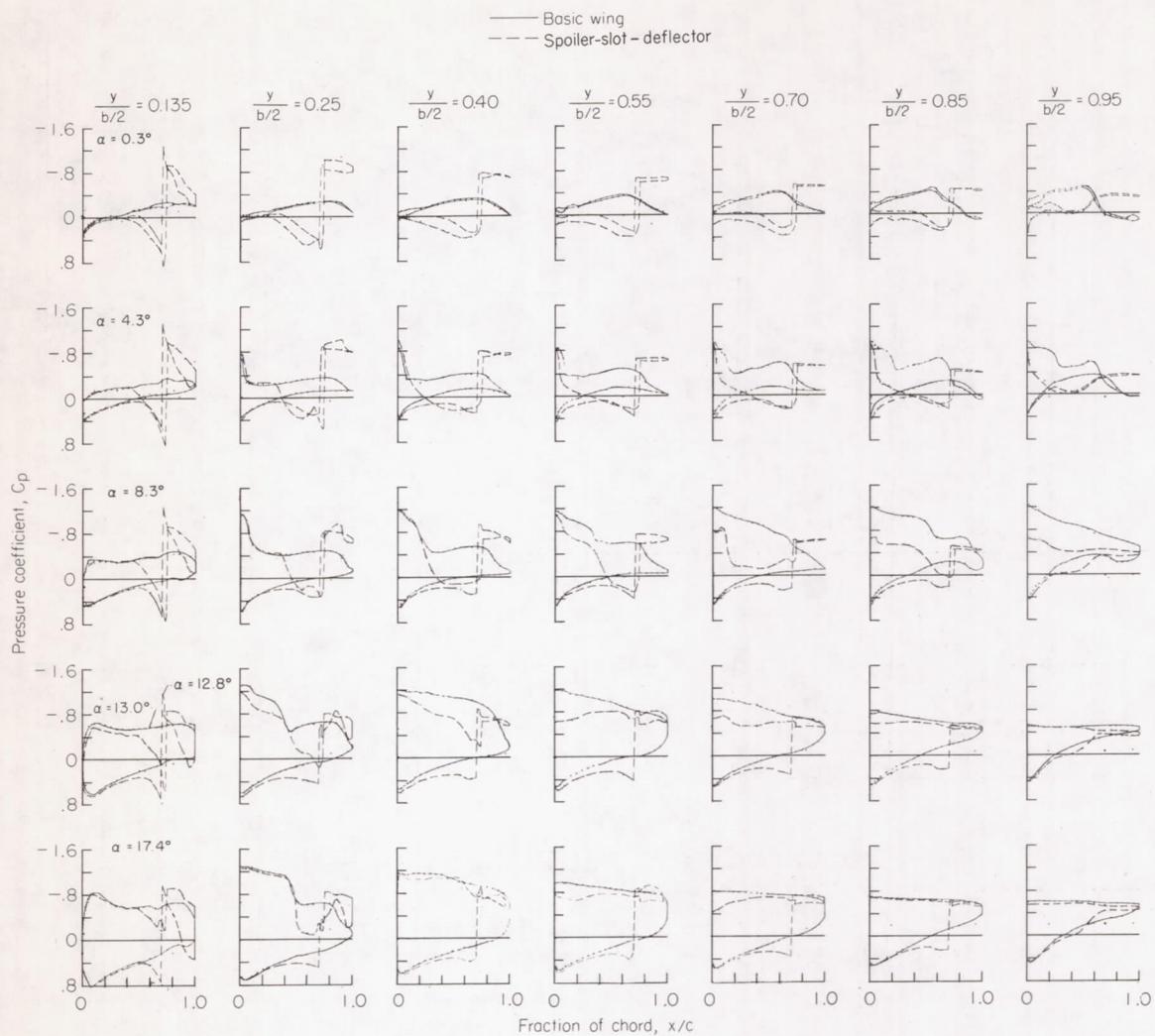
(e) $M = 0.98$.

Figure 7--Continued.

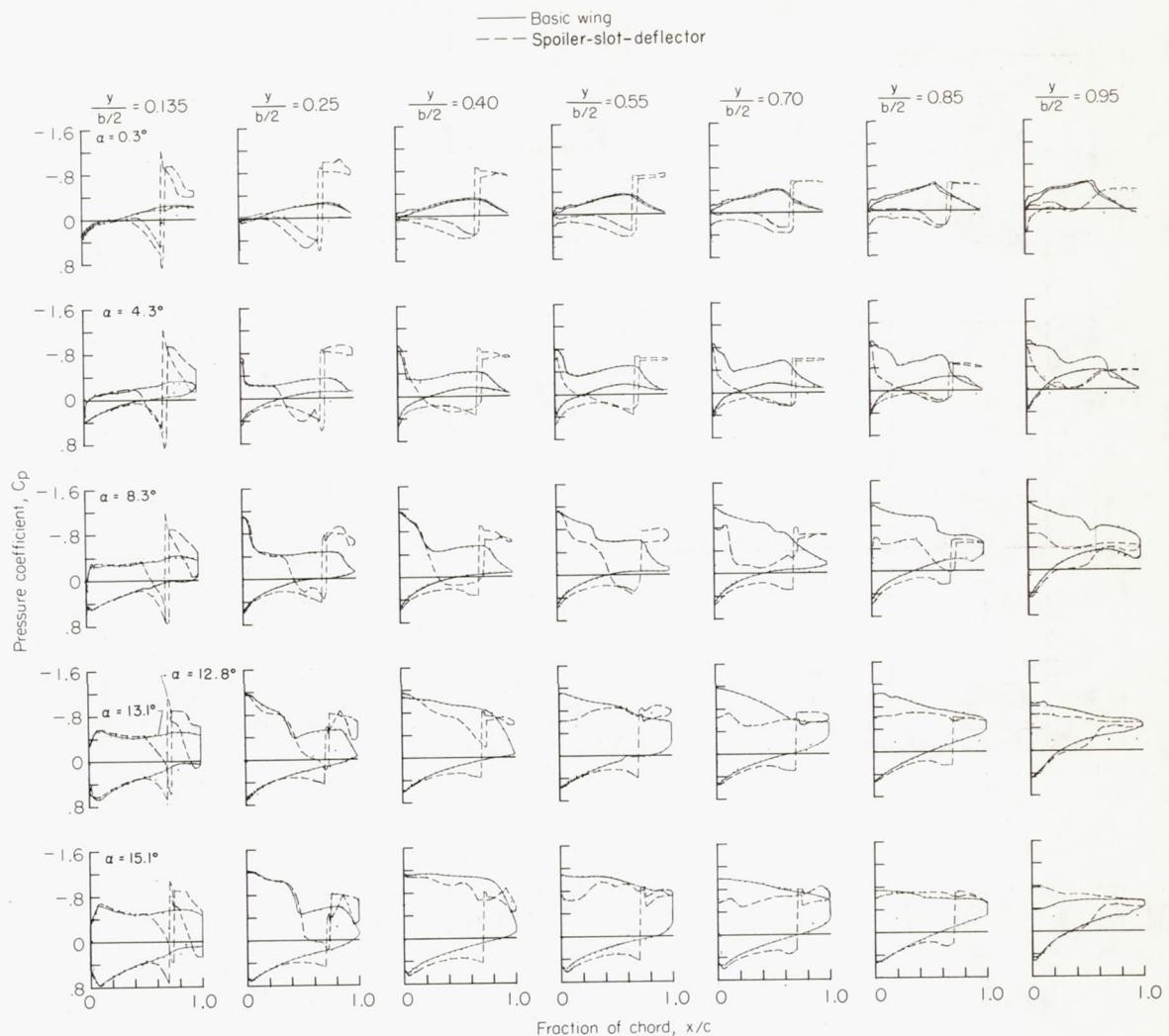
(f) $M = 1.00.$

Figure 7.- Continued.

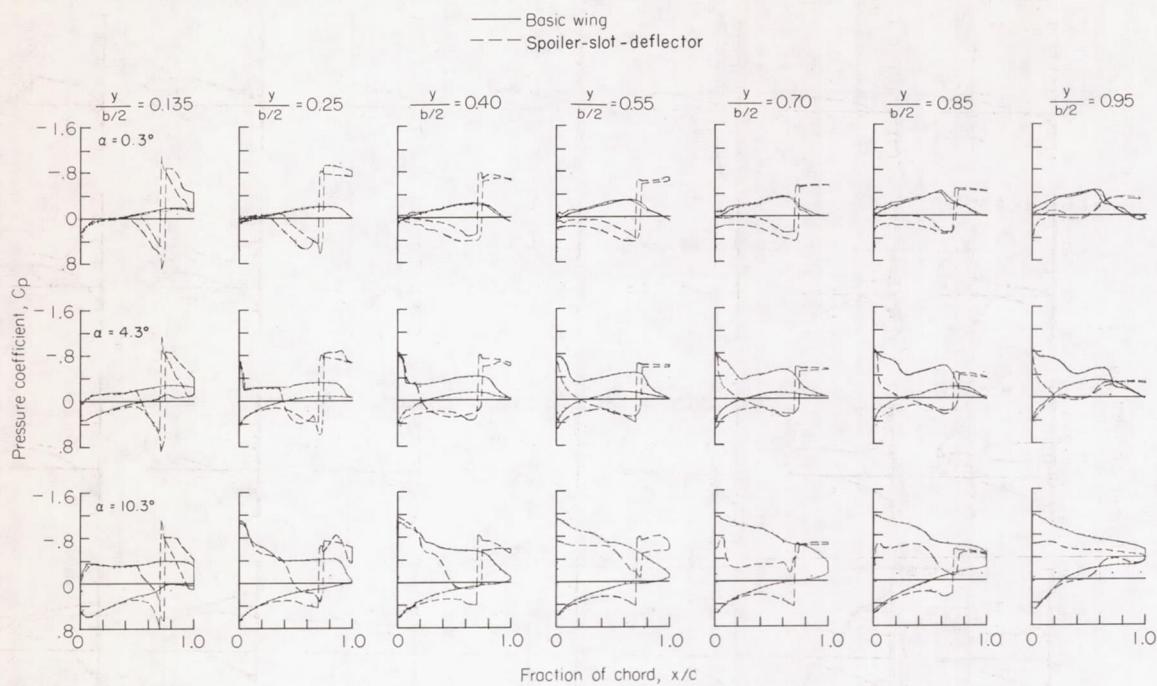
(g) $M = 1.03$.

Figure 7.- Concluded.

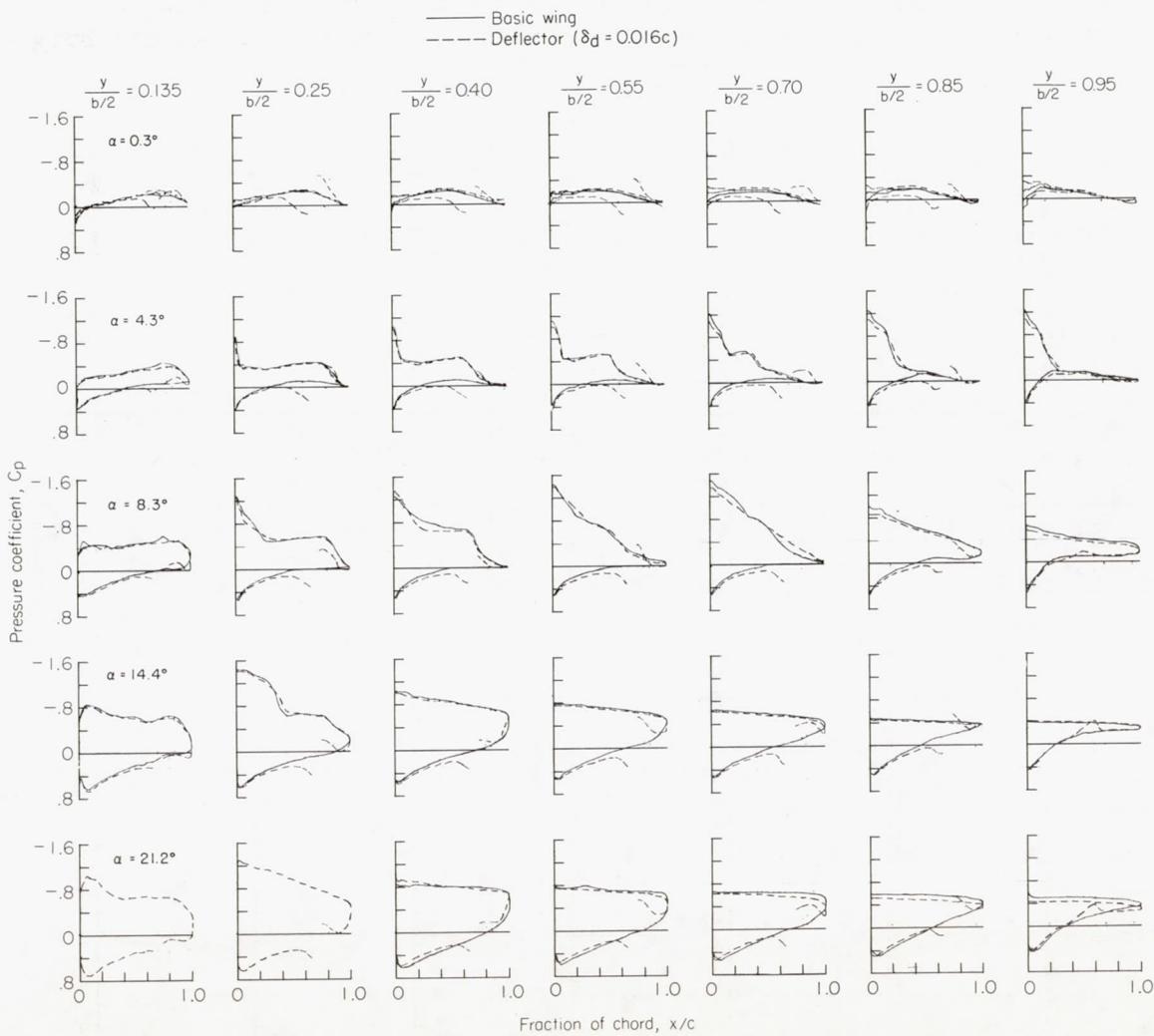


Figure 8.- Wing chordwise pressure distributions for the basic model and a deflector ($\delta_d = 0.016c$) aileron configuration at a Mach number of 0.90.

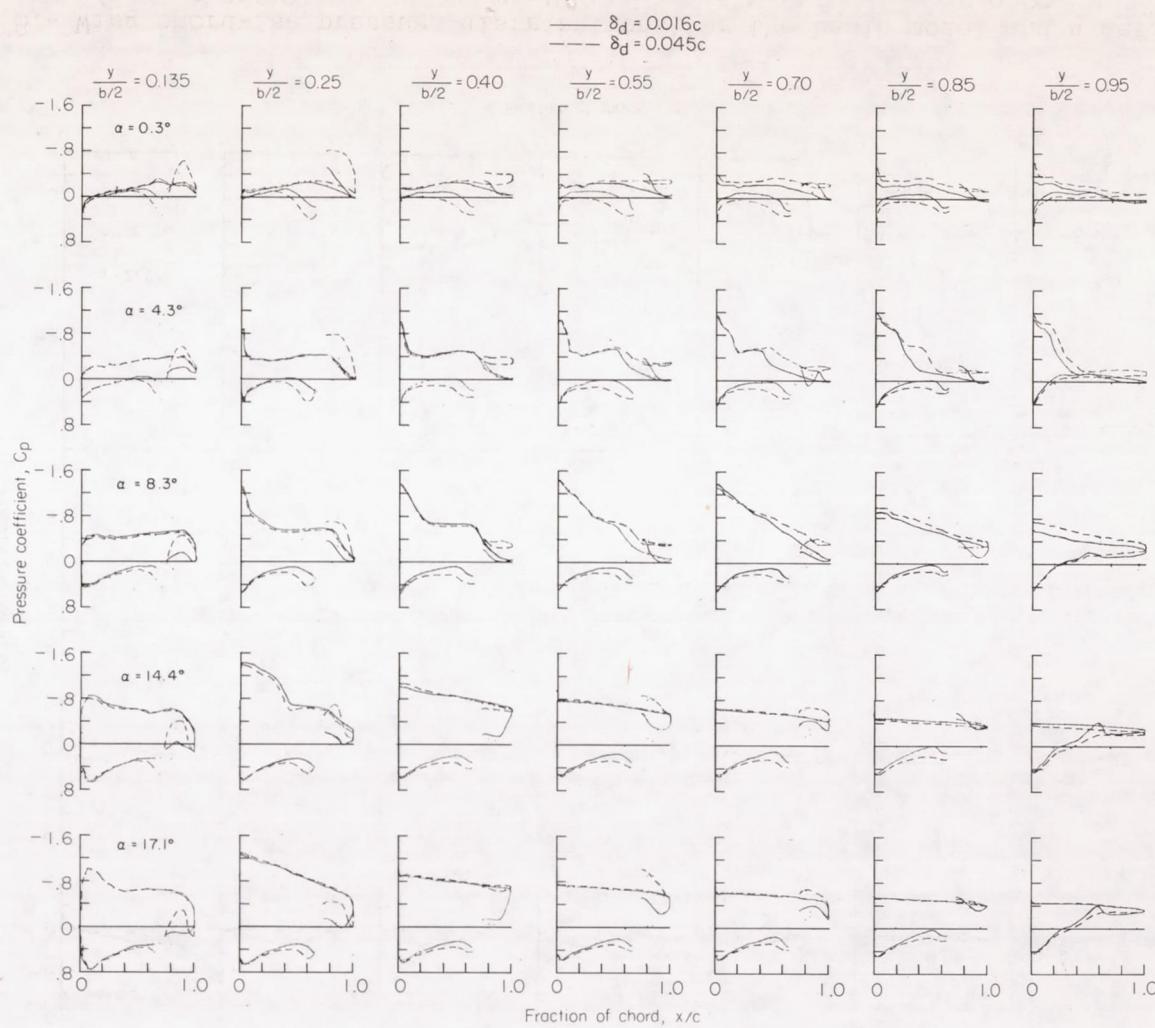


Figure 9.- Wing chordwise pressure distributions showing the effect of changing projection of a deflector aileron from $0.016c$ to $0.045c$ at a Mach number of 0.90.

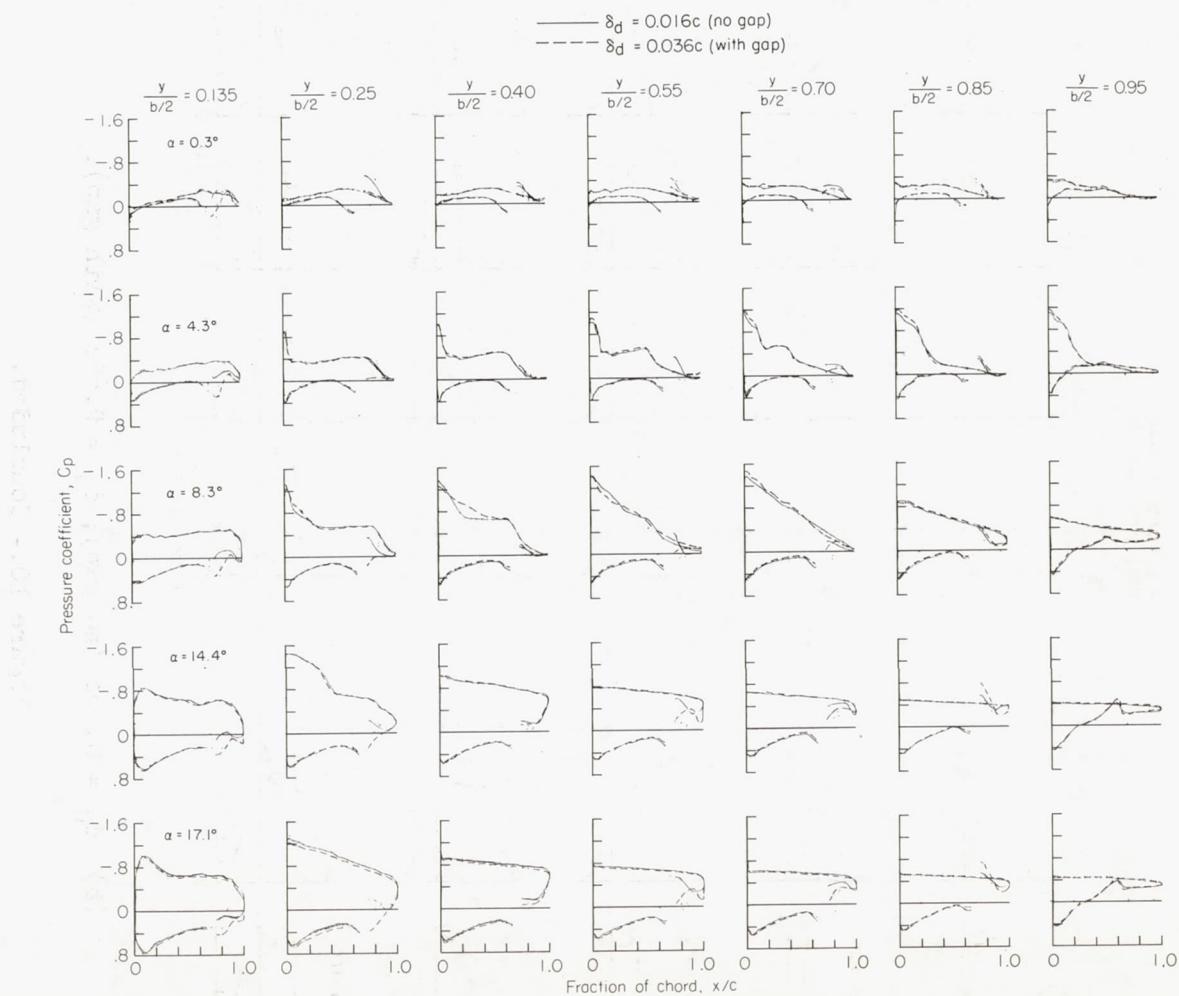
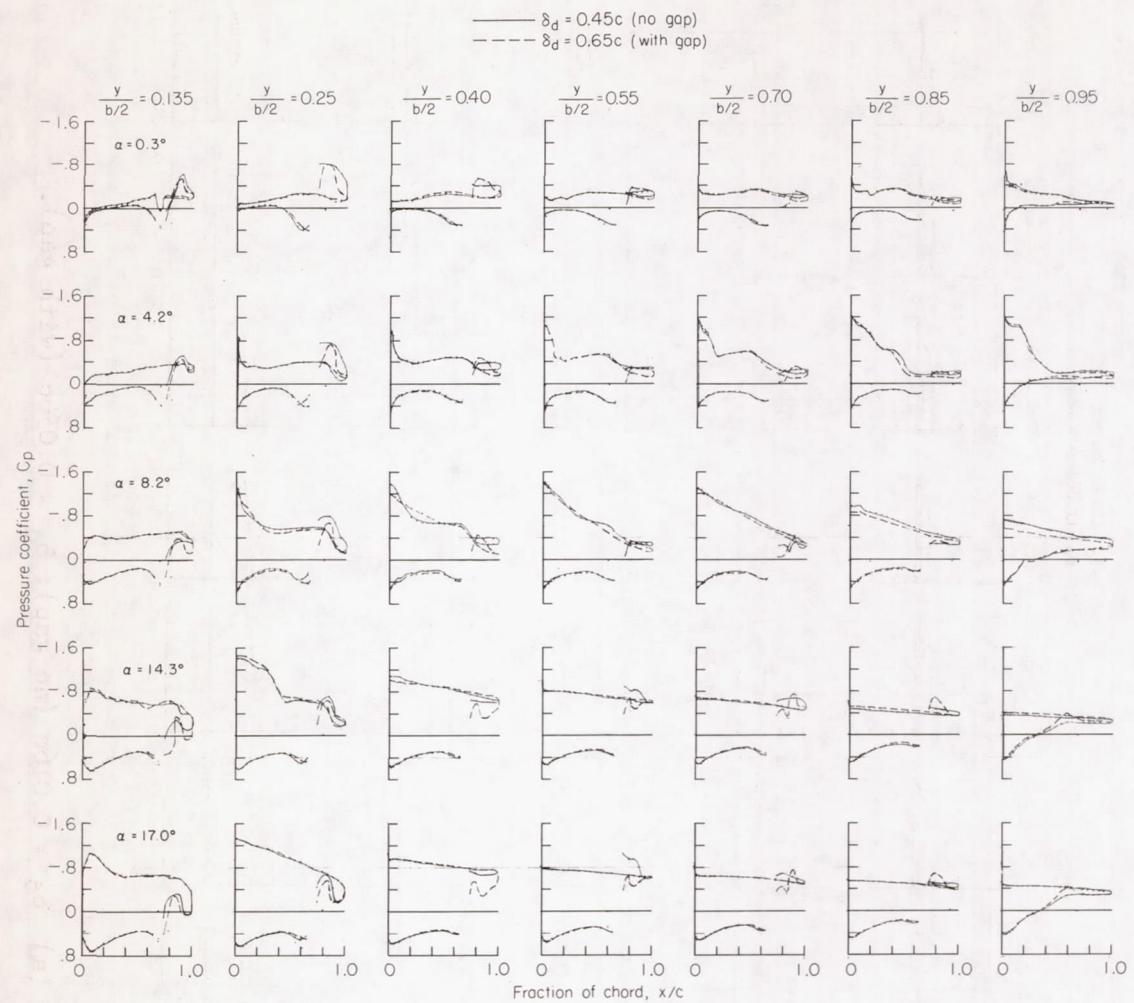
(a) $\delta_d = 0.016c$ (no gap); $\delta_d = 0.036c$ (with gap).

Figure 10.- Wing chordwise pressure distributions showing the effect of a $0.02c$ gap between a deflector aileron and the wing at a Mach number of 0.90.



(b) $\delta_d = 0.45c$ (no gap); $\delta_d = 0.65c$ (with gap).

Figure 10.- Concluded.

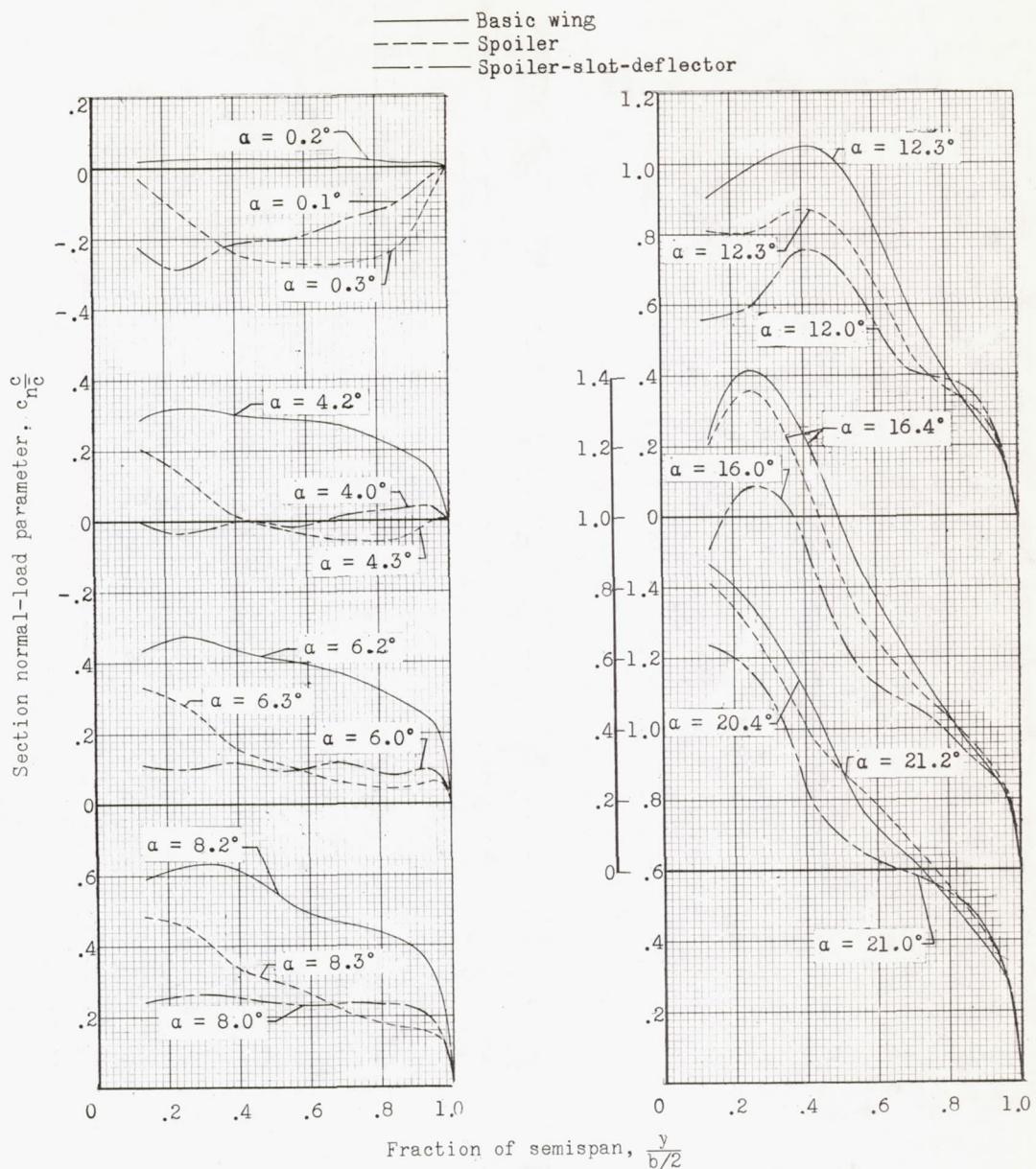
(a) $M = 0.60.$

Figure 11.- Wing semispan load distributions for the basic model, a spoiler aileron configuration, and a spoiler-slot-deflector aileron configuration.

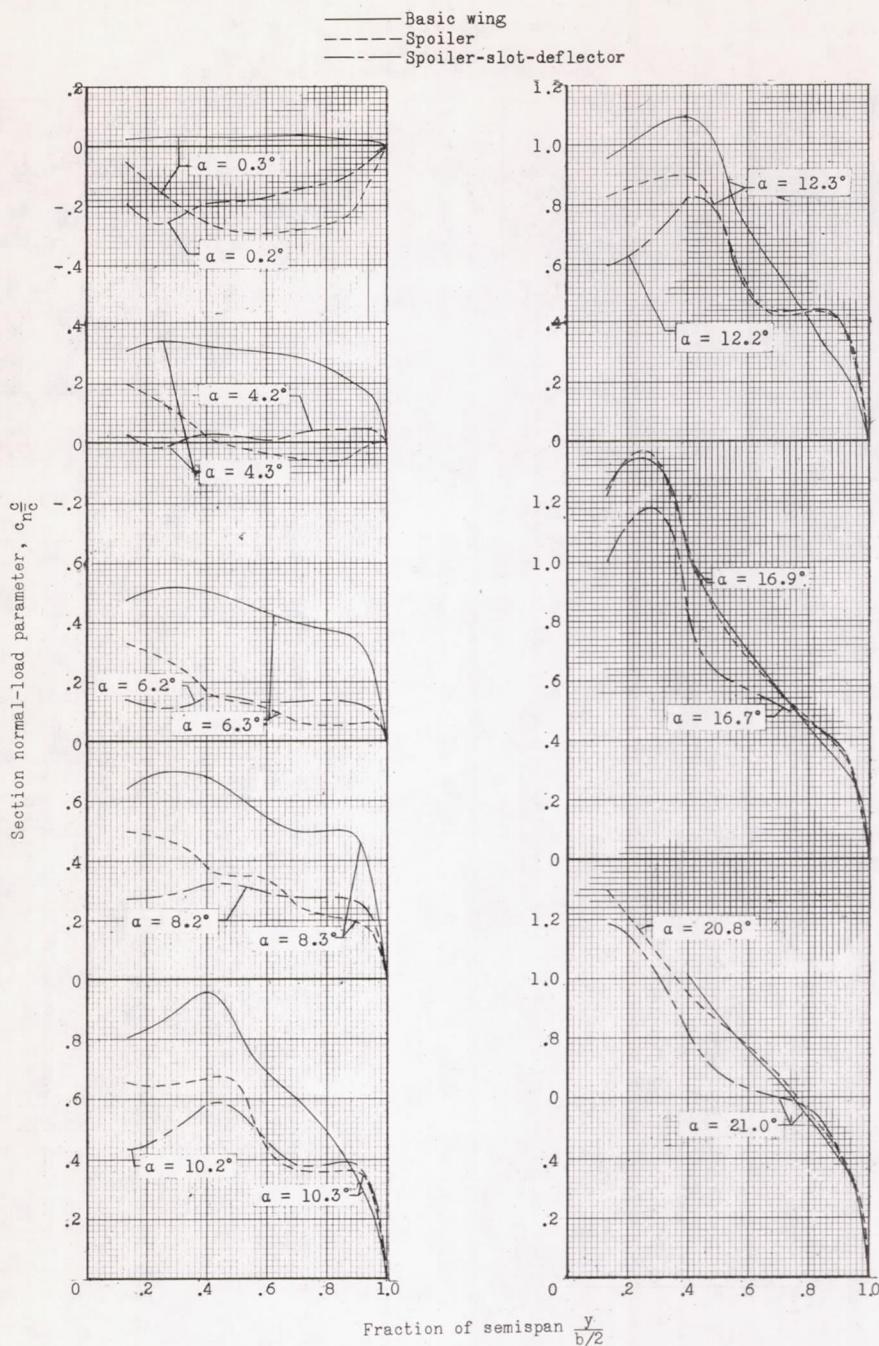
(b) $M = 0.80.$

Figure 11.- Continued.

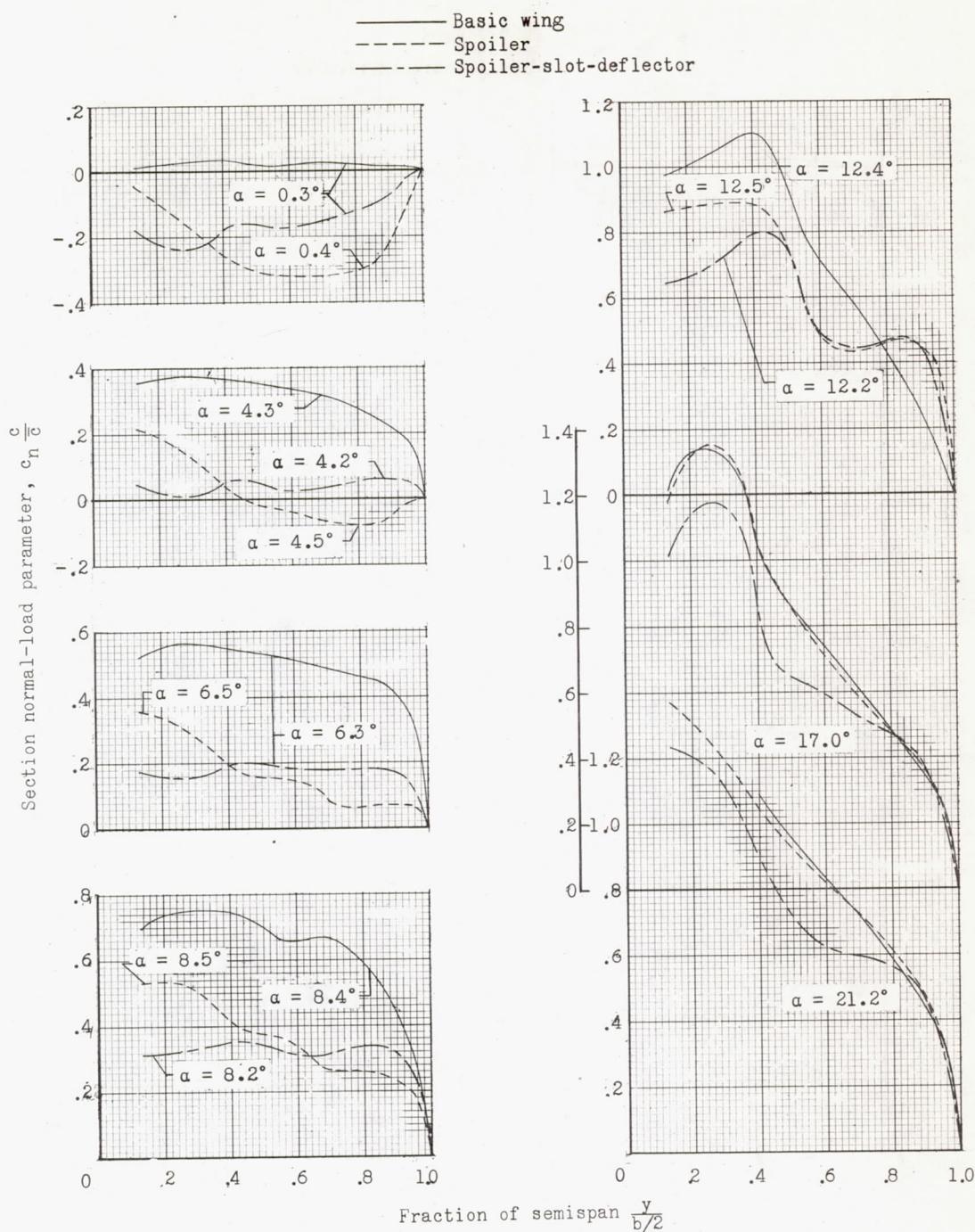
(c) $M = 0.90$.

Figure 11.- Continued.

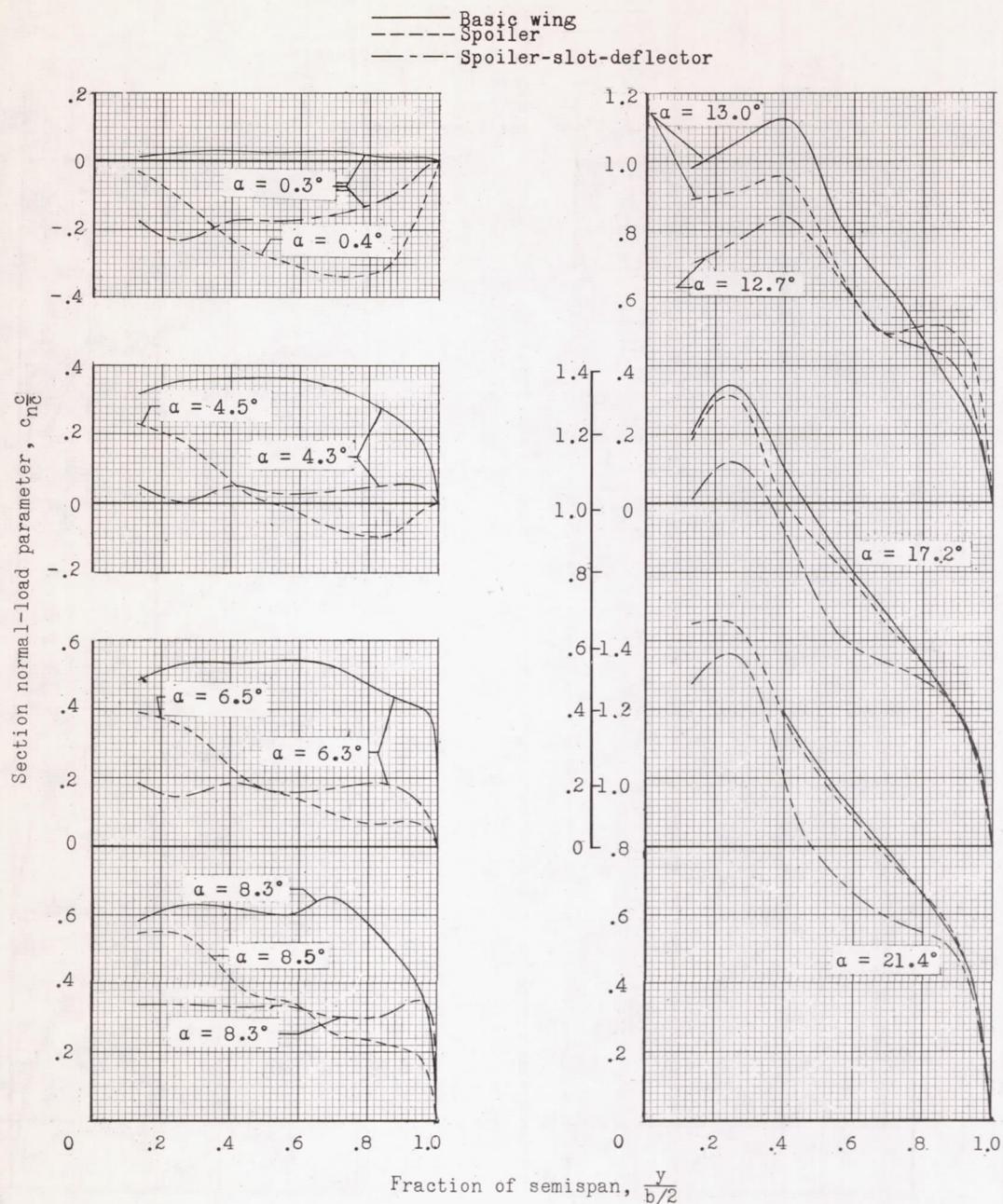
(d) $M = 0.94$.

Figure 11.- Continued.

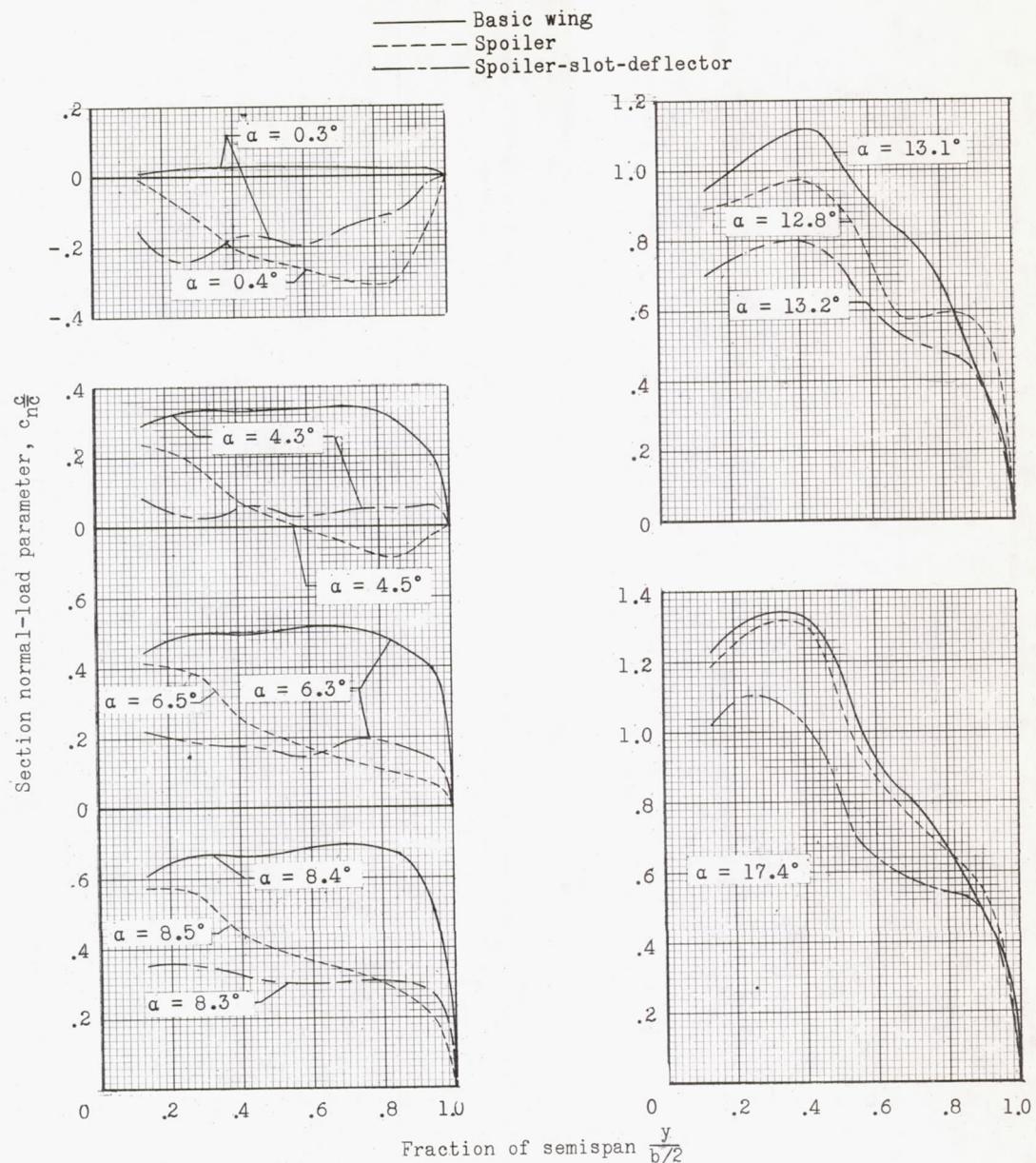
(e) $M = 0.98$.

Figure 11.- Continued.

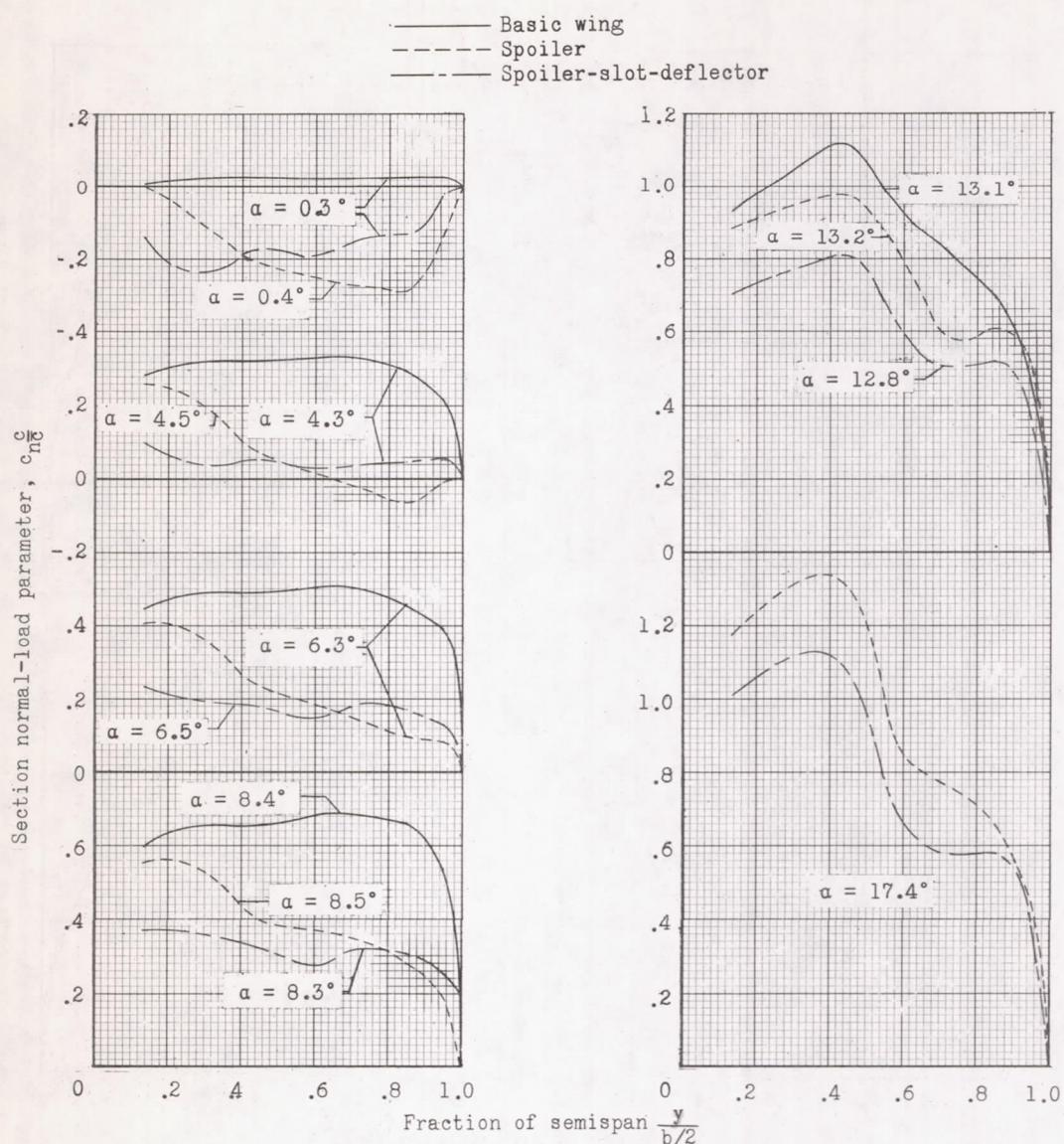
(f) $M = 1.00$

Figure 11.- Continued.

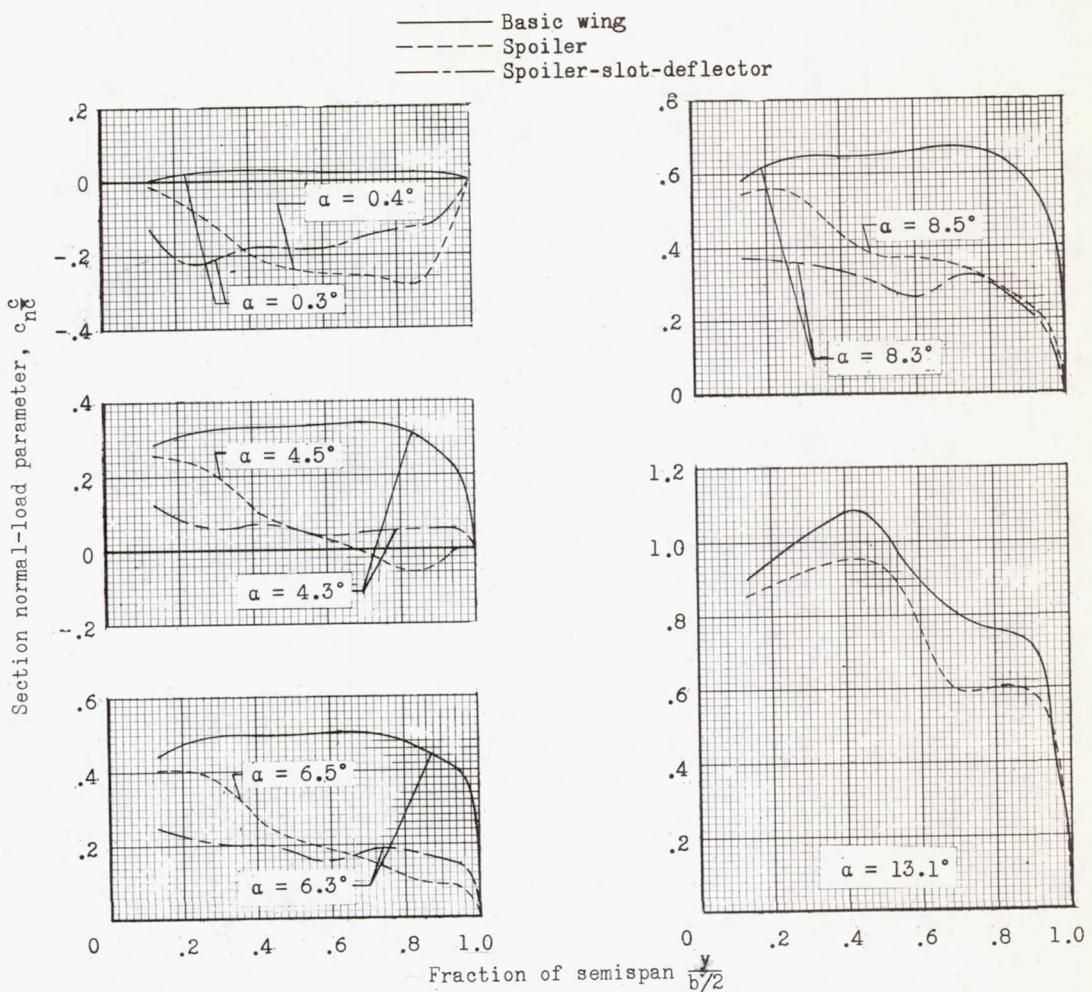
(g) $M = 1.03$.

Figure 11.- Concluded.

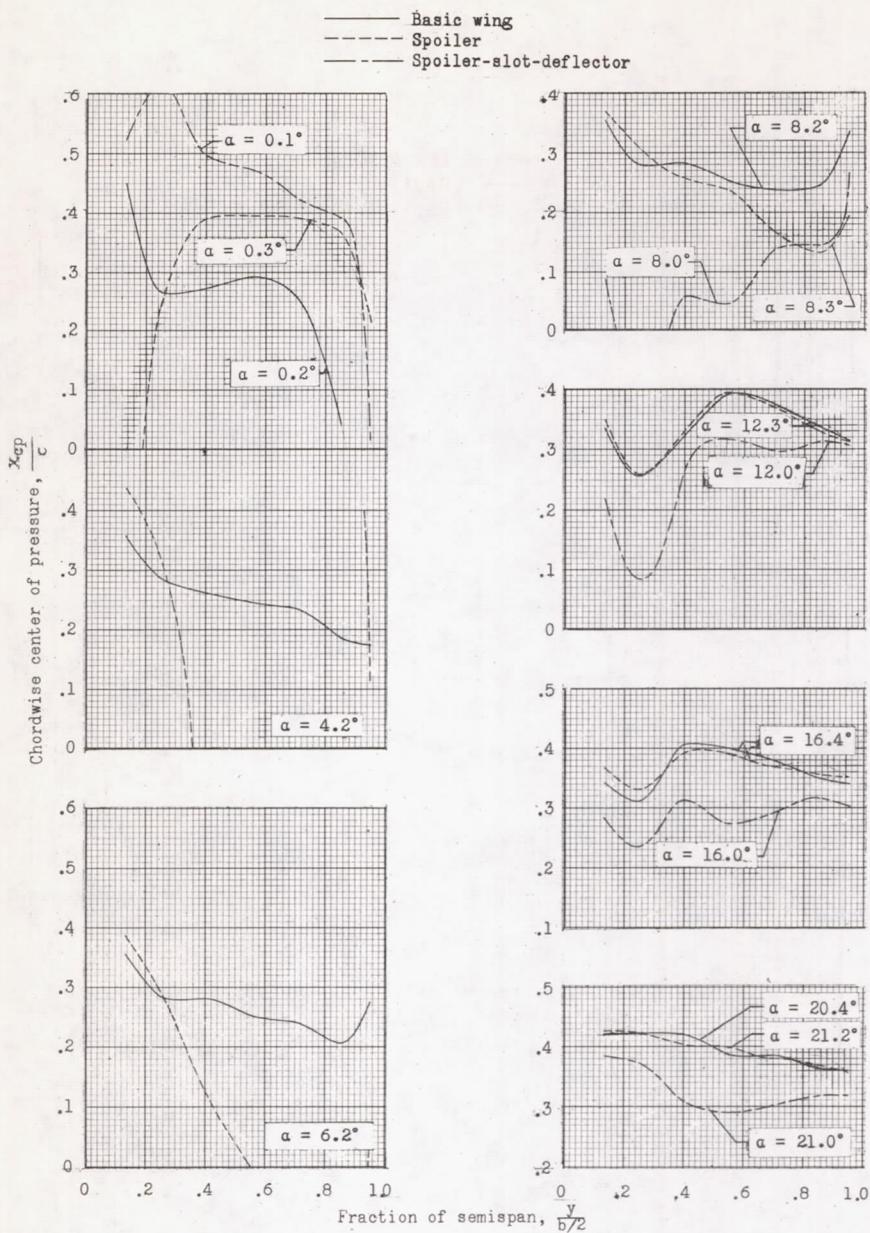
(a) $M = 0.60$.

Figure 12.- Wing section center of pressure for the basic model, a spoiler aileron configuration, and a spoiler-slot-deflector aileron configuration.

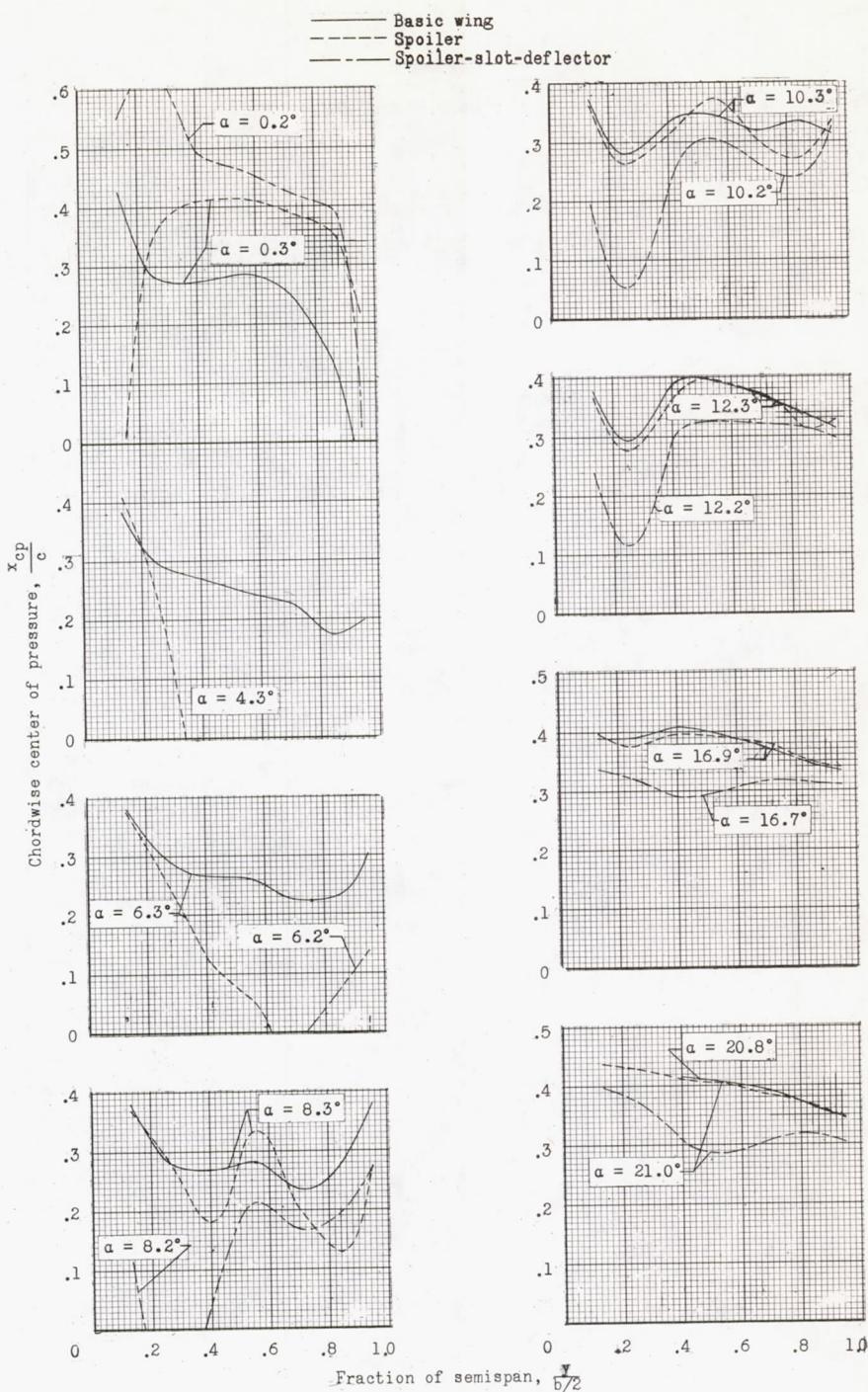
(b) $M = 0.80.$

Figure 12.- Continued.

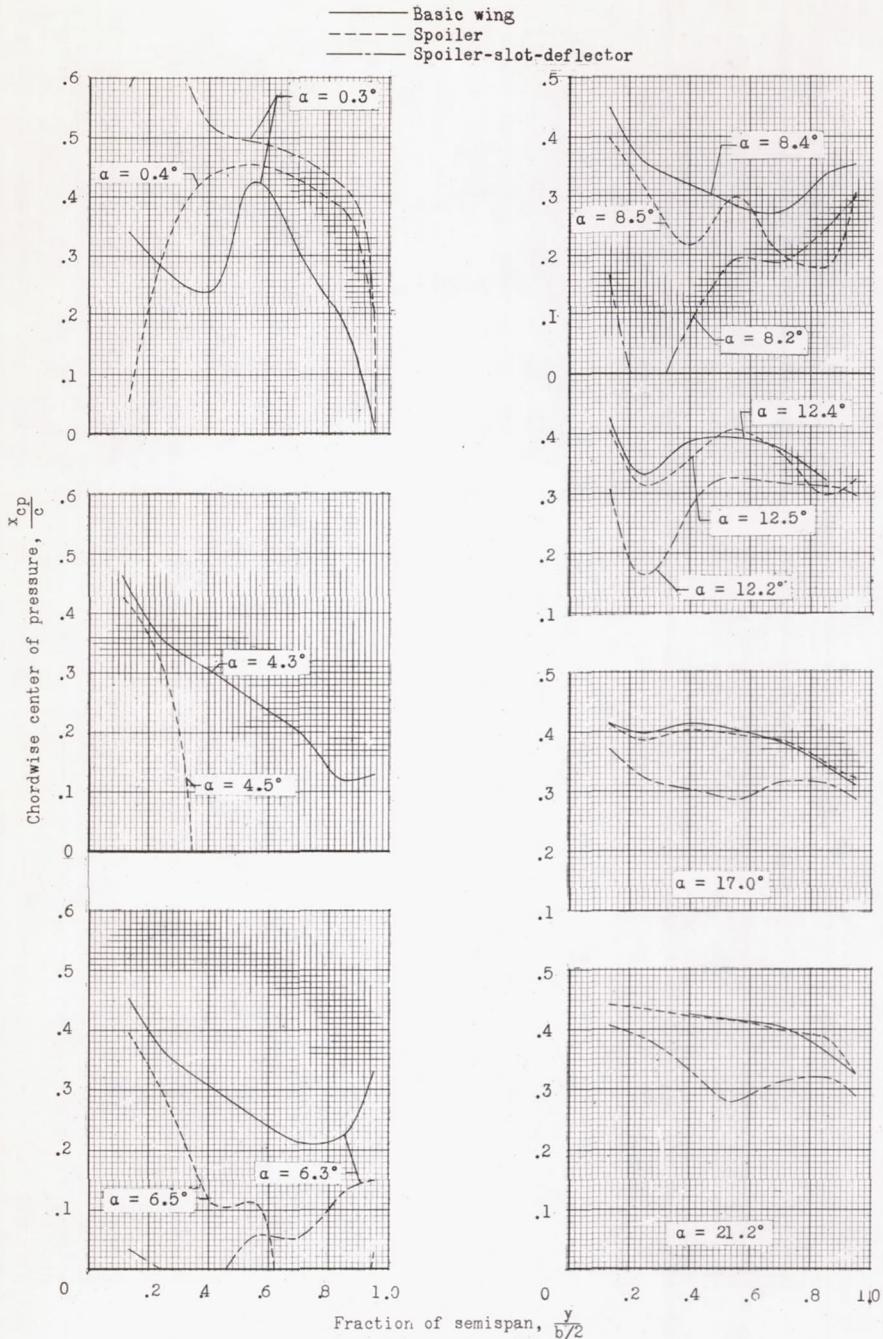
(c) $M = 0.90.$

Figure 12.- Continued.

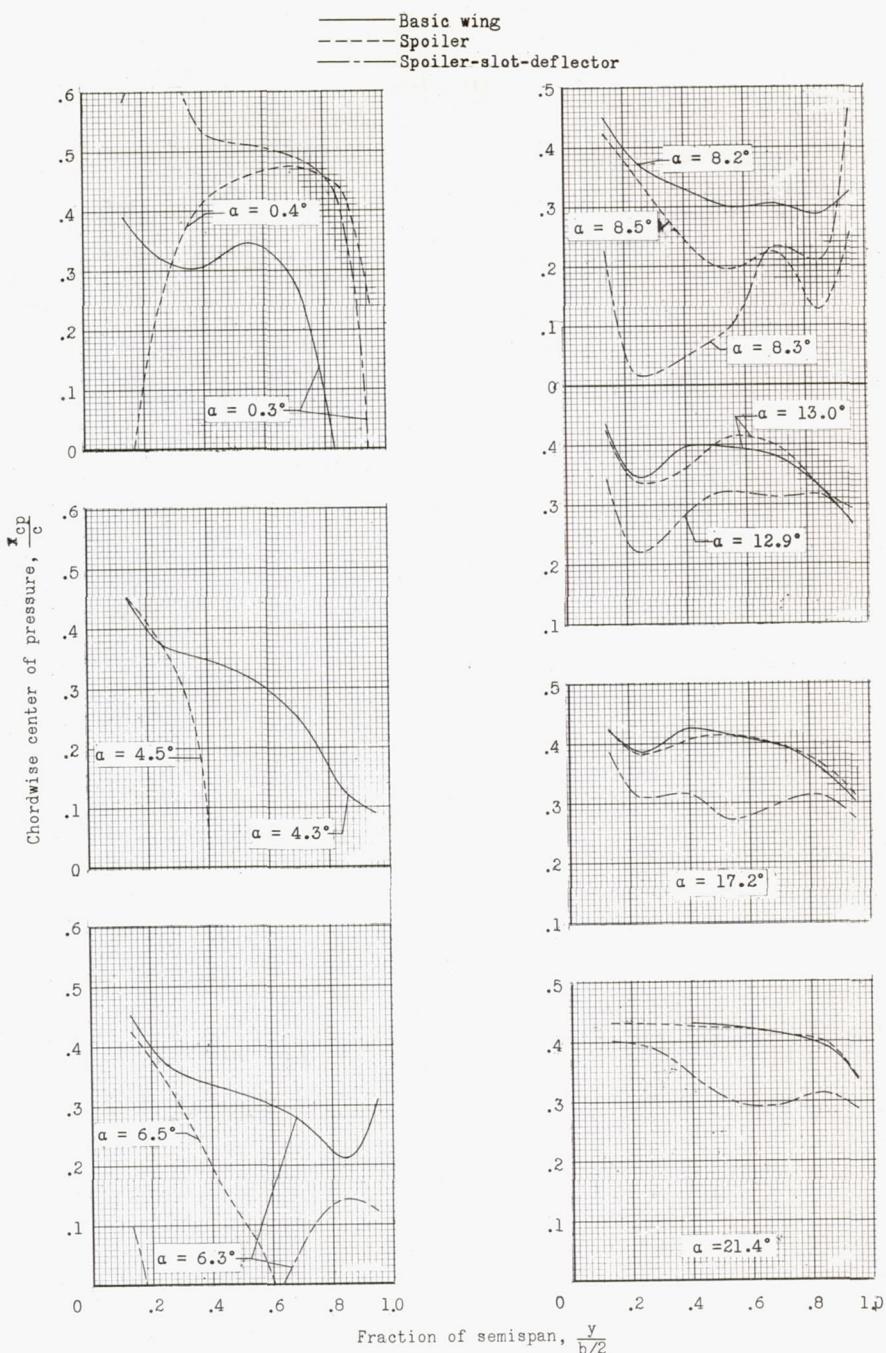
(d) $M = 0.94.$

Figure 12.- Continued.

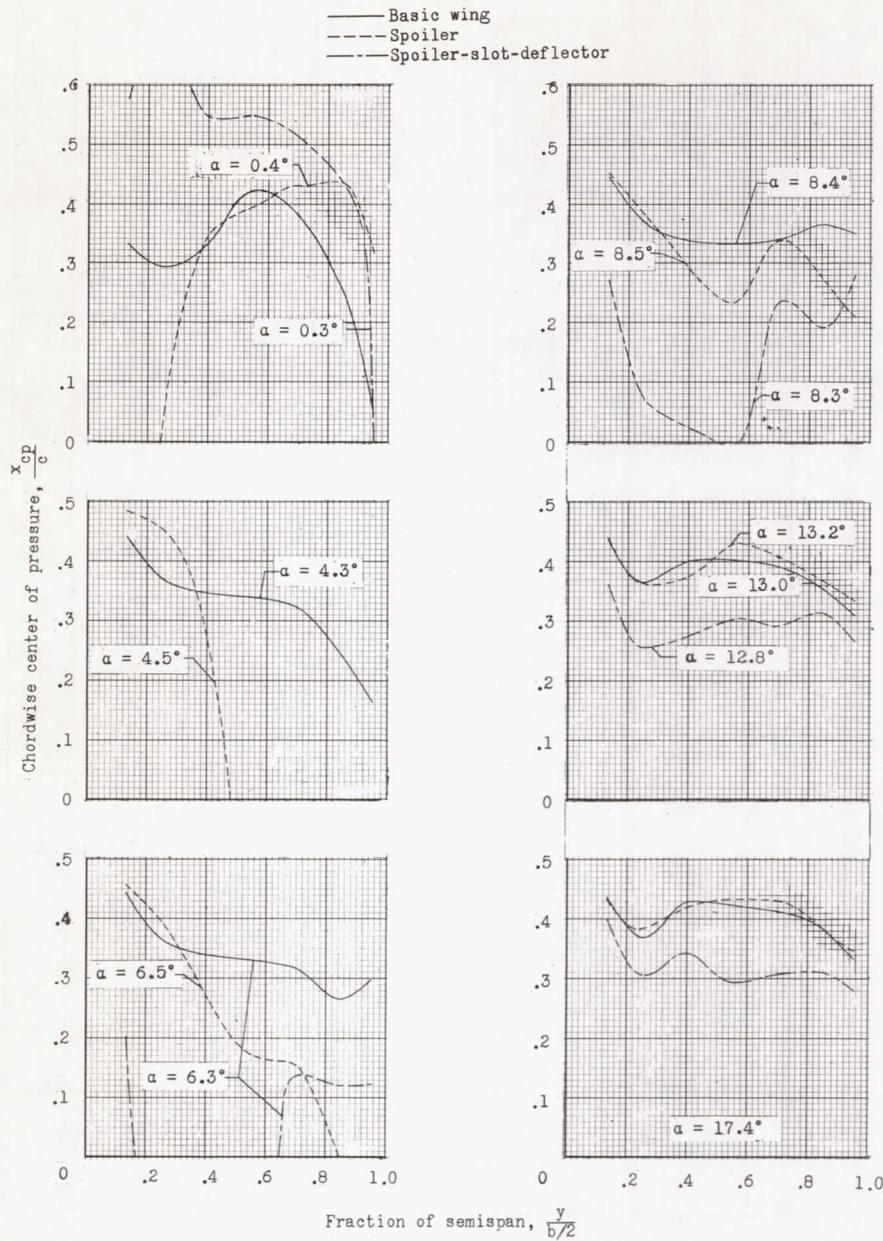
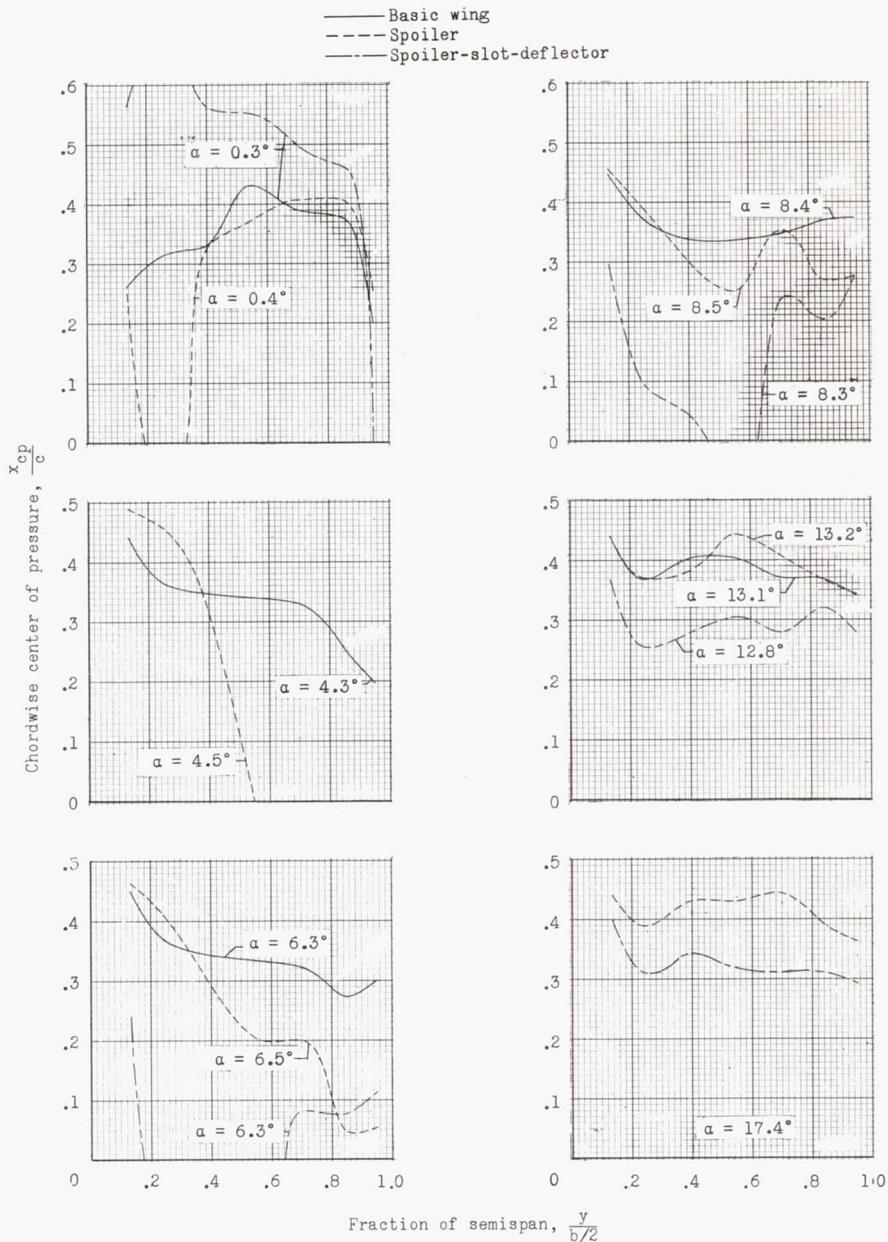
(e) $M = 0.98.$

Figure 12.- Continued.



(f) M = 1.00.

Figure 12.- Continued.

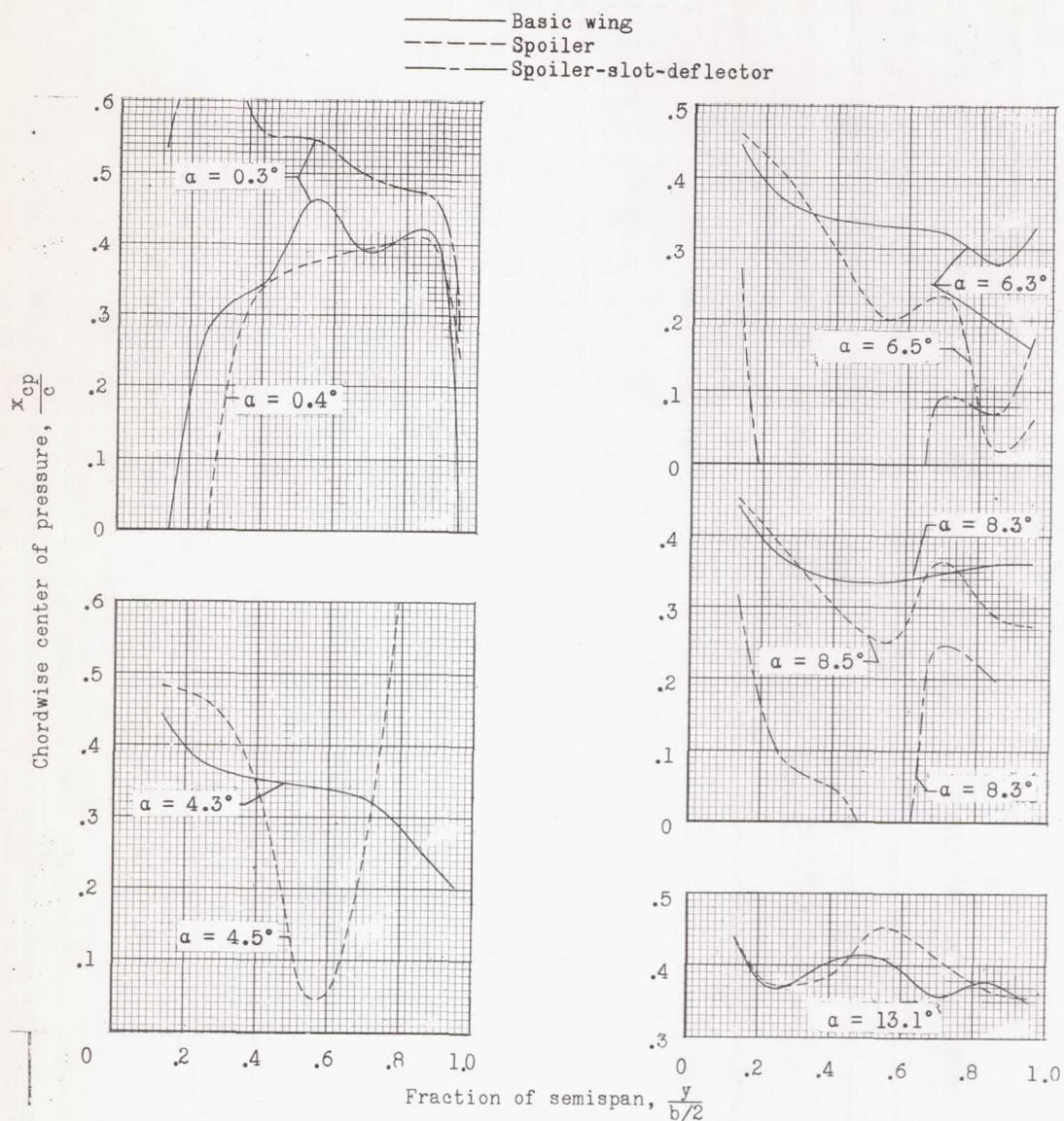
(g) $M = 1.03$.

Figure 12.- Concluded.

CONFIDENTIAL

NACA RM 157J11

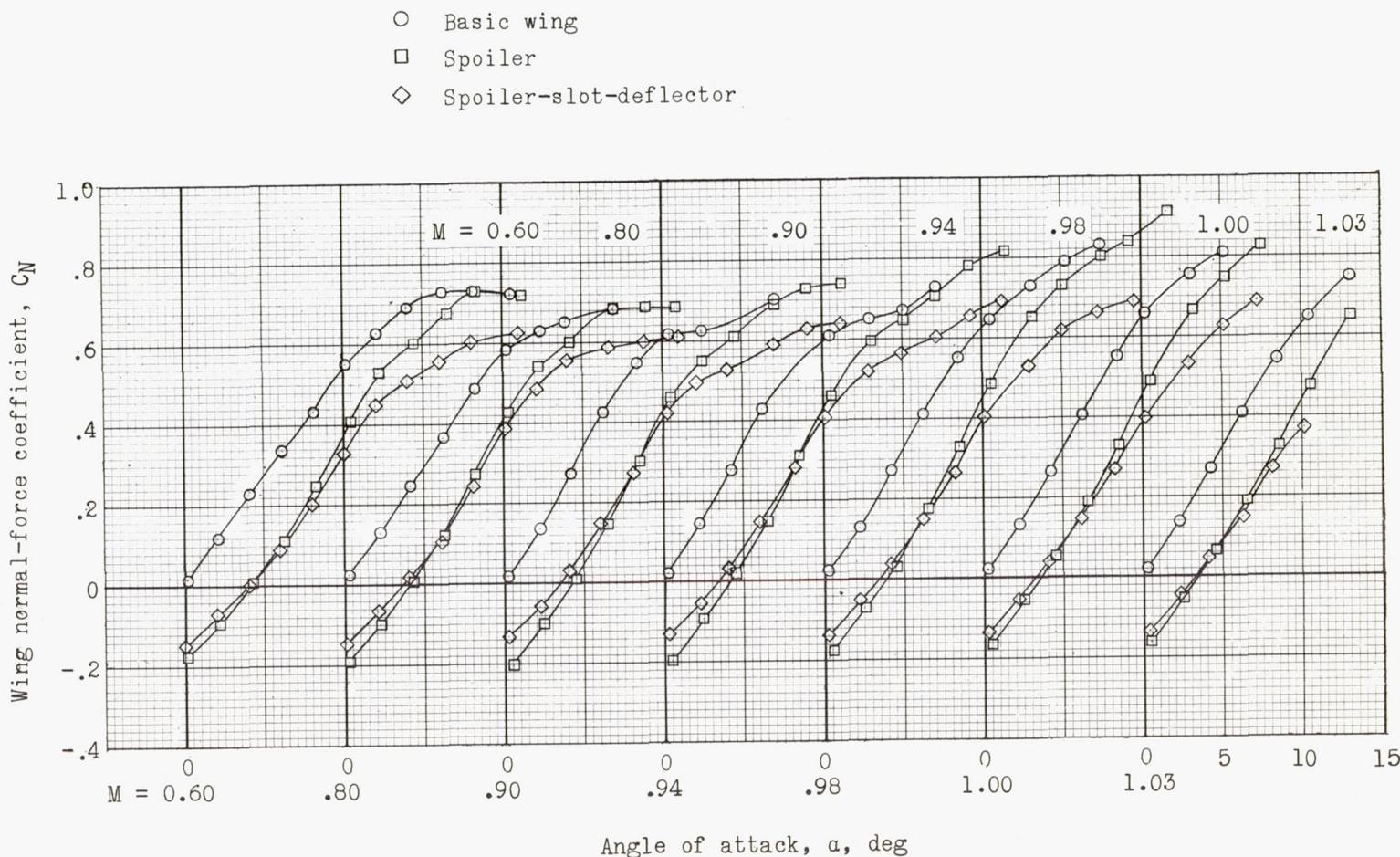


Figure 13.- Wing normal-force characteristics for the basic model, a spoiler aileron configuration, and a spoiler-slot-deflector aileron configuration.

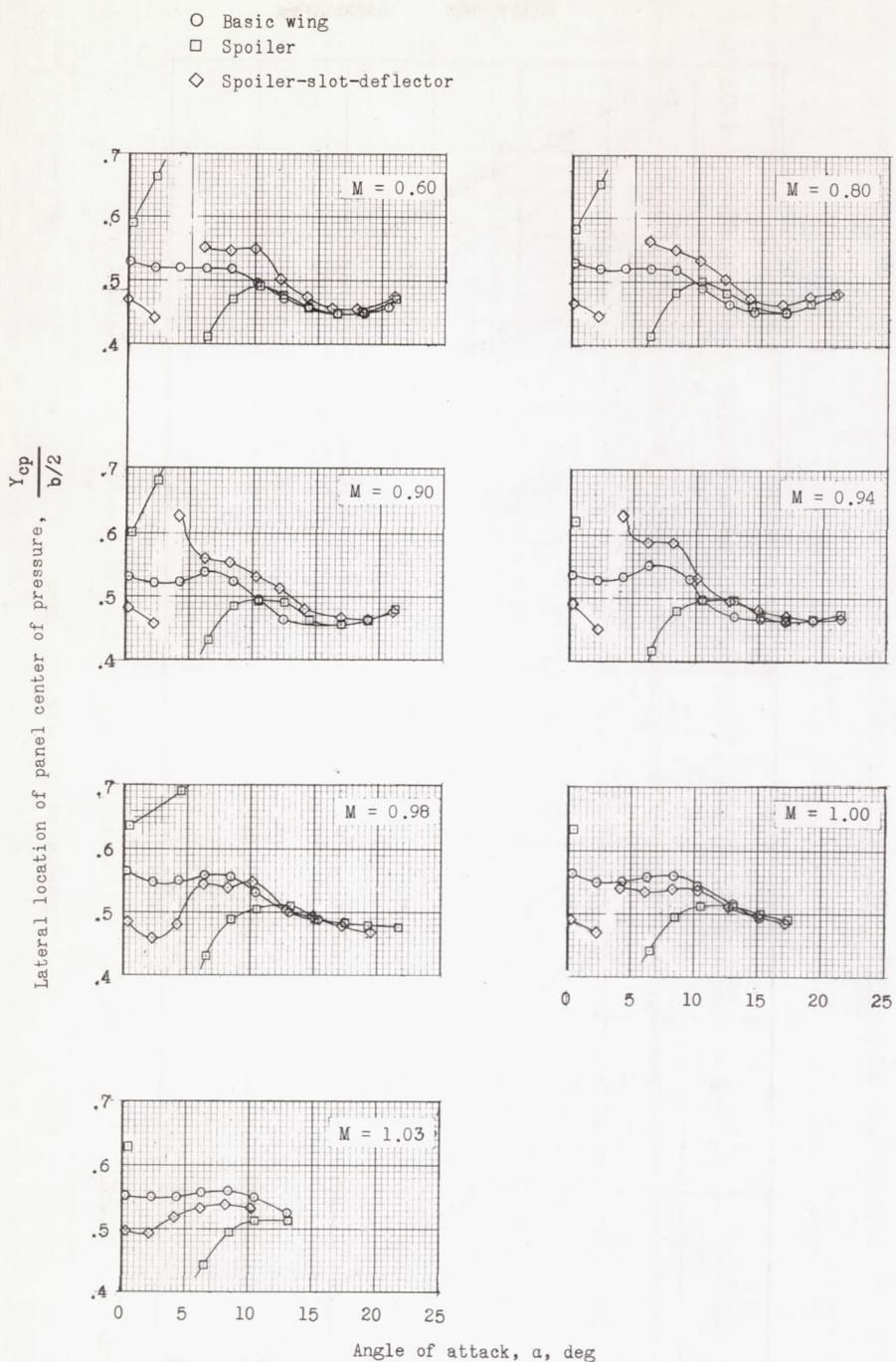


Figure 14.- Lateral position of wing center of pressure for the basic model, a spoiler aileron configuration, and a spoiler-slot-deflector aileron configuration.

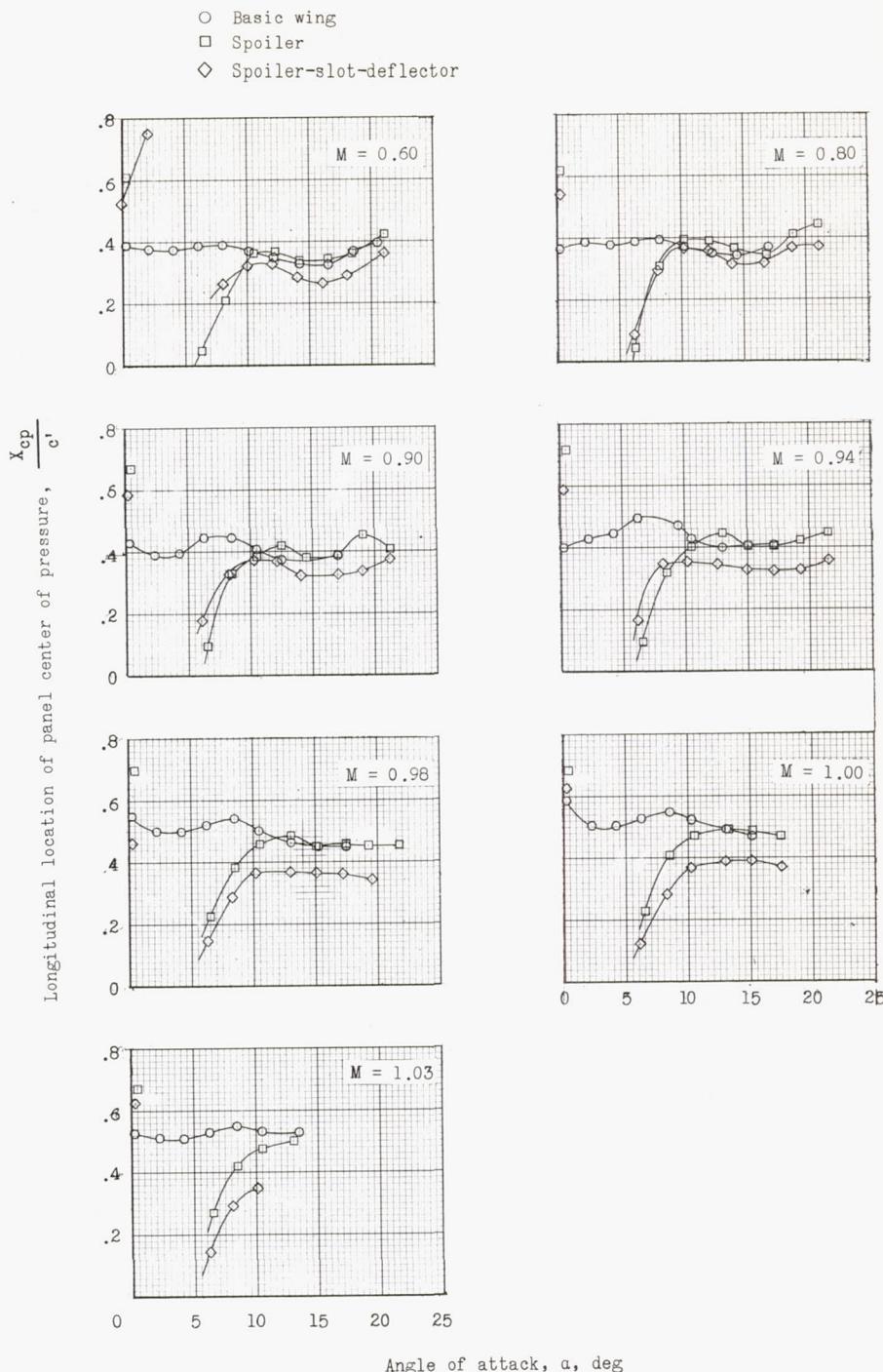


Figure 15.- Longitudinal position of wing center of pressure for the basic model, a spoiler aileron configuration, and a spoiler-slot-deflector aileron configuration.

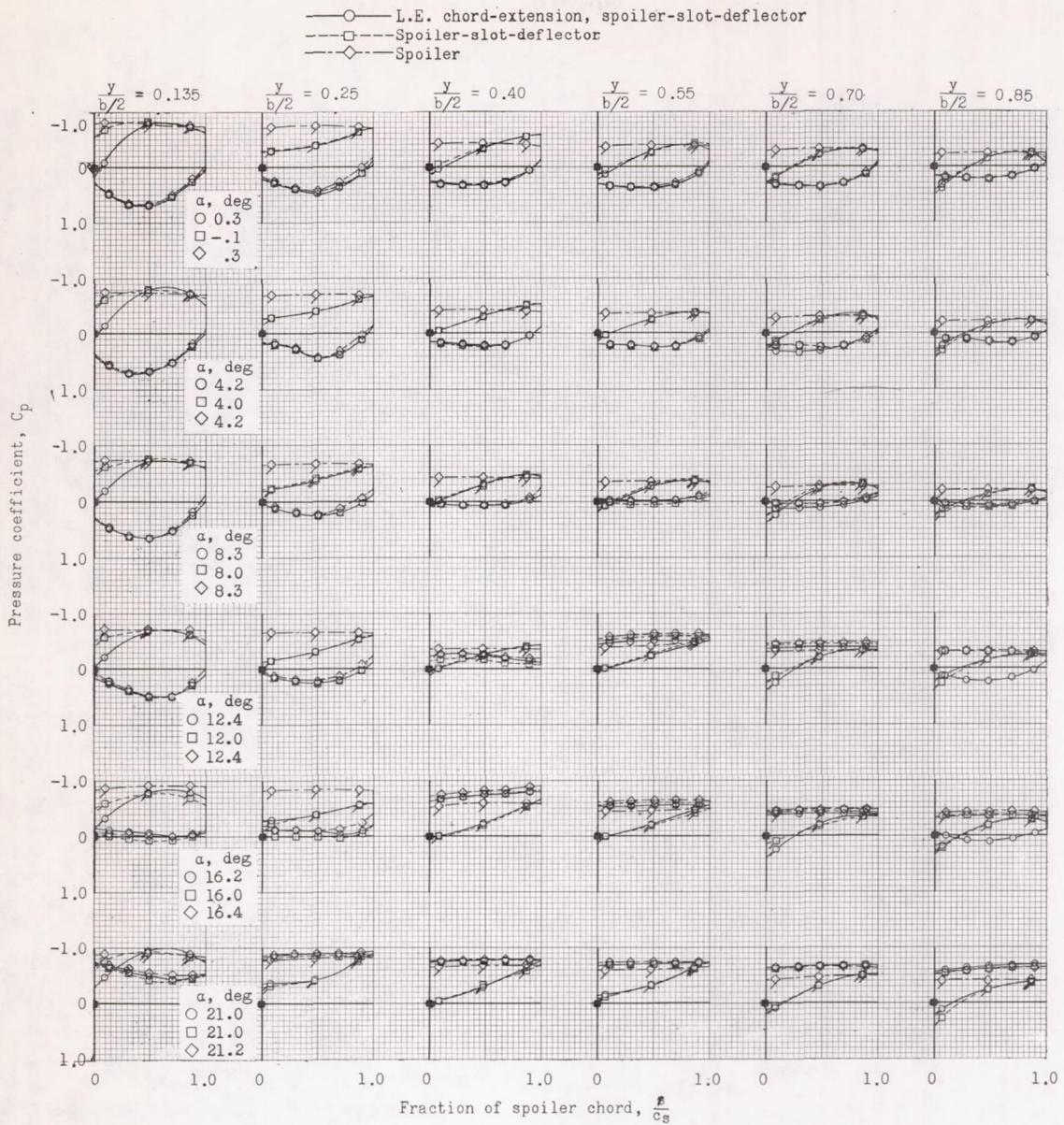
(a) $M = 0.60.$

Figure 16.- Section pressure distributions for a spoiler aileron and spoilers of basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations. (Flagged symbols indicate rear surface.)

(b) $M = 0.90$.

Figure 16.- Continued.

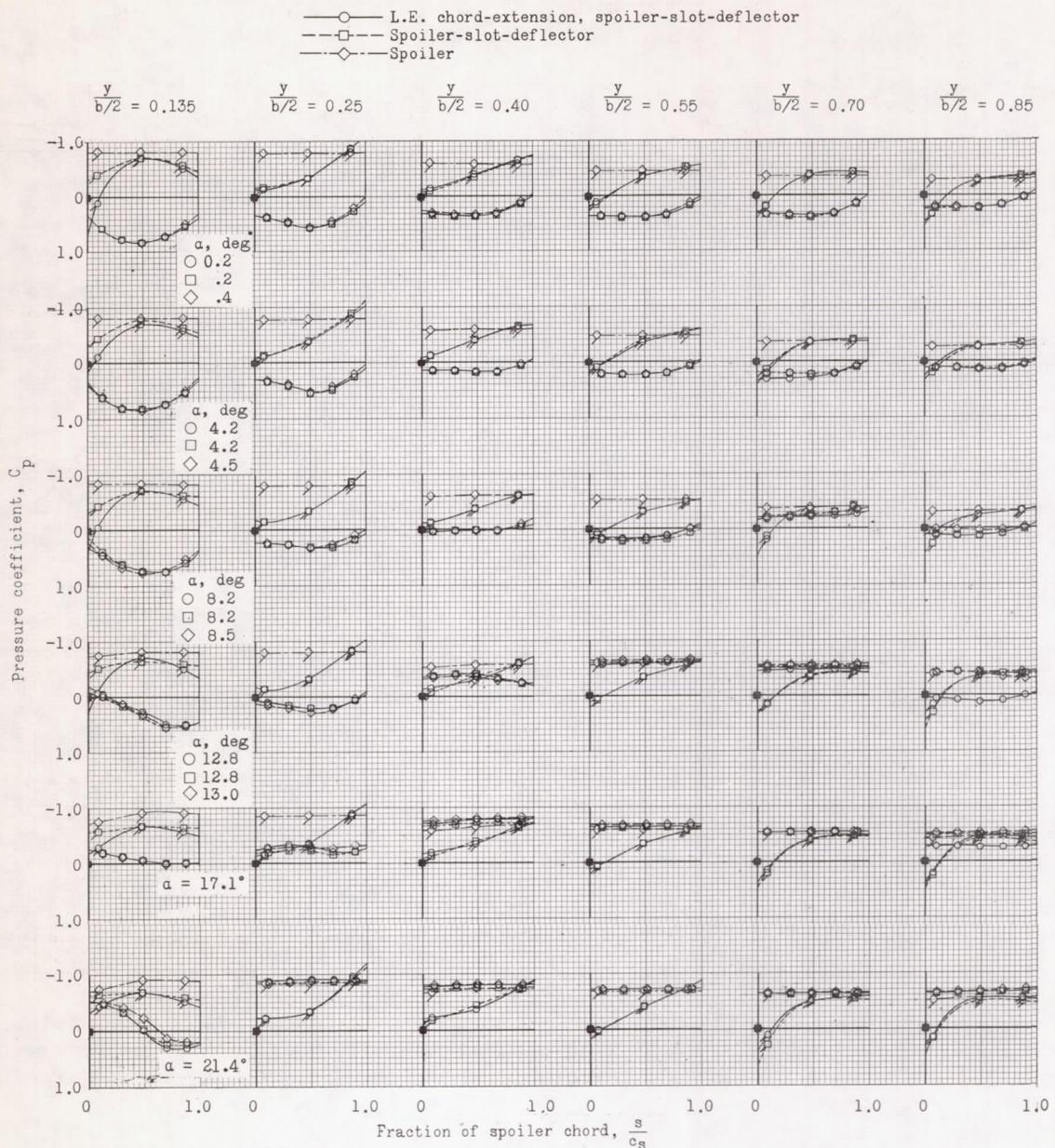
(c) $M = 0.94$.

Figure 16.- Continued.

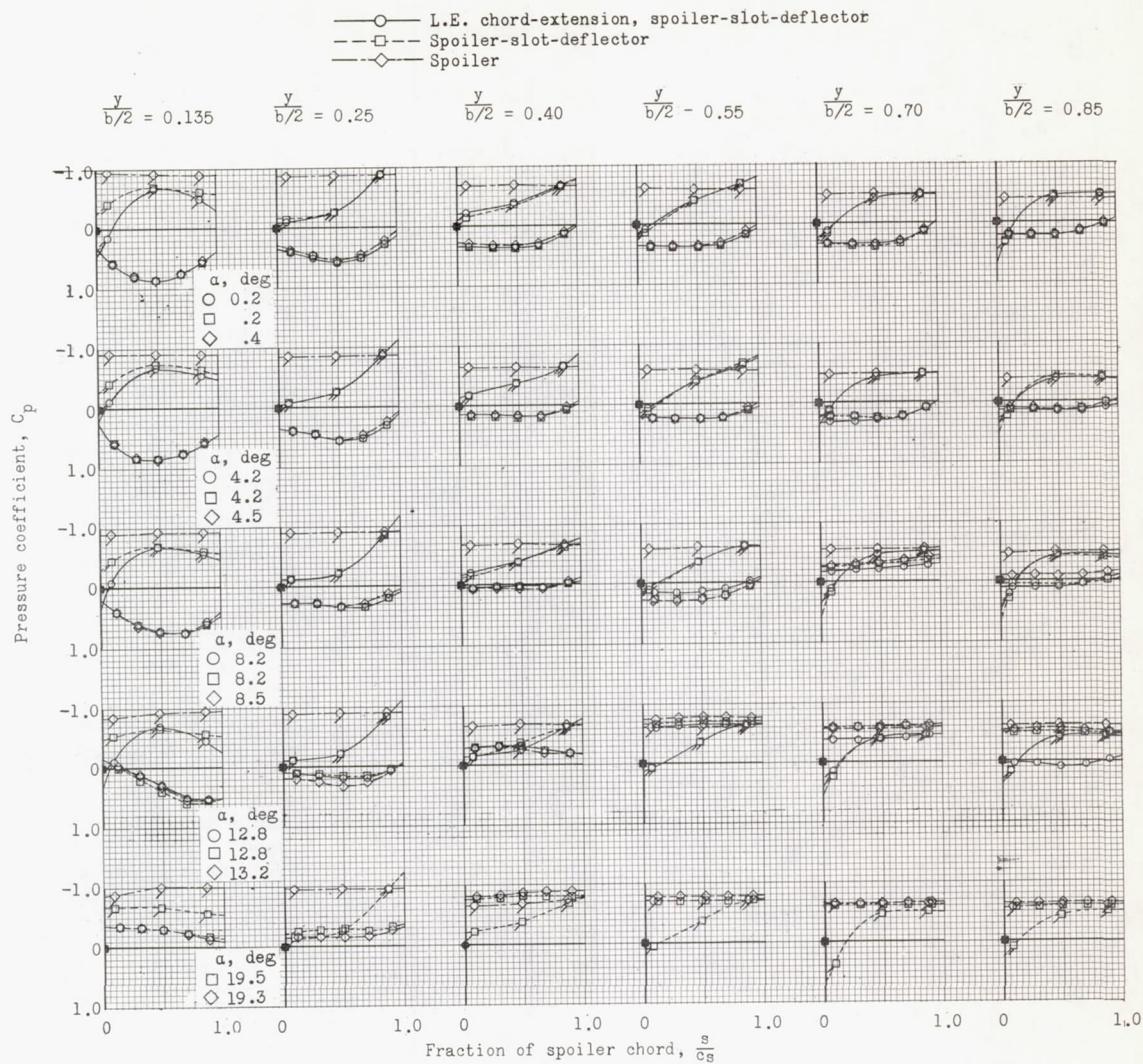
(d) $M = 0.98$.

Figure 16.- Continued.

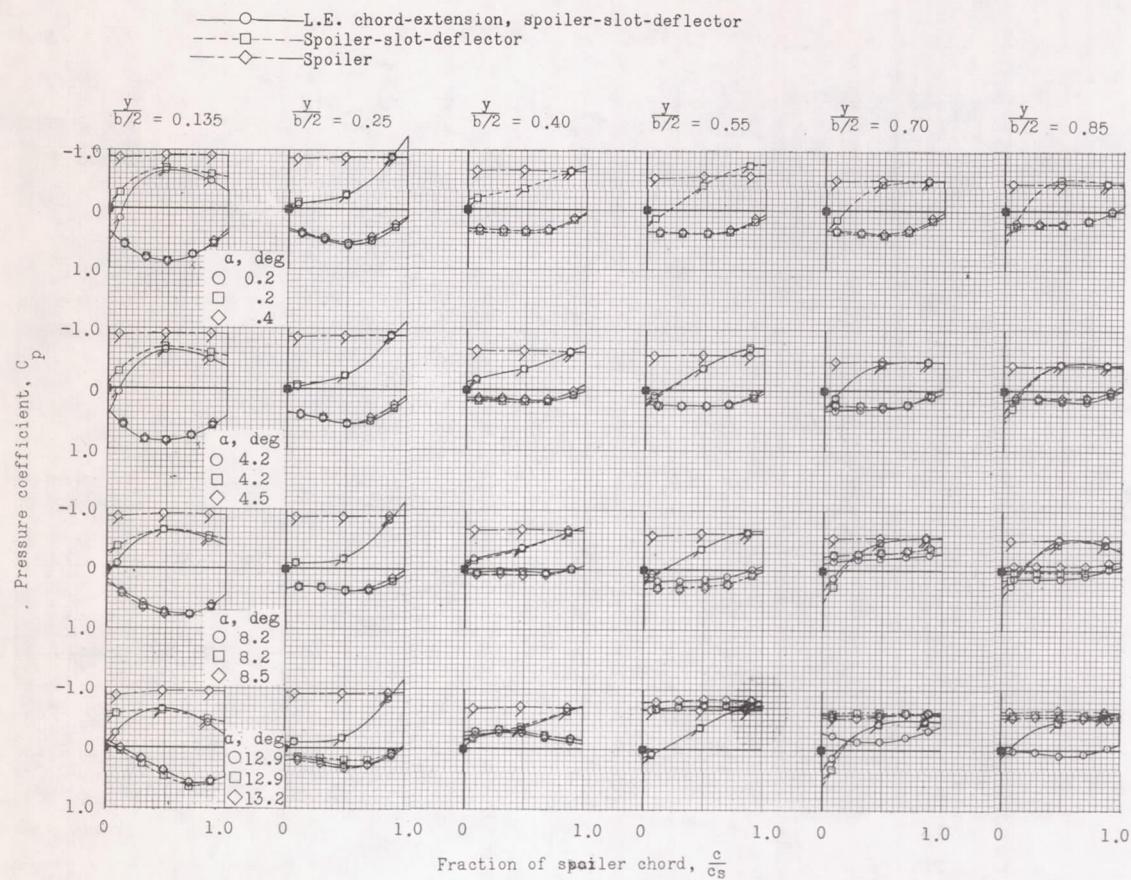
(e) $M = 1.00.$

Figure 16.- Concluded.

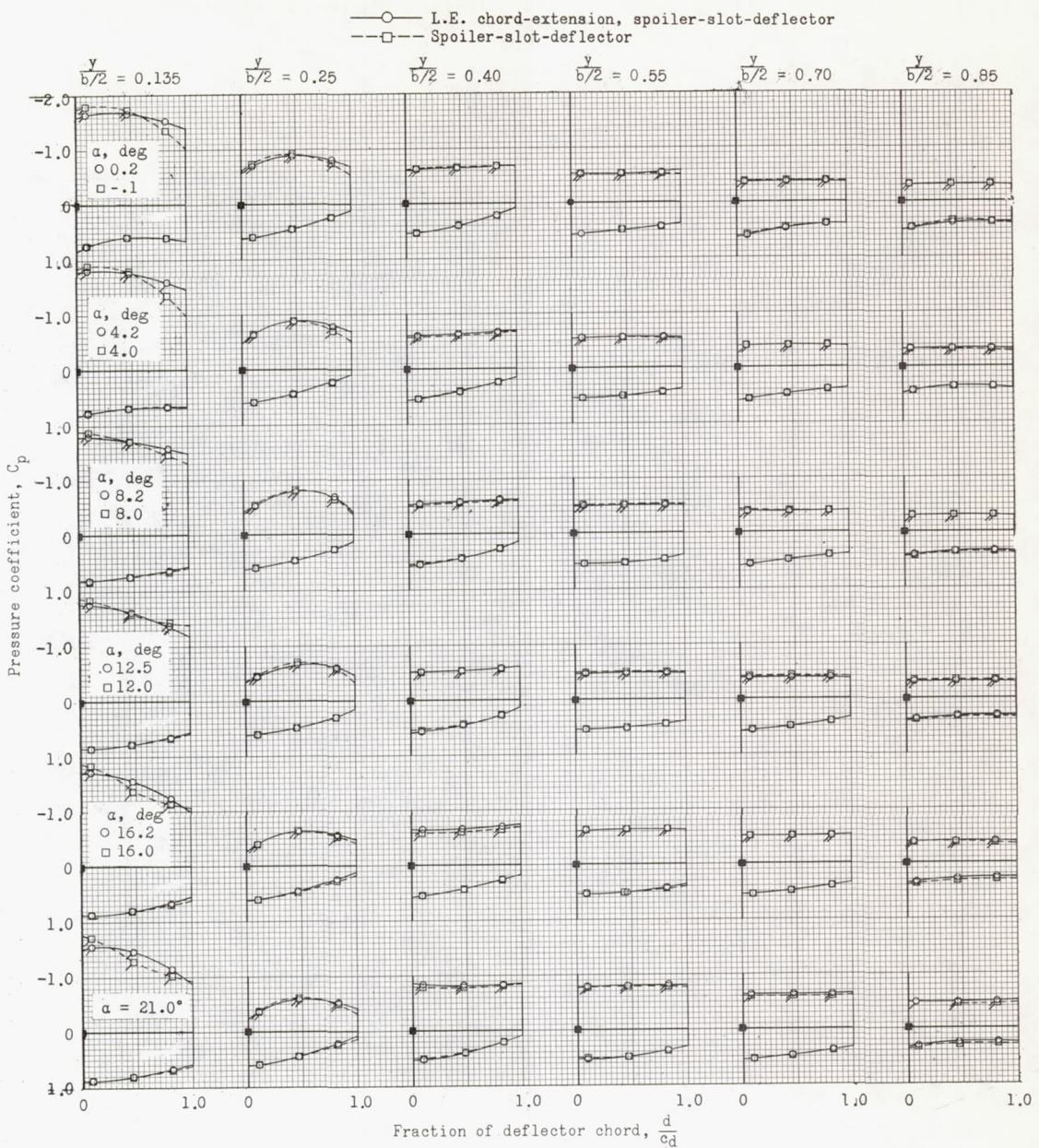
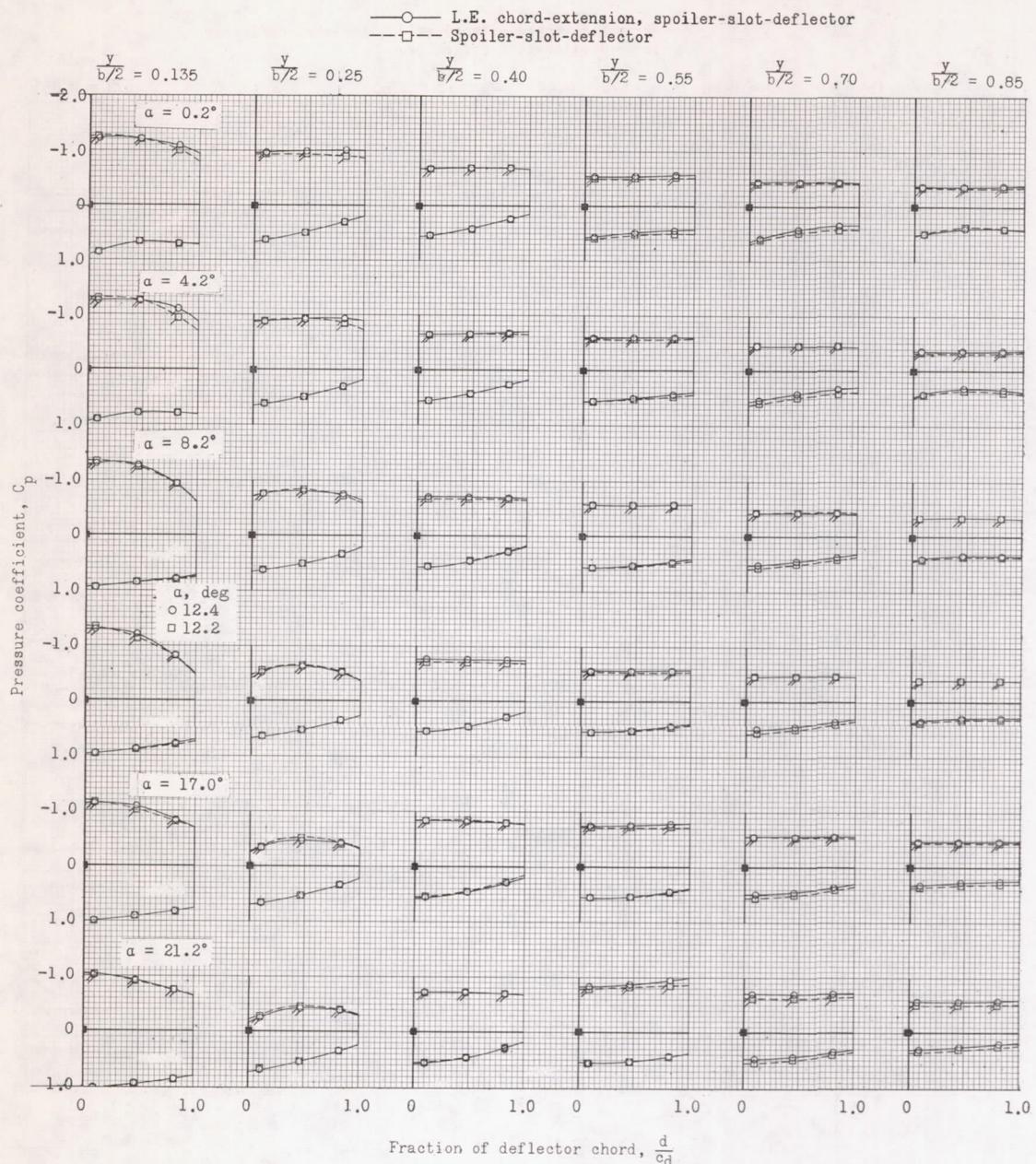
(a) $M = 0.60$.

Figure 17.- Section pressure distribution for deflectors of the basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations. (Flagged symbols indicate rear surface.)



(b) M = 0.90.

Figure 17.- Continued.

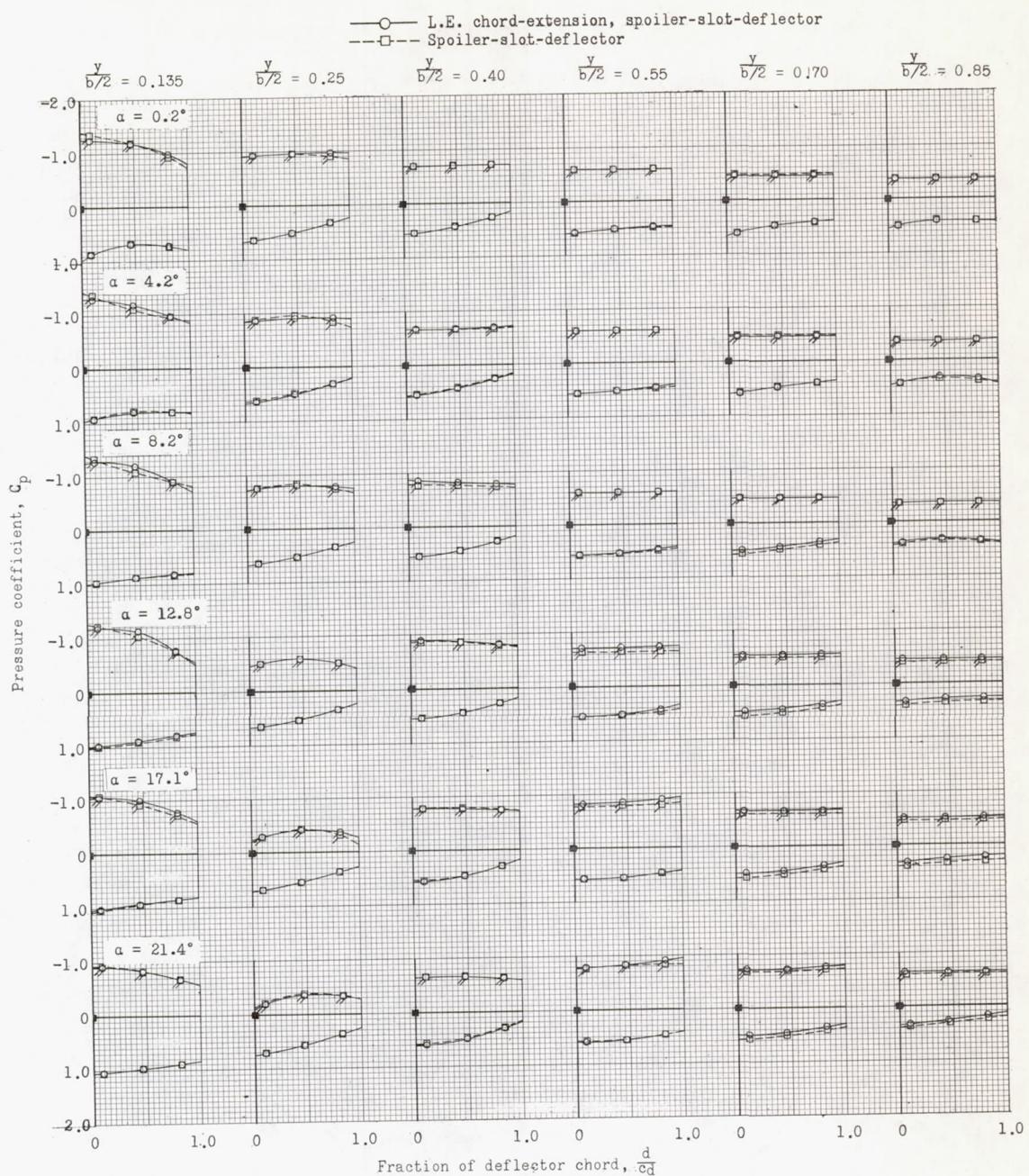
(c) $M = 0.94$.

Figure 17.- Continued.

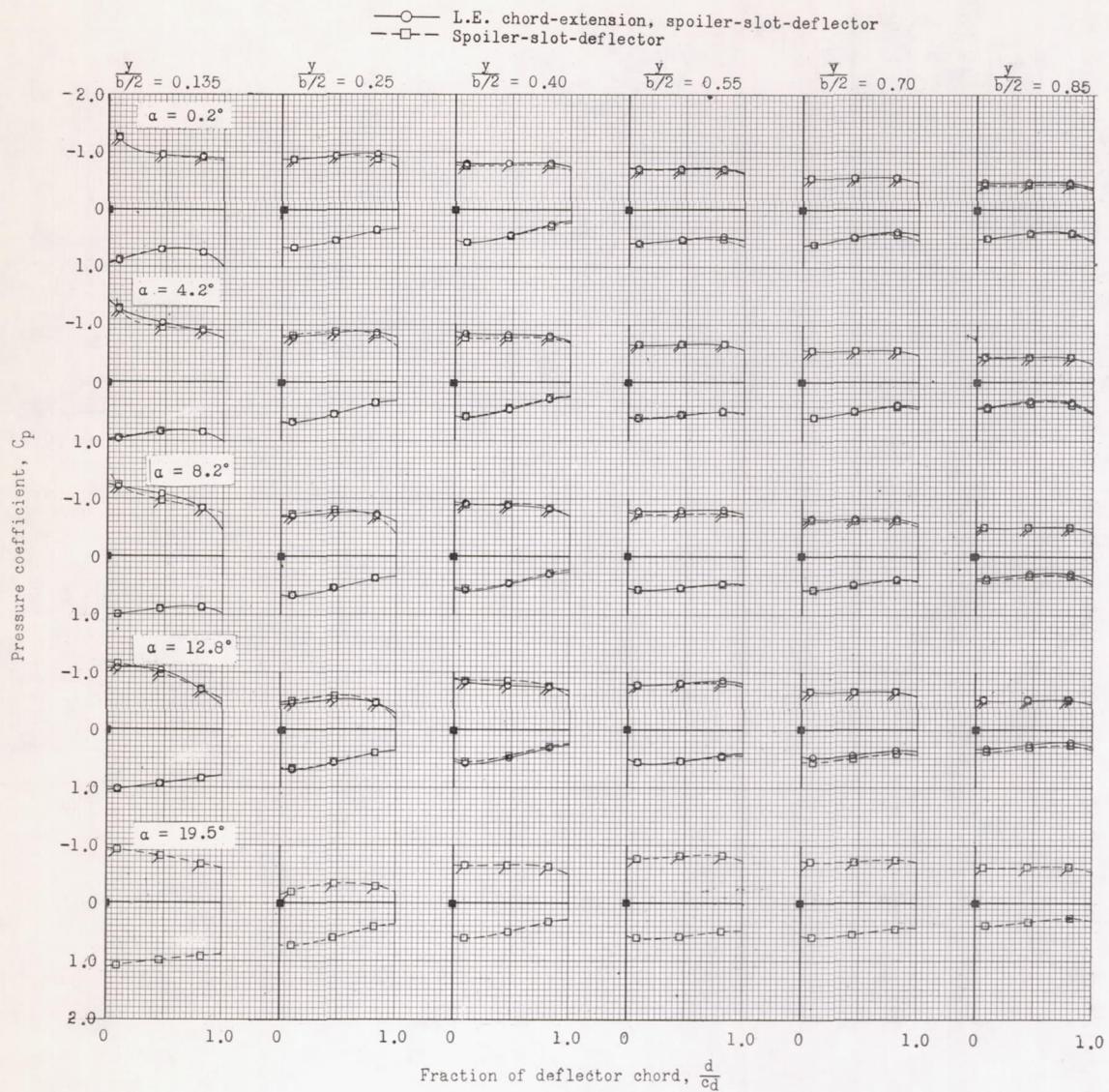
(d) $M = 0.98.$

Figure 17.- Continued.

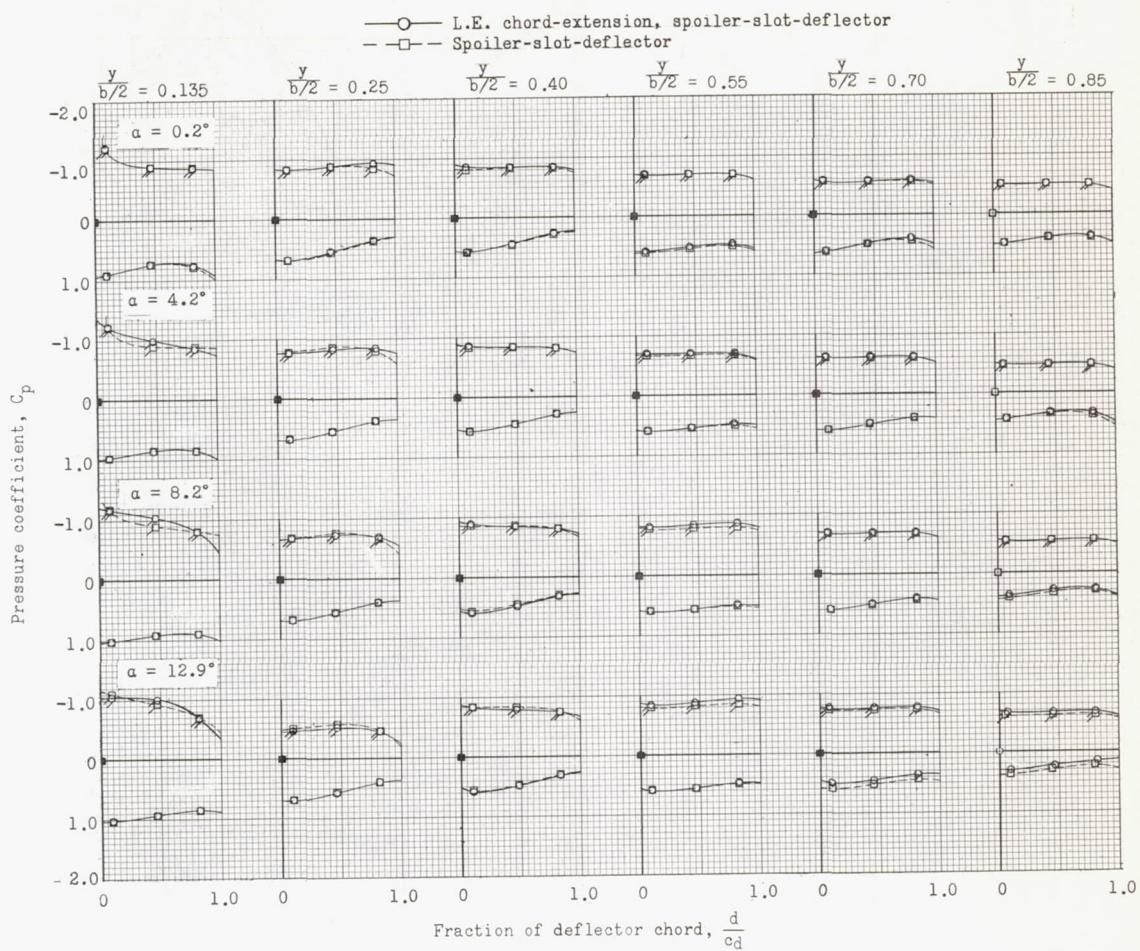
(e) $M = 1.00$.

Figure 17.- Concluded.

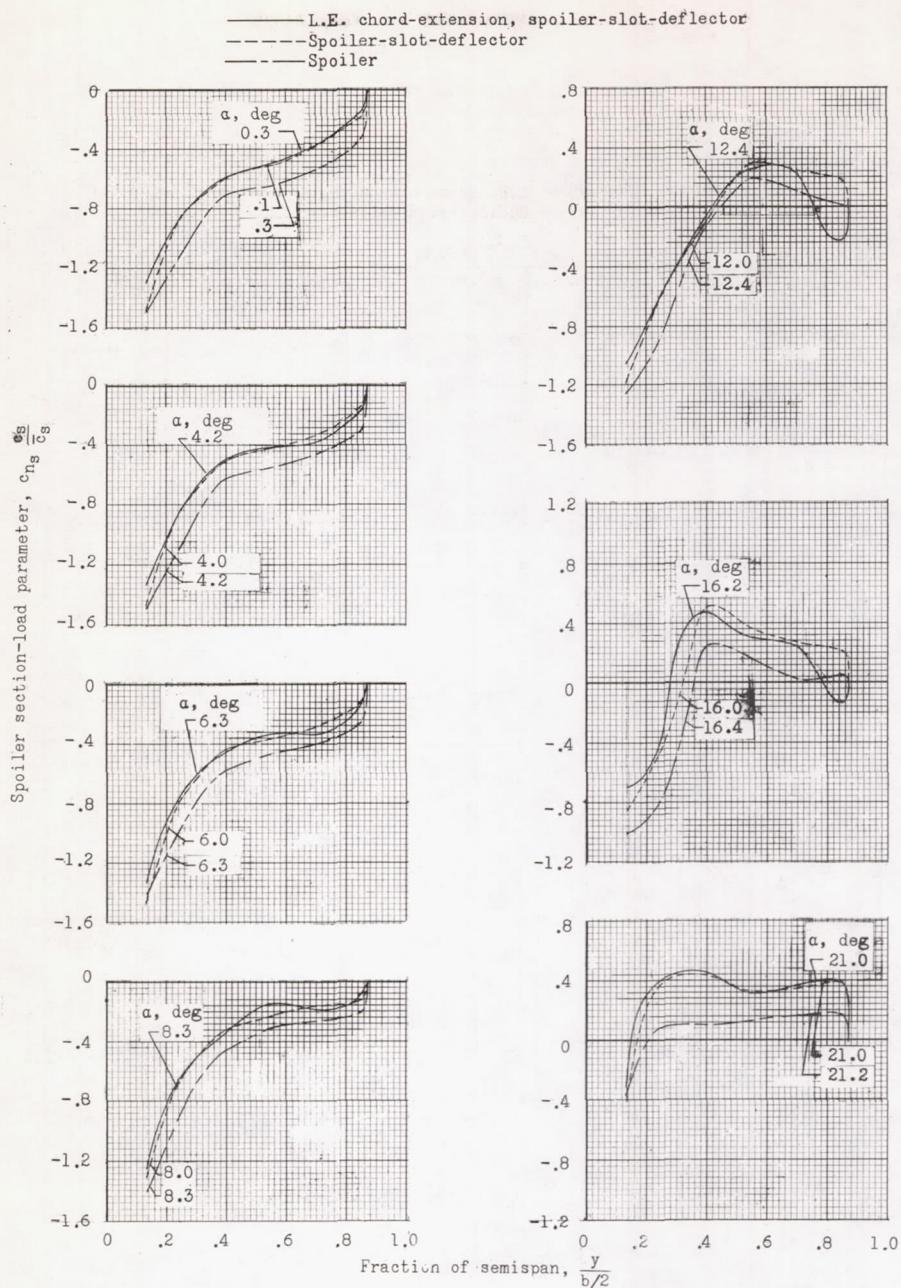
(a) $M = 0.60$.

Figure 18.- Span-load distributions for a spoiler aileron and spoilers of the basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations.

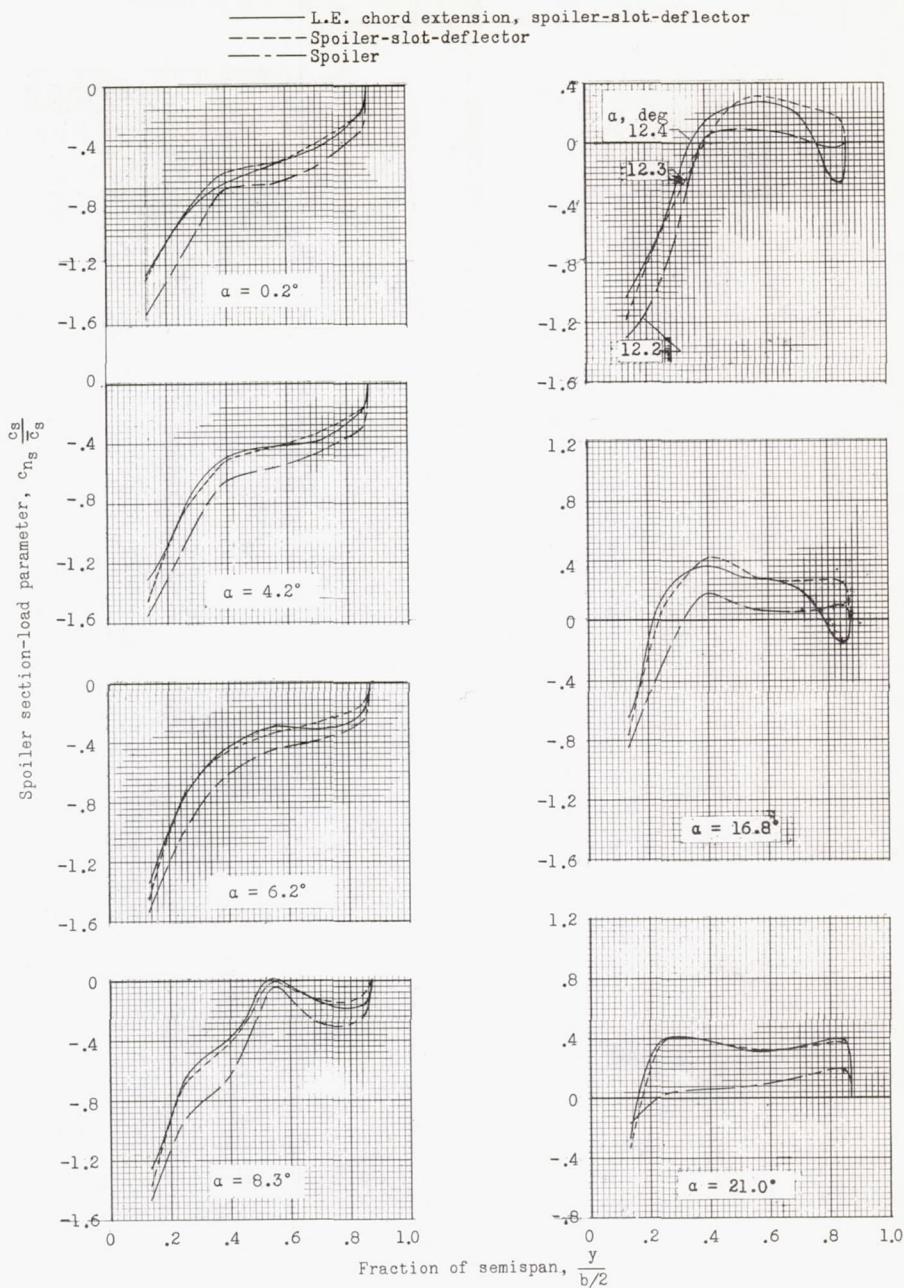


Figure 18.- Continued.

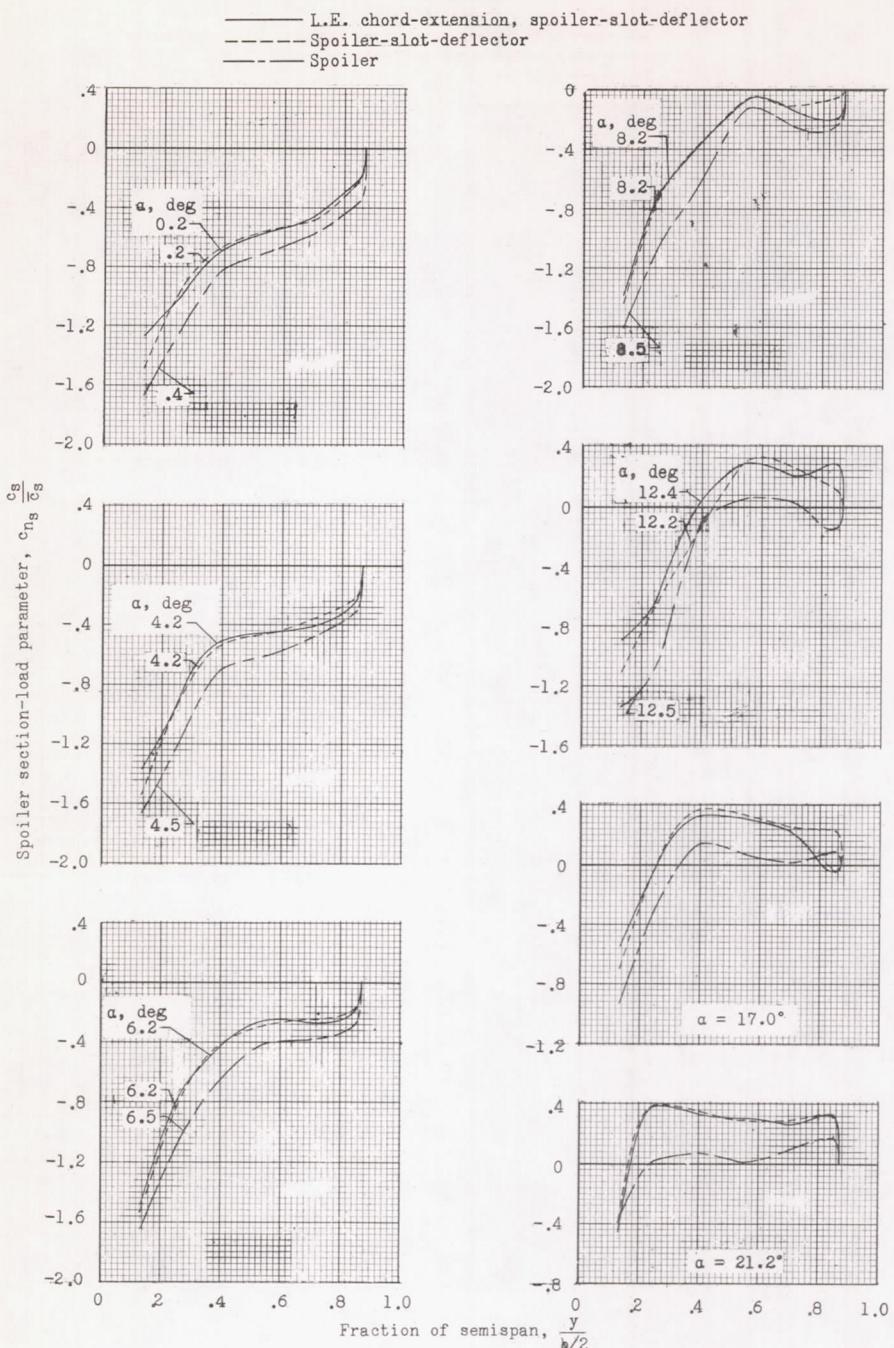
(c) $M = 0.90.$

Figure 18.- Continued.

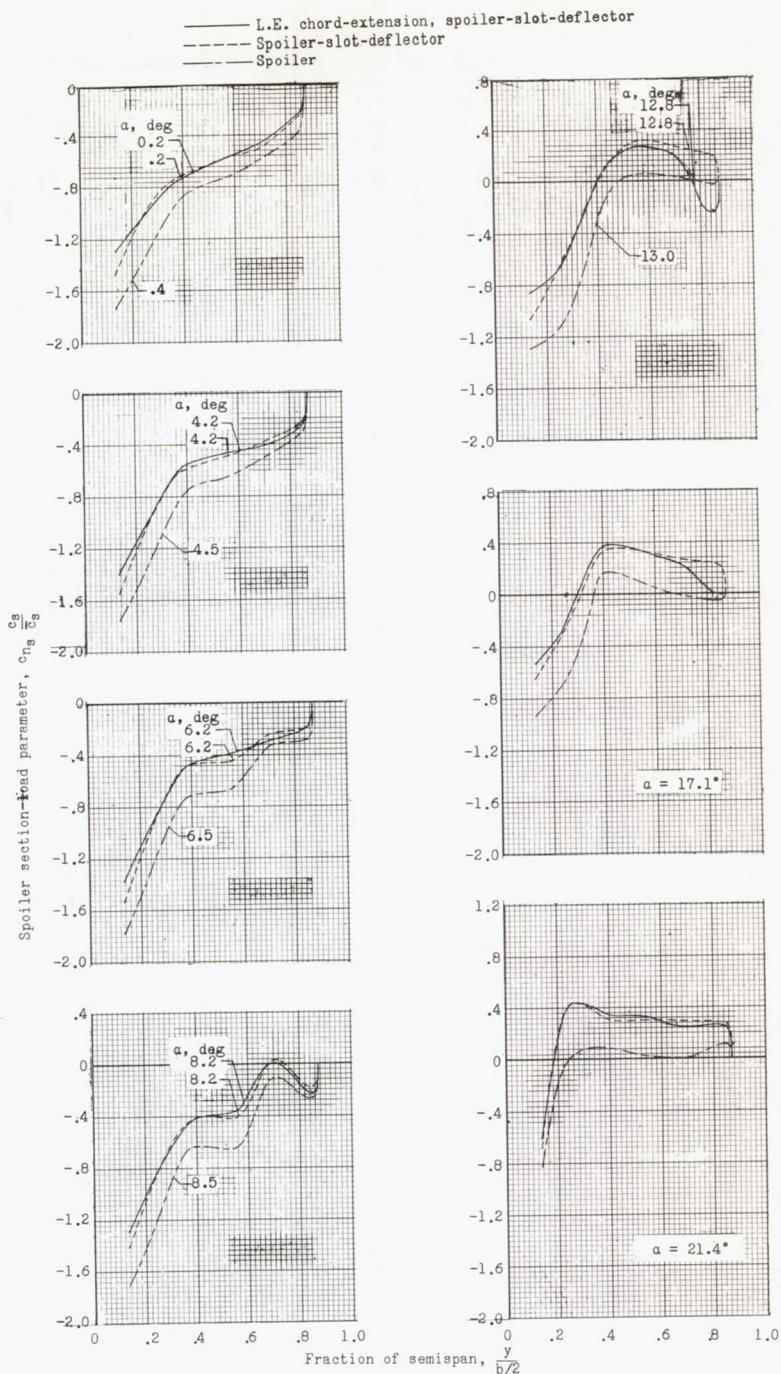
(d) $M = 0.94$.

Figure 18.- Continued.

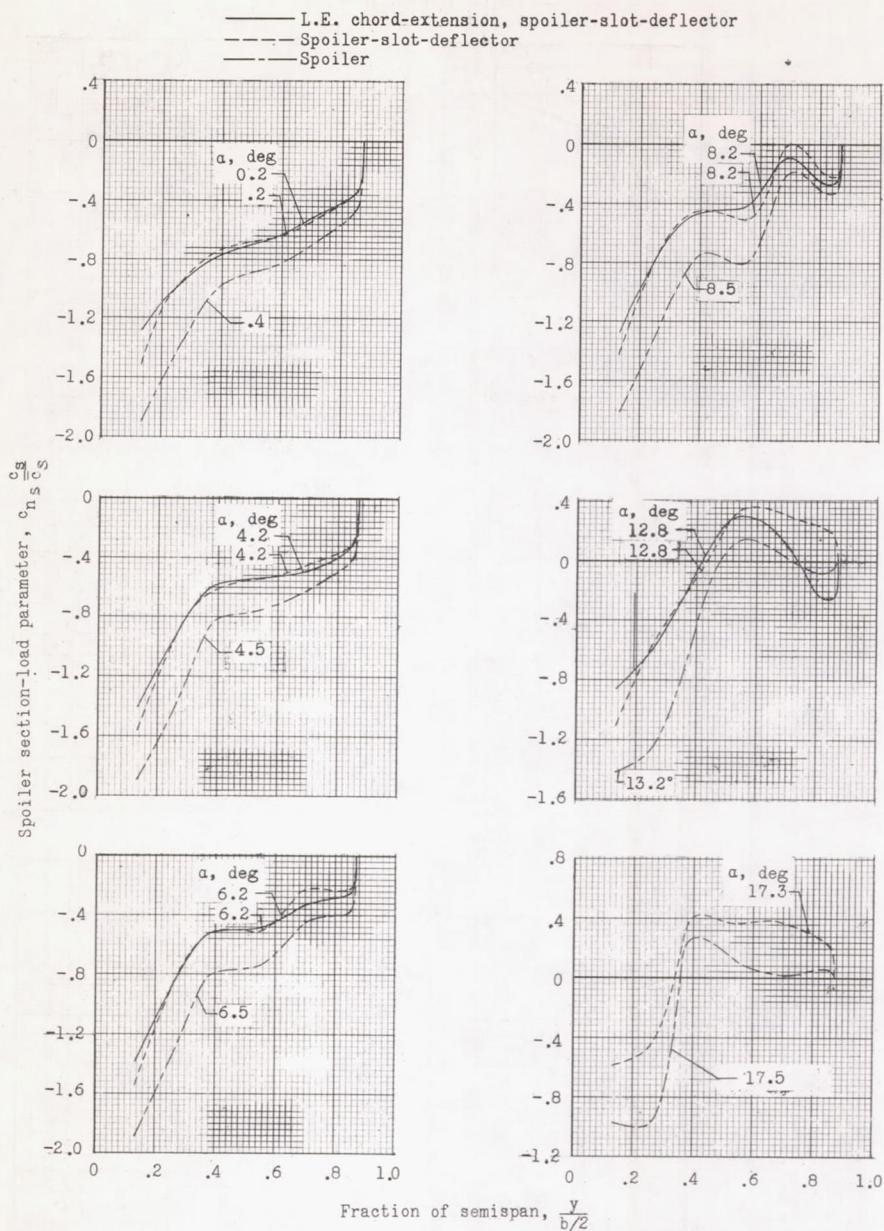
(e) $M = 0.98.$

Figure 18.- Continued.

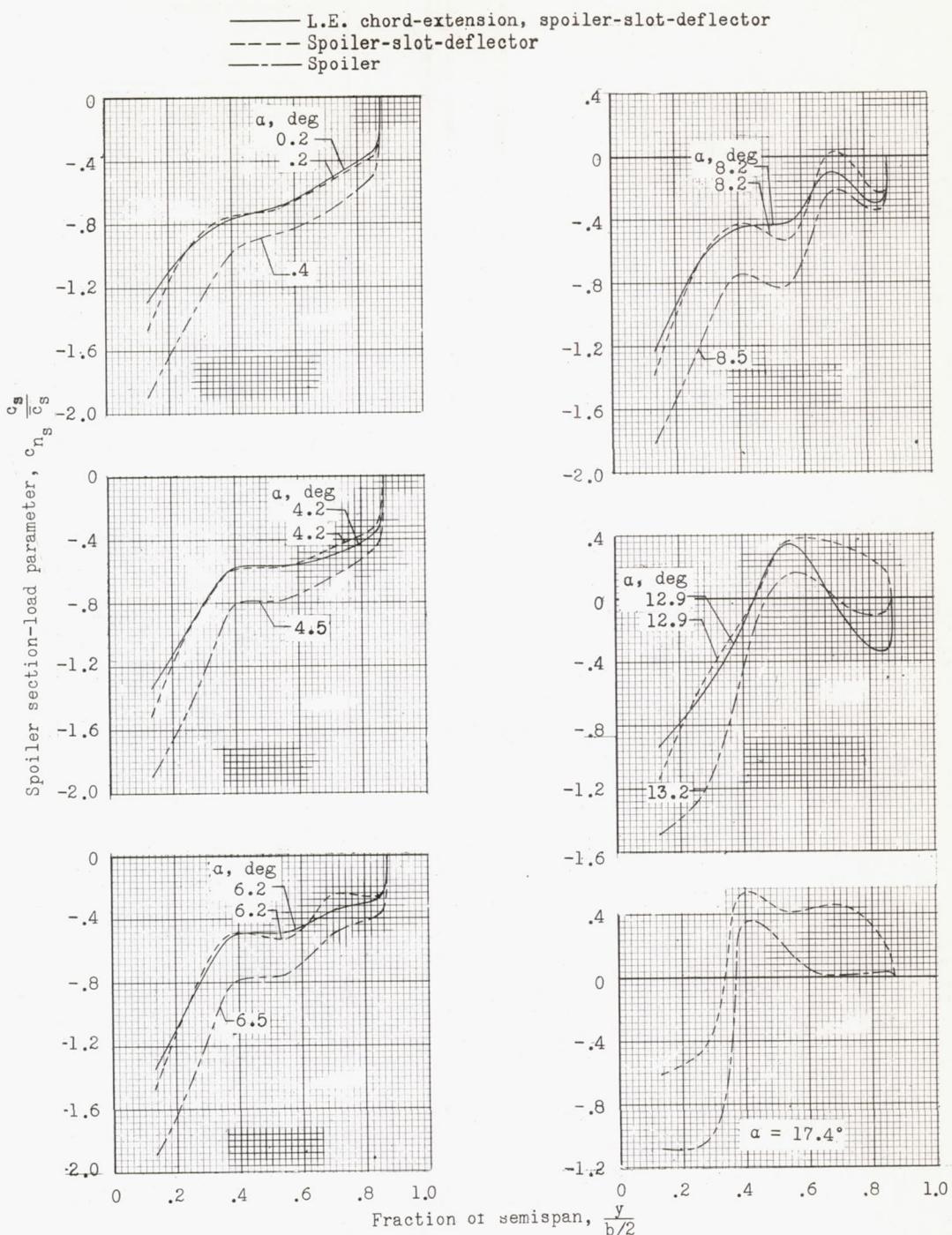
(f) $M = 1.00$.

Figure 18.- Continued.

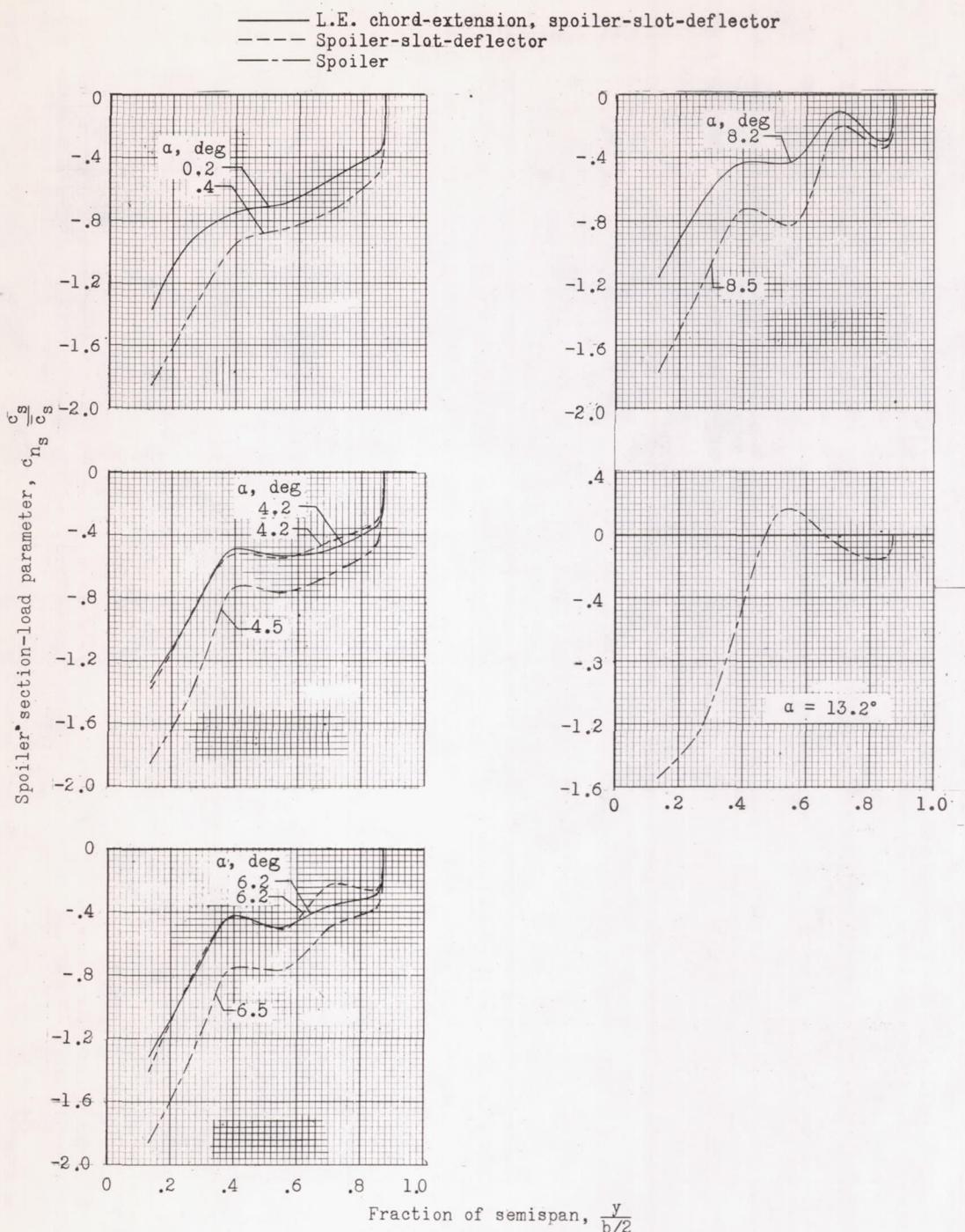
(g) $M = 1.03.$

Figure 18.- Concluded.

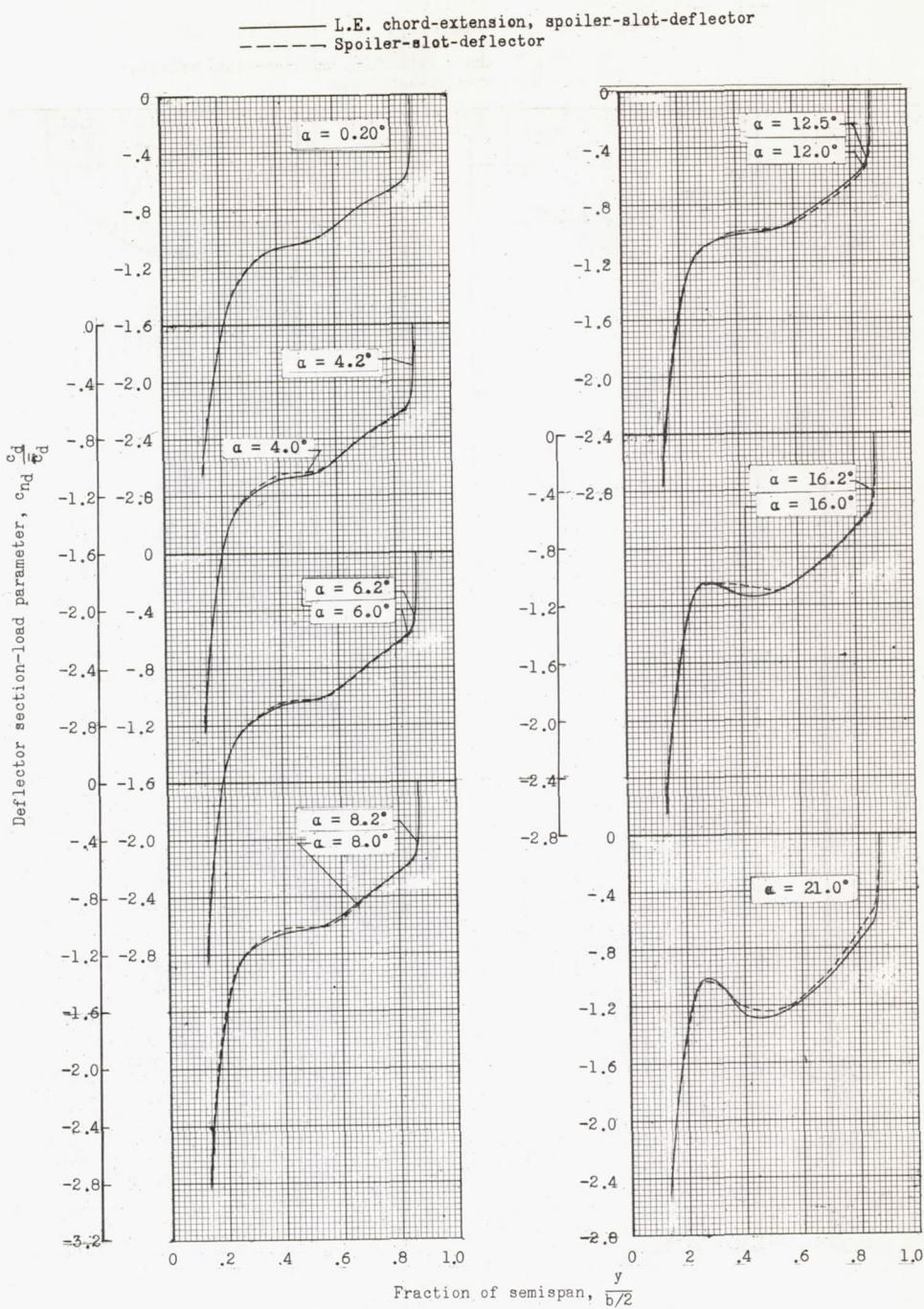
(a) $M = 0.60.$

Figure 19.- Span-load distributions for deflectors of basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations.

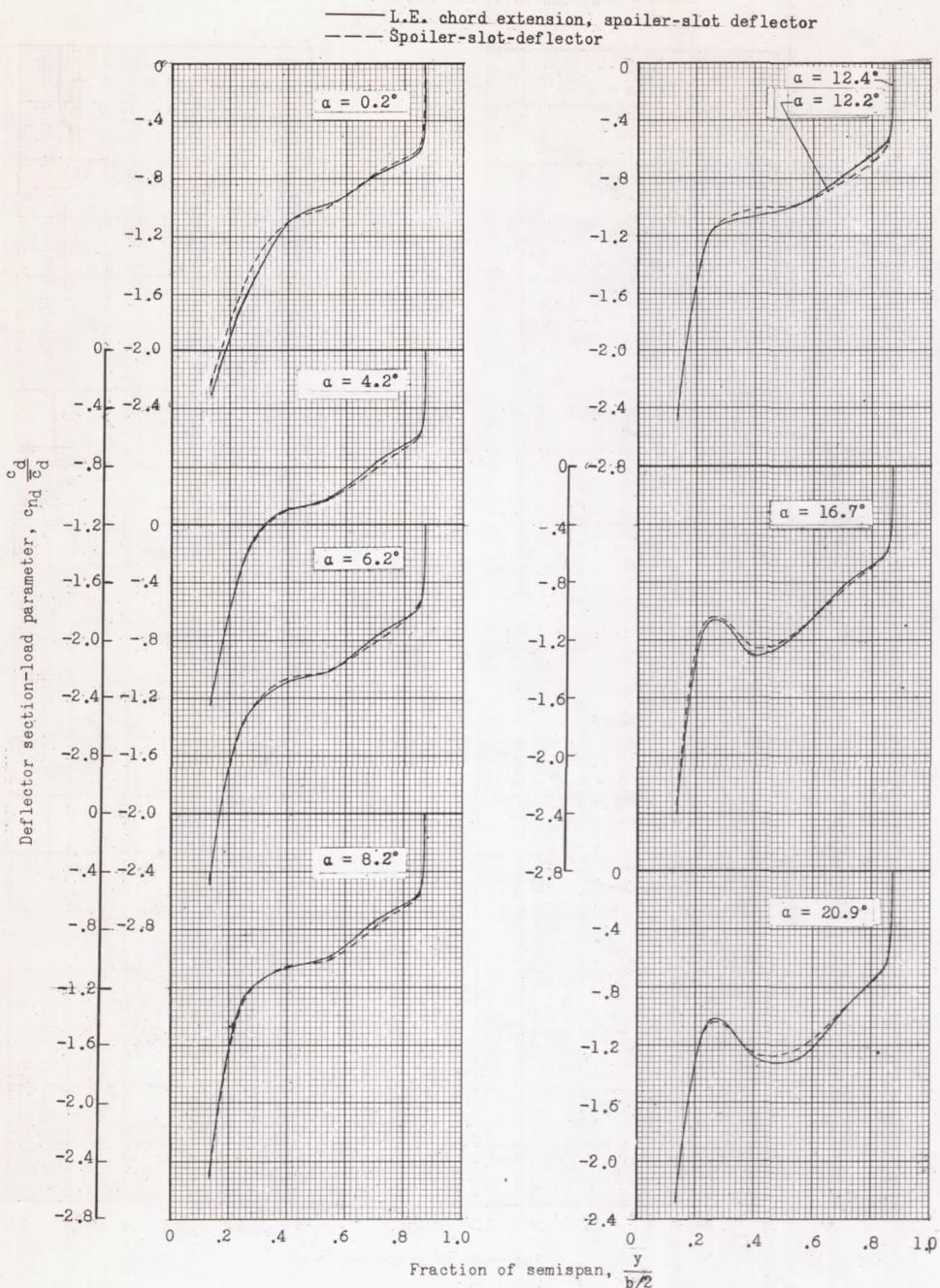
(b) $M = 0.80$.

Figure 19.- Continued.

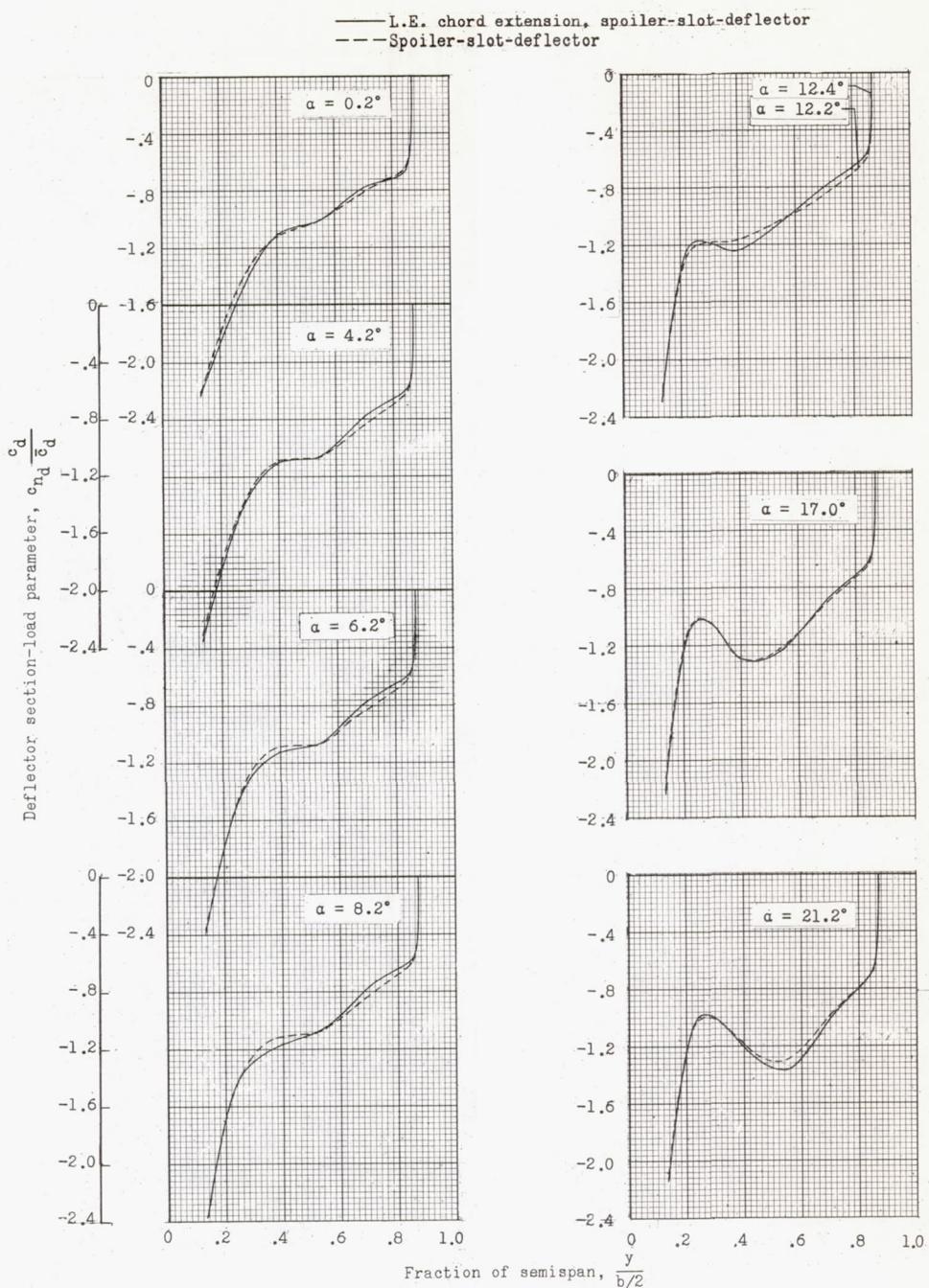
(c) $M = 0.90$.

Figure 19.- Continued.

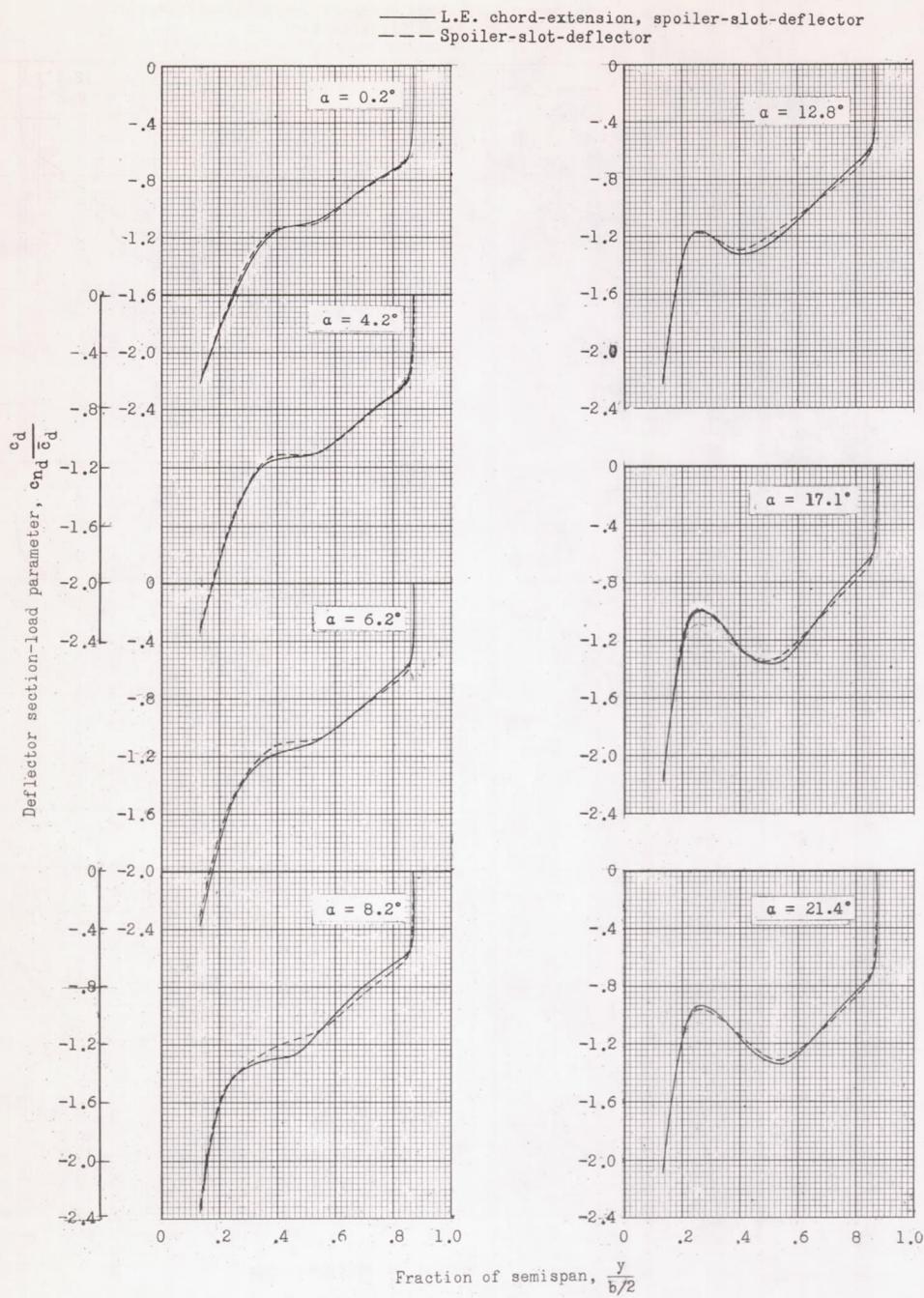
(d) $M = 0.94$.

Figure 19.- Continued.

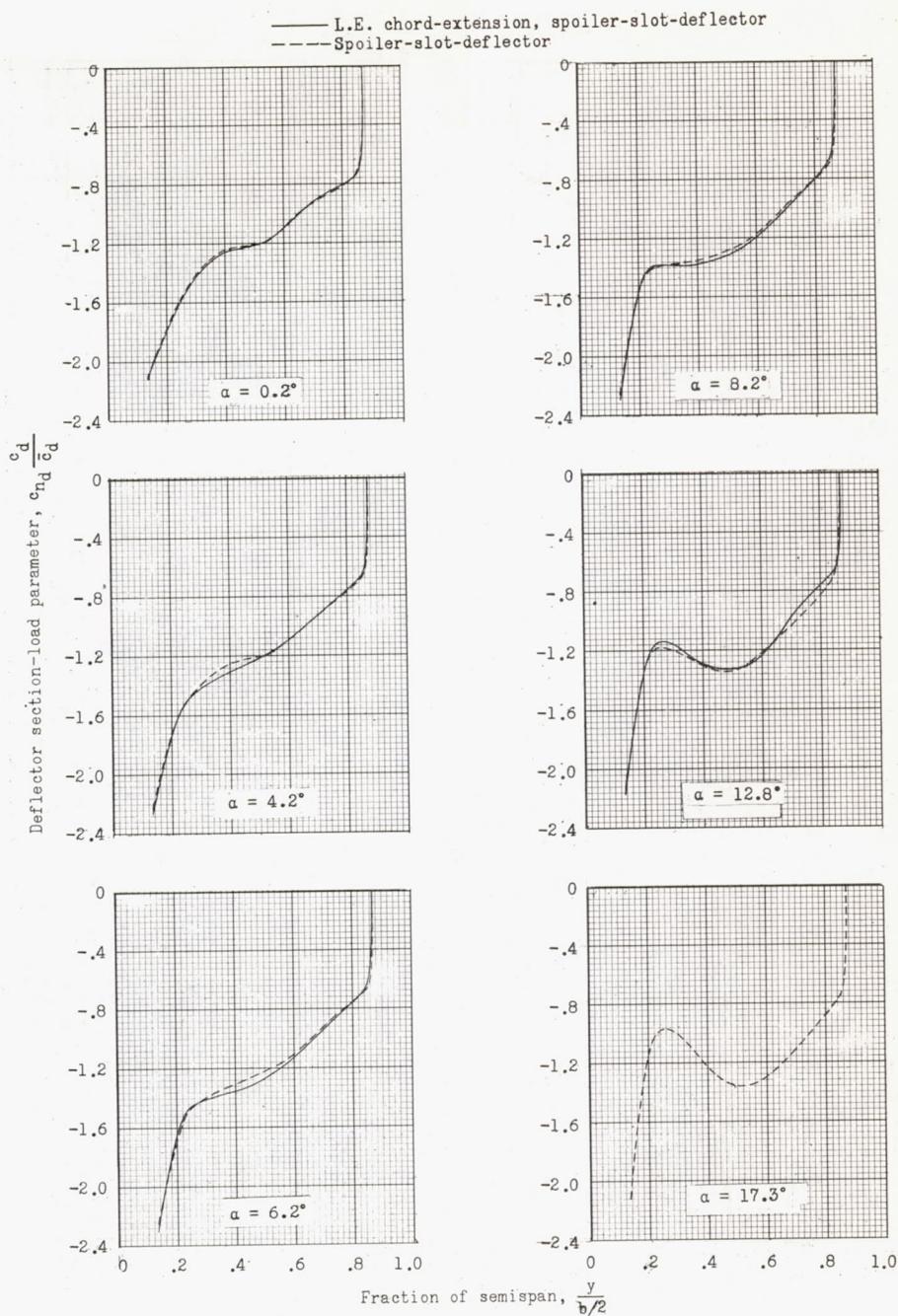
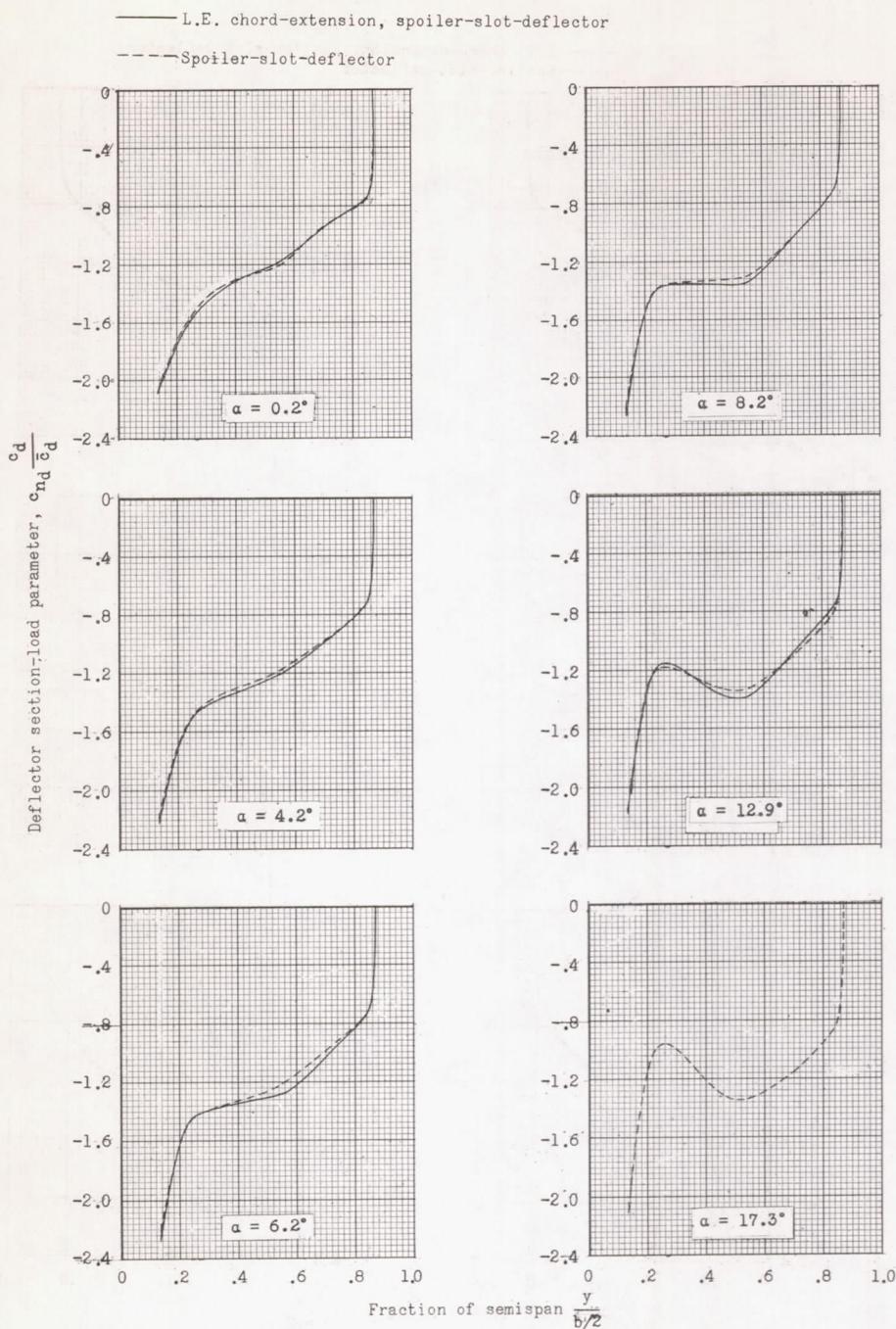
(e) $M = 0.98$.

Figure 19.- Continued.



(f) M = 1.00.

Figure 19.- Continued.

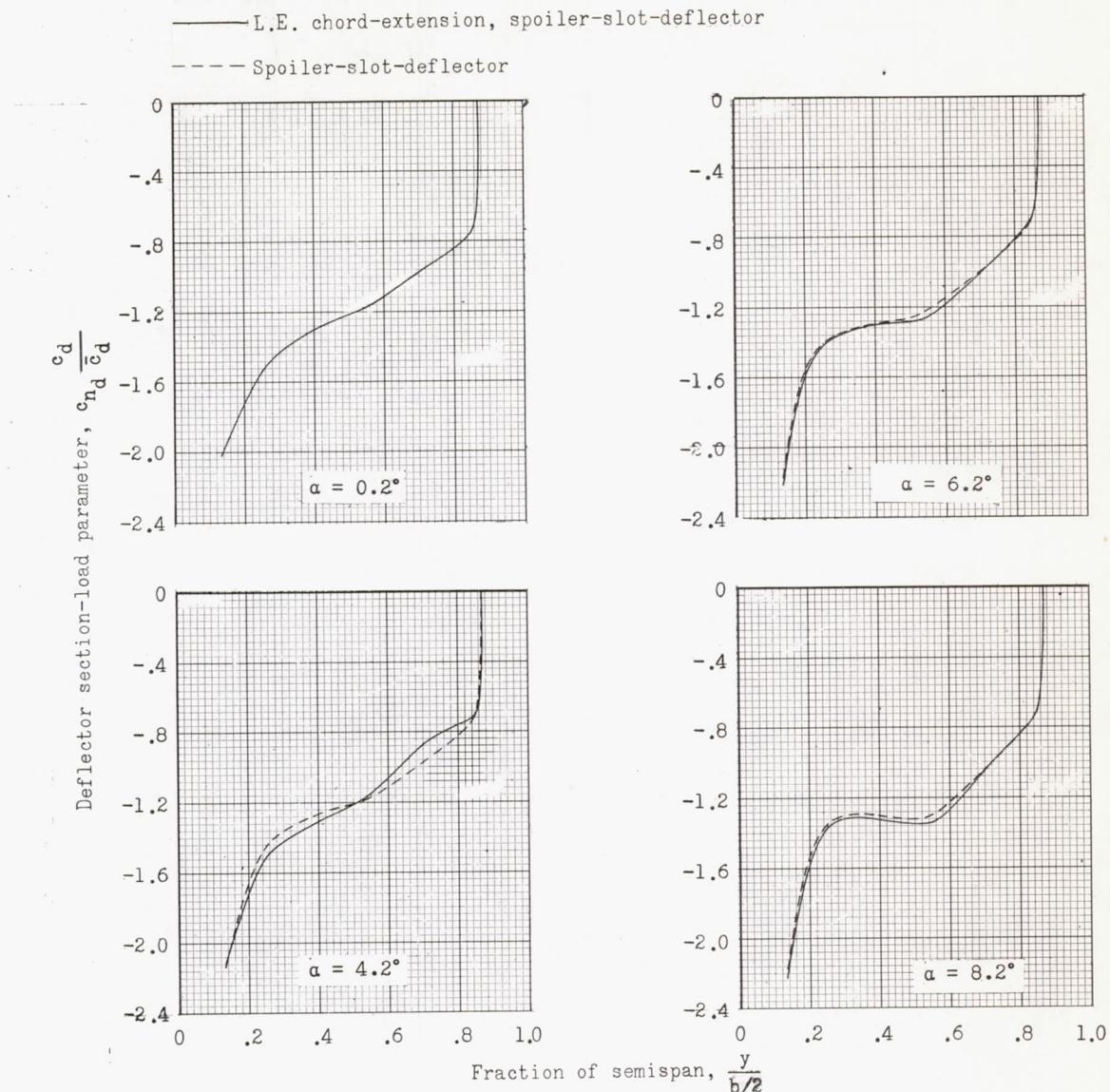
(g) $M = 1.03$.

Figure 19.- Concluded.

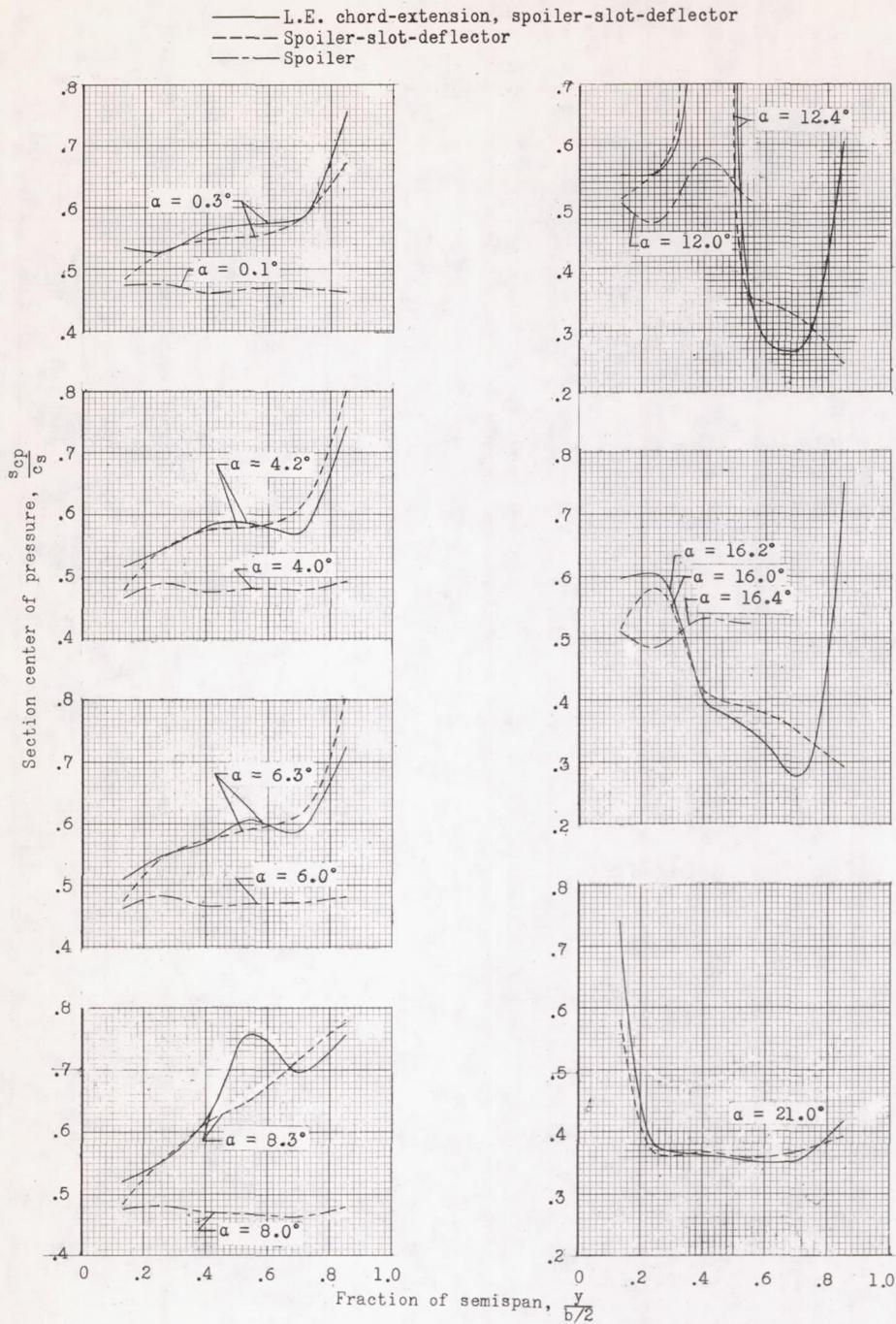
(a) $M = 0.60$.

Figure 20.- Section centers of pressure for a spoiler aileron and spoilers of basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations.

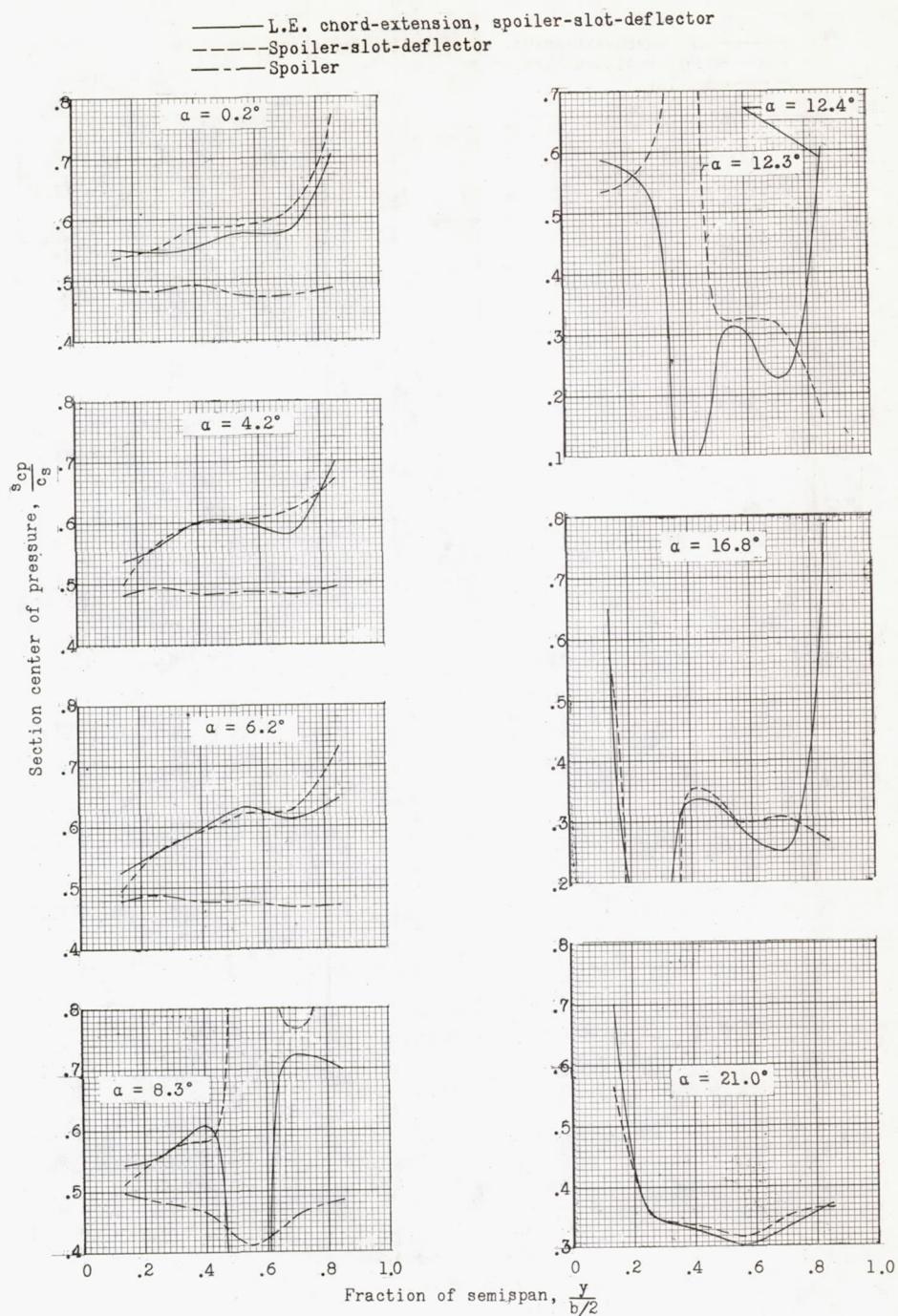
(b) $M = 0.80$.

Figure 20.- Continued.

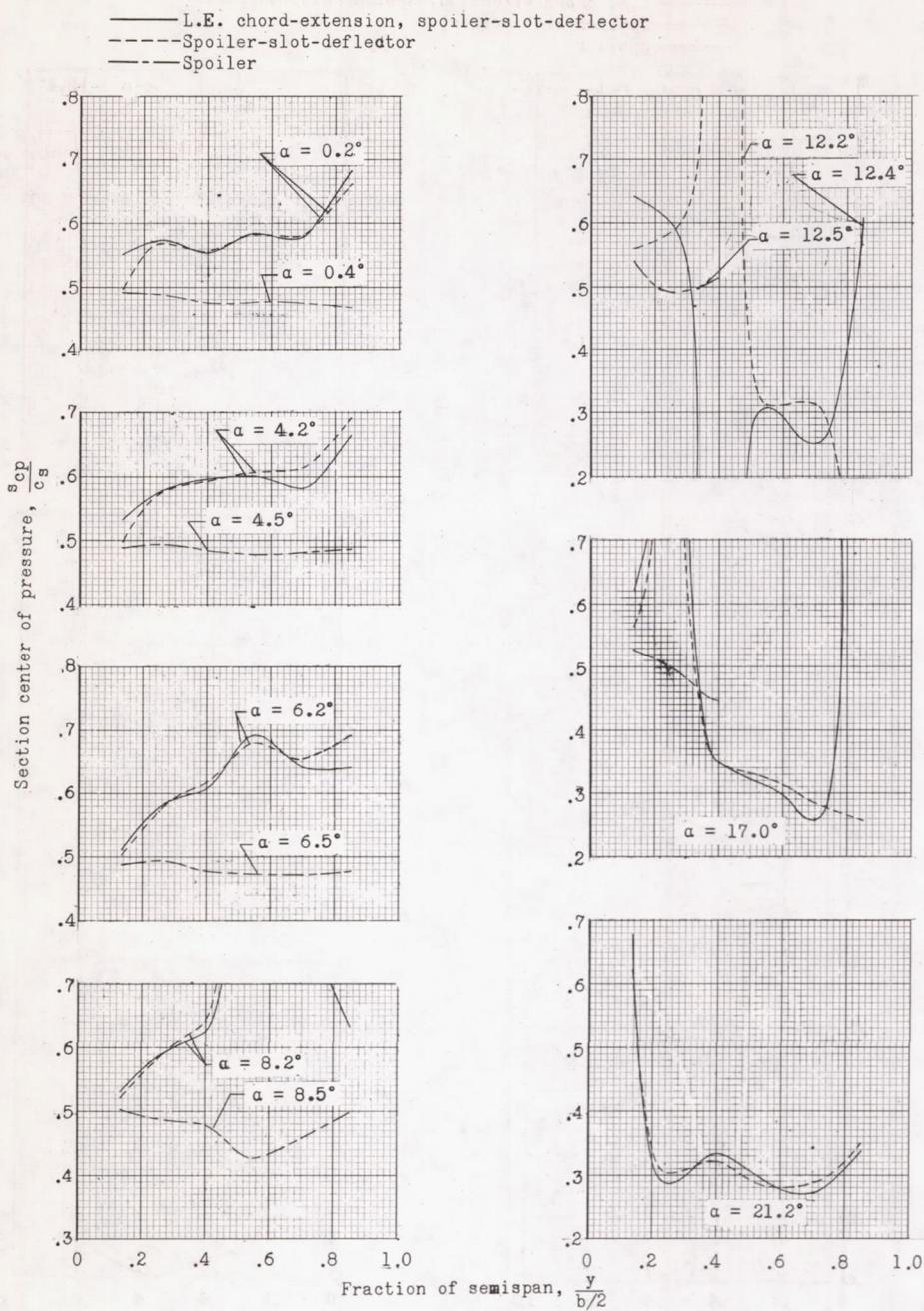
(c) $M = 0.90.$

Figure 20.- Continued.

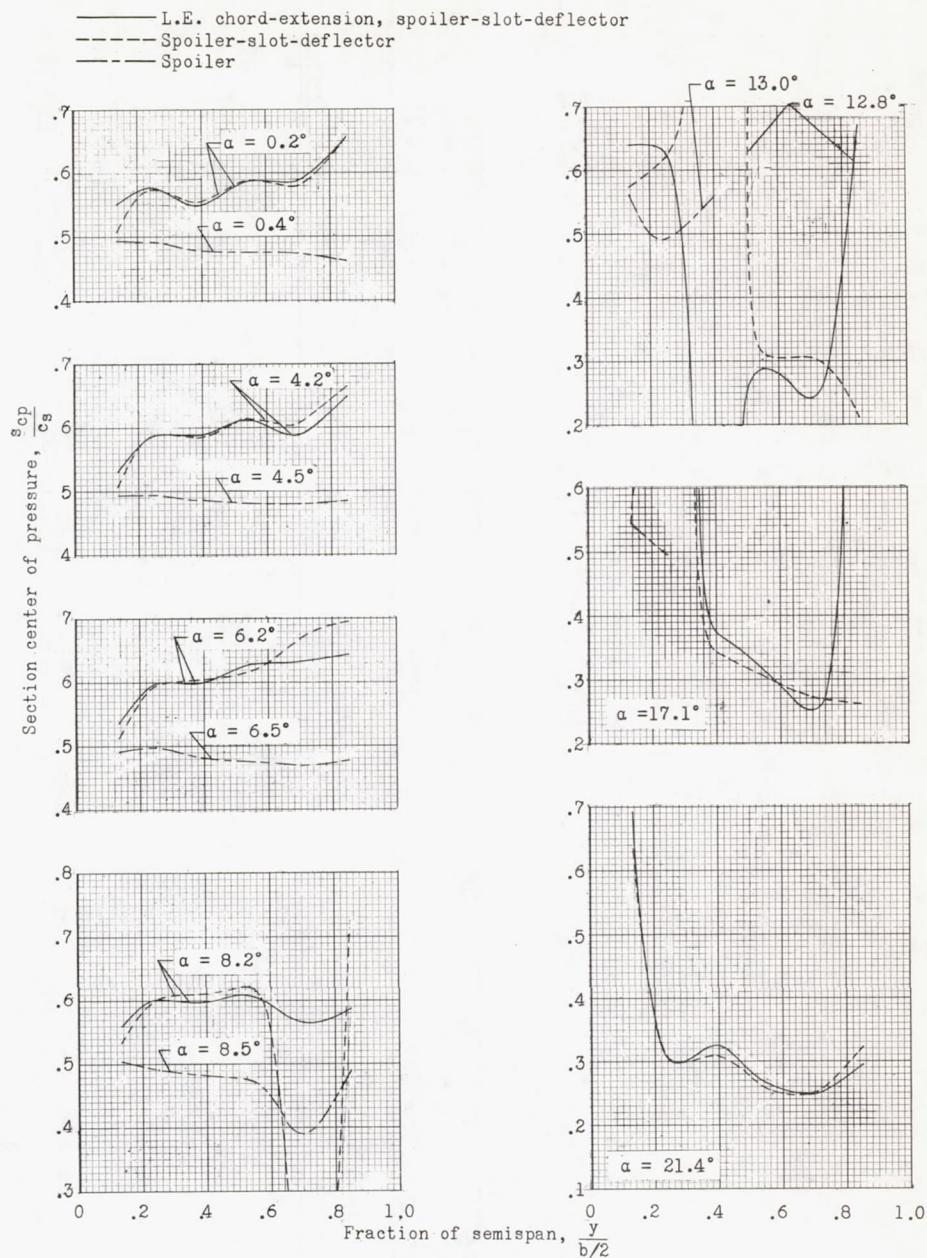


Figure 20.- Continued.

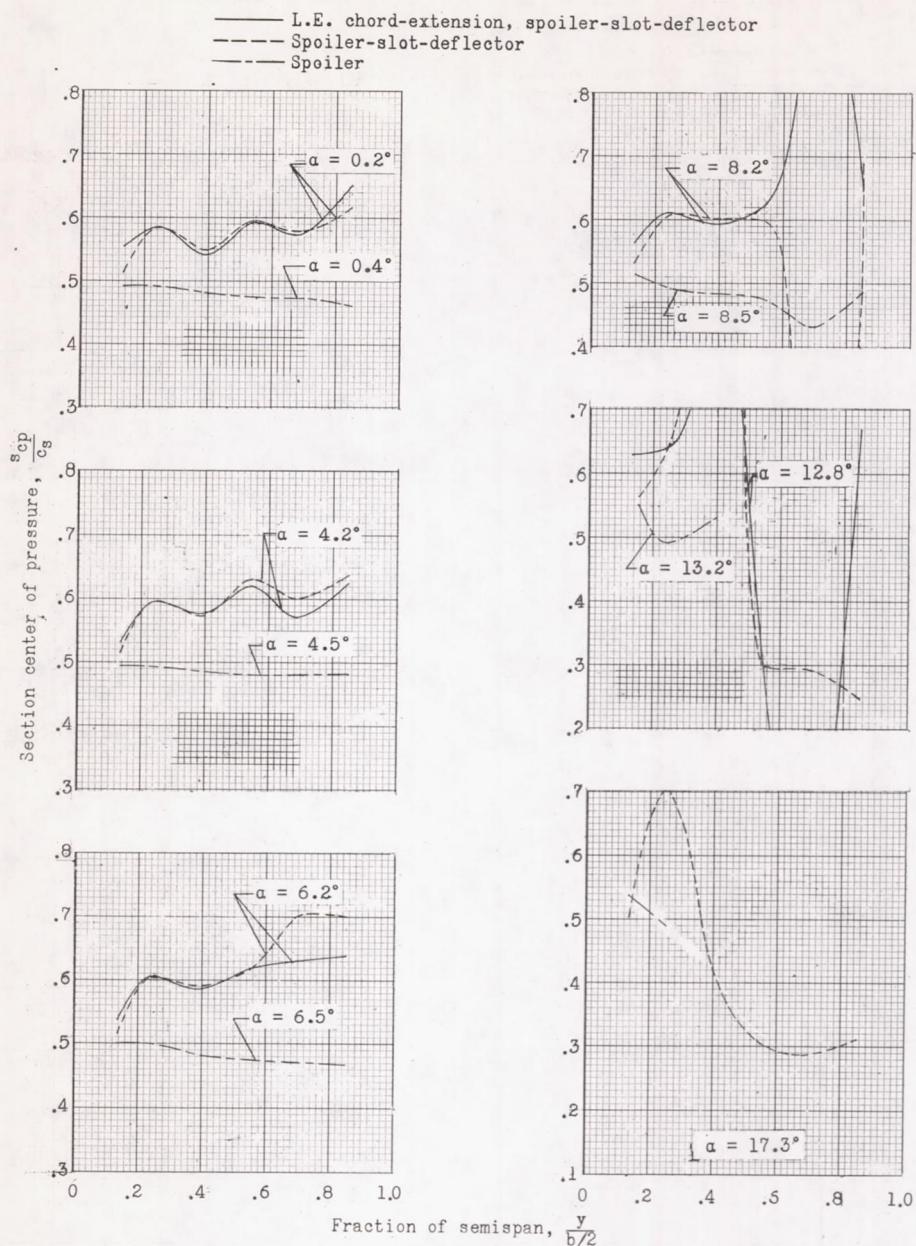


Figure 20.- Continued.

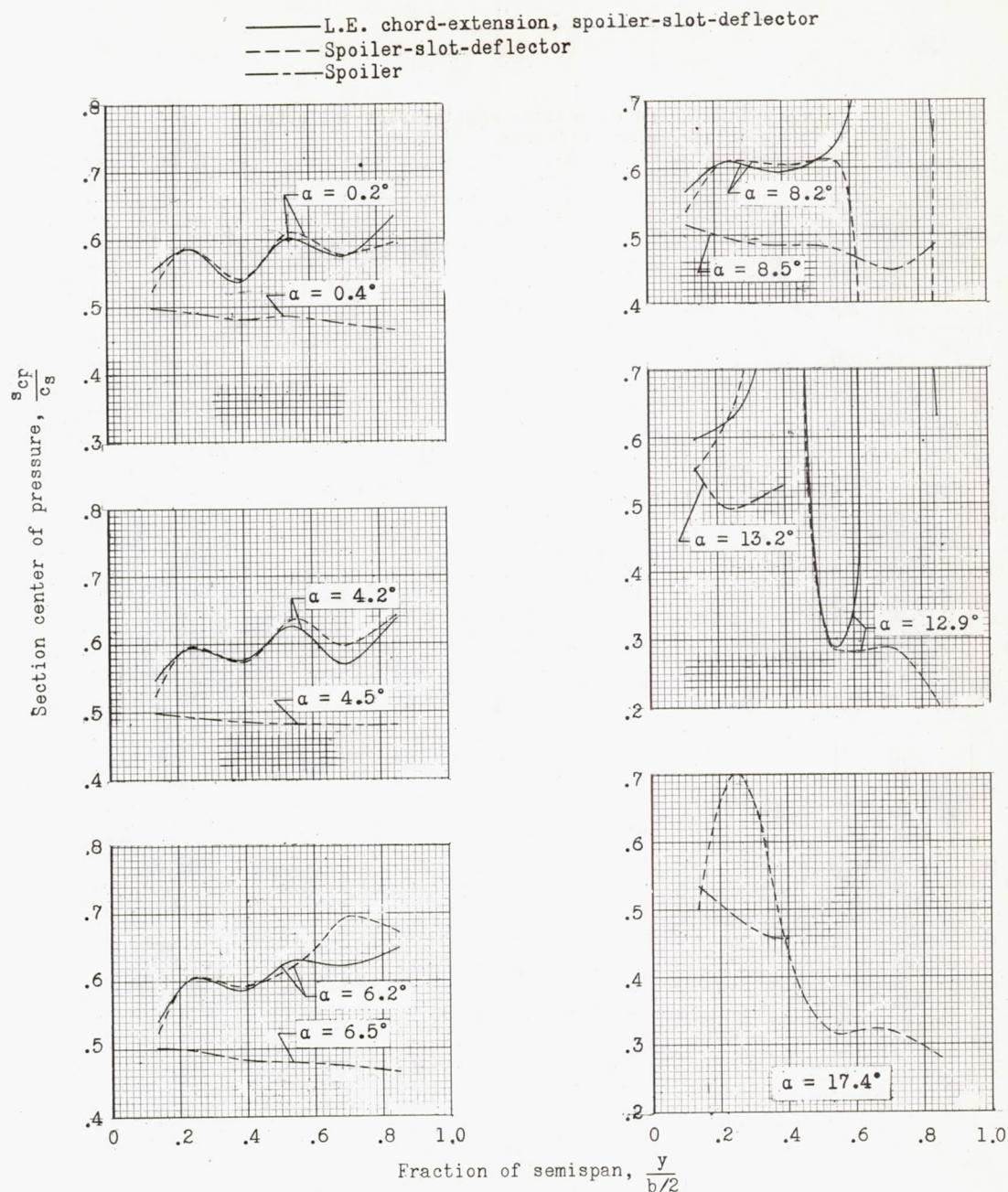
(f) $M = 1.00$.

Figure 20.- Continued.

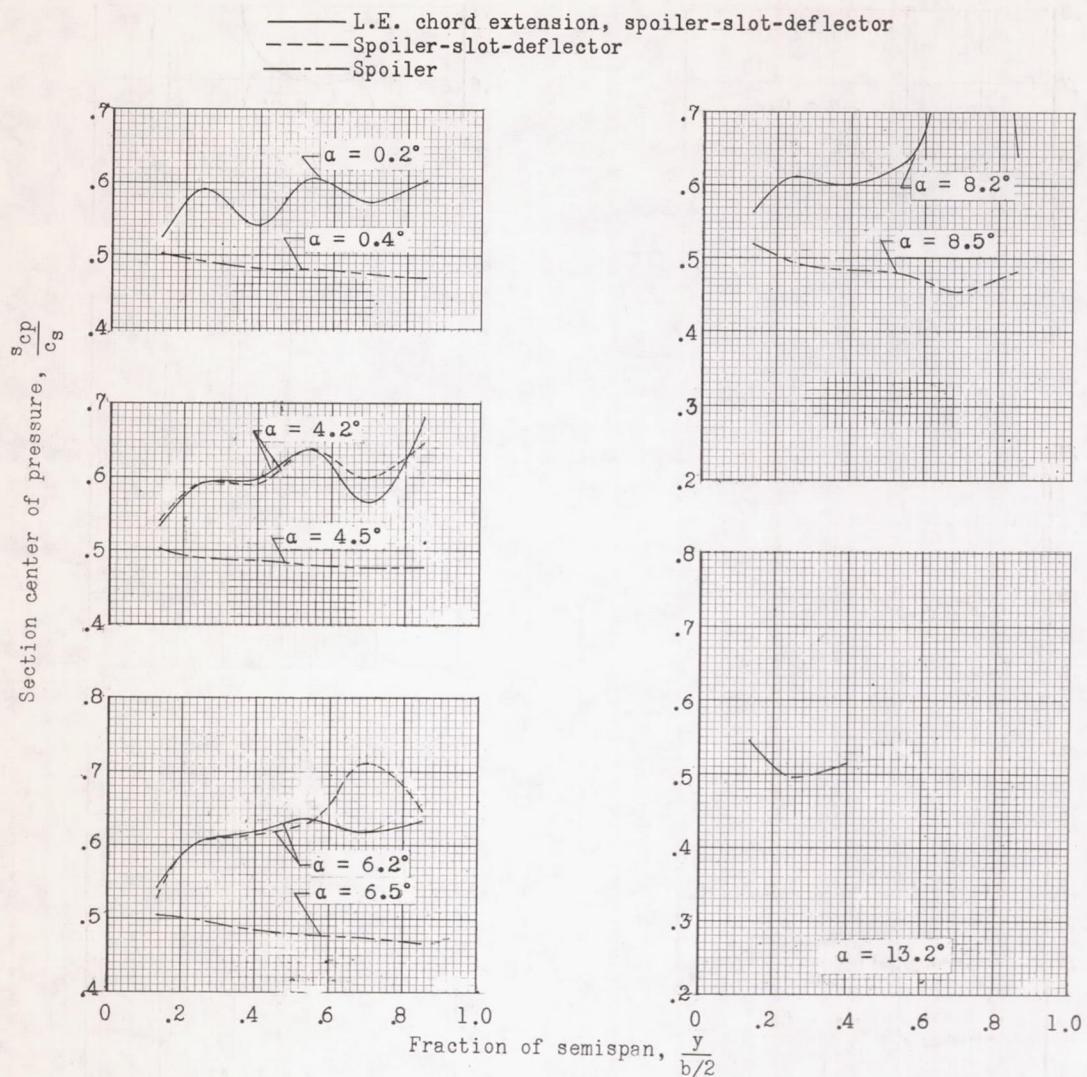


Figure 20.- Concluded.

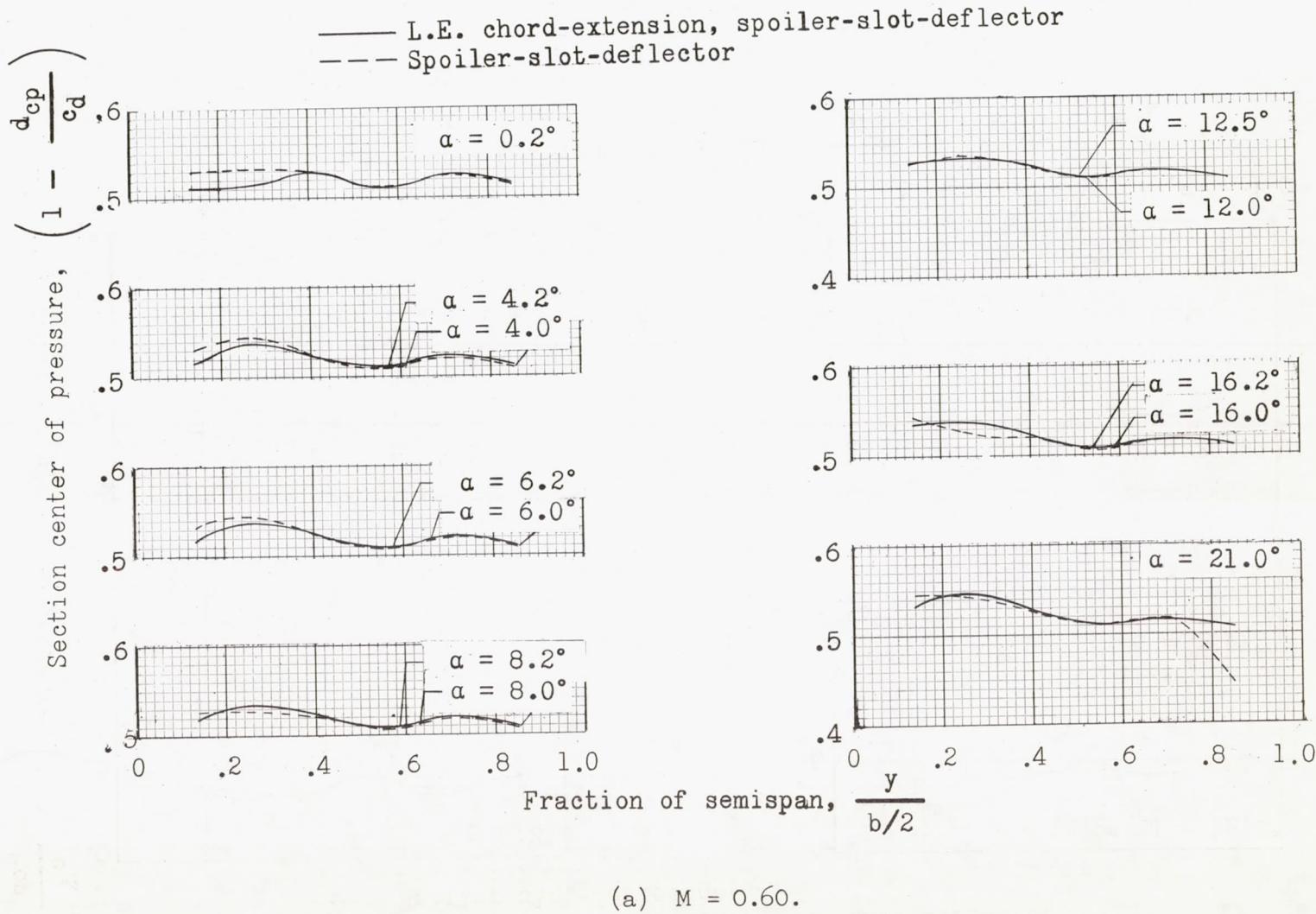
(a) $M = 0.60$.

Figure 21.- Section centers of pressure (measured from hinge line) for deflectors of the basic and leading-edge chord-extension spoiler-slot-deflector aileron configurations.

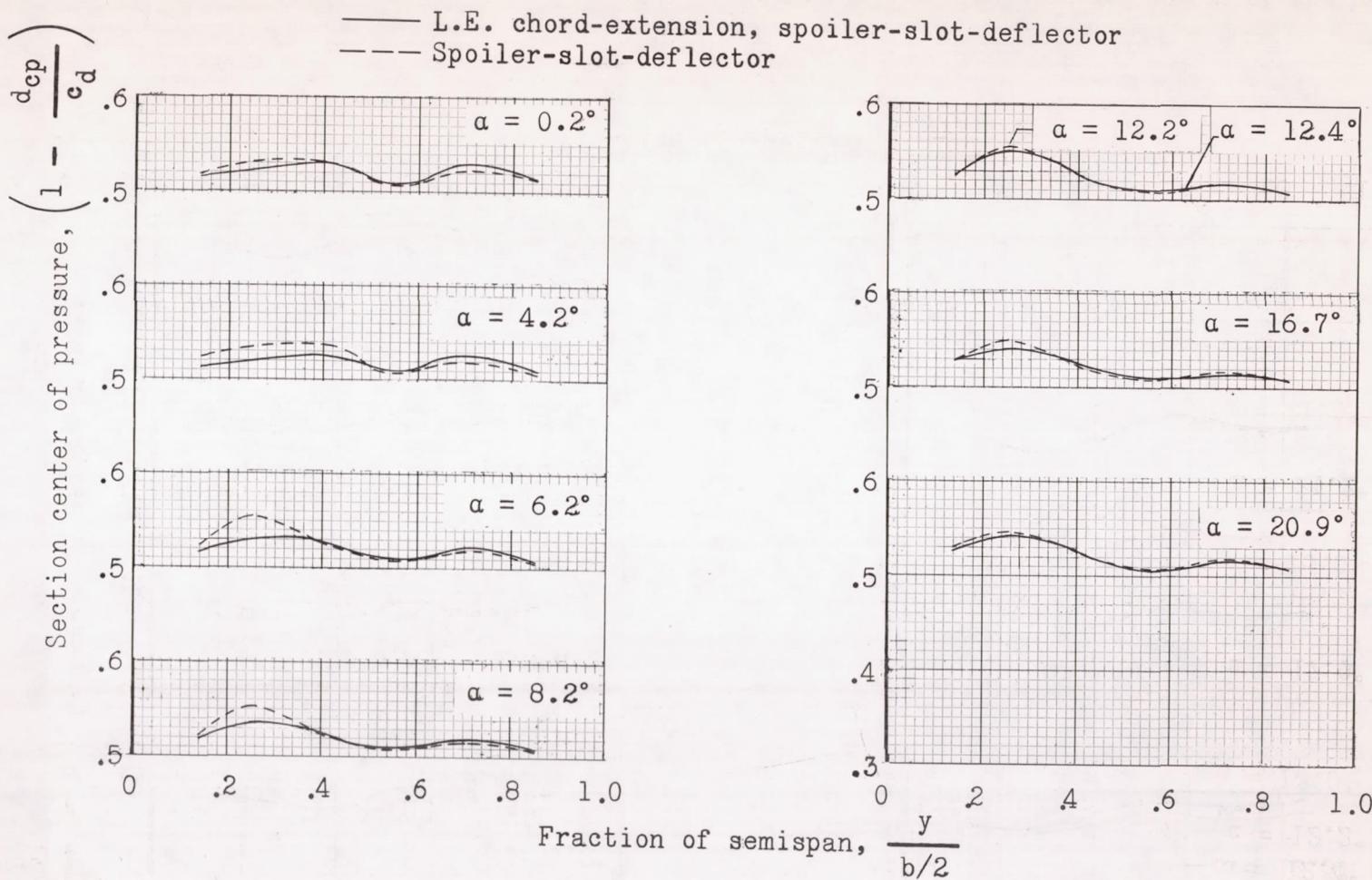
(b) $M = 0.80$.

Figure 21.- Continued.

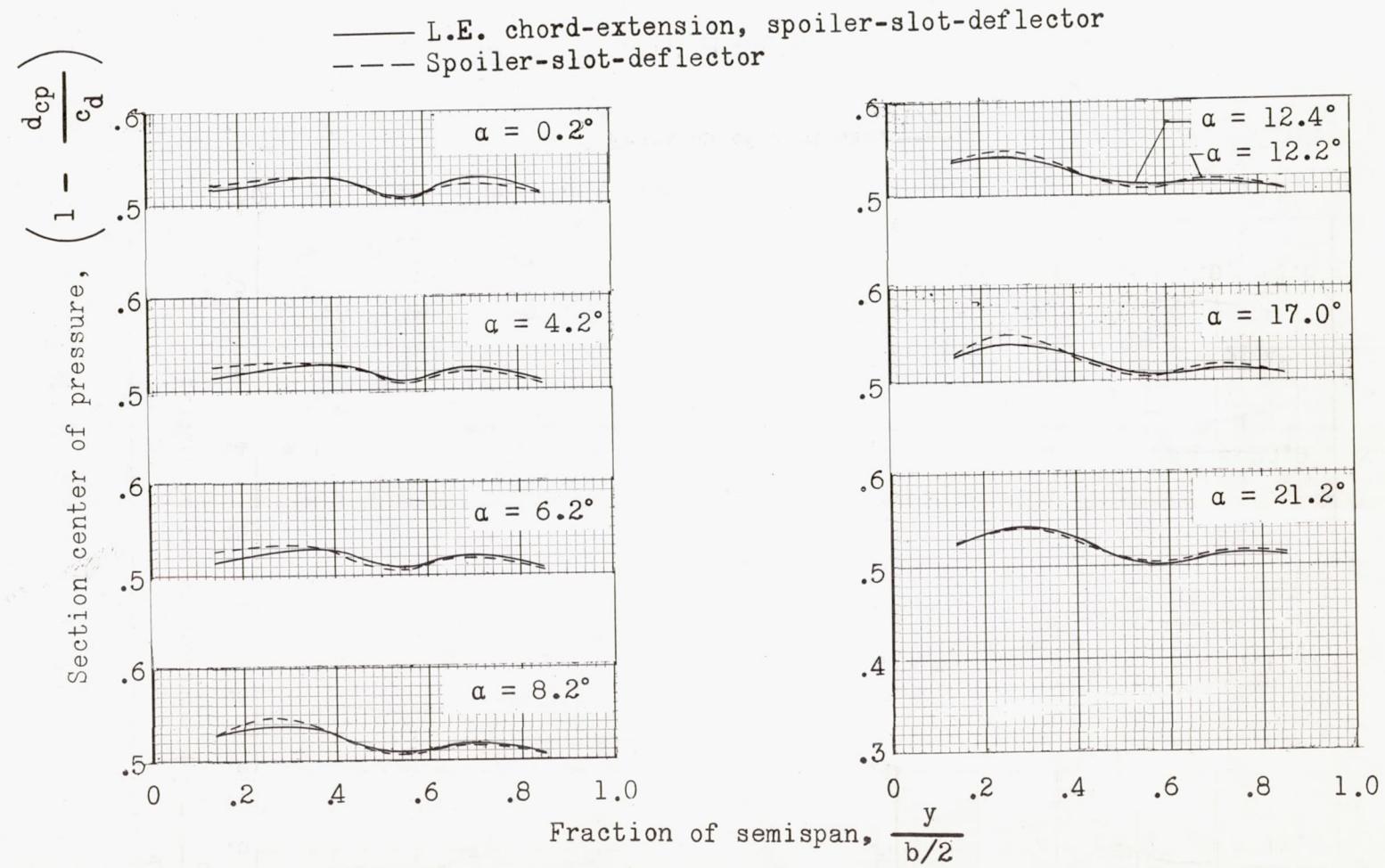
(c) $M = 0.90.$

Figure 21.- Continued.

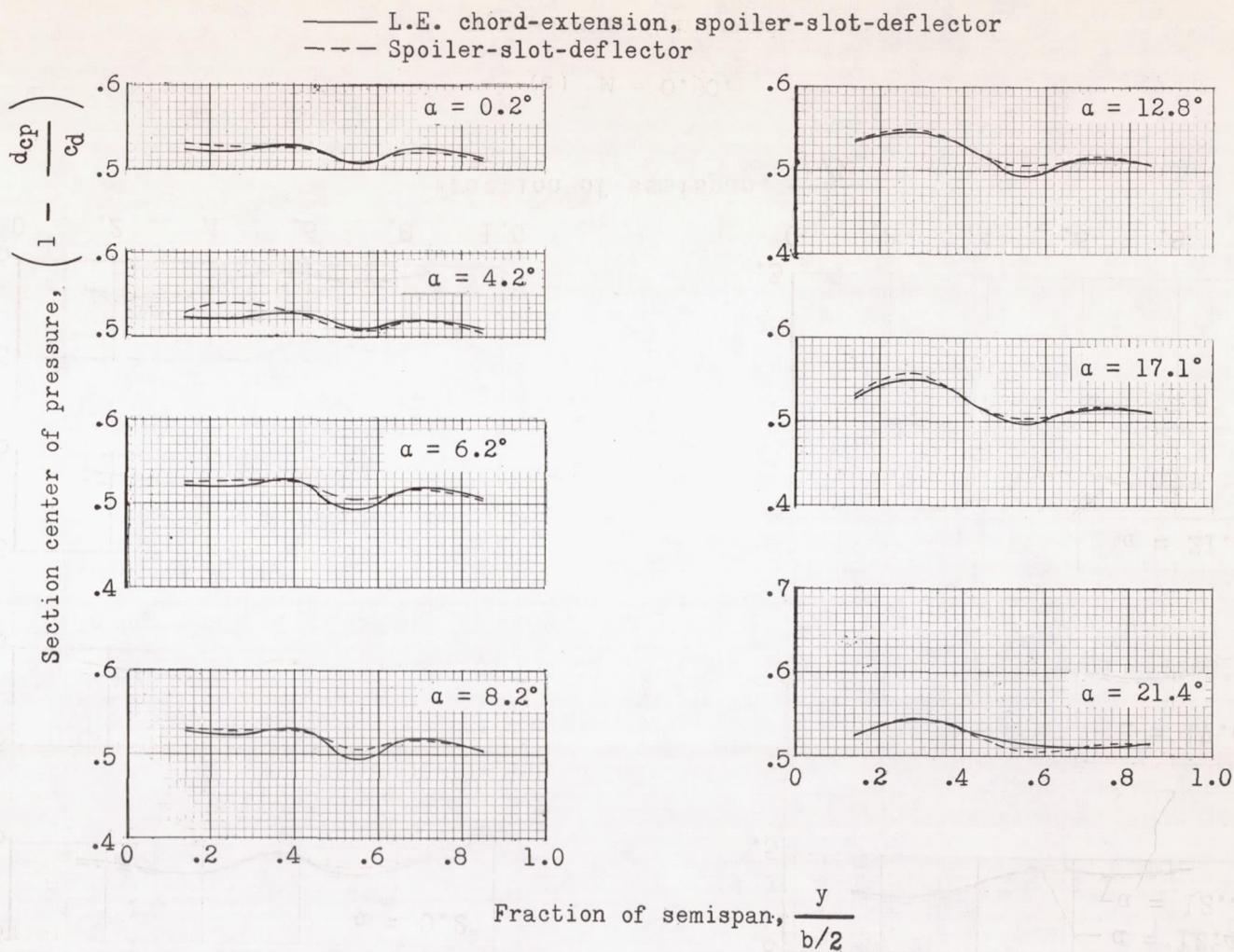
(d) $M = 0.94$.

Figure 21.- Continued.

CONFIDENTIAL

116

CONFIDENTIAL

NACA RM 15711

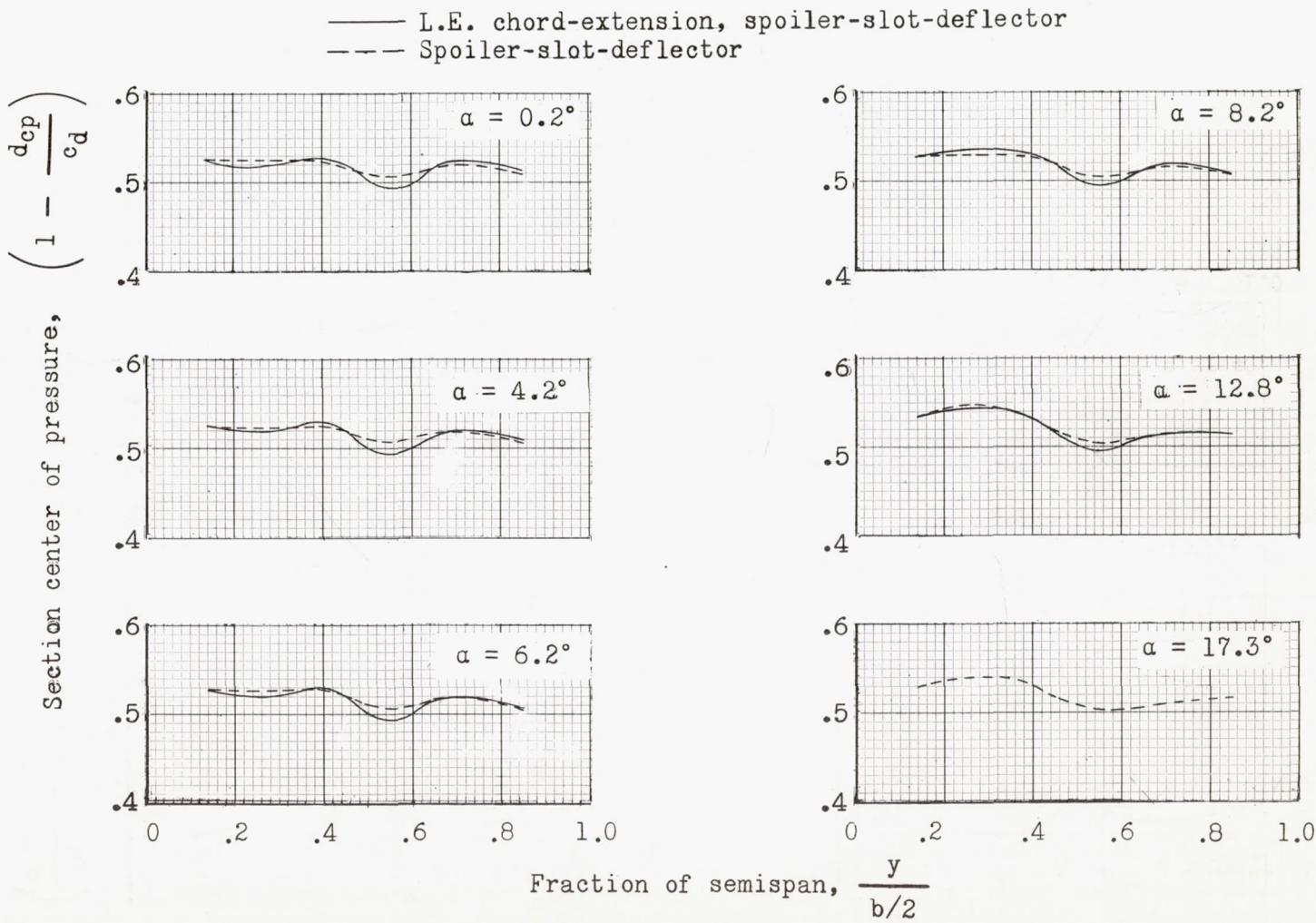
(e) $M = 0.98.$

Figure 21.- Continued.

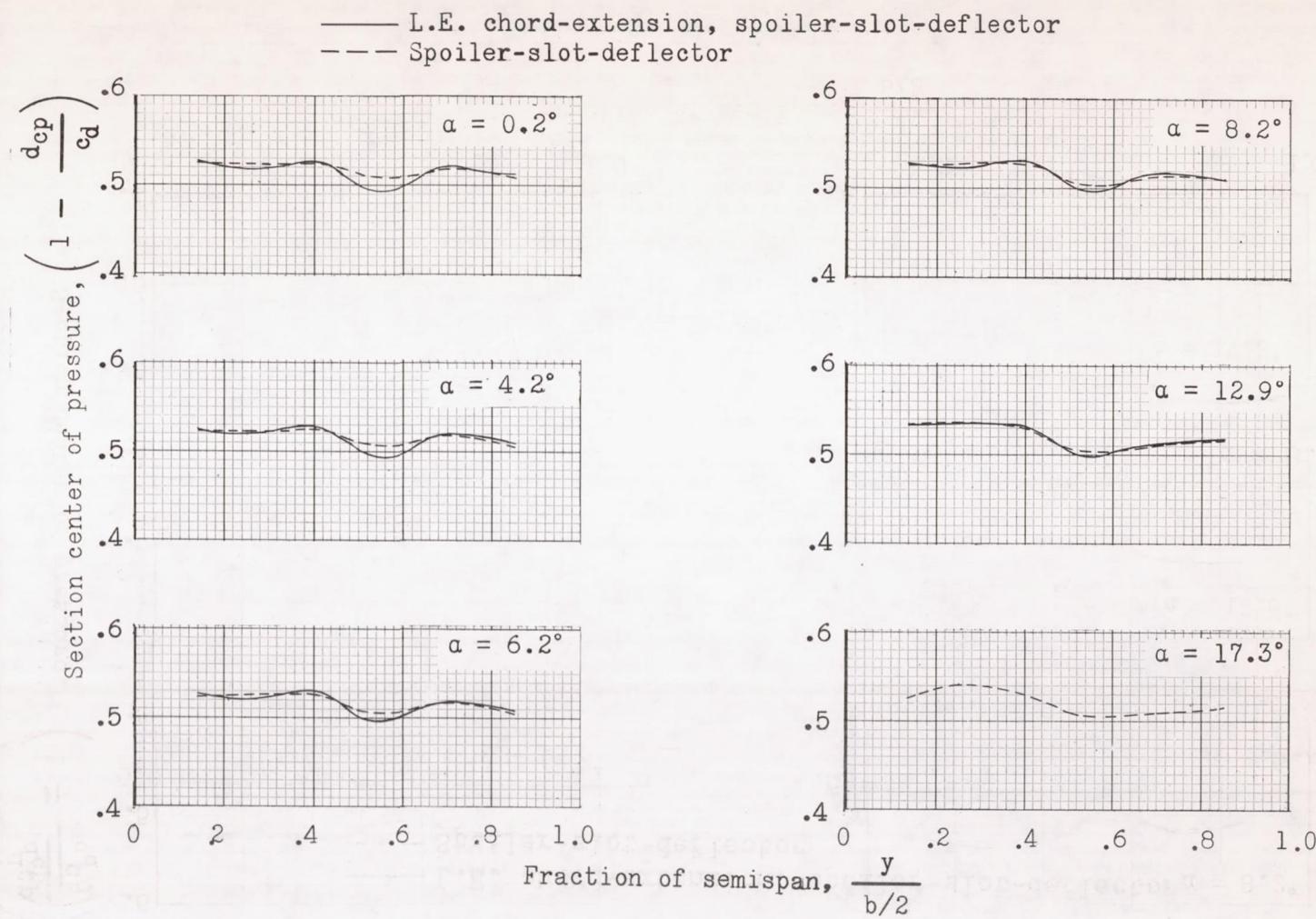
(f) $M = 1.00.$

Figure 21.- Continued.

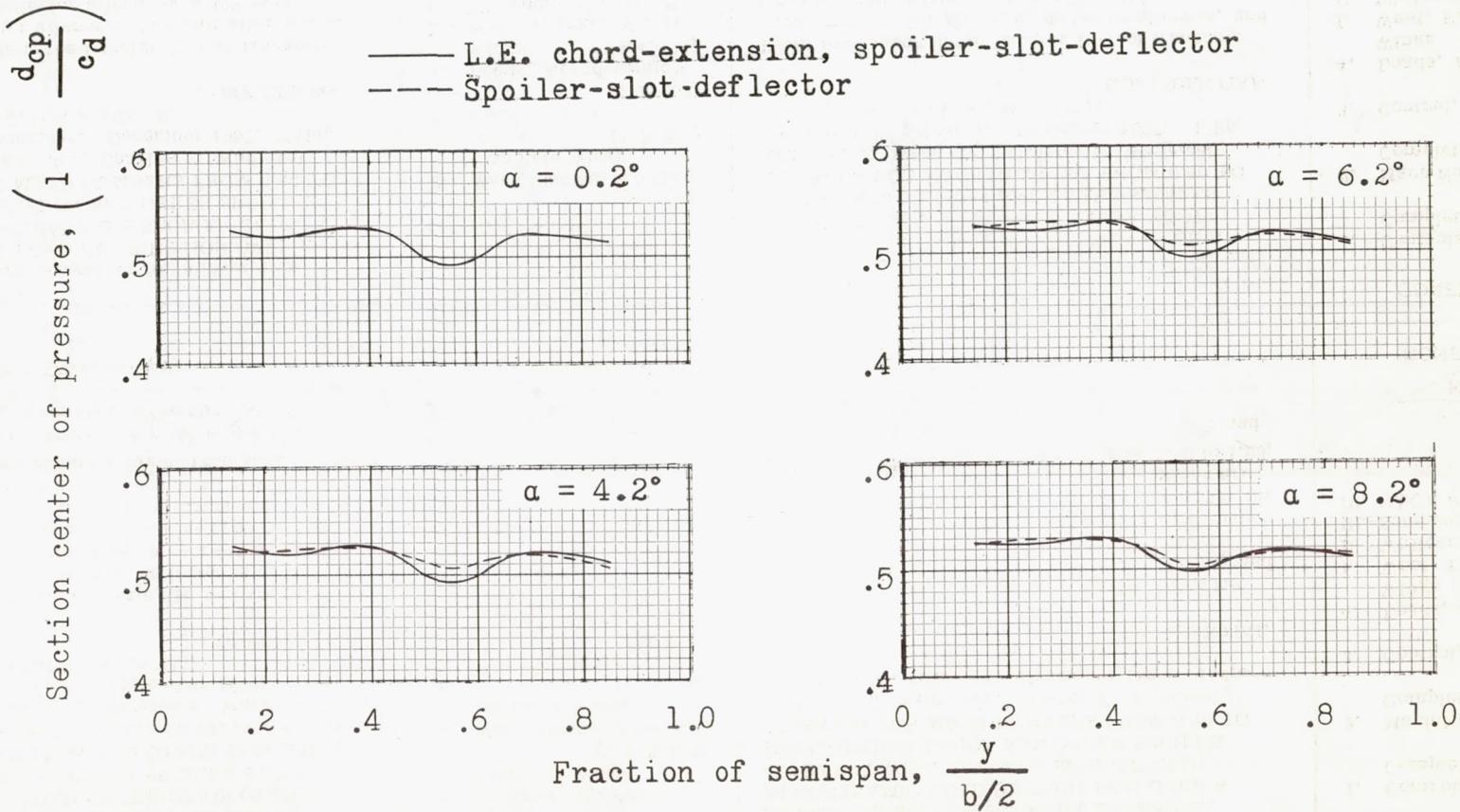
(g) $M = 1.03$.

Figure 21.- Concluded.

CONFIDENTIAL

CONFIDENTIAL